

# ***DIFFERENTIAL ALGEBRA SPACE TOOLBOX\**** ***FOR NONLINEAR UNCERTAINTY PROPAGATION IN*** ***SPACE DYNAMICS***

*6<sup>th</sup> International Conference on Astrodynamics Tools and Techniques (ICATT)*  
*Darmstadt, Germany*

M. Rasotto, A. Morselli, A. Wittig, M. Massari,  
P. Di Lizia, R. Armellin, C. Y. Valles and  
G. Ortega

March 17<sup>th</sup>, 2016

*\* Developed under ITT AO/1-7570/13/NL/MH: Nonlinear Propagation of Uncertainties in Space Dynamics based on Taylor Differential Algebra"*

- Uncertainty propagation is a crucial issue in spaceflight dynamics
  - Space surveillance and tracking
  - Reentry and casualty area computation
  - Robust design of space trajectories and systems
  - ...
- Most spaceflight mechanics problems involve **nonlinear behavior**



Need of efficient tools for **nonlinear propagation of uncertainties**

Linearized models

...

Monte Carlo

 low computational burden

 low accuracy

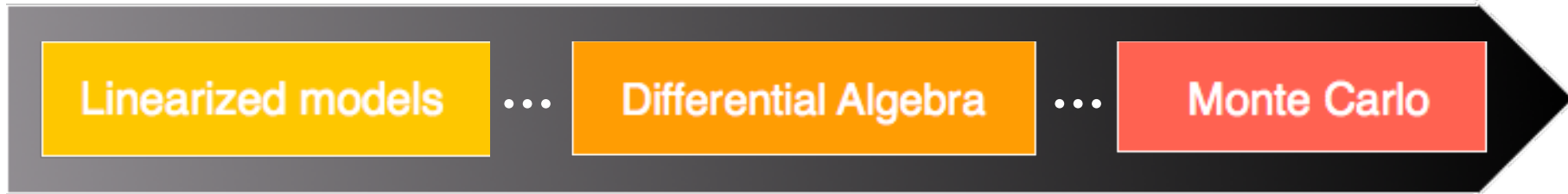
 computationally intensive

 high accuracy



Can we find a **compromise technique**?

- We need a technique to:
  - **Improve** accuracy of linearized models
  - **Reduce** computational cost of classical Monte Carlo



 low computational burden

 low accuracy

 computationally intensive

 high accuracy



Can we find a **compromise technique**?

- We need a technique to:
  - **Improve** accuracy of linearized models
  - **Reduce** computational cost of classical Monte Carlo

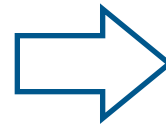


Differential Algebra



- Differential Algebra (DA) is an automatic differentiation technique

Algebra of  
real numbers



Algebra of  
Taylor polynomials

- DA can be implemented in a computer environment (DACE)
- Given any sufficiently regular function  $f(\boldsymbol{x})$ 
  - Initialize  $\boldsymbol{x}$  as a DA variable:  $[\boldsymbol{x}] = \bar{\boldsymbol{x}} + \delta\boldsymbol{x}$
  - Evaluate  $f$  in the DA framework:

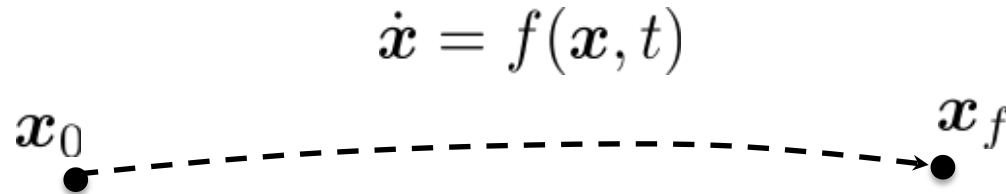
$$f([\boldsymbol{x}]) = \mathcal{T}_f(\delta\boldsymbol{x})$$




Taylor expansion of  $f$  around  $\bar{\boldsymbol{x}}$   
up to an **arbitrary order**  $k$

# EXPANSION OF THE FLOW OF ODEs

- Given any dynamics

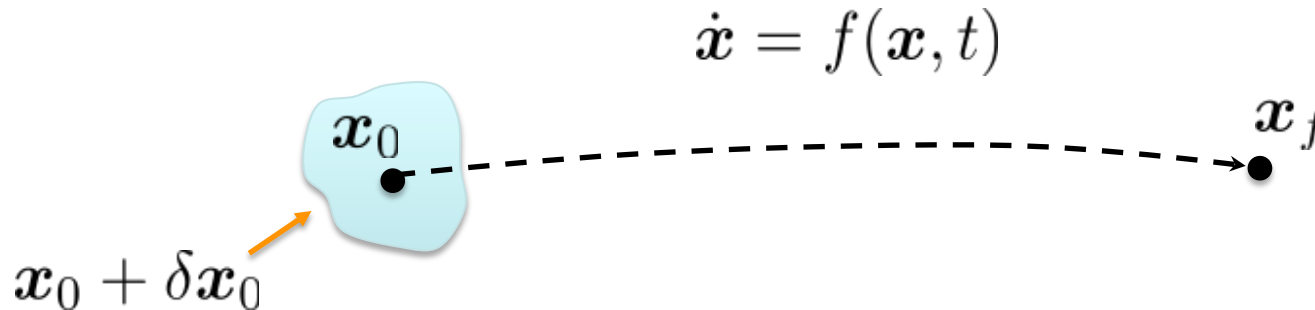


- Any numerical integrator is based on the evaluation of  $f$  and its algebraic manipulation
    - Initialize  $x_0$  as DA vector
    - Perform operations in DA
- 

Taylor expansion  
of  $x_f$  w.r.t.  $x_0$
- Uncertainty propagation can benefit from DA in different ways

# EXPANSION OF THE FLOW OF ODES

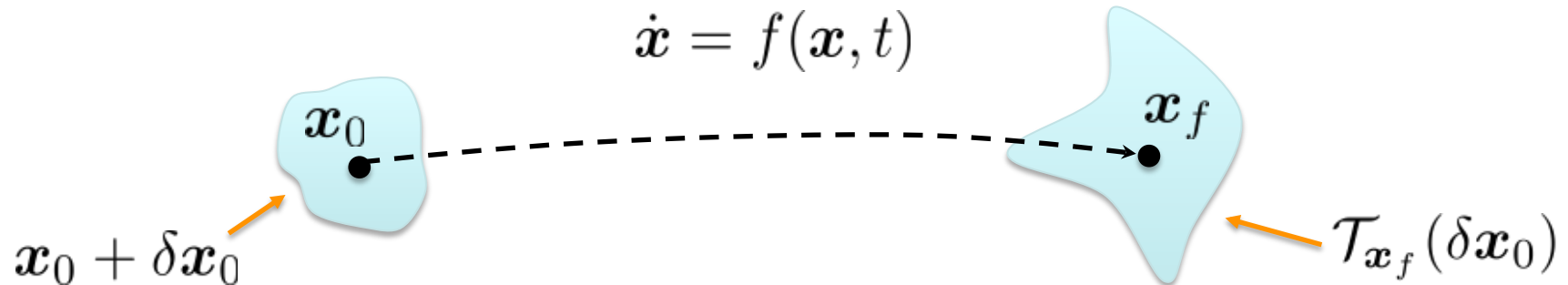
- Given any dynamics



- Any numerical integrator is based on the evaluation of  $f$  and its algebraic manipulation
    - Initialize  $x_0$  as DA vector
    - Perform operations in DA
- ➡
- Taylor expansion  
of  $x_f$  w.r.t.  $x_0$
- Uncertainty propagation can benefit from DA in different ways

# EXPANSION OF THE FLOW OF ODES

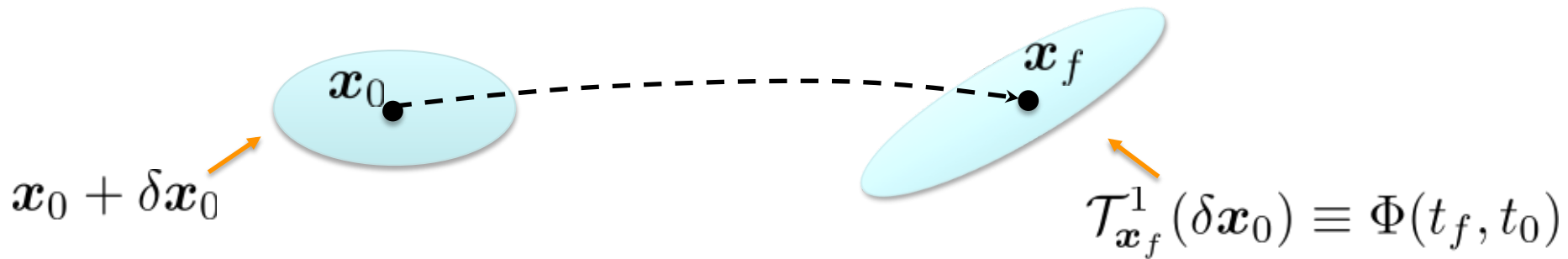
- Given any dynamics



- Any numerical integrator is based on the evaluation of  $f$  and its algebraic manipulation
    - Initialize  $x_0$  as DA vector
    - Perform operations in DA
- ➡
- Taylor expansion  
of  $x_f$  w.r.t.  $x_0$
- Uncertainty propagation can benefit from DA in different ways



# LINEAR COVARIANCE PROPAGATION



- If all computations are performed to **order 1**, the final Taylor expansion  $\mathcal{T}_{x_f}^1(\delta x_0)$  coincides with the **state-transition matrix**  $\Phi(t_f, t_0)$

Perform a single DA  
integration with  
expansion order 1



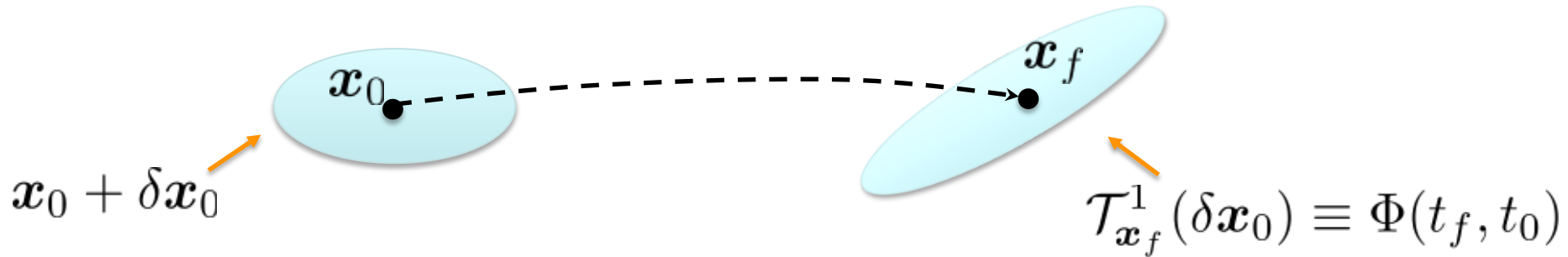
Extract the STM from  
the Taylor polynomial  
map



Map the initial  
covariance:

$$\mathcal{C}_f \equiv \Phi \mathcal{C}_0 \Phi^T$$

# LINEAR COVARIANCE PROPAGATION



- If all computations are performed to **order 1**, the final Taylor expansion  $\mathcal{T}_{x_f}^1(\delta x_0)$  coincides with the **state-transition matrix**  $\Phi(t_f, t_0)$

Perform a single DA  
integration with  
expansion order 1



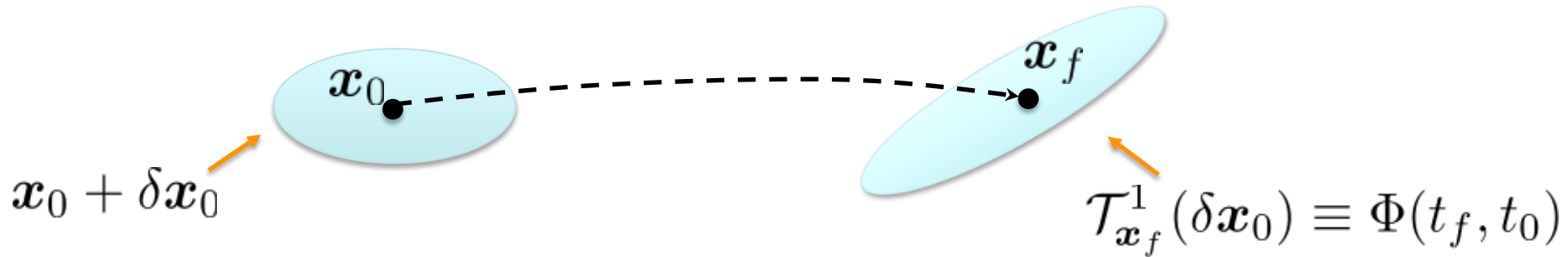
Extract the STM from  
the Taylor polynomial  
map



Map the initial  
covariance:

$$\mathcal{C}_f \equiv \Phi \mathcal{C}_0 \Phi^T$$

# LINEAR COVARIANCE PROPAGATION



- If all computations are performed to **order 1**, the final Taylor expansion  $\mathcal{T}_{x_f}^1(\delta x_0)$  coincides with the **state-transition matrix**  $\Phi(t_f, t_0)$

Perform a single DA  
integration with  
expansion order 1



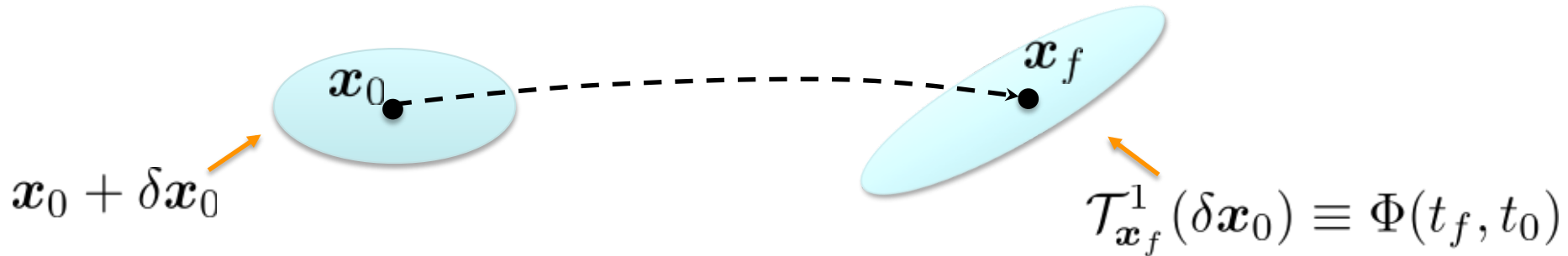
Extract the STM from  
the Taylor polynomial  
map



Map the initial  
covariance:

$$C_f = \Phi C_0 \Phi^T$$

# LINEAR COVARIANCE PROPAGATION



- If all computations are performed to **order 1**, the final Taylor expansion  $\mathcal{T}_{x_f}^1(\delta x_0)$  coincides with the **state-transition matrix**  $\Phi(t_f, t_0)$

Perform a single DA  
integration with  
expansion order 1



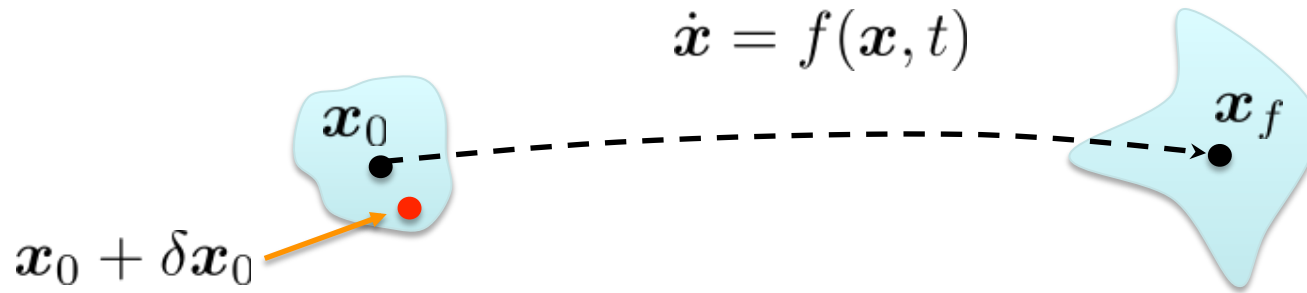
Extract the STM from  
the Taylor polynomial  
map

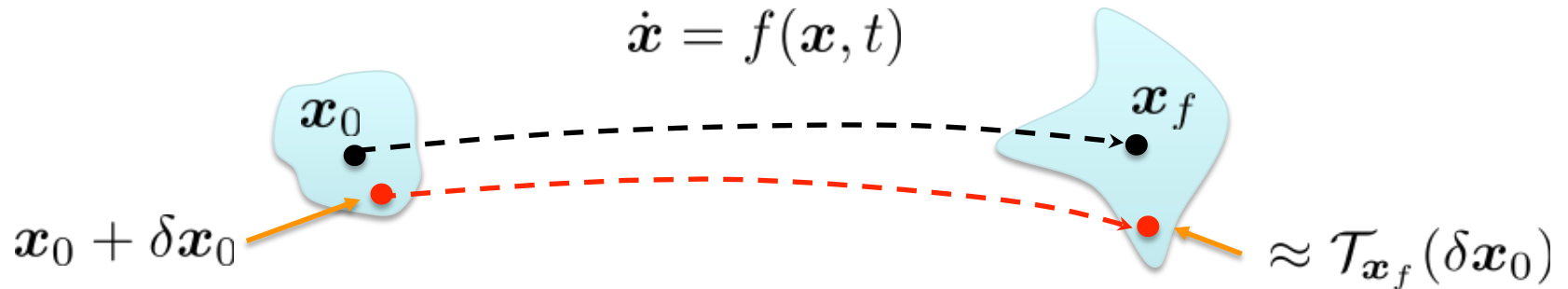


Map the initial  
covariance:

$$C_f = \Phi C_0 \Phi^T$$

No need of **variational equations!**

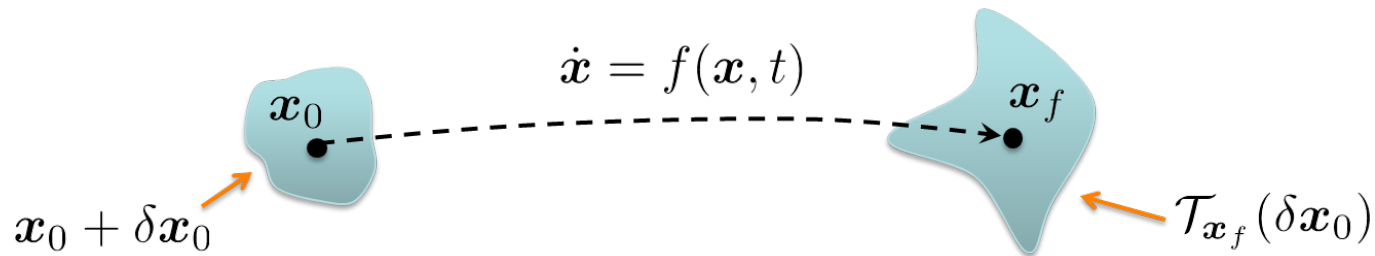
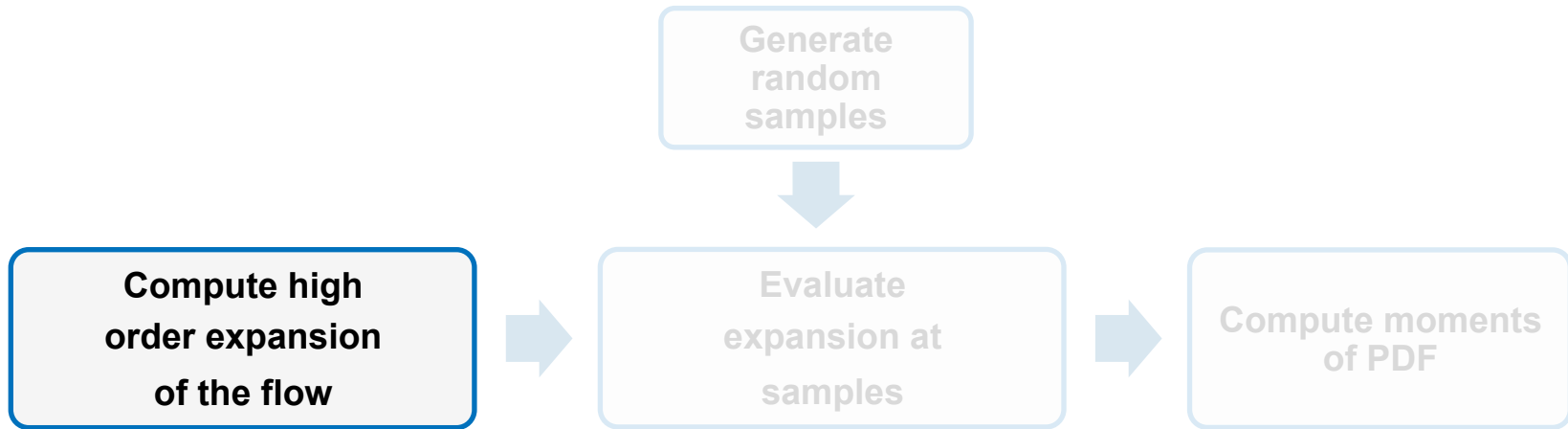


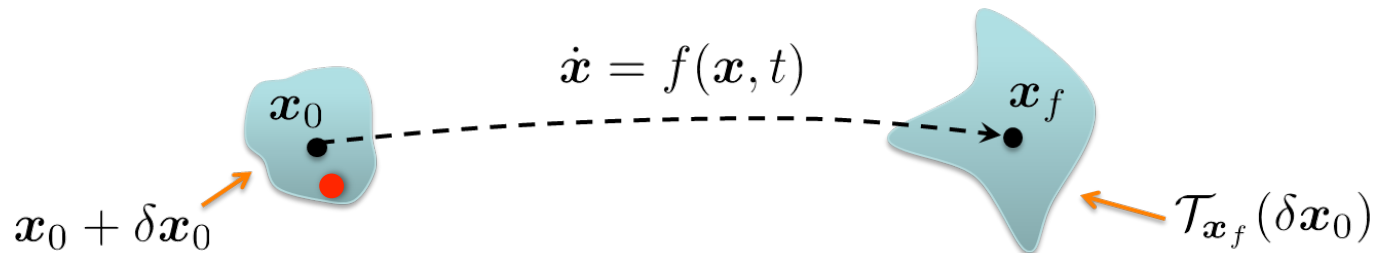
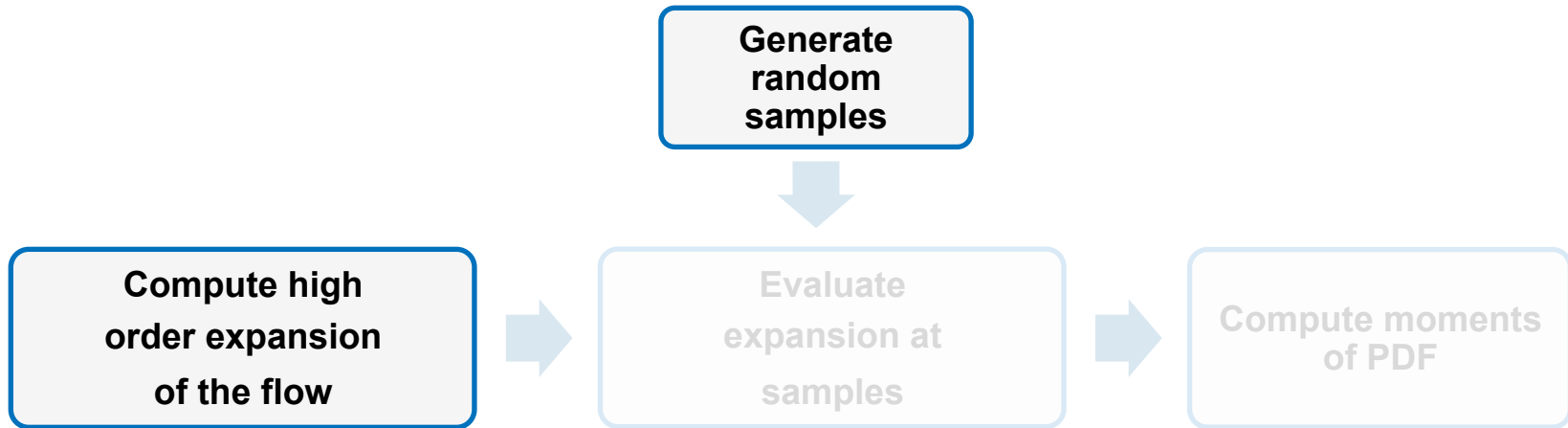


- Any pointwise integration can be replaced by the evaluation of the polynomial  $\mathcal{T}_{x_f}(\delta x_0)$

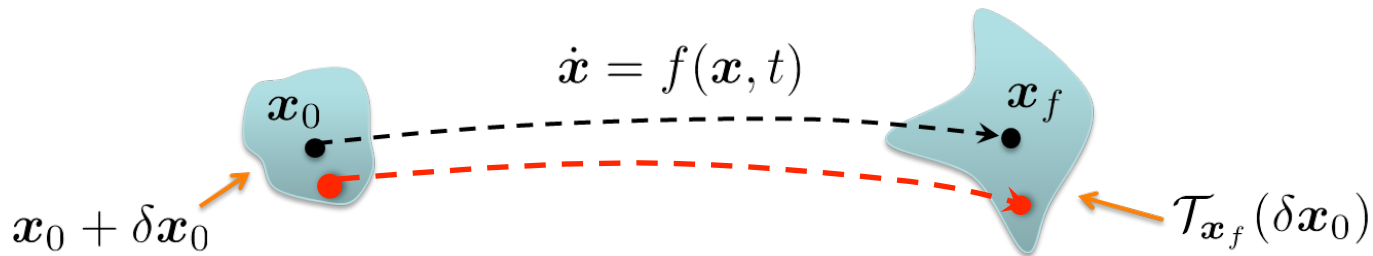
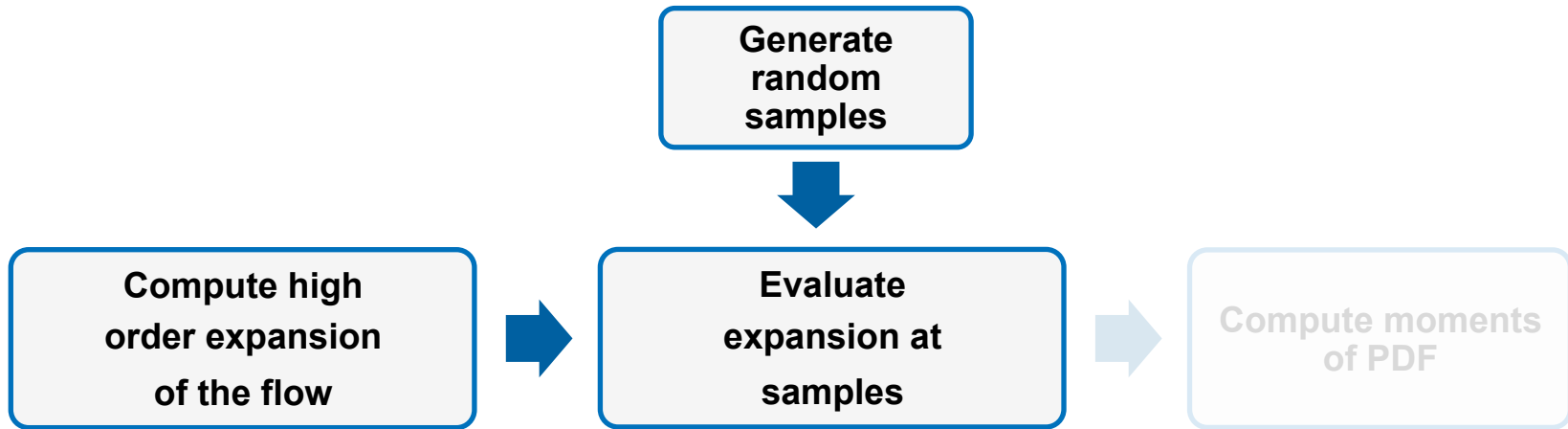


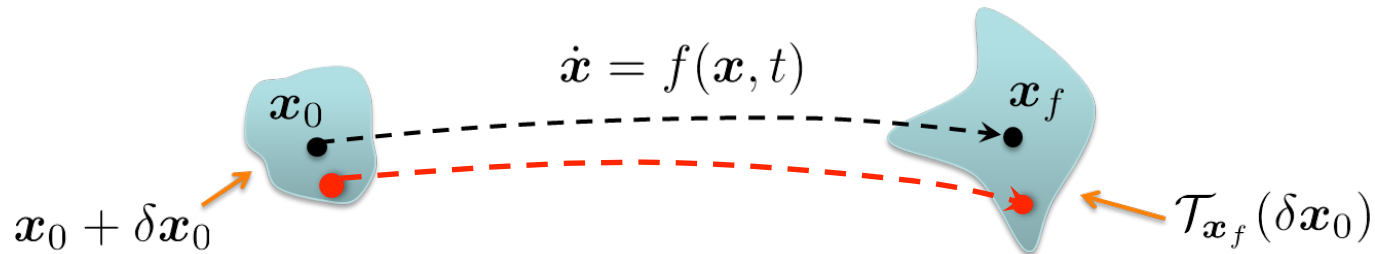
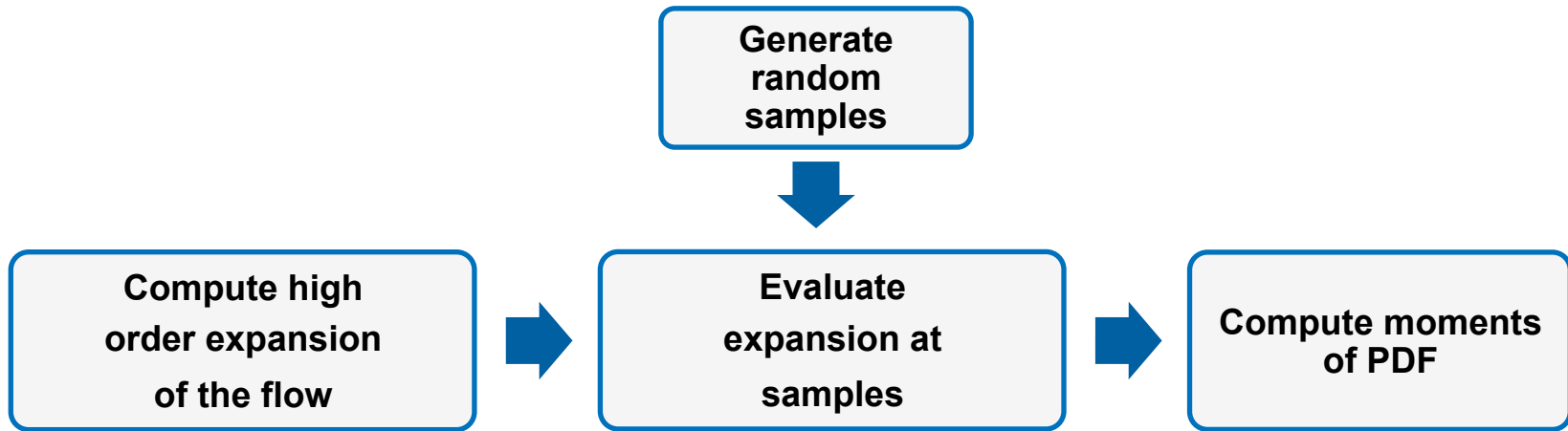
Saving in computational time w.r.t. classical MC

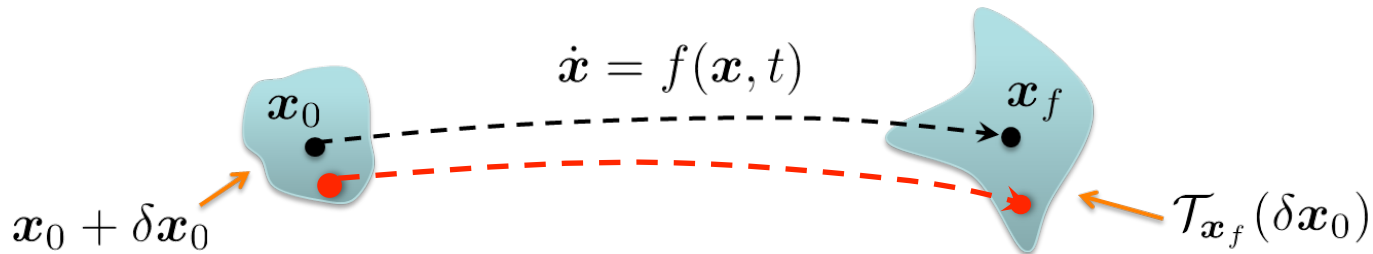
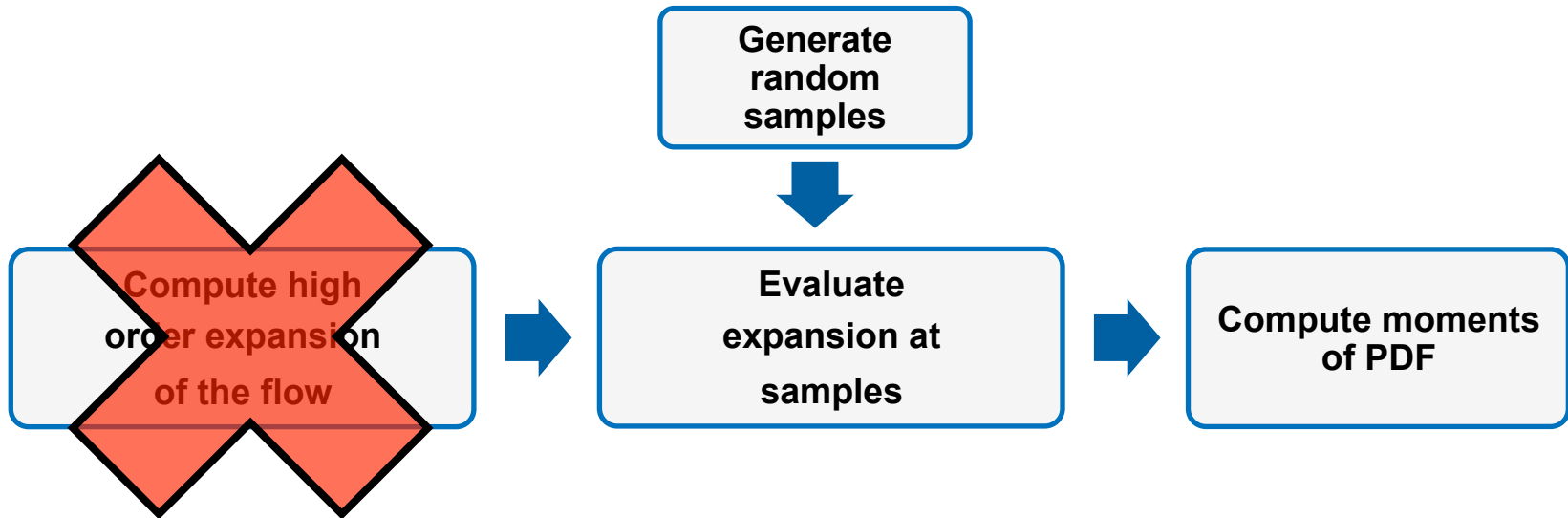






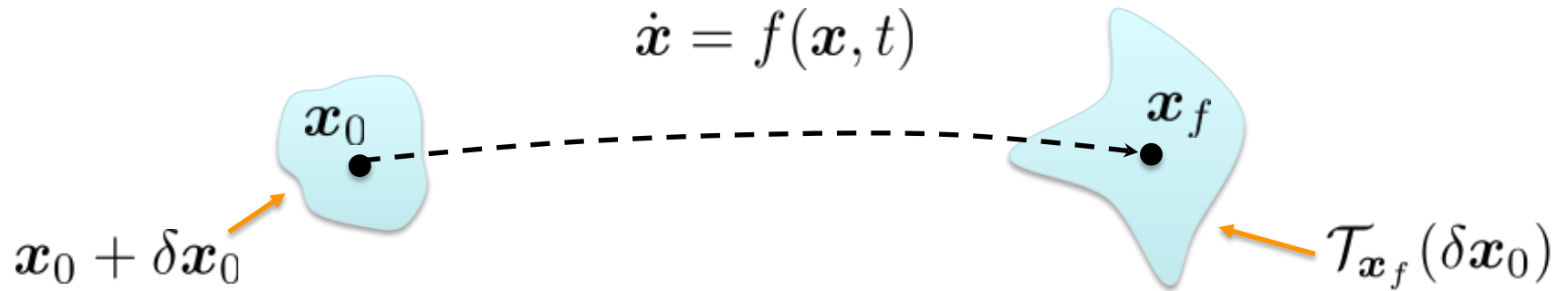


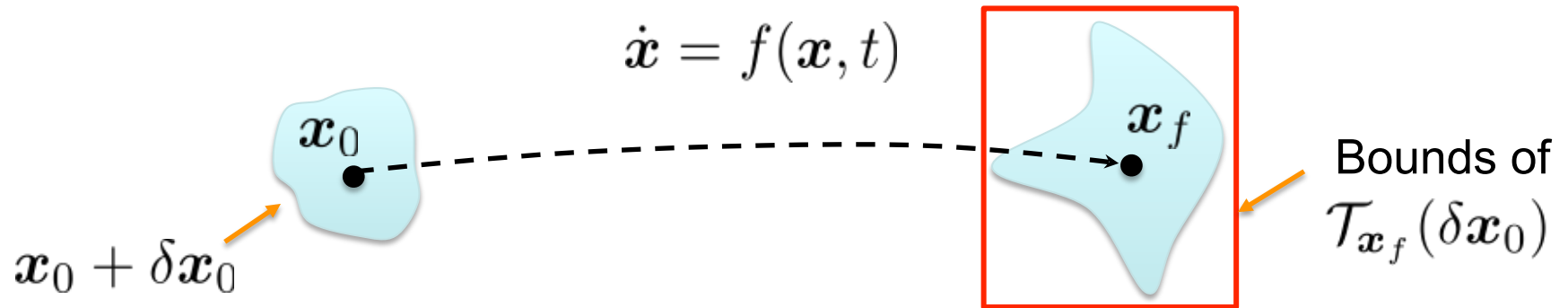




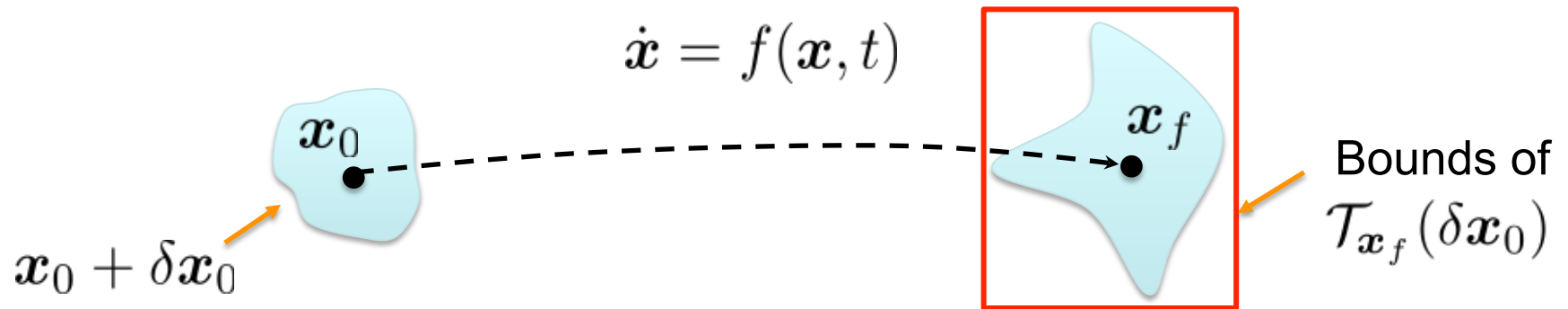
Same polynomial can be used to map different uncertainties

# POLYNOMIAL BOUNDER





- Use **polynomial bounders** to estimate the range of the propagated uncertainties by bounding  $\mathcal{T}_{x_f}(\delta x_0)$



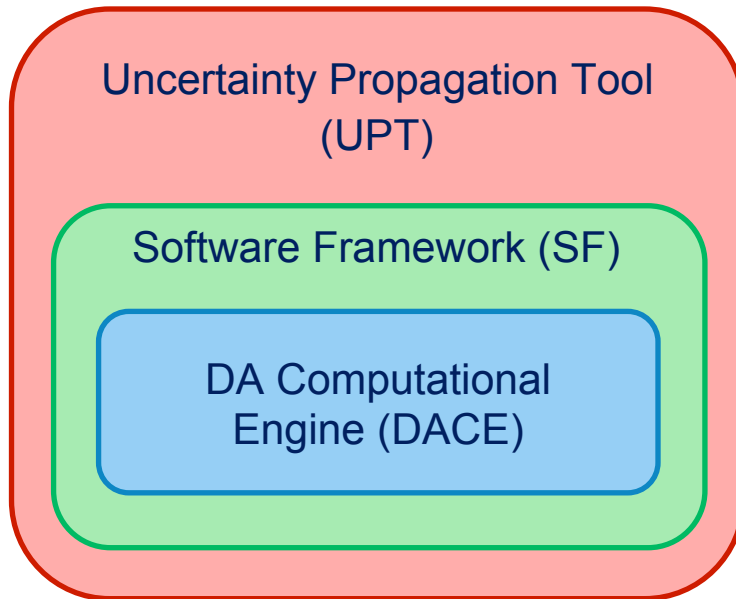
- Use **polynomial bounders** to estimate the range of the propagated uncertainties by bounding  $\mathcal{T}_{x_f}(\delta x_0)$




- Useful in some applications to **verify constraints satisfaction** (no need to map a statistical distribution)

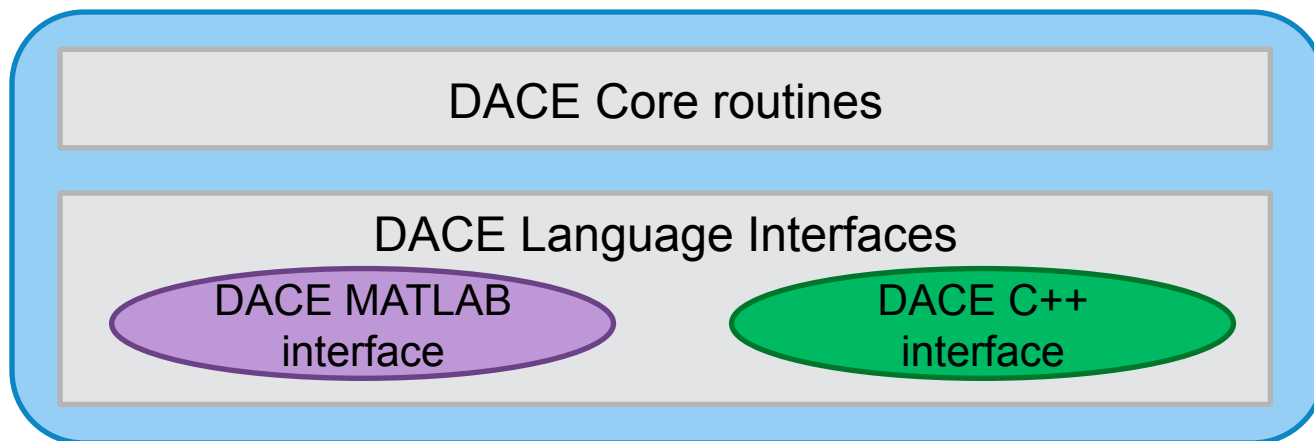


## Differential Algebra Space Toolbox



- **DA Computational Engine**   
Implements Taylor DA arithmetic to handle polynomial operations
- **Software Framework**  
Provides all routines to perform DA based propagation in astrodynamics
- **Uncertainty Propagation Tool**  
Provides all routines and an interface for DA-based propagation of uncertainties

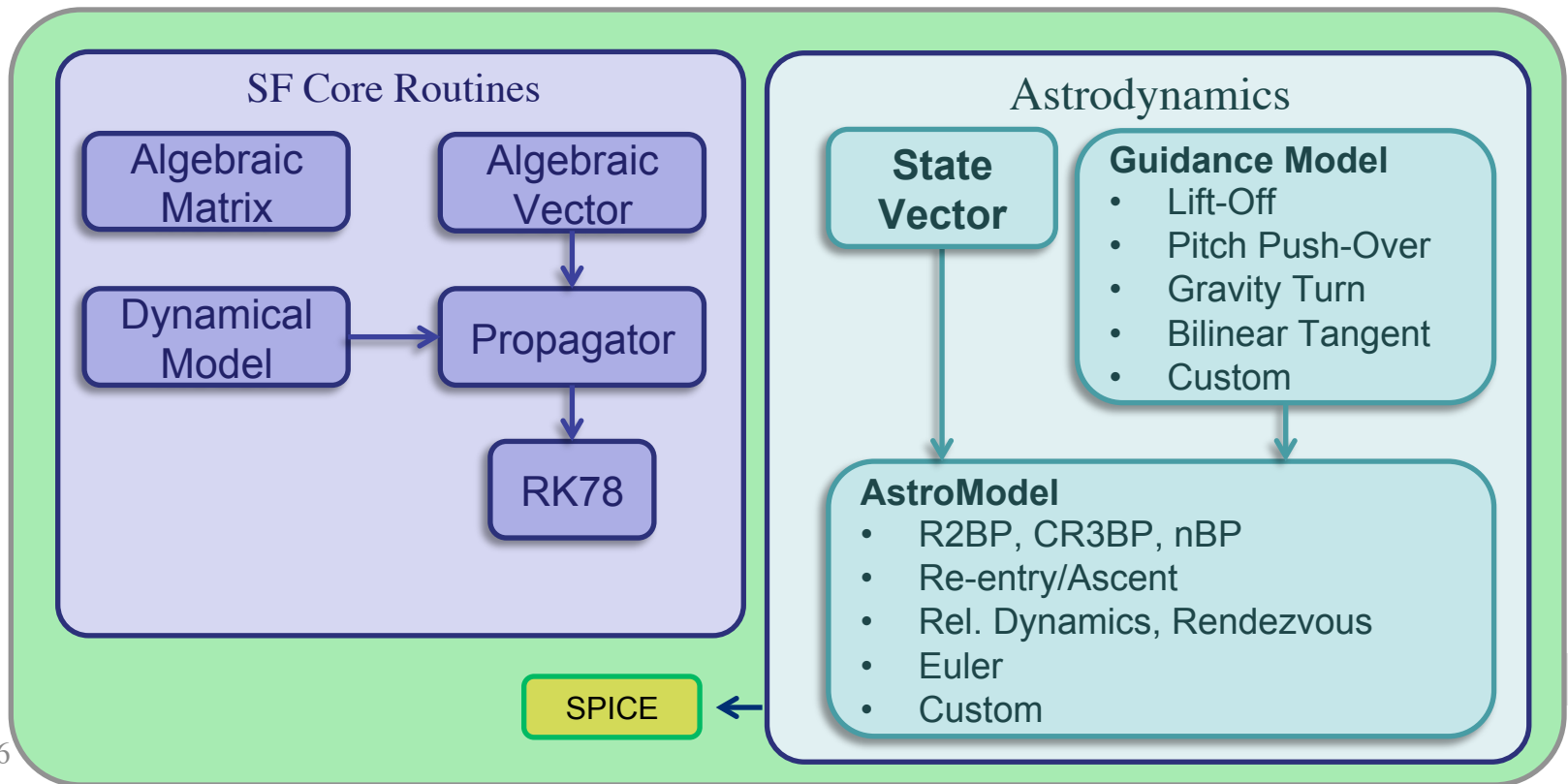
- The DA Computational Engine (DACE) contains **the implementation of the basic DA routines**
- DACE Core routines: **Fortran 95**
  - Initialization, Memory management, error handling, DA operations
  - Each routine approximates the result of an operation by its Taylor exp.
- Interfaces: **C++, MATLAB**



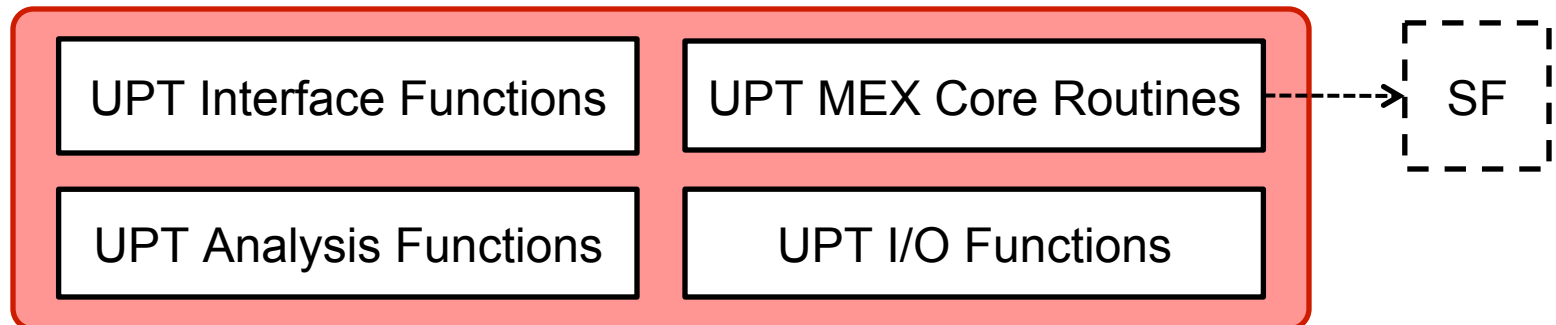


# SF MODULE OVERVIEW

- The SF includes more advanced features:
  - DA operations between **vectors** and/or **matrices** of DA
  - DA implementation of the numerical **integration schemes**
  - **Dynamical** models and **guidance** models



- The UPT is aimed at performing **uncertainty propagations** and **statistical analyses**
- The UPT comes with a set of Matlab routines to:
  - Easily **interface with the SF** (run DA based computations)
  - Managing simulation results
  - Easy **graphical representations** of the performed analyses



## Standard user



**Uncertainty  
Propagation Tool  
(UPT)**

Uncertainty  
propagation in various  
dynamical systems in  
space-related  
applications

## Developer



**Software Framework  
(SF)**

Implementation of  
custom DA-based  
astrodynamics  
applications

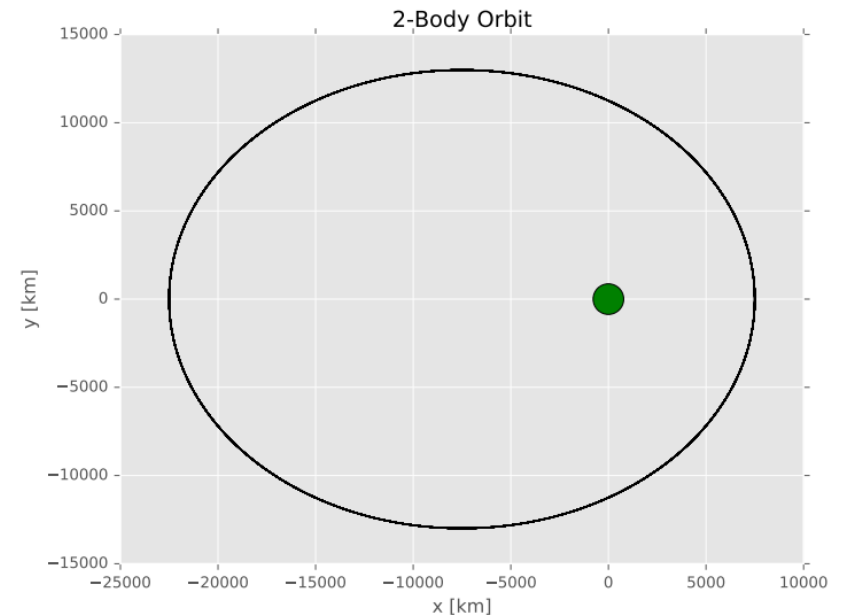
**DACE**

Use of DA  
operations for any  
custom application

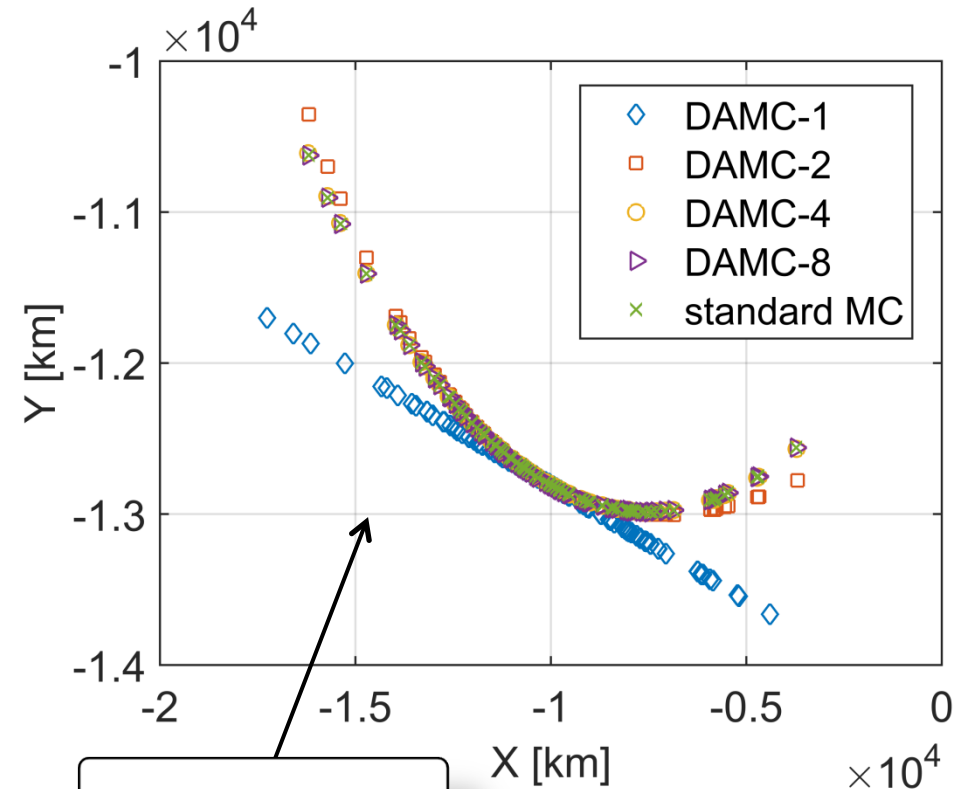
- Initial conditions

Initial Conditions			
State	Value	$\sigma$	Units
$r_x$	7.5e3	5e-2	km
$r_y$	0.0	5e-2	km
$r_z$	0.0	0.0	km
$v_x$	0.0	0.0	km/s
$v_y$	8.9286	0.0	km/s
$v_z$	0.0	0.0	km/s

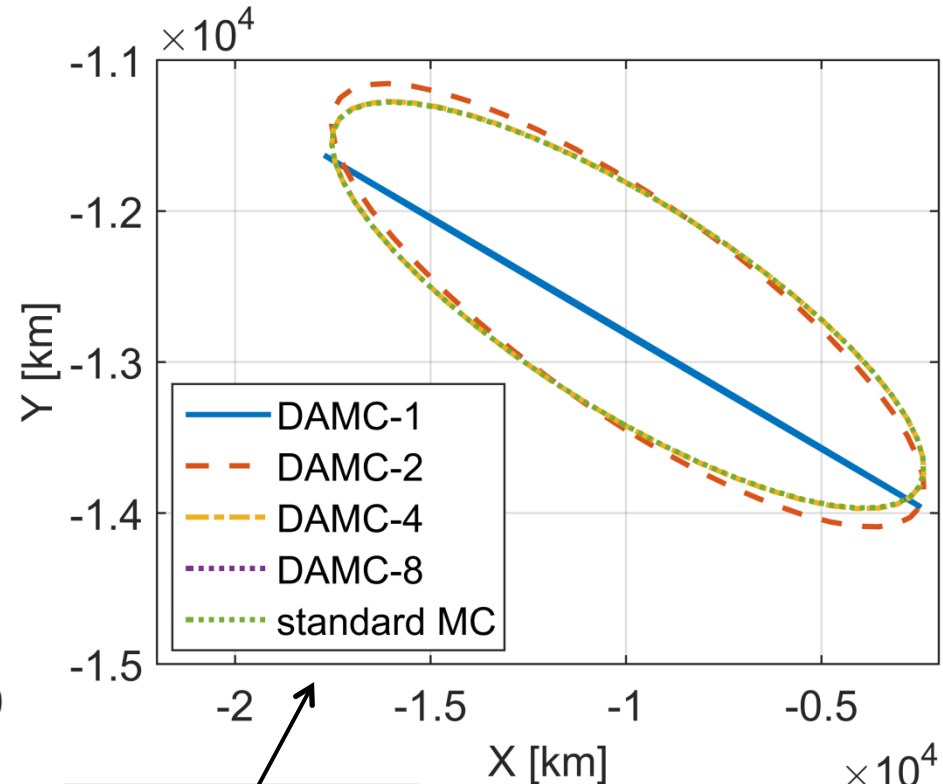
Propagation Time
$T_0 = 0.0$
$T_F = 30.8$ orbital periods



## DA based Monte Carlo

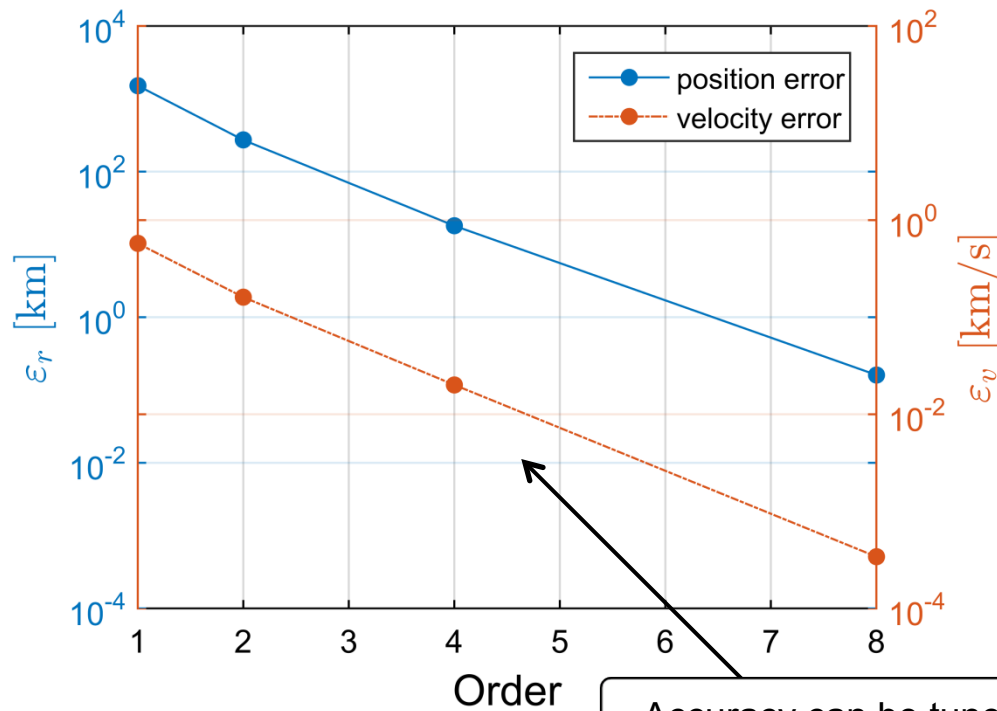


Final distribution



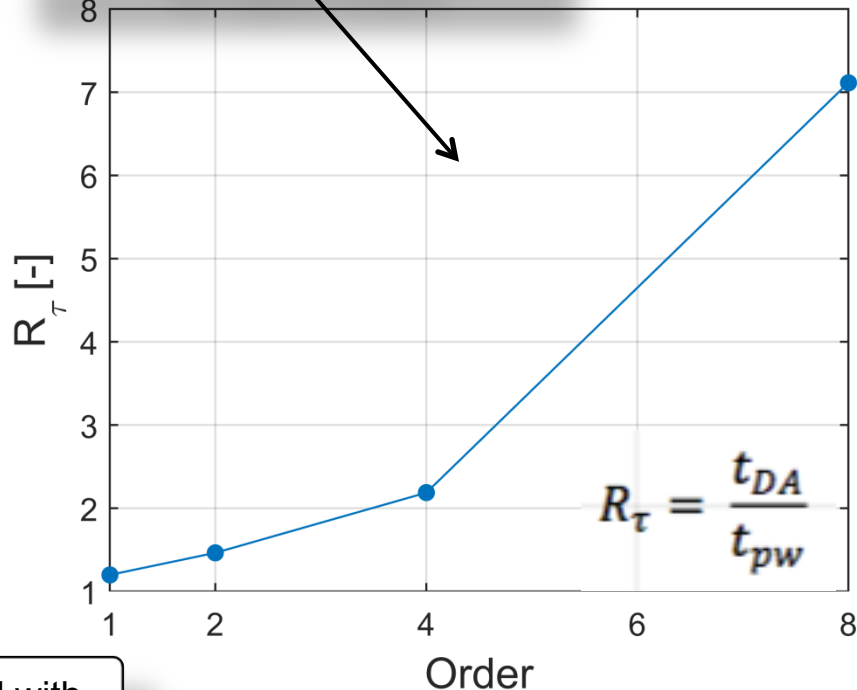
Final covariance  
ellipsoids

- DA based Monte Carlo



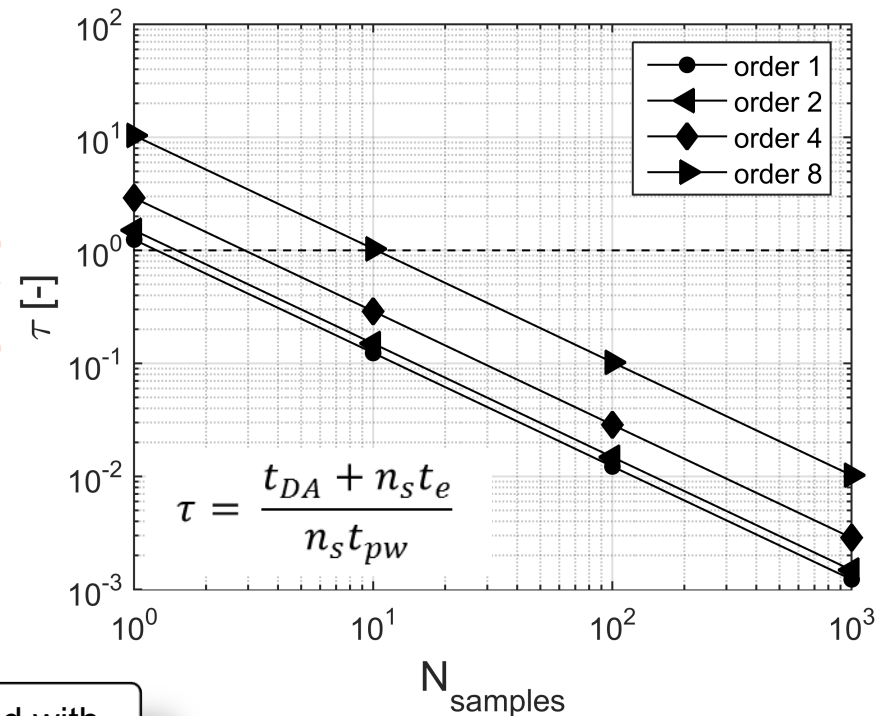
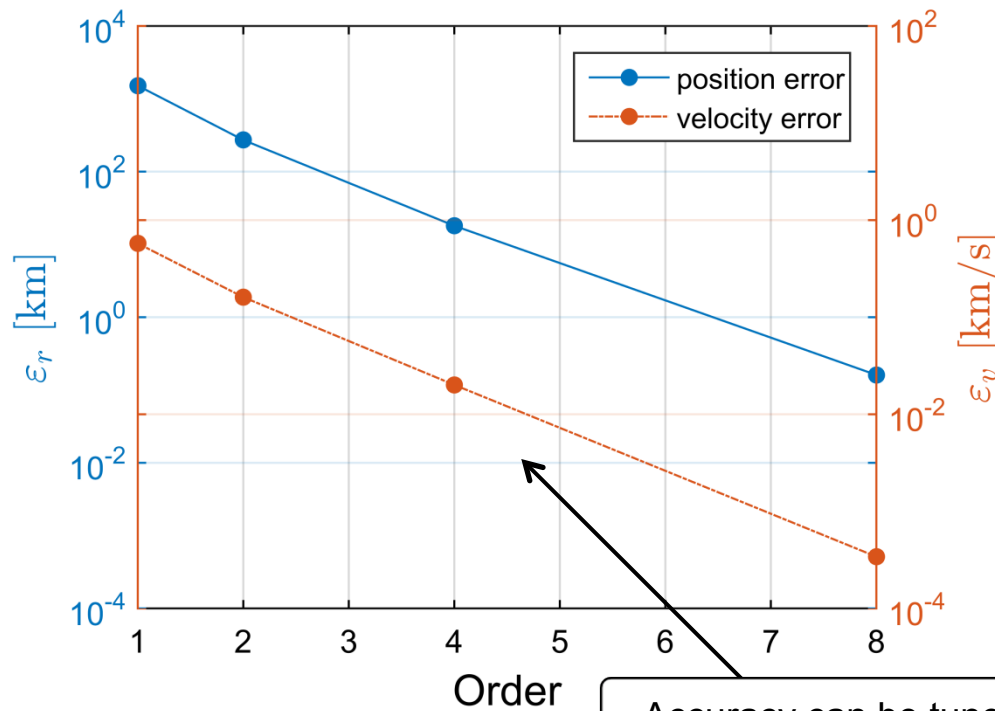
Accuracy can be tuned with the polynomial order

Computational cost increases for higher orders



$$R_\tau = \frac{t_{DA}}{t_{pw}}$$

- DA based Monte Carlo

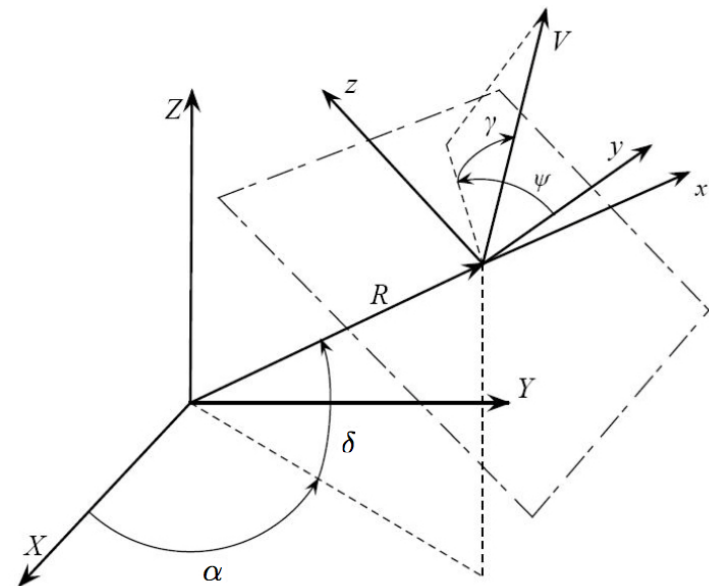


- Re-entry dispersion analysis for a ballistic re-entry vehicle

## Equations of motion

$$\left\{ \begin{array}{l} \dot{r} = v \sin \gamma \\ \dot{\alpha} = \frac{v \cos \gamma \cos \psi}{r \cos \delta} \\ \dot{\delta} = \frac{v \cos \gamma \sin \psi}{r} \\ \dot{v} = -\frac{D}{m} - g \sin \gamma \\ v\dot{\gamma} = \frac{L \cos \sigma}{m} - g \cos \gamma + \frac{v^2 \cos \gamma}{r} \\ v\dot{\psi} = \frac{L \sin \sigma}{m \cos \gamma} - \frac{v^2 \tan \delta \cos \gamma \cos \psi}{r} \end{array} \right.$$

## Reference frame



Exponential density model is used:  $\rho = \rho_0 e^{-\frac{h}{\beta}}$  (NRLMSISE available)



- Hayabusa re-entry\*

Initial Conditions			
State	Value	$\sigma$	Units
h	201.992	0.1085	km
$\alpha$	-124.28	0.0074	deg
$\delta$	-27.33	0.008	deg
v	12.035	0.002	km/s
$\gamma$	-12.35	0.0044	deg/s
$\psi$	-22.06	0.0119	deg/s

Model Parameters			
Parameter	Value	$\sigma$	Units
$C_D$	1.30	3.3%	-
$C_L$	0	-	-
m	18	-	kg
S	0.126	-	m <sup>2</sup>
$\rho_0$	1.217	6.6%	kg/m <sup>3</sup>
$\beta$	8.5	-	km

Propagation Time
$T_0 = 2010-06-13, 13:51:11.47$ UTC
$T_F = 2010-06-13, 15:00:00.00$ UTC

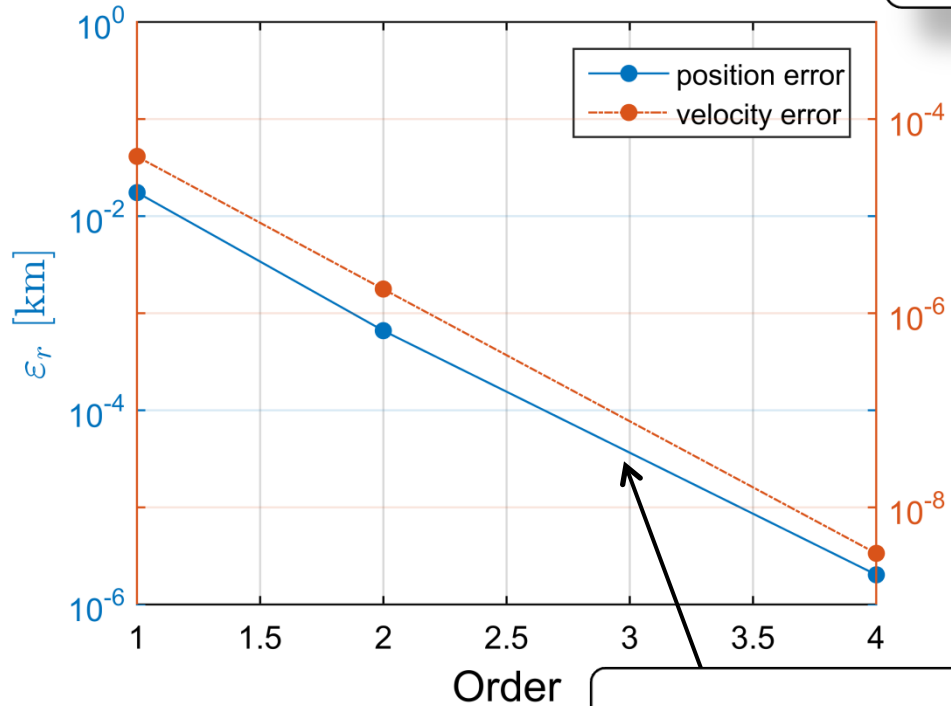
\*Cassel et al., 2011

**Propagation stopped at 25 km!**

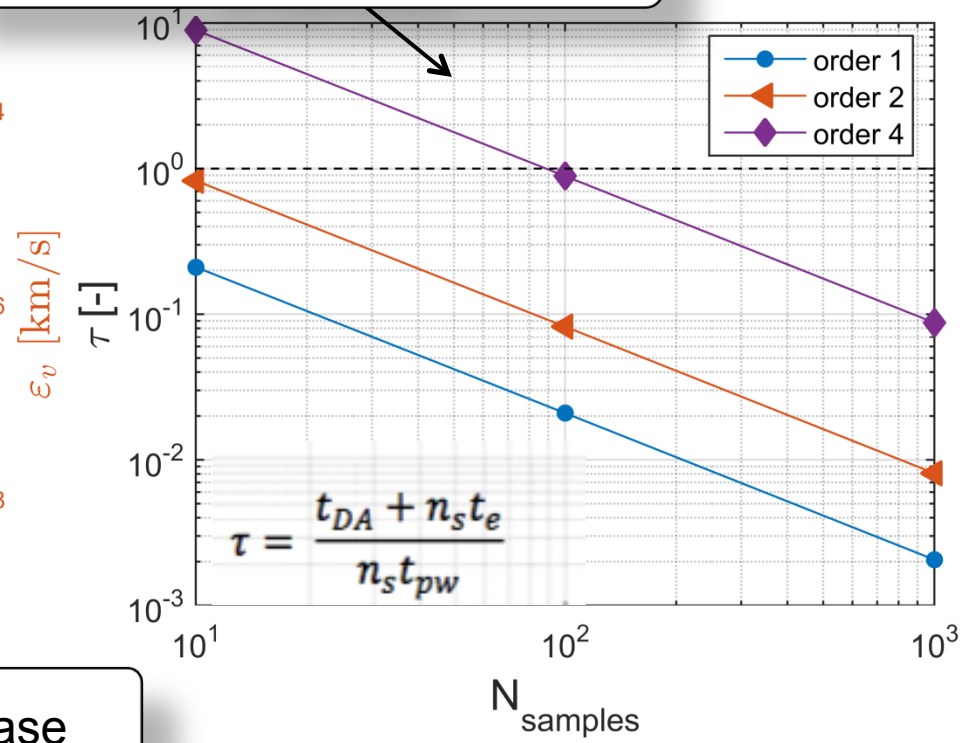
# APPLICATIONS: RE-ENTRY DYNAMICS

- DA based Monte Carlo

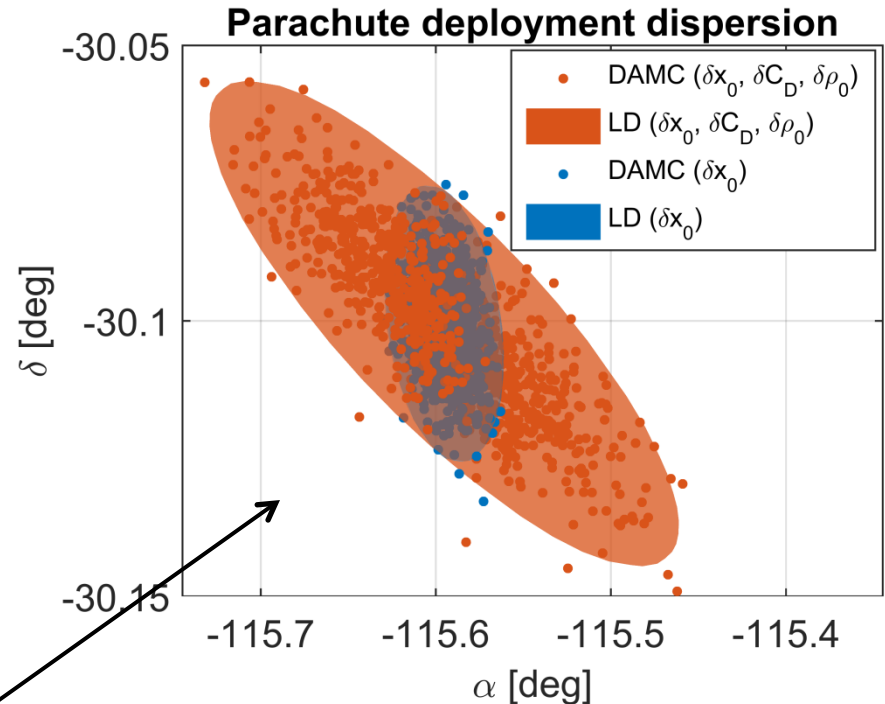
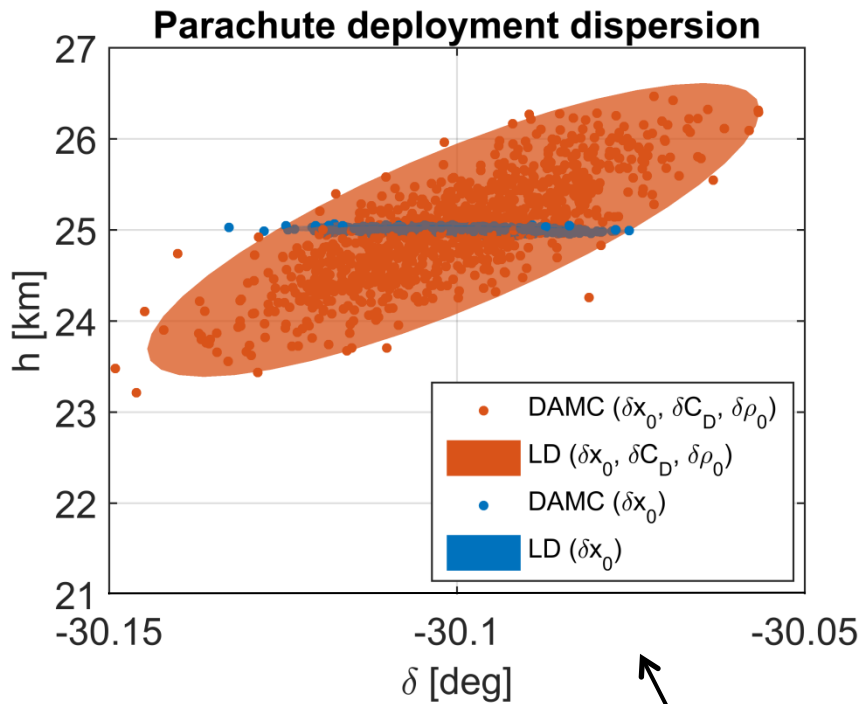
DAMC-k more efficient than classical MC for large number of samples



Errors decrease with higher orders



## ■ Covariance Propagation



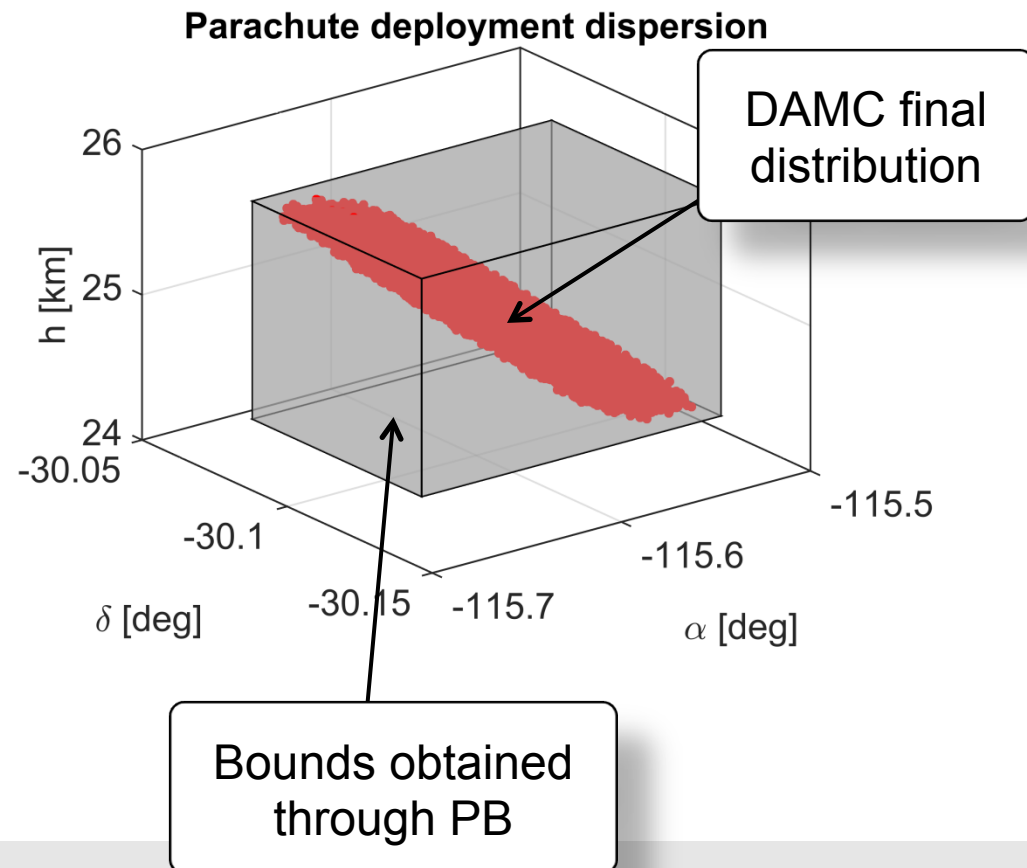
Uncertainty on density  $\rho_0$  and drag coefficient  $C_D$  ➡ increased dispersion

## Polynomial Bounder

$$\varepsilon_{LB} = \frac{|\min(x_{f,DAMC}) - x_{f,LB}|}{x_{f,LB}}$$

$$\varepsilon_{UB} = \frac{|\min(x_{f,DAMC}) - x_{f,UB}|}{x_{f,UB}}$$

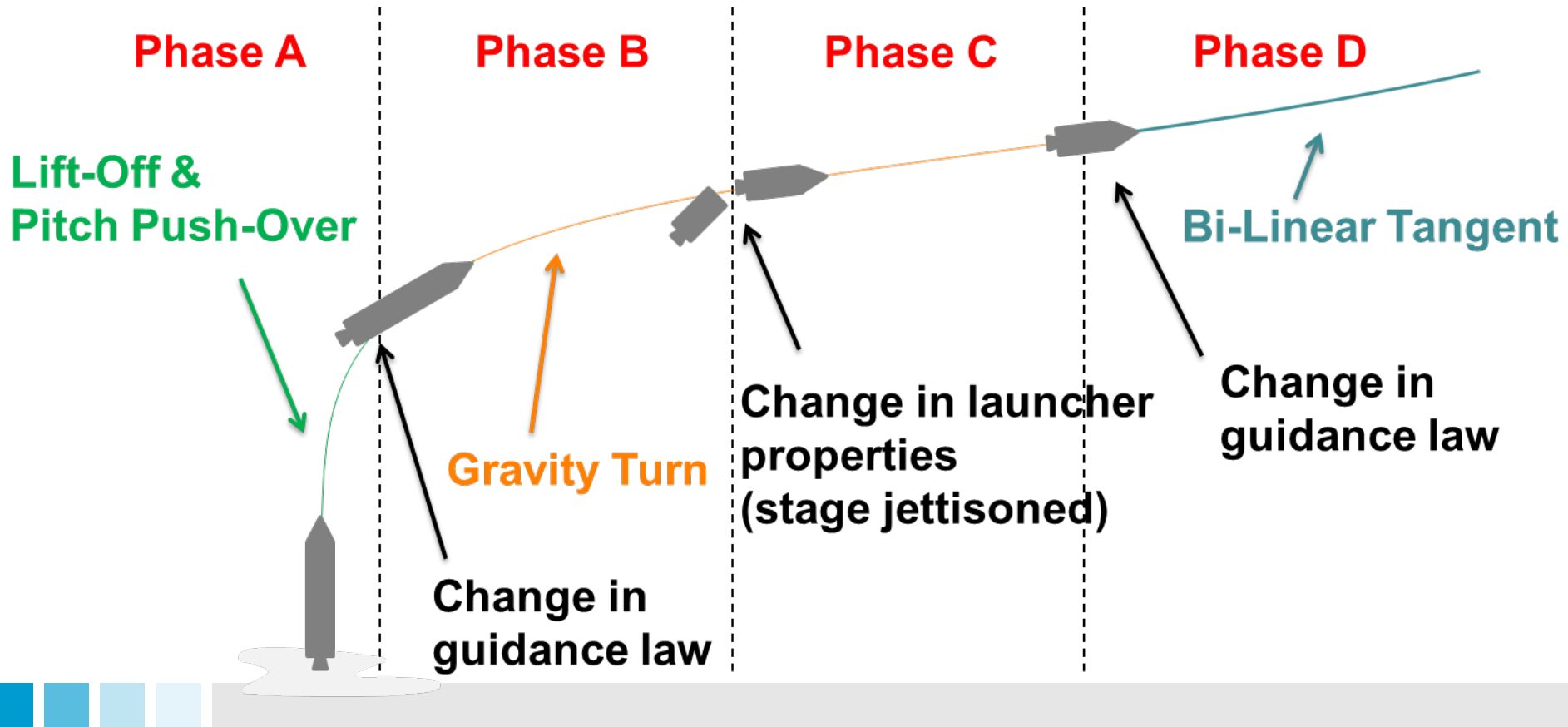
#samples	max( $\varepsilon_{LB}$ )	max( $\varepsilon_{UB}$ )
10 <sup>1</sup>	8.457e-03	4.104e-03
10 <sup>2</sup>	3.215e-03	1.979e-03
10 <sup>3</sup>	1.338e-03	8.025e-04
10 <sup>4</sup>	7.859e-04	4.634e-04
10 <sup>5</sup>	3.355e-04	4.457e-04



- The dynamical models available in the DAST are:
  - Two-Body
  - Three-Body
  - N-body
  - Re-entry
  - Relative
  - Attitude
  - Ascent
  - Rendezvous
  - Custom

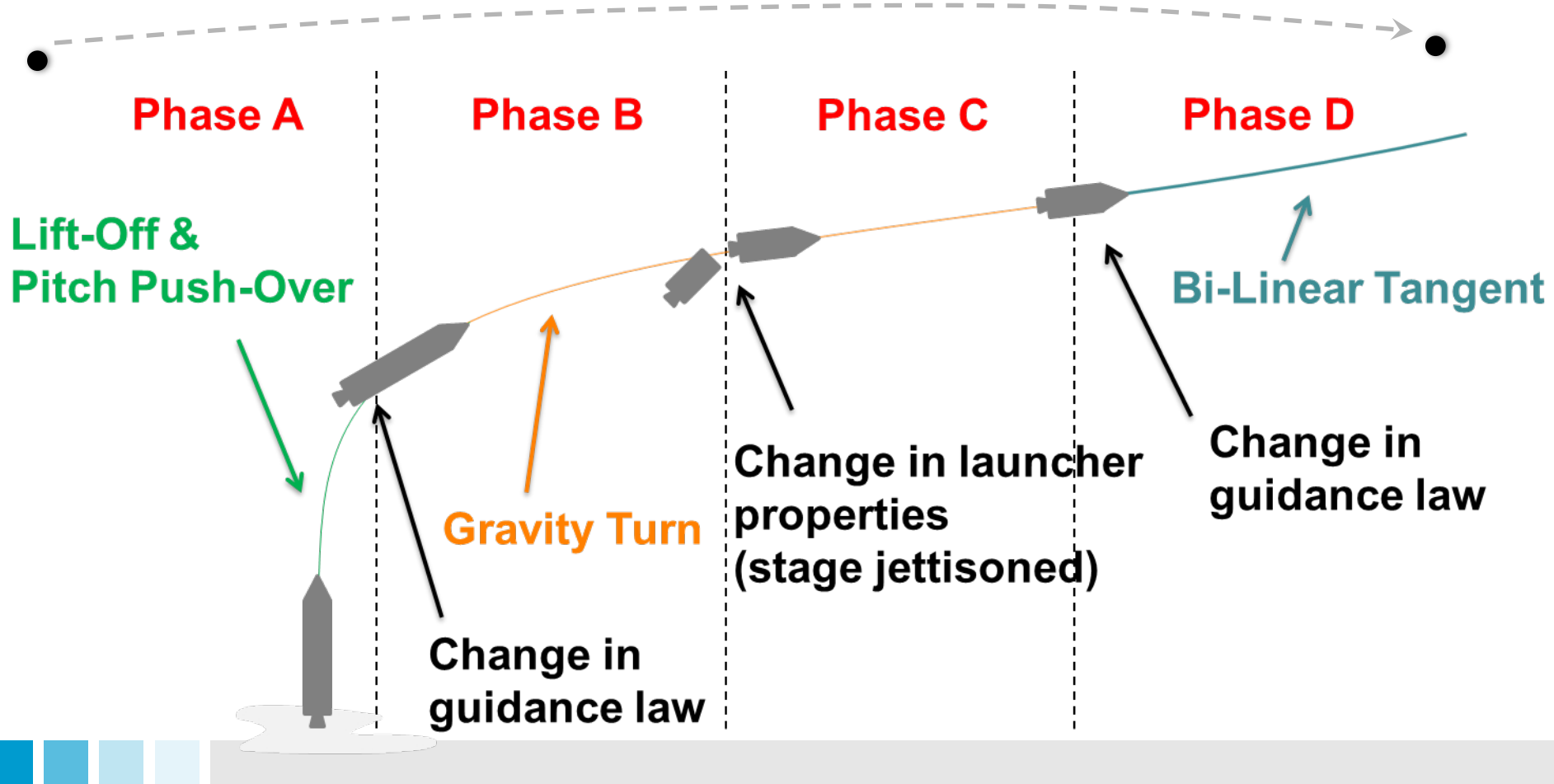
*Multi-phase approach*

# MULTI-PHASE: ASCENT DYNAMICS

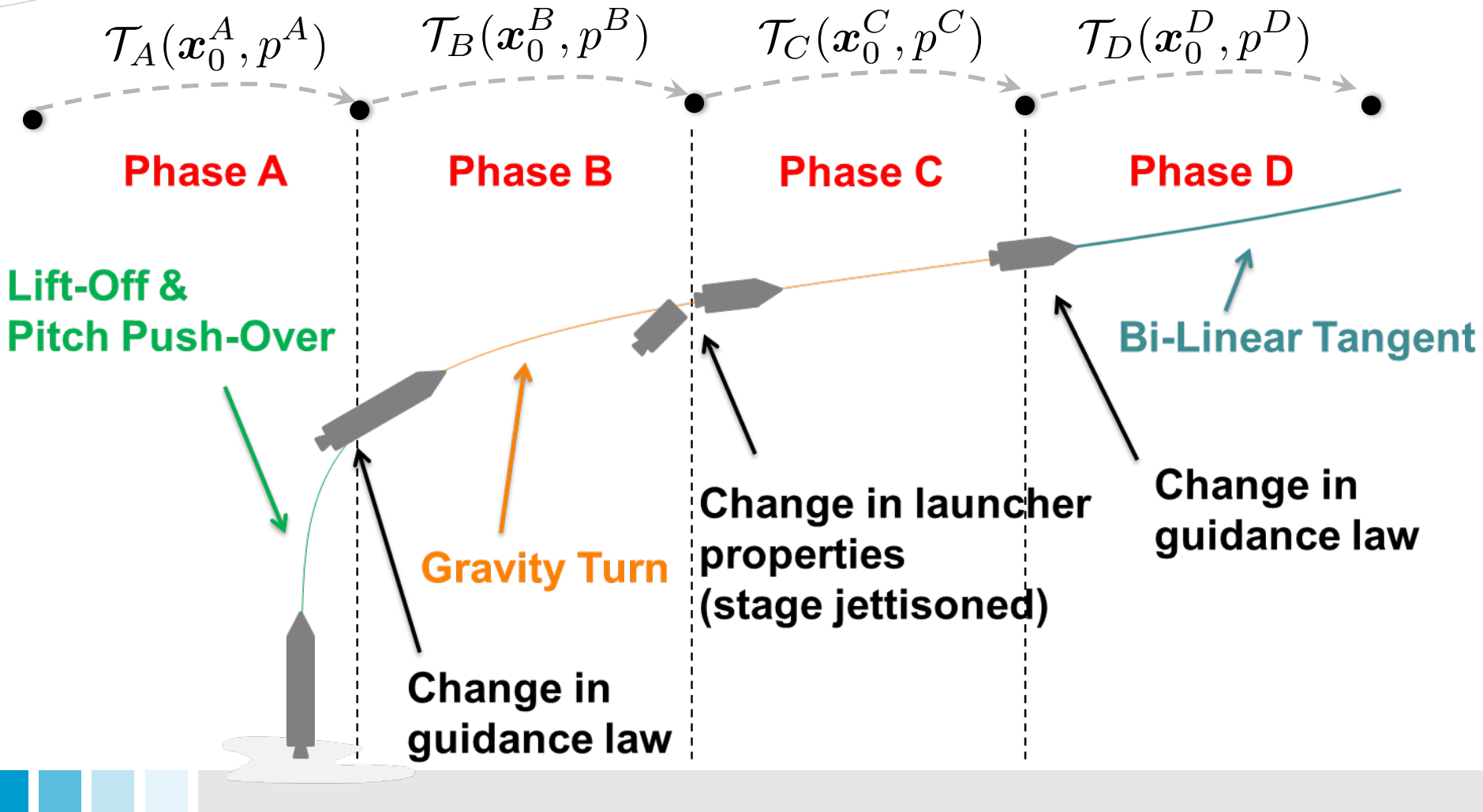


# MULTI-PHASE: ASCENT DYNAMICS

$$\mathcal{T}(x_0, p^A, p^B, p^C, p^D)$$



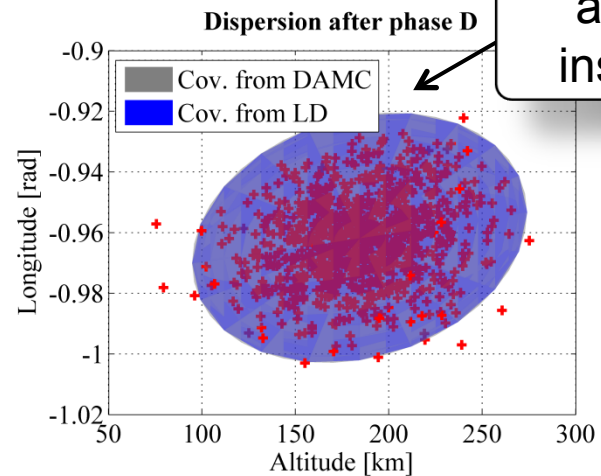
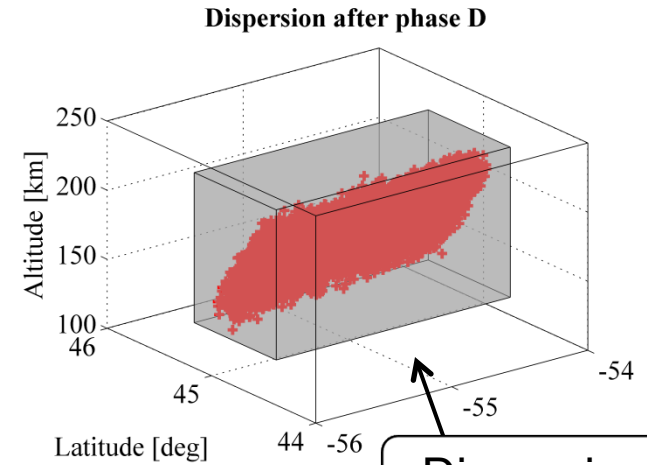
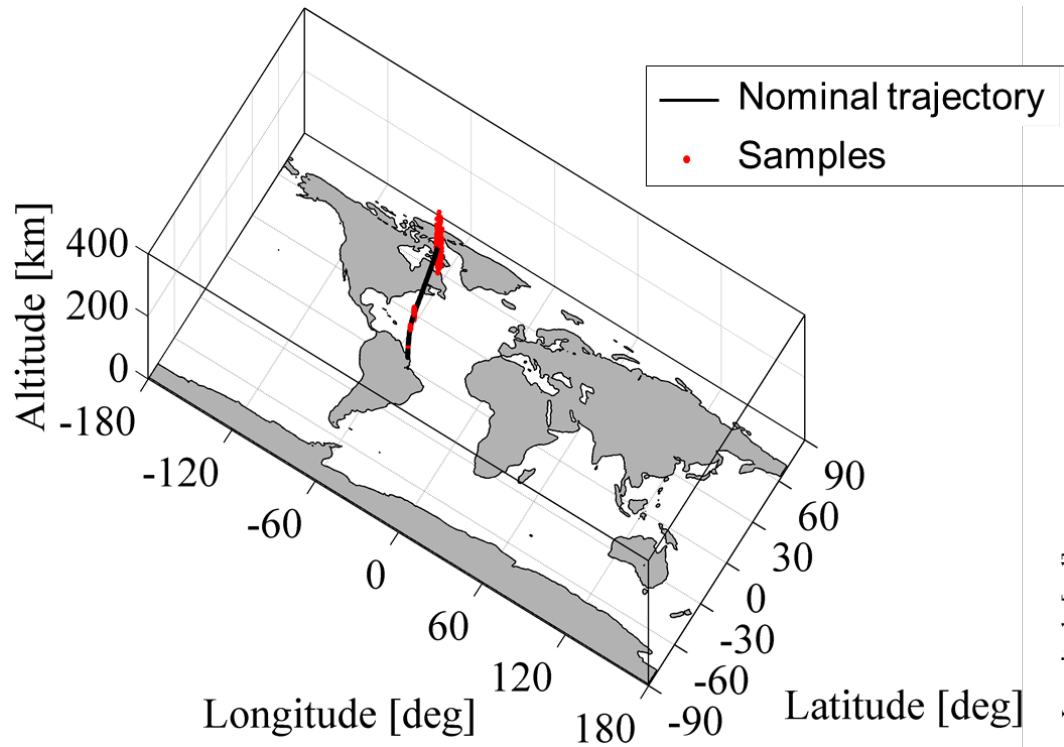
# MULTI-PHASE: ASCENT DYNAMICS






# MULTI-PHASE: ASCENT DYNAMICS

## ■ VEGA launcher



Dispersion at orbit insertion

- DAST is an efficient tool for nonlinear uncertainty propagation:
  - Propagations can be run on several **ready-to-use dynamical models** and any **DA-compatible custom dynamical model**
  - **More efficient than standard Monte Carlo** for typical number of samples
  - Analytical information available at the end of the propagation
- **Note:** method based on Taylor approximations  **Size of uncertainty set and order shall guarantee sufficient accuracy**
- If linear methods are sufficiently accurate for your application, you may not need to increase order, however...
- ...DA relieves you from the “pain” of **writing variational equations**

# ***DIFFERENTIAL ALGEBRA SPACE TOOLBOX\**** ***FOR NONLINEAR UNCERTAINTY PROPAGATION IN*** ***SPACE DYNAMICS***

*6<sup>th</sup> International Conference on Astrodynamics Tools and Techniques (ICATT)*  
*Darmstadt, Germany*

M. Rasotto, A. Morselli, A. Wittig, M. Massari,  
P. Di Lizia, R. Armellin, C. Y. Valles and  
G. Ortega

March 17<sup>th</sup>, 2016

*\* Developed under ITT AO/1-7570/13/NL/MH: Nonlinear Propagation of Uncertainties in Space Dynamics based on Taylor Differential Algebra"*