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RAPID DEPLOYMENT OF DESIGN ENVIRONMENT FOR EUCLID AOCS DESIGN

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Introduction

- Euclid is a cosmology mission dedicated to study the geometry and the nature of the Dark Universe with unprecedented accuracy
 - Observe a 15000 deg² wide area of the sky from L2
 - Scientific goals result in very demanding performances for the AOCS subsystem:
RPE < 75 mas over 700 sec during observations
- Euclid S/C: procured by ESA and supplied by TAS-I
- SENER is the prime contractor of the AOCS sub-system
 - Work executed in partnership with ADS-NL
 - SENER responsibility is for the whole AOCS. Design, implementation and verification of GNC/AOCS modes and their SW, units subcontracts and overall verification.
- Current AOCS status is fully in line with expectations
 - AOCS SRR completed in 1 month and AOCS PDR in 7 additional months
 - PDR colocation was held at the end of November
 - Units' subcontractors activities are being launched with good progress

Euclid Design Simulator and AOCS Development

- In the frame of EUCLID AOCS development two simulators are employed:
 - **ESE: Engineering Simulation Environment**, responsibility of Elecnor-Deimos
 - *Functional Engineering Simulator, one of the formal test environments supporting overall Euclid AOCS verification and validation process*
 - **EDS: Euclid Design Simulator**
 - *The AOCS design environment*
- **Euclid Design Simulator**
 - Developed during Phase-B2 activities, will be evolved and maintained over the whole project life-cycle
 - Environment employed in design iterations: it's the framework in which AOCS modes, functions, interfaces and databases are being created
 - Units models: initial models in EDS → evolution through suppliers info → revert to EDS for update and test → specification

SENERIC

- SENERIC is a suite for high-fidelity tools and models for the design, analysis, and validation of an AOCS
 - Initially funded by Spanish Ministry of Technology, turned in internal R&D project for continuation
- SENERIC is at the core of DKE, mathematical functions, sensors, and actuators of EDS. Models and libraries available for:
 - AOCS Equipment: sensors and actuators
 - AOCS/GNC Functions: for rapid prototyping, analysis and design
 - Environment: force-torque perturbations, ephemerides, time references
 - Mathematical operations & transformations
 - S/C Dynamics and kinematics
- Employed in the design of Planck and IXV AOCS, in PROBA-3, and for the design and auto-coding of OPTOS Attitude Control System
- Math library taken as basis of EUCLID AOCS math library

Dynamics

- Dynamics: joint integration of the S/C and reaction wheels, sloshing, and telescope Filter Wheel Assembly (FWA/GWA) dynamics equations

$$\begin{bmatrix} I_{tot} & I_{RW}U_{RW} \\ I_{RW}U_{RW}^T & I_{RW}I_{n \times n} \end{bmatrix} \begin{pmatrix} \dot{\bar{\omega}}_{SC} \\ \dot{\bar{\omega}}_{RW} \end{pmatrix} = \begin{pmatrix} -\bar{\omega}_{SC} \times (I_{tot}\bar{\omega}_{SC} + I_{RW}U_{RW}\bar{\omega}_{RW} + \bar{u}_{FWA}I_{FWA}\omega_{FWA}) \\ -\bar{T}_f(\bar{\omega}_{RW}) \end{pmatrix} + \begin{pmatrix} \bar{T}_{ext} \\ \bar{T}_{RW} \end{pmatrix} + \begin{pmatrix} -\bar{u}_{FWA}I_{FWA}\dot{\omega}_{FWA} \\ \bar{0} \end{pmatrix}$$

- The FWA/GWA is modelled by explicitly imposing a kinematic motion by directly injecting the angular rate and acceleration profiles
 - Acceleration obtained respecting integral relation with rate
 - This ensures a correct torque profile and conservation of angular momentum

Simulation Core - Actuators

- MPS (Micro-Propulsion System). Cold gas thrusters for attitude control in scientific observations.
 - Accurate model needed in line with strict pointing requirements
 - Modelling of thrust noise frequency profile by filtering a white noise source
- RCS (Reaction Control System). Employed for attitude and orbit control
 - Effects included: integration & sampling, thrust time profile (rise+tail), noise, plume impingement
 - RCS pulse forced scaled over the AOCS cycle to model correct linear impulse independently of the sample time
- RWA (Reaction Wheels Assembly)
 - “Distributed” model: separate integration of equation of motions and simulation of dynamic behaviour
 - The RWA will have to be stopped and restarted hundred of thousands of times during the mission
 - Wheels braking modelling must ensure conservation of angular momentum
 - Dead time included during torque reversals
 - RWA dynamics simulates static+viscous+Coulomb friction

Simulation Core - Sensors (I)

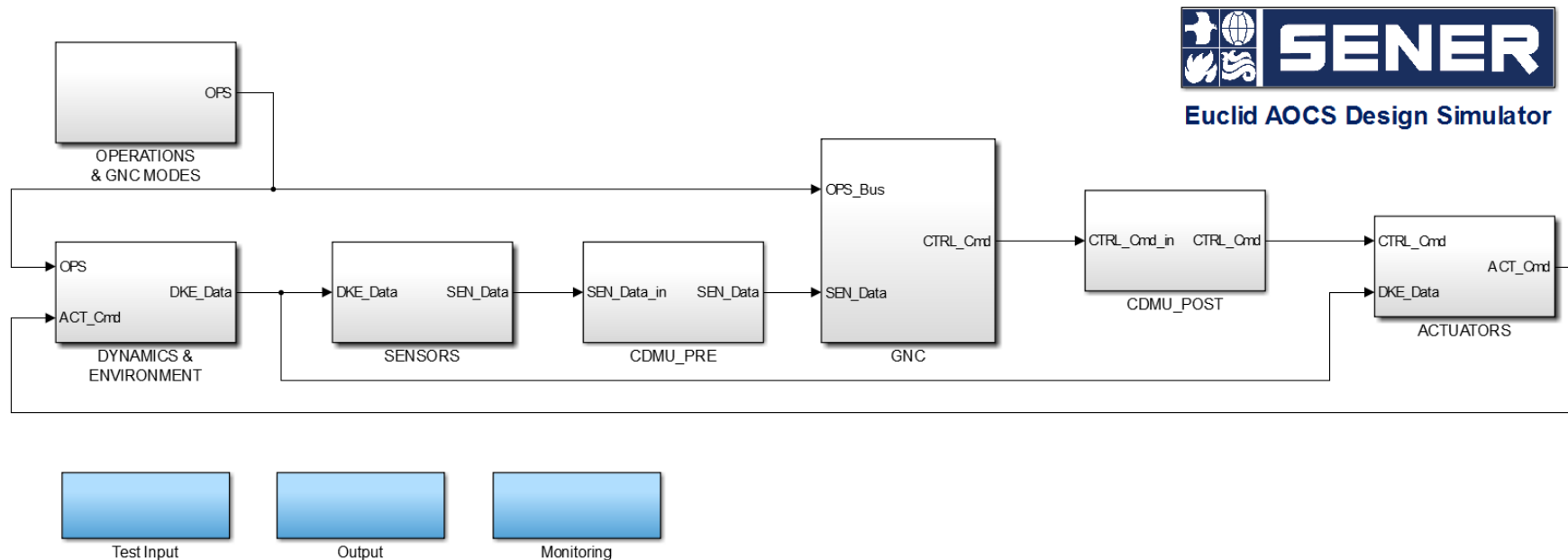
- Gyroscope
 - Rate integrating gyro model employed for 4 independent heads
 - Ad hoc filtering of White noise sources employed to reconstruct Allan variance profile
- Accelerometers
 - Approach similar to gyro.
 - Linear accelerations due to sensors accommodation + rotational dynamics included
- Coarse Rate Sensors
 - Analogous to gyros
- Sun Sensors (SAS)
 - Non-ideal cosine output simulated
 - Albedo included based on offline computation taking into account Earth illumination and sensor FOV

Simulation Core - Sensors (II)

- FGS (Fine Guidance Sensor)
 - CFI for the AOCS; physically mounted on telescope
 - Most relevant sensor for science observation phase
 - Complex functioning: sensor modes logic, attitude lock, acquisition delays, data outages, measurement delays, absolute/relative measurement, performances function of angular velocity
 - Custom in-house model built implementing specified behaviours and characteristics
 - Wrapping built to allow EDS to swap in-house model for customer-provided model
 - Customer-provided model incrustrated in EDS with minimum modifications
- Star trackers
 - Integration+process delay taken into account
 - Velocity-dependent noise included
 - Spatial effects due to FOV and detector characteristics (pixel error) modelled by first and second order transfer functions with variable (velocity and rotation axis dependent) gains

Simulation Core

- CDMU (Central Data Management Unit)
 - AOCS is executed as part of the onboard software, running on the CDMU
 - CDMU pre-GNC and post-GNC functional subsystems included in model in order to represent correctly the timely data flow due to the finite AOCS execution time

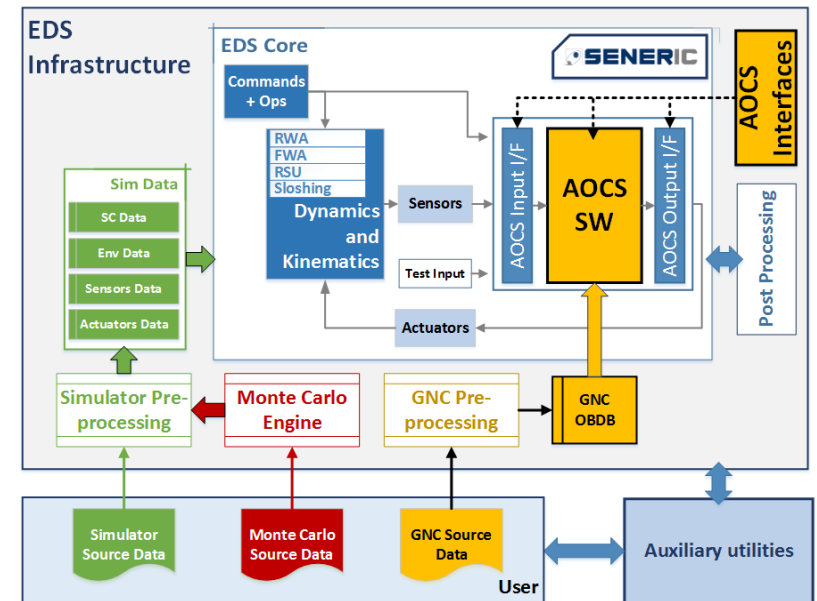


EDS Workflow and Infrastructure

- The EDS was developed taking into account the following usage guidelines:
 - fast input/parameters set-up
 - Allow quick algorithms design loops: modify AOCS algorithm or unit model, run test case, check output, modify again
 - Allow systematic execution: manage large data sets, run and store in an orderly fashion different sim cases, including Monte Carlos
- EDS high level components:
 - Simulink simulation core
 - MATLAB simulation and data management infrastructure
 - Utilities for workflow efficiency and ancillary operations

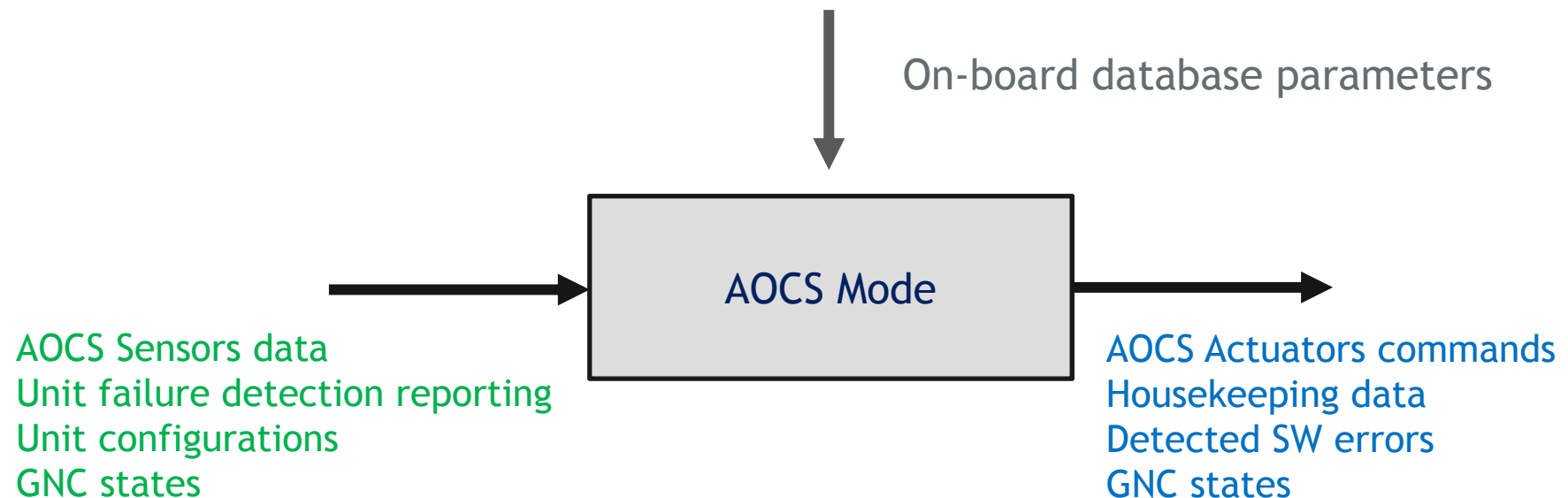
Simulations execution

- Three execution modes allowed:
 - Design Run
 - allow rapid iteration on algorithms design and simulator parameters, minimizing the set-up effort.
 - aim: configure the simulation with a single MATLAB command
 - Systematic Run
 - Monte Carlo
- Data management
 - Source databases: user-defined parameters
 - Pre-processing: transforming user-friendly parameters into data needed by simulation core
 - XML databases and m-files used: m-files for quick setup and debug, xml for more structured runs
 - XML files can be generated directly from m-files, to reduce data preparation effort



AOCS Interfaces (I)

- Interfaces definition plays a relevant role in the design and development process
- All AOCS modes except Safe will be auto-coded
- The AOCS modes will have software I/F only with AASW-MAN (manually coded AOCS software). Logic interfaces are with
 - Sensors and actuators through the AASW-MAN
 - Directly with AASW-MAN itself
- The transferred information



AOCS Interfaces (II)

- Strong data typing approach is employed in the definition of the AOCS modes models
- Input/Output AOCS-AUTO data are organized into Simulink Buses, analogous to C data structures
 - The Bus Objects are autocoded into C-code type definitions, which constitute part of the actual flight software
- The AOCS interfaces are defined in terms of such data buses and their elements, with their data types and dimensions
- The interfaces must be iterated and documented, and will be needed by the AOCS-MAN SW and by the ESE to embed the AOCS-AUTO code
- EDS contains a utility that allows generating the interfaces minimizing the iteration effort
 - Interface signals, data types and sizes are defined in a specific Simulink model
 - Resulting Bus Objects for all modes are automatically generated (single command)
 - The Bus definition libraries revert in the Simulator interface blocks, ensuring consistency with what expected by GNC

AOCS Software Development (I)

- The AOCS is being developed according to the Model-Based Design (MBD) philosophy
 - The model is an executable specification that is continually refined throughout the development process
- MATLAB/Simulink allow to
 - Link models with requirements
 - Generate C code with comments and tags that trace with the model and the requirements
- AOCS algorithms:
 - Part of the model to be included in the AOCS Application Software
 - Are being evolved in the EDS
 - Will eventually be developed as auto-generated code on the embedded processor
 - Will be verified in SIL, SVF and HIL facilities

AOCS Software Development (II)

- Model-Based design development and validation plan follows the evolutionary model
 - Iterative approach where progressive versions of the model/software are planned
 - For each version a model validation is foreseen
- The AOCS model will evolve based on simulation loops performed in the EDS environment
 - Simulations on the EDS will be the start point to develop the formal validation tests
- Each AOCS mode is created as a referenced model
 - Root class representing the overall mode
 - Decomposition in smaller elements according to functionalities (submodes, functions) and assignment to specific C code functions during autocoding
 - Simulink architecture defined and analysed taking into account Simulink to C mapping



The way to see the future

Thank you