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Celestial and Spaceflight Mechanics Laboratory

# High-Fidelity Small-Body Lander Simulations

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6<sup>th</sup> International Conference on Astrodynamics Tools and Techniques Darmstadt, 16 March 2016





# → Increase mission return with lander and surface mobility operations

→ Design requires **high-fidelity modeling** 

## Contents



- 1. Framework
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# Framework

- *SAL* software package
- Previously: spherical landers
- Currently: extension to arbitrary shapes
- Integration with RK5(4) in C++, event capability
- Propagation relative to rotating target frame







## Gravity



- Constant-density polyhedron model, expensive
- <u>Method 1</u>: Linearization
  - Acceleration:  $\mathbf{g}(\mathbf{r}) \simeq \mathbf{g}(\mathbf{r}_0) + (\mathbf{r} \mathbf{r}_0) \cdot \mathbf{\Gamma}(\mathbf{r}_0)$
  - Error control with  $\Delta r_{max}$
- <u>Method 2</u>: Reduced-resolution gravity model
  - Error control with resolution



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# Surface

- Surface interactions govern dissipation, high-resolution model necessary
- Challenge: collision detection with global surface and rocks
- <u>Method 1</u>: Atlas
  - Latitude/longitude grid of small *local* worlds
  - Fast collision detection with single active local world







## Surface



- <u>Method 2</u>: Surface rock distribution
  - Affects energy and topographic dissipation
  - Previously: stochastic model, limitations
  - Full model is necessary, large # of rocks
  - Procedural generation on *active* local world
  - Modified icosahedron





## Surface



- <u>Method 3</u>: Bounding spheres
  - Defined for local worlds, rocks, and lander
  - Fully encompass target
  - Trigger finer collision detection
  - Fast detection for large # of features



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# Collisions

- Impact of arbitrary shapes:
  - Non-eccentric, coupling of normal force and friction
  - Integrated using time-like normal impulse **p**
  - Governed by *e* and μ
  - Sliding velocity s, slip vs. stick
  - Contact when  $v_N \sim 0$







### Sample deployment to Itokawa



## Results



- Verification of deployment strategy
- Effect of uncertainty in:
  - Release conditions, *e.g.* altitude
  - Surface interaction properties, *e.g.* e and  $\mu$
  - Rock distribution
- Surface mobility operations
  - Control strategy
- Lander/rover shape optimization
- Visualization in OpenGL

### Lander shape comparison



### Impact locations



## Conclusions



- Elements required for high-fidelity simulations
- Methods to reduce numerical burden:
  - Gravity (linearization/resolution)
  - Surface (atlas/rocks/spheres)
- Collision handling of arbitrary shapes
- Enables design and verification of lander/rover
- Future work:
  - Shape optimization
  - Contact motion
  - Mission scenarios

### Sample Mascot-2 deployment to Didymoon

### video placeholder

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