

# ***DESIGN OF OPTIMAL OBSERVATION STRATEGY FOR RE-ENTRY PREDICTION IMPROVEMENT OF GTOs UPPER STAGES***

*6<sup>th</sup> International Conference on Astrodynamics Tools and Techniques (ICATT)  
Darmstadt, Germany*

G. Di Mauro, M. Rasotto, M. Massari,  
P. Di Lizia, R. Armellin, Q. Funke and T. Flohrer

March 16<sup>th</sup>, 2016

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Improvement of their re-entry prediction is a key issue

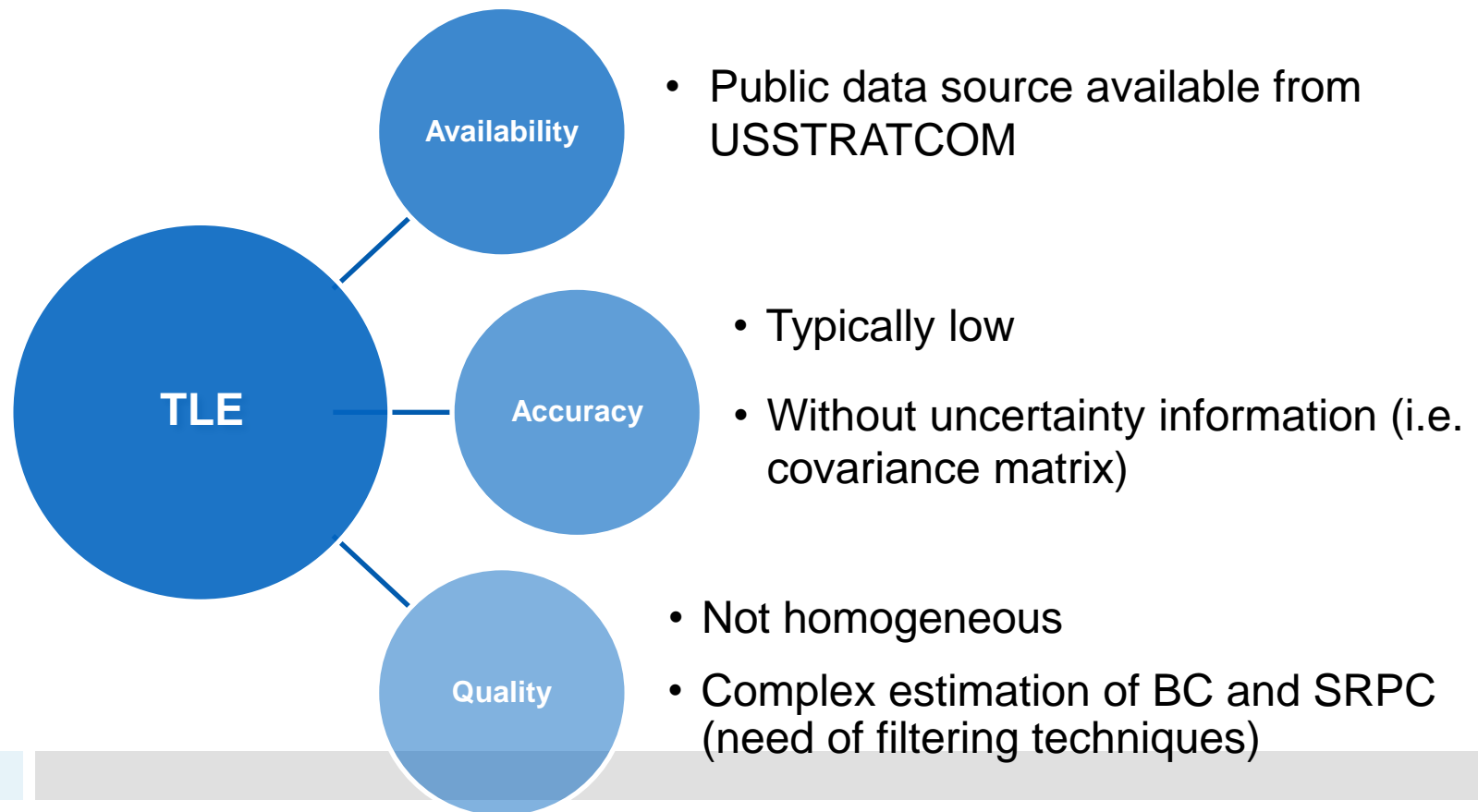
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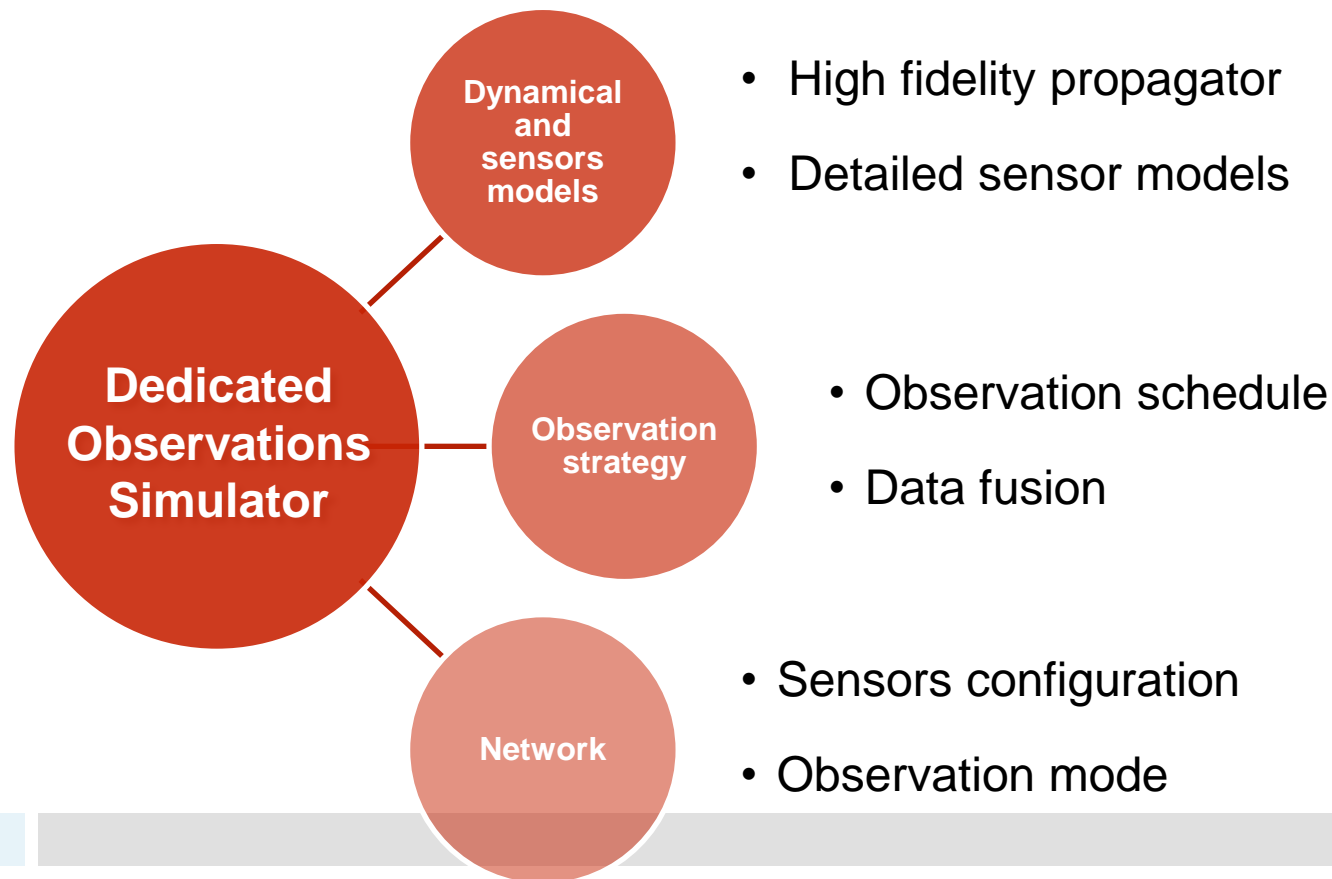
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*"ITT AO/1-8155/15/D/SR Technology for Improving Re-Entry Predictions of European Upper Stages through Dedicated Observations"*

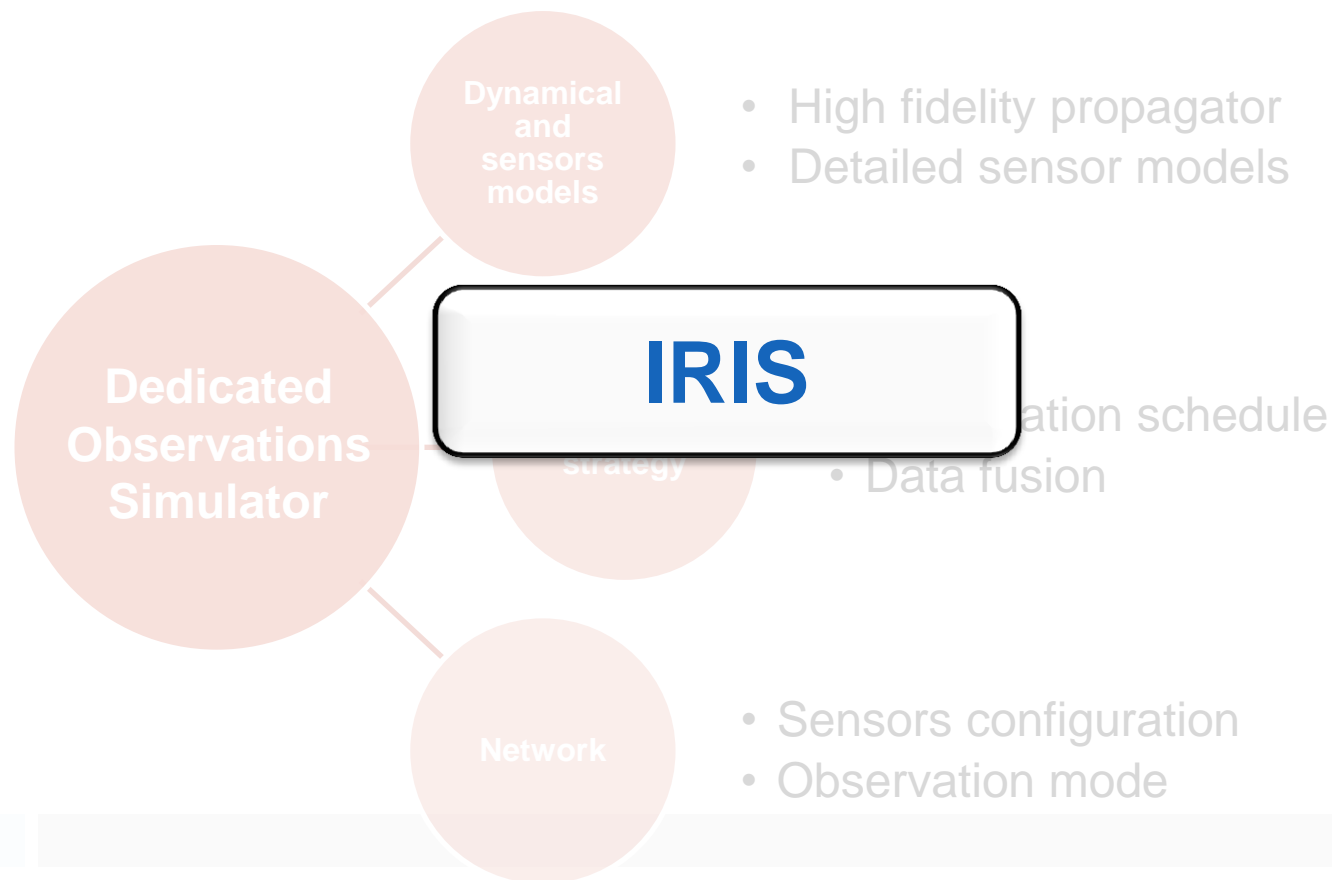
- Improvement based on Two Line Elements (TLE)



- Improvement through dedicated observations



- Improvement through dedicated observations



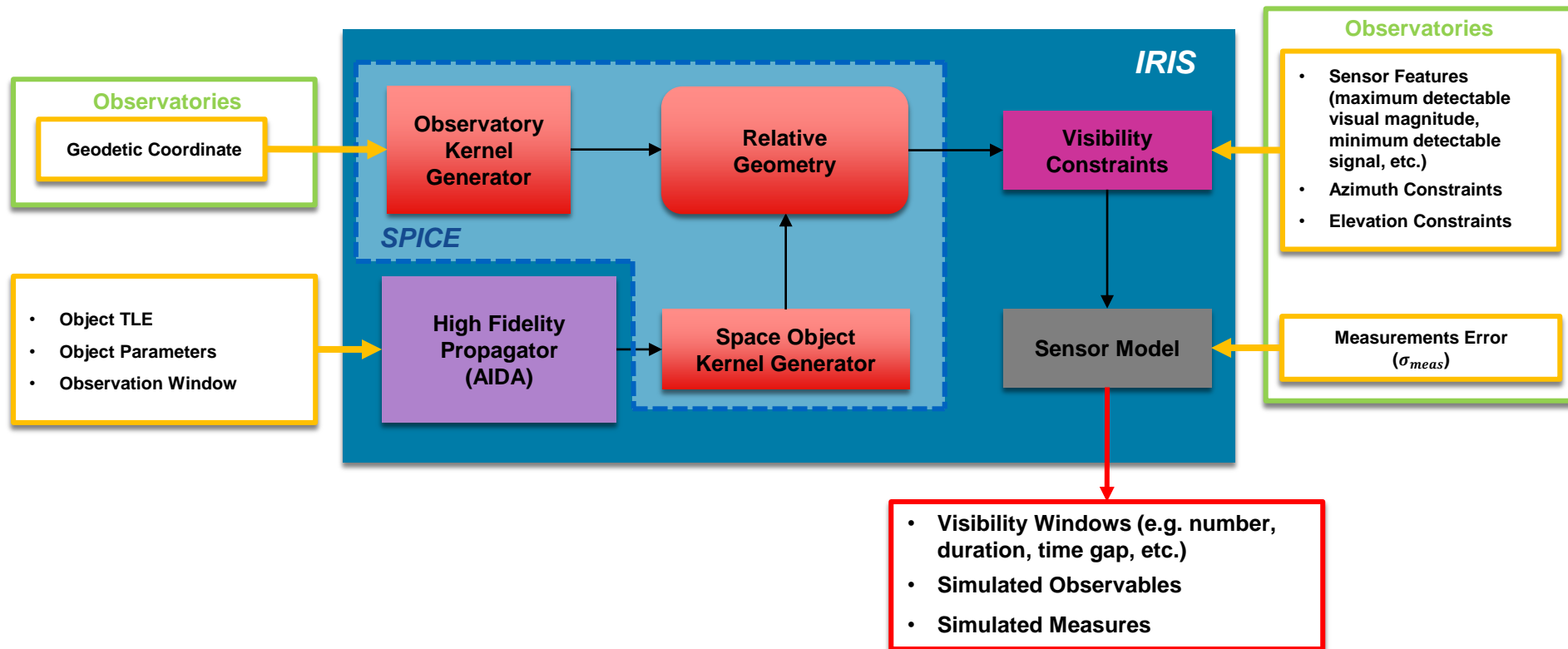


## IRIS

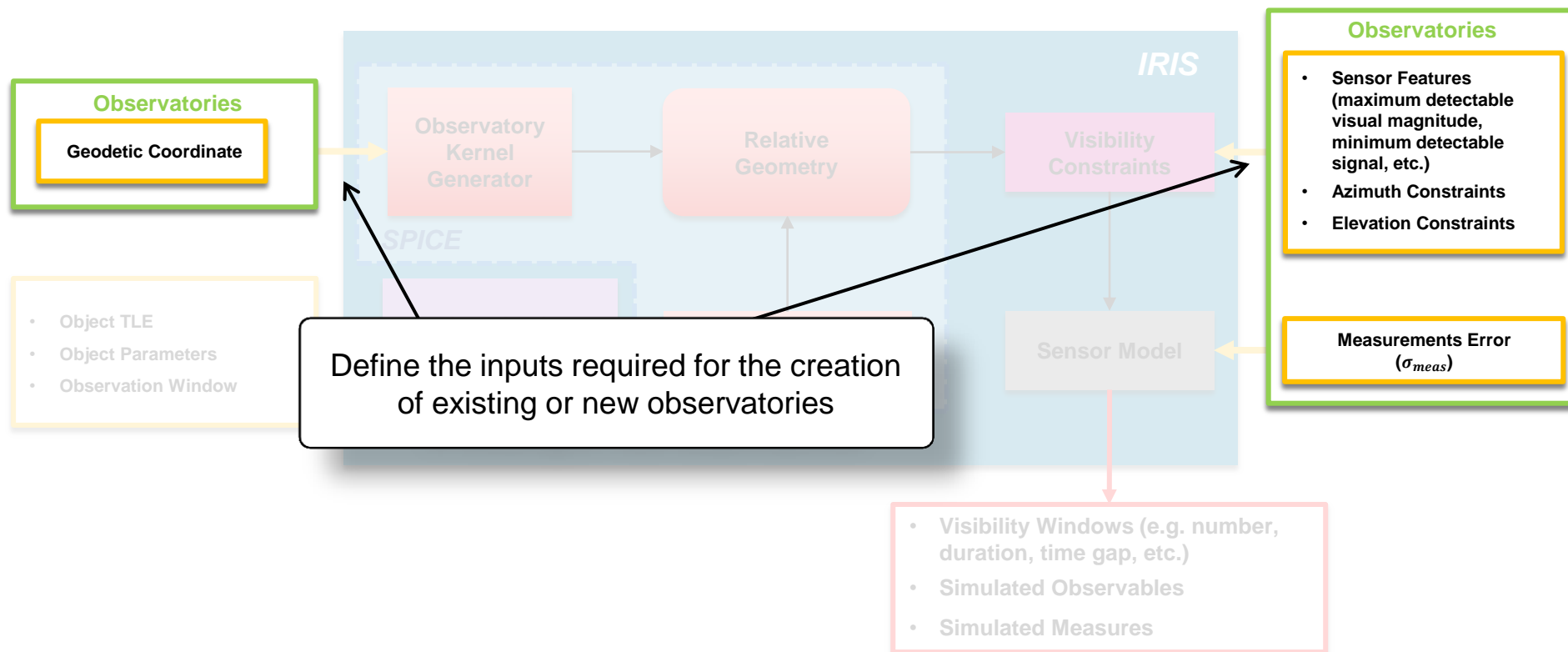
### ■ Goals:

- Accurate simulation of observation campaigns through existing and ideal sensors
- Computation of visibility windows, observables, ground track coverage etc.
- Design of observation strategies for the improvement of re-entry predictions

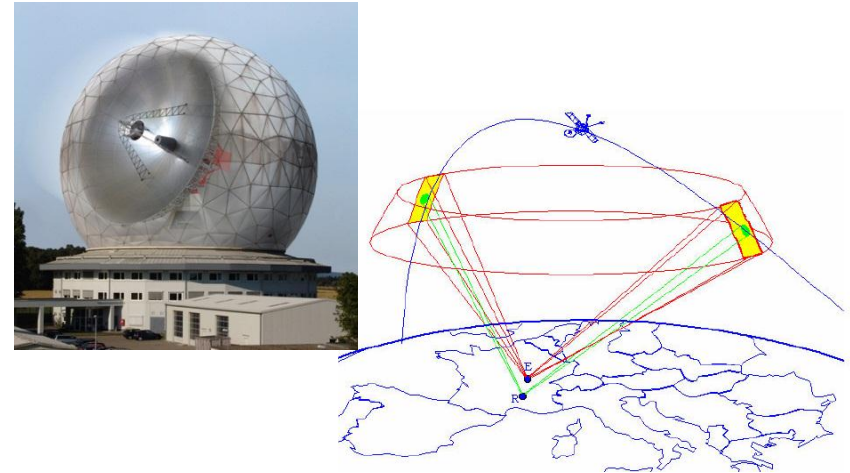
- Fully implemented in Matlab



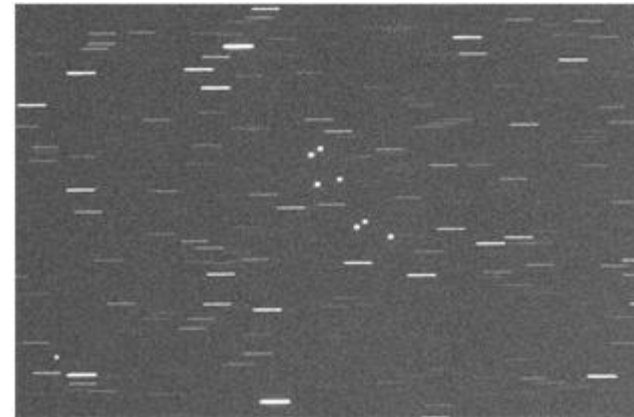
- Observatory class



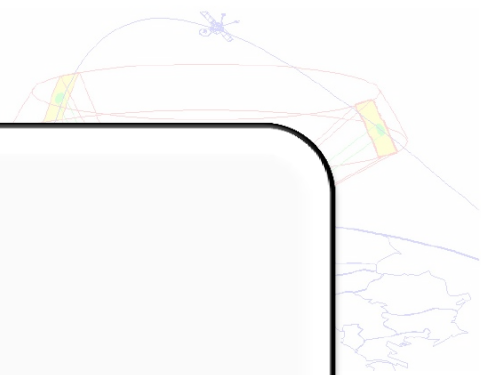
- **Active systems** such as the radar and LADAR (Lidar radar) illuminate space objects with radio (VHF/UHF) or laser beams:  $d^{-4}$  dependency



- **Passive systems** such as optical sensors which use the SUN as source of illumination:  $d^{-2}$  dependency



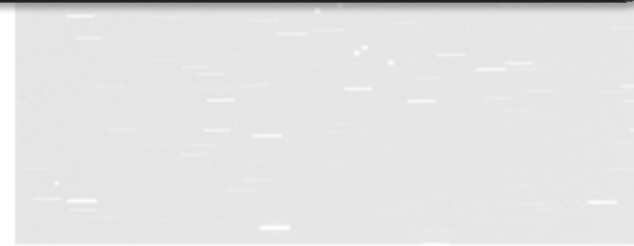
- Active systems such as the radar and LADAR (Lidar radar)



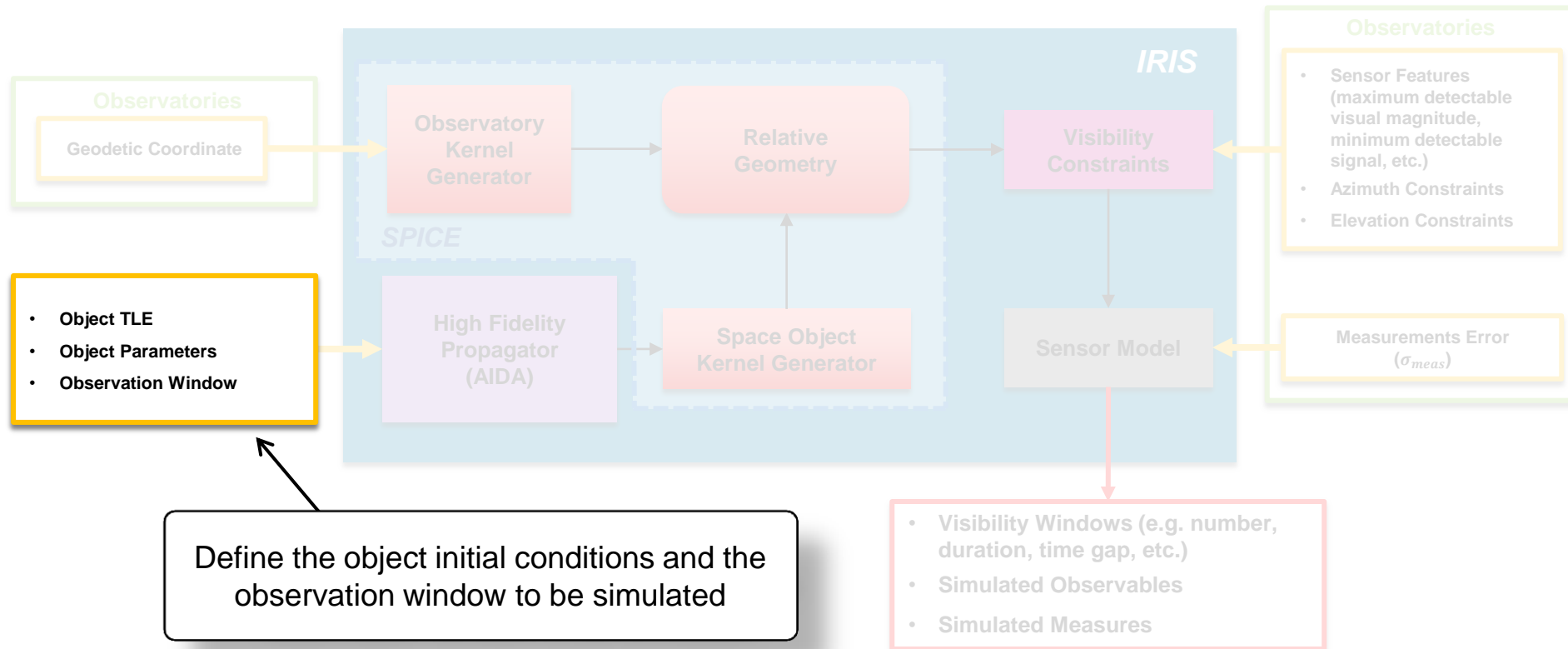
Ideal observatories can also be defined:

- Laser sensors
- Infrared sensors
- ...

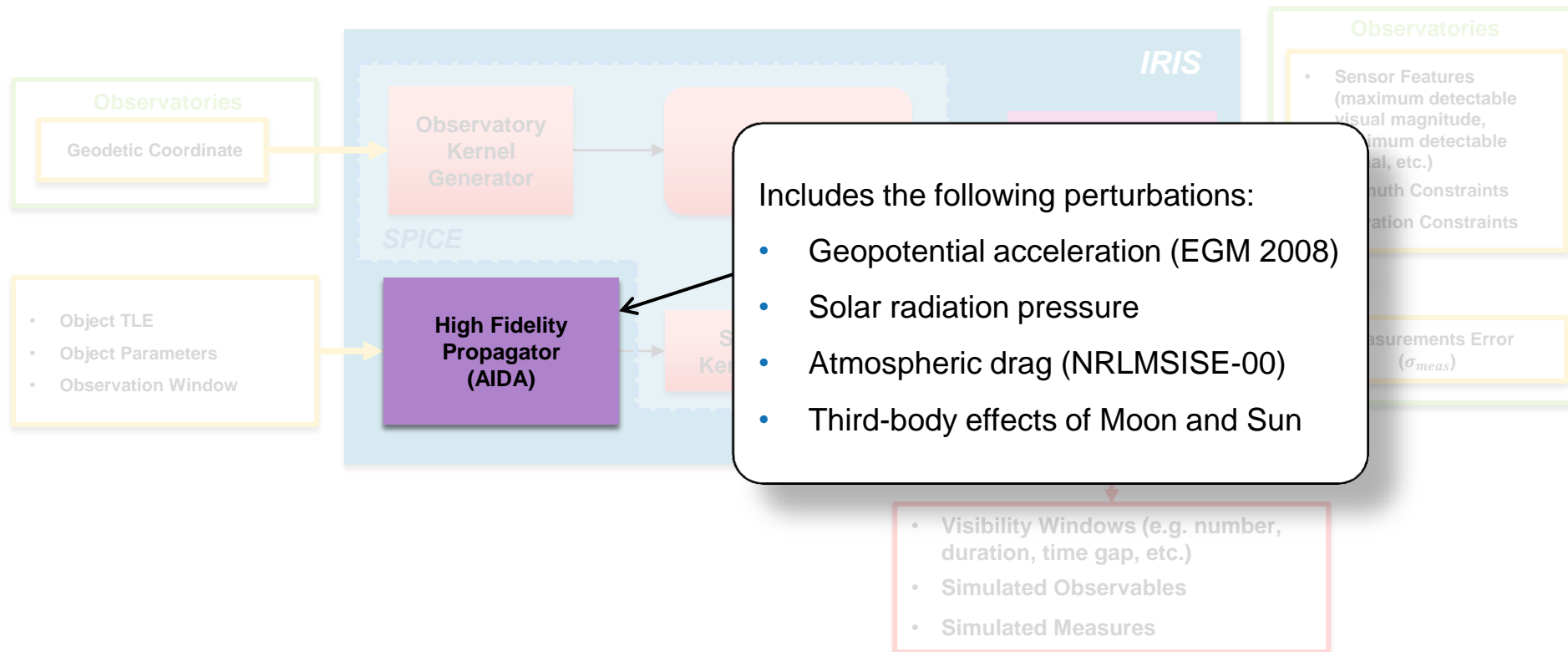
Source of illumination:  
source of illumination:  $d^{-2}$   
dependency



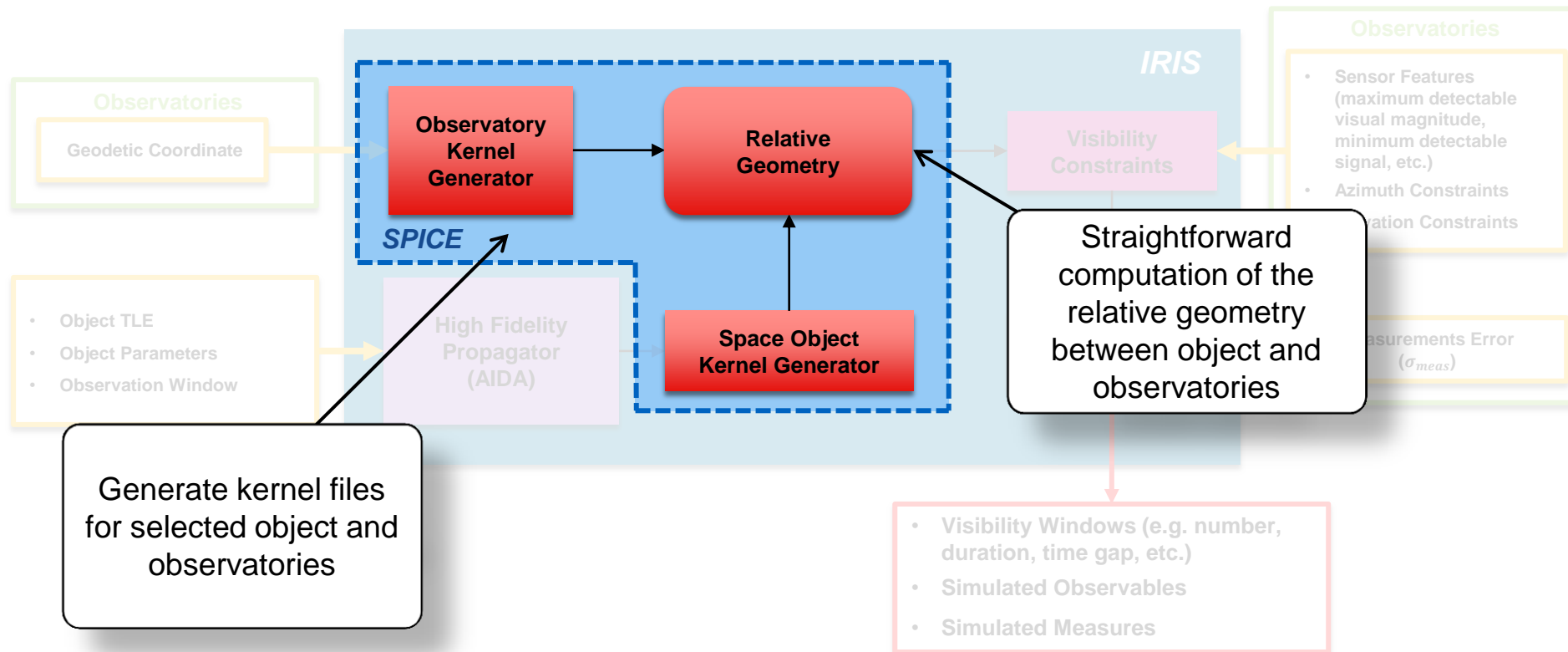
- Object class



- High fidelity propagator (AIDA) module



- SPICE module





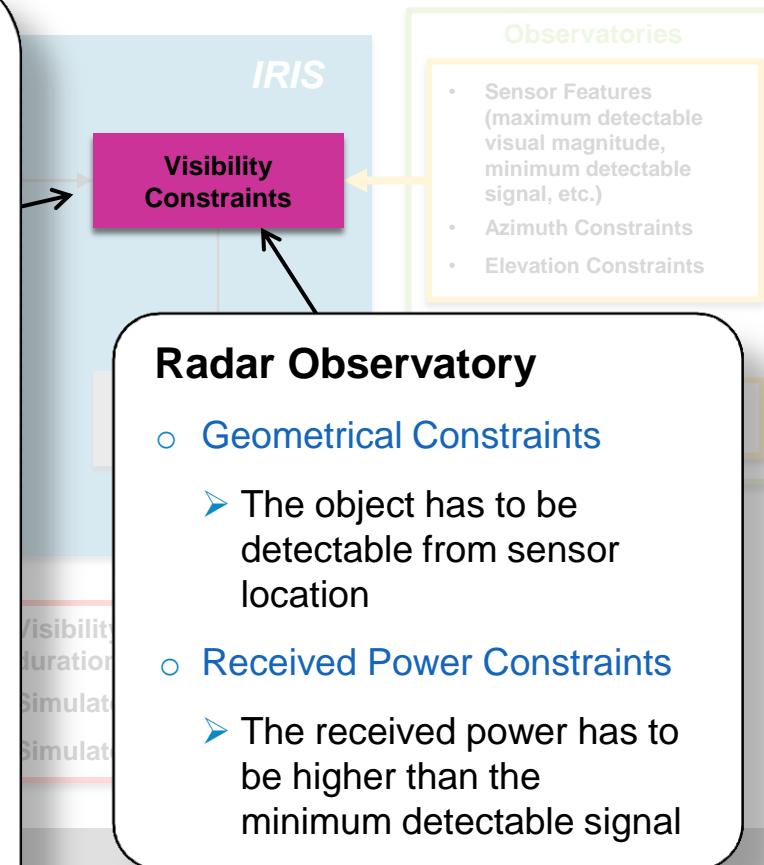
## ■ Visibility module

### Optical Observatory

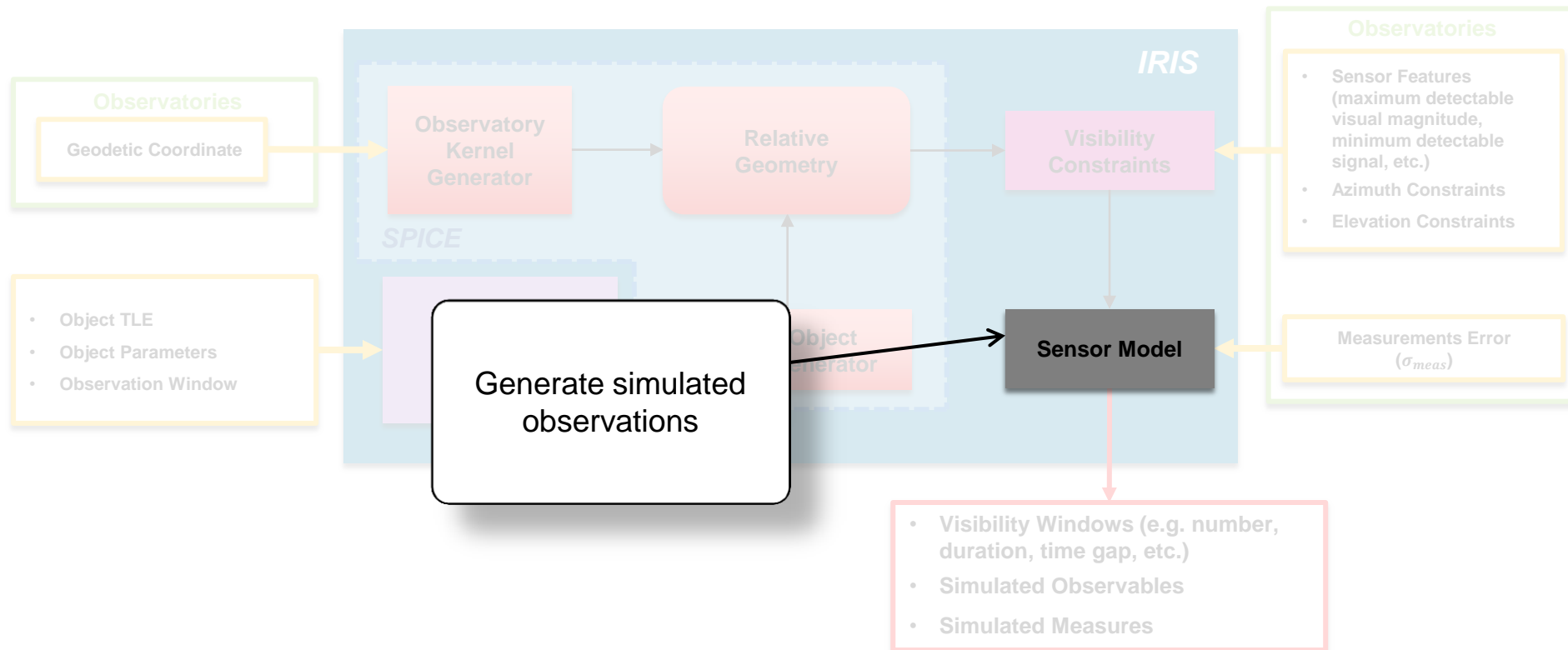
- Geometrical Constraints
  - The object has to be detectable from sensor location
- Illumination Constraints
  - Object must be illuminated by the Sun
  - Object brightness must exceed that of the background sky by a certain margin
  - Dark background during observations
  - Observation must occur during the night
- Cloud Coverage Constraint
  - The sky has to be clear to perform the measurement
- Angular Velocity Constraints
  - Object velocity doesn't exceed a maximum value

### Radar Observatory

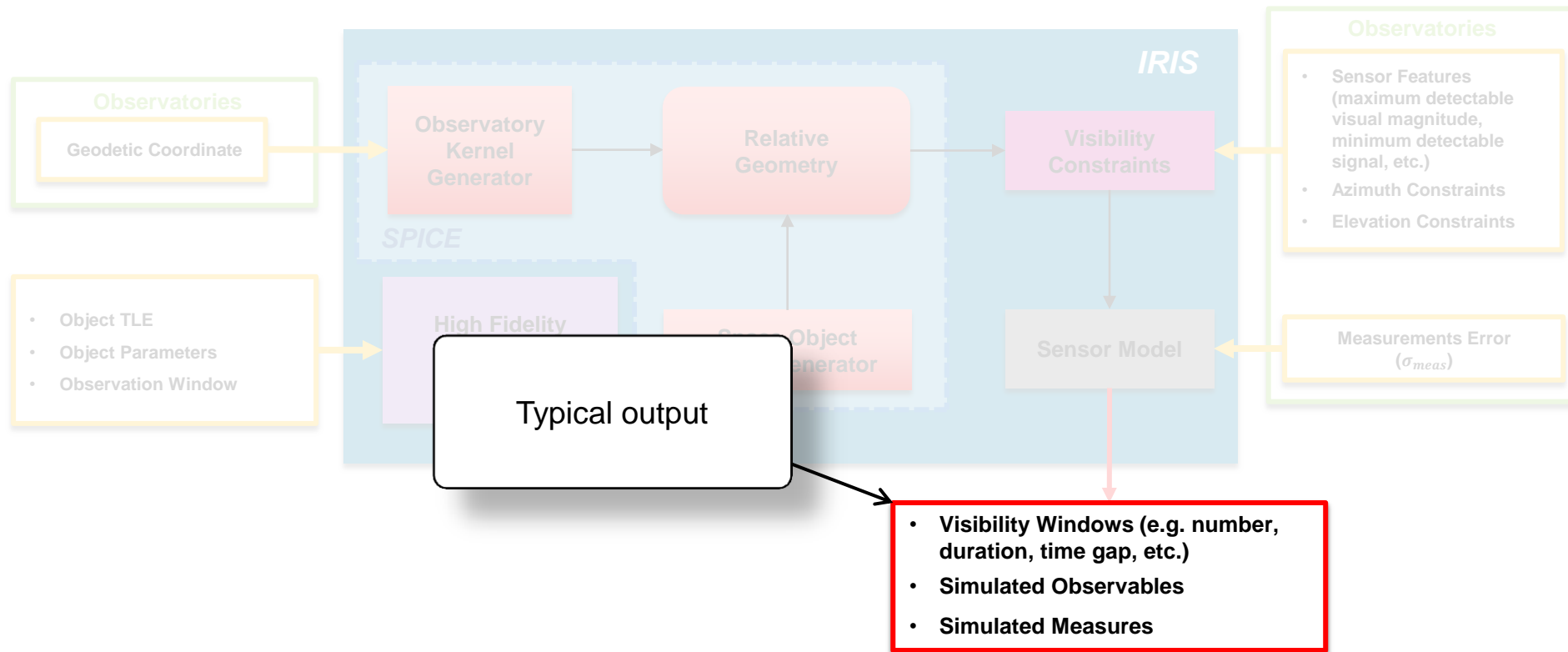
- Geometrical Constraints
  - The object has to be detectable from sensor location
- Received Power Constraints
  - The received power has to be higher than the minimum detectable signal



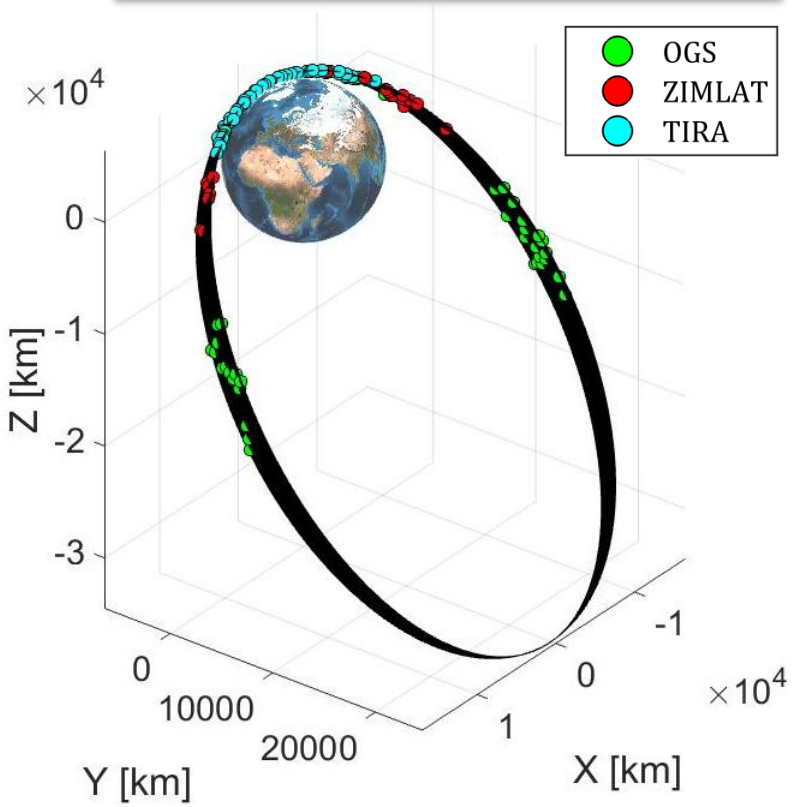
- Sensor module



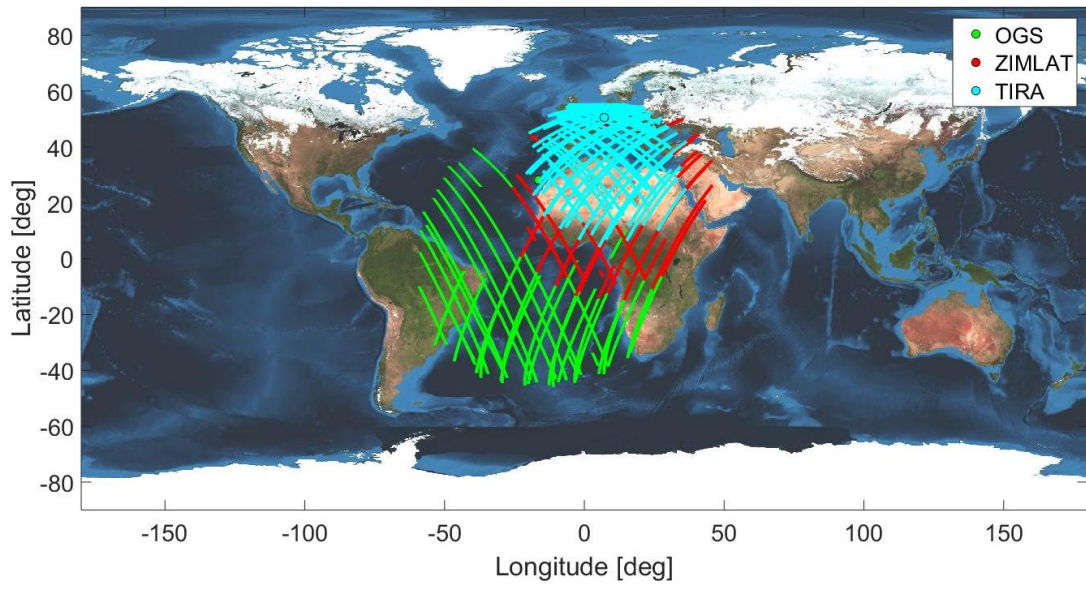
## ■ Output



**Measures along the orbit**



**Ground Track of the observable**



- Observational data represent the inputs for the design of optimal observation strategies



**Optimization Approach**

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## Optimization Approach

- Automatic method for observation strategy design

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## Optimization Approach

- Automatic method for observation strategy design
- Strategy design problem as a multi-objective optimization problem:

$$\mathbf{X}_{opt} = [1,0,0,1, \dots, 0,0,1] \quad \mathbf{X}_{opt} \in \mathfrak{R}^{N_{Obs}}$$

$$\mathbf{f}(\mathbf{X}_{opt}) = [f_1(\mathbf{X}_{opt}) \quad f_2(\mathbf{X}_{opt})] = [N_{Obs} \quad \max(\lambda_i)]$$

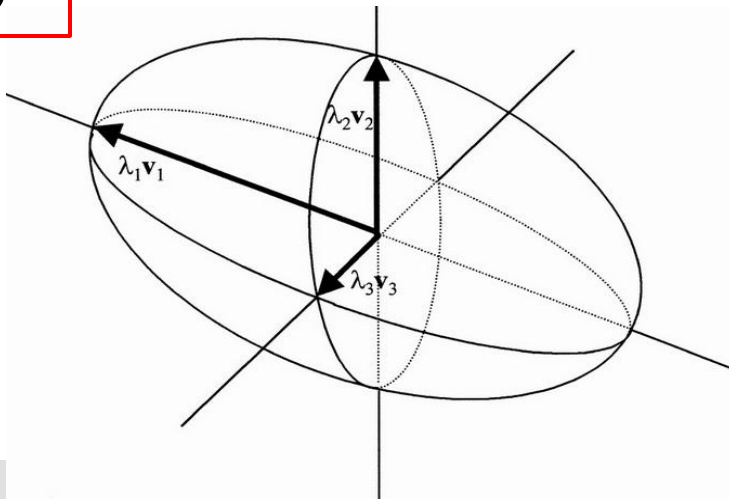
$$Cov = (H^T W H)^{-1}$$



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**Spectral  
decomposition**

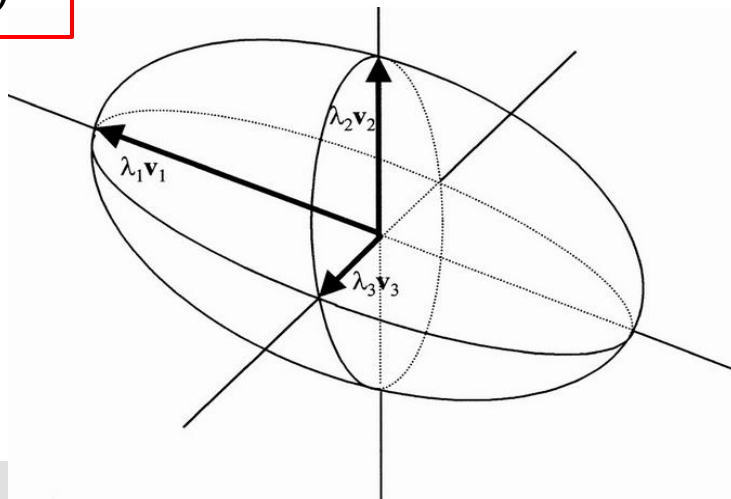
$$\max(\lambda_i)$$



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**Spectral  
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$$\max(\lambda_i)$$



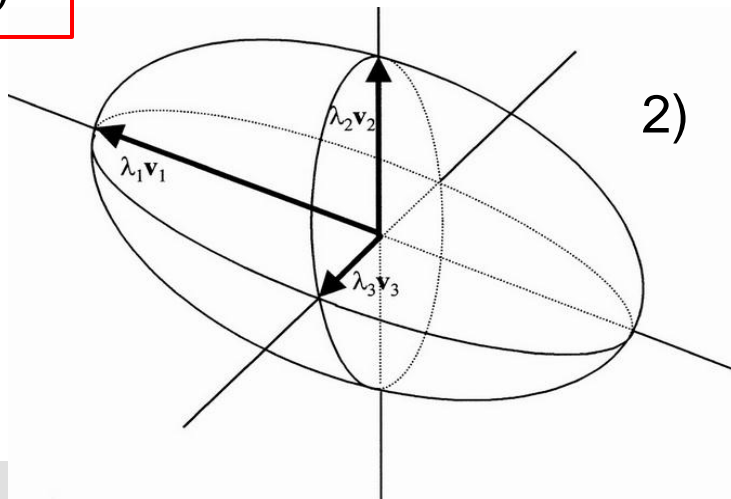
1) Other possibilities?

→  $\det(Cov)$ ,  $\text{tr}(Cov)$ , ...

$$Cov = (H^T W H)^{-1}$$

**Spectral decomposition**

$$\max(\lambda_i)$$



- 1) Other possibilities?  
 →  $\det(Cov)$ ,  $\text{tr}(Cov)$ , ...
- 2) Relation with re-entry prediction?  
 → Assessment of improvement is required

$$Cov = (H^T W H)^{-1}$$

**Jacobian  
matrix**

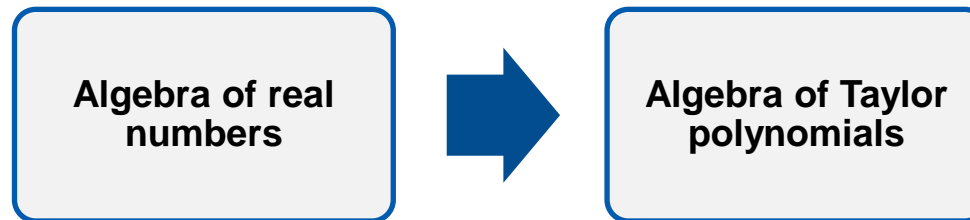
$$Cov = (H^T W H)^{-1}$$

**Jacobian  
matrix**



Jacobian matrix can be computed through  
Differential Algebra (DA)

- Substitution of classical real algebra with a new algebra of Taylor polynomials



- Any function of  $n$  variables is expanded into its Taylor polynomial up to an arbitrary order  $k$
- Unlike standard automatic differentiation tools, the analytic operations of differentiation and antiderivation are introduced

## Pros

- Jacobian computed at machine precision
- Efficient computation of J for any dynamical and sensor models
- No finite differences or variational equations used
- Reduction of computational efforts

CPU Time	
MATLAB (finite differences)	~200 s
DA	~15 s

## Cons

- Need of a DA framework:

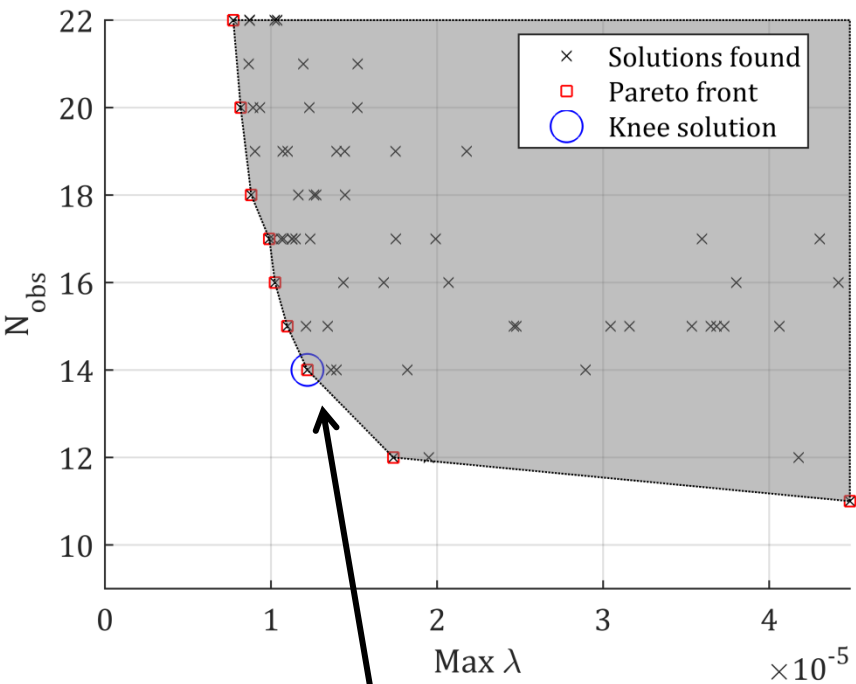


In C++, interface  
through Mex files

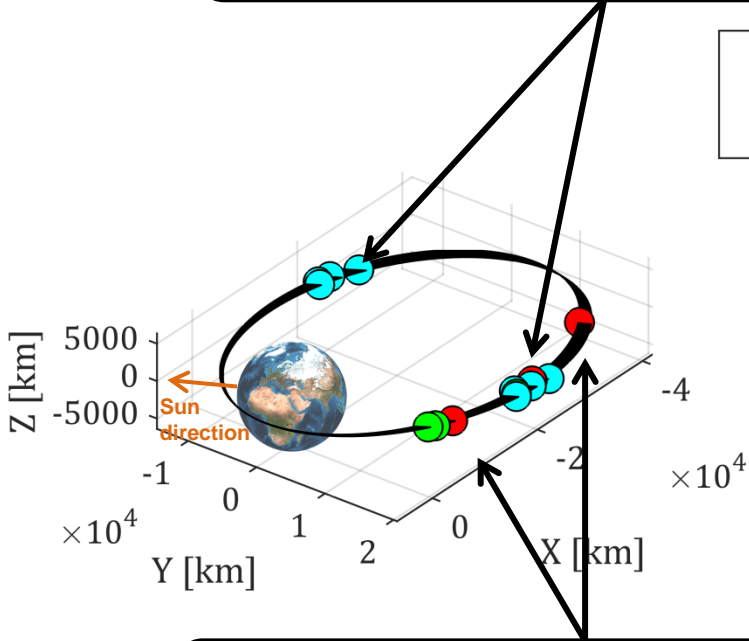
## ➤ Ariane 5 Second Stage (37239)

Radar measurements are taken at quite large distance from the perigee and close to the maximum available bound for this type of sensors

- OGS
- ZIMLAT
- TIRA



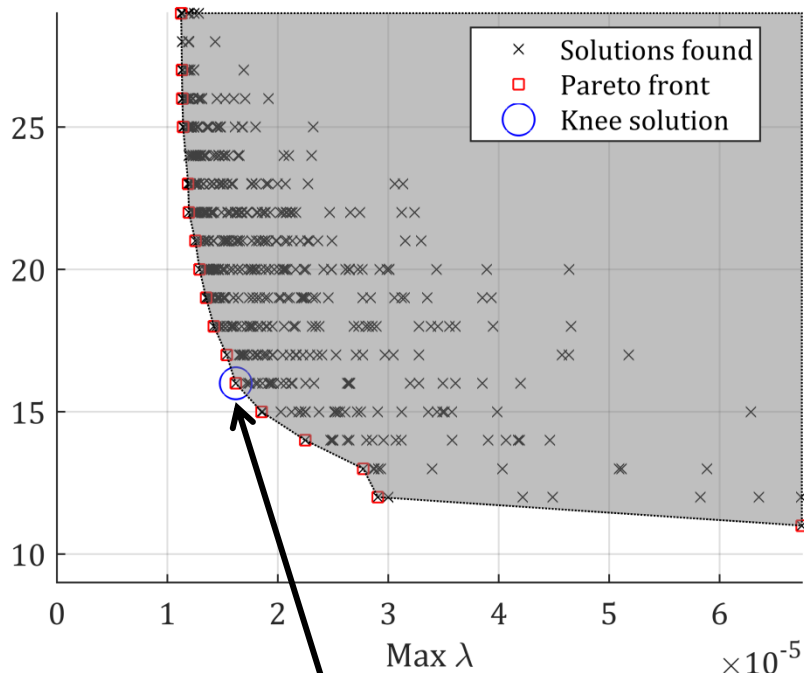
The knee-solution corresponds to  $N_{obs} = 14$



Optical observations are all on the same side of the orbit because of the position of the Sun relative to the observed object



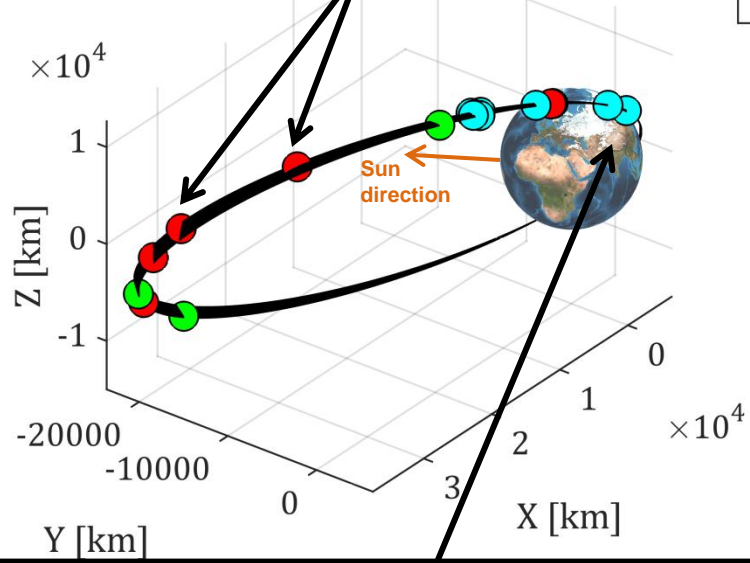
## ➤ CZ-3A Third Stage (37949)



The knee-solution corresponds to  $N_{obs} = 16$

OGS and ZIMLAT produce the most of measurements when the object moves around the orbit apogee, whereas TIRA can observe the object when it is close to orbit perigee

- OGS
- ZIMLAT
- TIRA



TIRA provide measures only when the sub-satellite point is on north hemisphere. This is mainly due to high inclination of orbit ( $i_0 = 55^\circ$ ) combined with the high latitude of TIRA sensor

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  - Computation of visibility windows, observables, ground track coverage etc.
- An optimization approach can be used to obtain the optimal observation strategy for given object and sensors network
- Study not yet completed...
  - No evidence of common patterns that can be used as guidelines for the definition of a general optimal strategy
  - Other approaches are currently under investigation
  - Covariance analyses

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