

DESEO

Design Engineering Suite for Earth Observation



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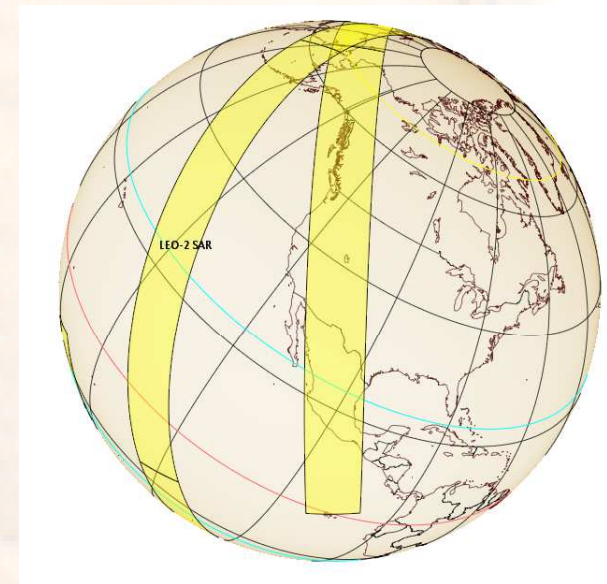
DEIMOS Space S.L.U., Spain

- **DESEO Overview**
- **Toolkit Heritage**
- **Software Architecture**
- **Analysis Capabilities**
- **Applicability to Earth Explorers Phase A/B1 Studies**
- **Showcase: DEIMOS-2 Mission**
- **Conclusions**

- **DESEO is a toolkit designed to support EO missions**
 - Mission analysis
 - Preliminary system/subsystem design activities
- **DESEO has been designed to be used**
 - by mission and system engineers
 - throughout all phases of an EO mission (from Phase 0 to E)
 - whenever accurate & fast quantitative results are needed
 - to support design trade-offs and assessment analyses
- **DESEO has been designed**
 - to be a modular, flexible and self-standing toolkit
 - to provide the user with
 - a comprehensive set of mission and system related computation modules
 - visualization capabilities to yield meaningful numerical and graphical results

- **DESEO has been conceived as a tool**
 - in continuous evolution
 - suitable to be upgraded with further modules
 - able to be interfaced with external software

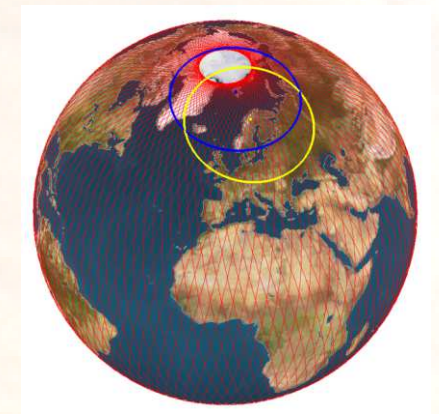
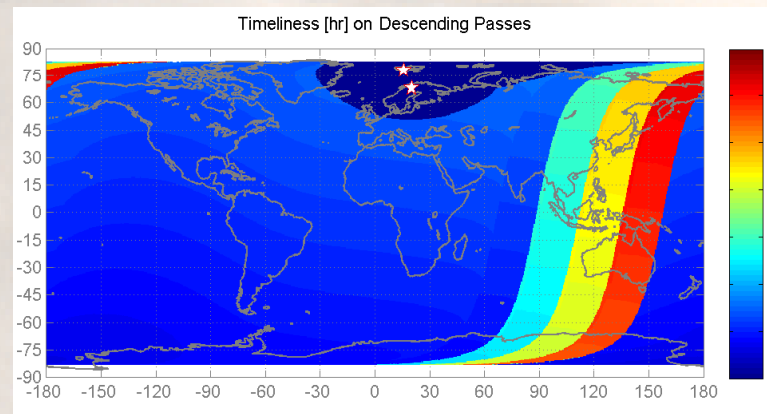
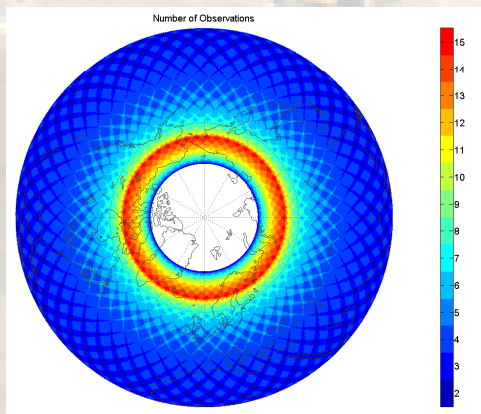
- **DESEO currently embeds in a Graphical User Interface**
 - 38 different analyses
 - an EO missions data repository
 - Satellites
 - Ground Stations
 - Thematic Areas of Interest
 - a results visualization module
 - 3D interactive visualizations
 - Gantt charts
 - Cartesian plots
 - Cartographic maps representations
 - Tables



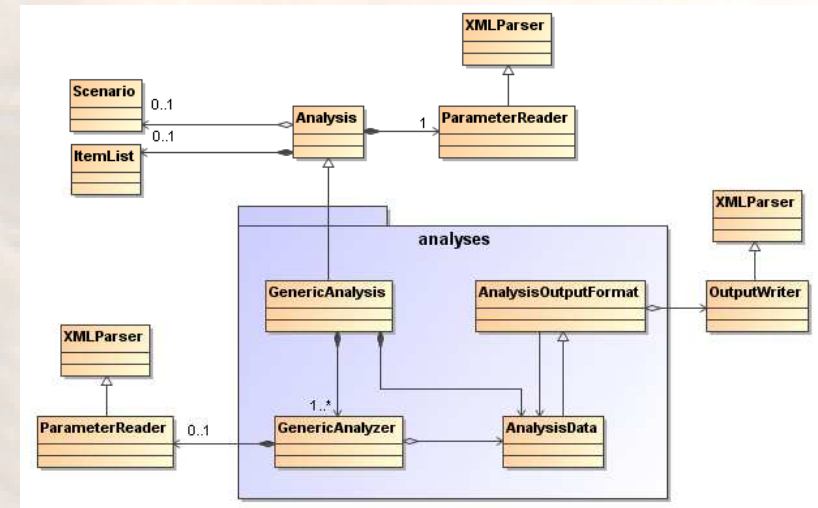
- **Mission and system analysis are continuously evolving disciplines**
 - connected with the evolution of the EO missions concepts
- **Analyst expertise is a key asset**
 - the analyst shall dynamically interpret new requirements and propose solutions through creative analysis approaches
- **A very large history of analyses allowed the experts to define best practices and optimal approaches to face a large number of recursive problems**
- **DESEO tries to implement all those best practices**
 - to provide the analyst with assessment and design means tailored on the basis of his/her specific needs

- **EO Customer Furnished Items (CFIs) as key asset**
 - Orbit propagation
 - Satellite attitude modes
 - Pointing laws

- **DEIMOS internal tools and algorithms represent the DESEO backbone**
 - State-of-the-art expertise in the mission analysis field
 - Robustness
 - applied and tested in a large number of analyses and in a wide spectrum of EO missions



- **DESEO has been developed**
 - following the principles of flexibility and modularity
 - guaranteeing reduced maintenance and upgrading efforts
- **Re-engineering of existing algorithms in a unique but modular architecture**
- **Extended use of XML Schema Documents (XSD)**
 - external interfaces definition
- **Very flexible infrastructure**
 - integration of many different analyses in a unique paradigm
 - easy extendibility
 - prone to future improvements
- **Object-Oriented approach**



- **DESEO embeds 38 different analyses, organized in 9 core computational groups**
 - Orbit Propagation
 - Coverage Analyses
 - Ground Station Analyses
 - Orbit Control
 - System Analyses
 - Orbit Analyses
 - Manoeuvres Computations
 - Pointing Analyses
 - Tools

Analysis	CarbonSat	FLEX	Biomass
Orbit Propagation	Orbit Altitude Evolution	Formation Flying Stability (*)	Orbit Altitude Evolution
- Attitude	N/A	N/A	N/A
Coverage	Coverage, Sunlint Tracking (*)	Coverage, Duty Cycle	Coverage, Duty Cycle (*)
Ground Stations Visibility	GS Network	GS Network, Interference (*)	GS Network LEOP
Ground Stations Conflict	GS Network	GS Network	GS Network LEOP
Timeliness	N/A	On-board Timeliness	Data Latency (*)
Swath Properties	Coverage	Coverage	Coverage
Sun-Zenith Angle	Coverage	Coverage	N/A
Observation-Zenith Angle	Coverage	Coverage	Coverage
SC Topocentric Coordinates	GS-to-S/C Viewing Geometry (*)	GS-to-S/C Viewing Geometry	N/A
- Pointing Analysis	N/A	N/A	N/A
Semi-analytical Propagation	Delta-V Budget	Delta-V Budget	Delta-V Budget
Atmospheric Properties	Atmospheric Density Profile Delta-V Budget	Atmospheric Density Profile Delta-V Budget	Atmospheric Density Profile Delta-V Budget
Altitude Control	Delta-V Budget	N/A	N/A
Inclination Control	Delta-V Budget (*)	N/A	Delta-V Budget
Eq. Ground Track Control	N/A	GTE Control (*)	Delta-V Budget
- OA + OI Control	N/A	N/A	N/A
- EGT + OI Control	N/A	N/A	Delta-V Budget
EOL Decay	Delta-V Budget	Delta-V Budget	Delta-V Budget
Beta Angle	Beta Angle	Beta Angle	Beta Angle
Eclipses	Eclipse	Eclipse	Eclipse
- Ground Illumin.	N/A	N/A	N/A
- Time Tran.	N/A	N/A	N/A
- Coordinates Tran.	N/A	N/A	N/A
Injection Errors Correction	Delta-V Budget	Delta-V Budget	Delta-V Budget LEOP
Collision Avoidance	Delta-V Budget	Delta-V Budget	Delta-V Budget
Orbit Transfer	N/A	Formation Acquisition	Delta-V Budget
Master-Drone Control	N/A	Formation Control (*)	N/A
- Orbit Wizard	N/A	N/A	N/A
LEO Selection	Orbit Selection	N/A	Orbit Selection
- SSO Inclination	N/A	N/A	N/A
- Frozen Eccentricity	N/A	N/A	N/A
- RAAN Drift Rate	N/A	N/A	N/A
Basic Swath Geometry	Orbit Selection	N/A	N/A
- Geodetic Distance	N/A	N/A	N/A
ΔV & Fuel Budget	Delta-V Budget	Delta-V Budget	Delta-V Budget
OBDH	N/A	N/A	Data Volume
- Power Budget	N/A	N/A	N/A

Traceability matrix between DESEO analyses and Earth Explorer Phase A/B1 Mission Analysis Reports (MAR)

DESEO analyses cover almost all the EE MAR

- Few analyses partially cover the MAR^(*). Little extensions or basic post-processing are needed for a full MAR^(*) coverage
- Only specific mission related analyses are not included in the scope of DESEO

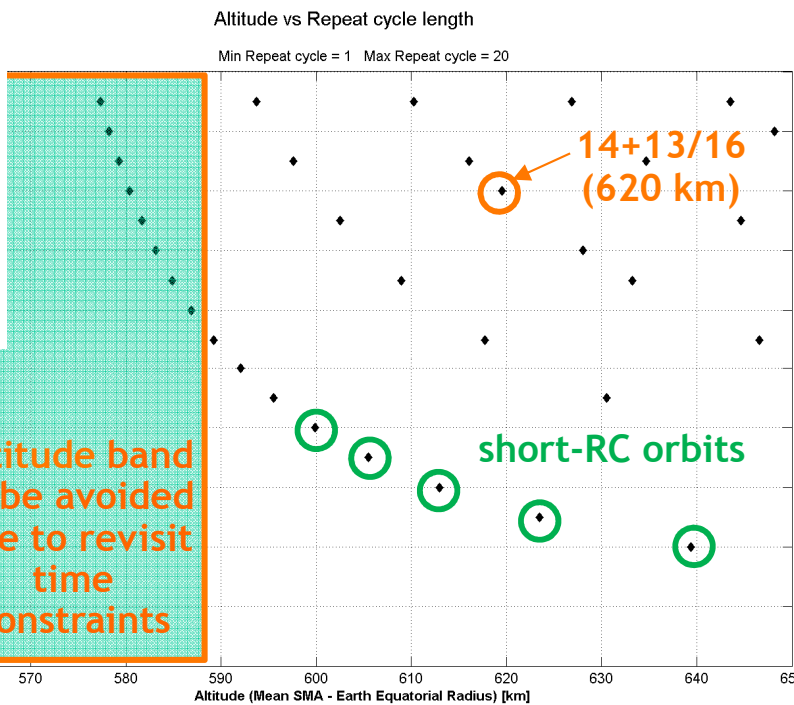
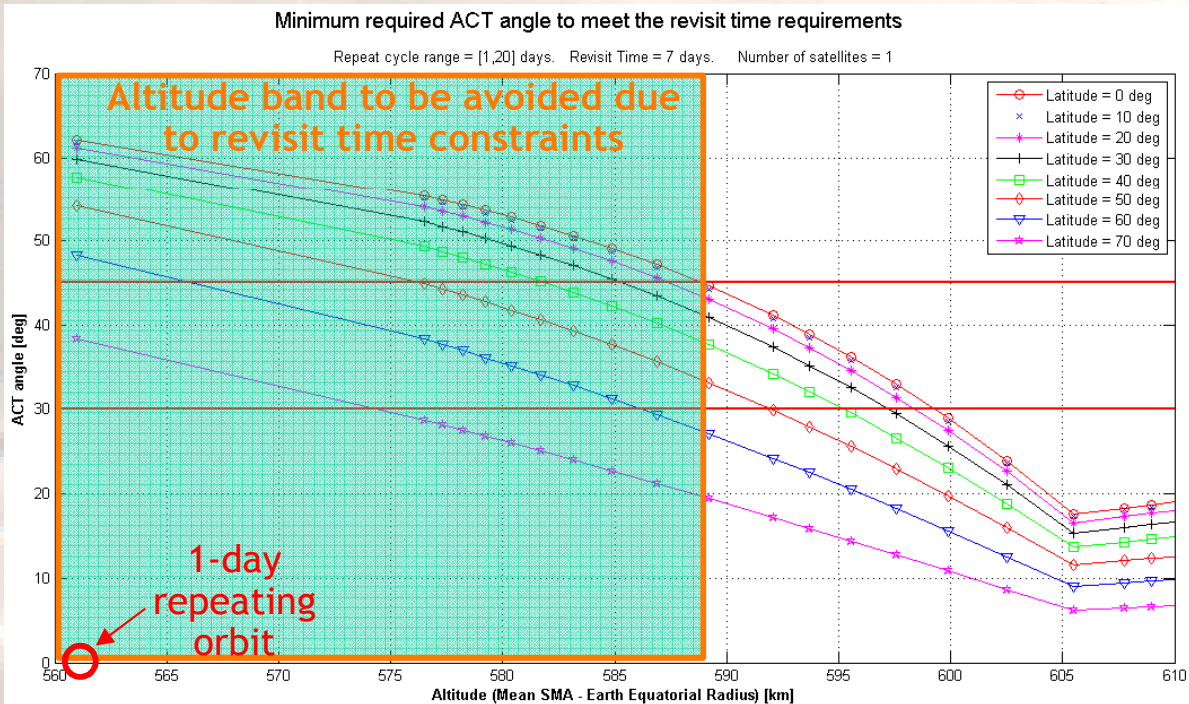
Analysis	DEIMOS-2	Analysis	DEIMOS-2
Orbit Propagation	Orbit Evolution	Beta Angle	Orbit Selection
Attitude	Agility (*)	Eclipses	Orbit Selection
Coverage	Coverage, Duty Cycle	Ground Illumin.	N/A
Ground Stations Visibility	GS Network	Time Tran.	N/A
Ground Stations Conflict	GS Network	Coordinates Tran.	Launch Scenario (*)
Timeliness	N/A	Injection Errors Correction	Delta-V Budget
Swath Properties	Coverage	Collision Avoidance	Delta-V Budget
Sun-Zenith Angle	Coverage	Orbit Transfer	Launch Scenario, Delta-V Budget
OZA	Coverage	Master-Drone Control	N/A
SC Topocentric Coordinates	GS Network	Orbit Wizard	Orbit Selection
Pointing Analysis	N/A	LEO Selection	Orbit Selection
Semi-analytical Propagation	Long-term Orbit Evolution	SSO Inclination	N/A
Atmospheric Properties	Atmospheric Density Profile Delta-V Budget	Frozen Eccentricity	N/A
Altitude Control	Delta-V Budget	RAAN Drift Rate	N/A
Inclination Control	N/A	Basic Swath Geometry	Orbit Selection
Eq. Ground Track Control	N/A	Geodetic Distance	N/A
OA + OI Control	N/A	ΔV & Fuel Budget	Delta-V Budget
EGT + OI Control	N/A	OBDH	Mass Memory Occupation (*)
EOL Decay	Delta-V Budget	Power Budget	Power Budget (*)

Traceability matrix between DESEO analyses and DEIMOS-2 Mission Analysis (Phase A to E)

DESEO analyses cover almost all the needs!

- Only specific mission related analyses have been performed with post-processing

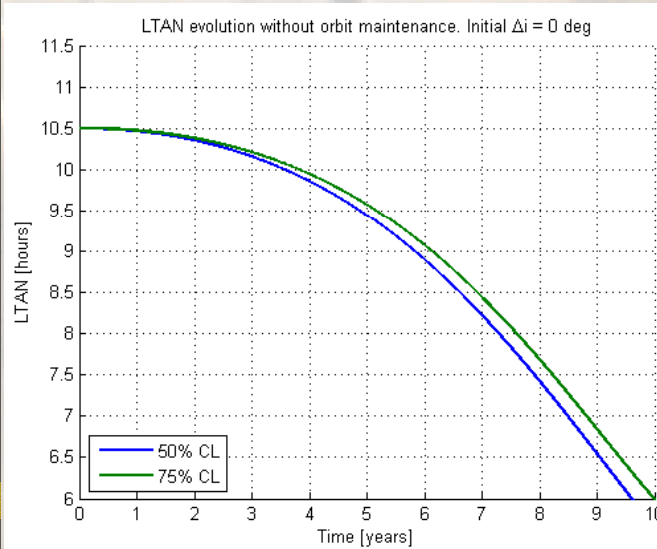
• Candidate Orbit Identification



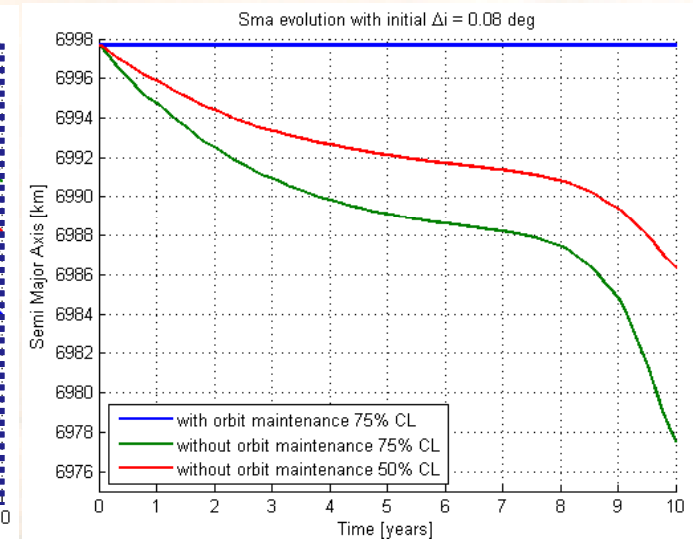
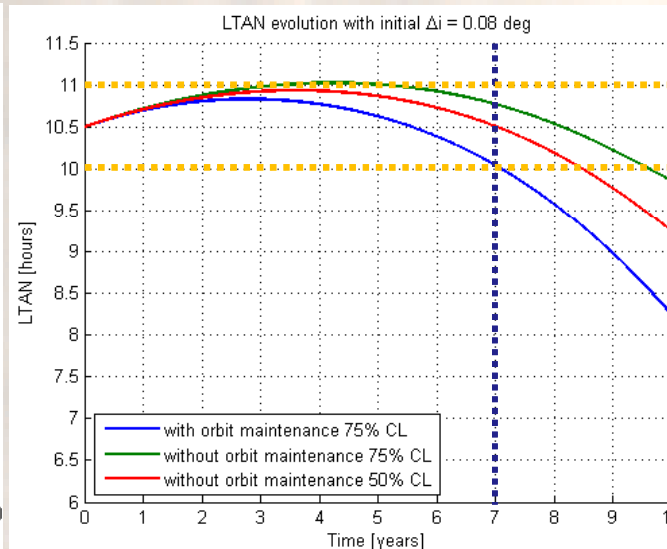
• Injection Orbit Selection

- Optimise LTAN evolution applying initial Δi w.r.t. **SSO** inclination
 - ✓ Initial inclination affects LTAN drift and provides **better LTAN profile**
 - ✓ LTAN & SMA profiles with **small deviation from 10h30**
 - ✓ **Avoid dedicated LTAN control** by out-of-plane manoeuvres
 - ✓ Controlled orbit (altitude maintenance) and uncontrolled orbit (free orbit decay) with different solar & geomagnetic activity levels

Uncontrolled orbit



Controlled & uncontrolled orbit

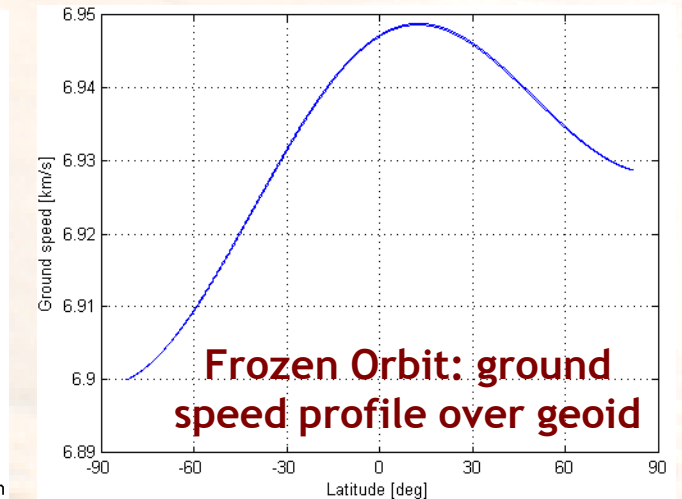
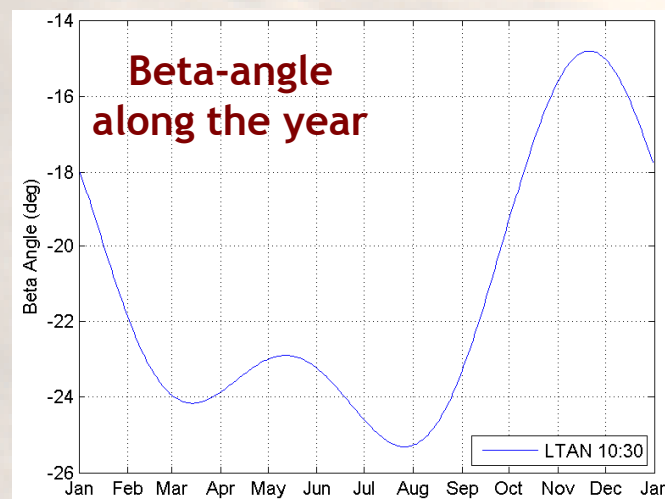
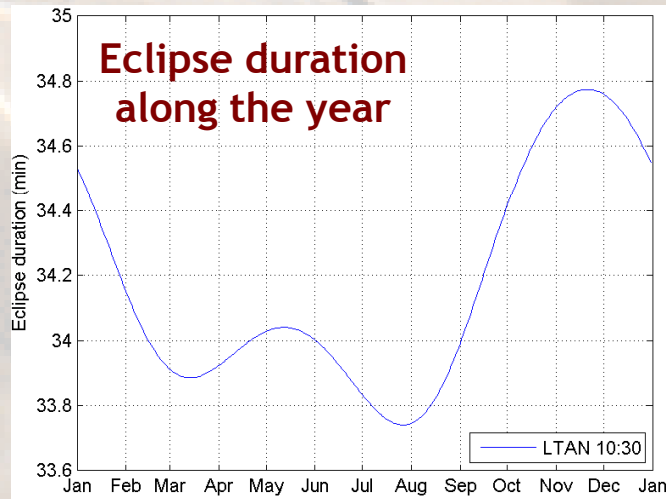
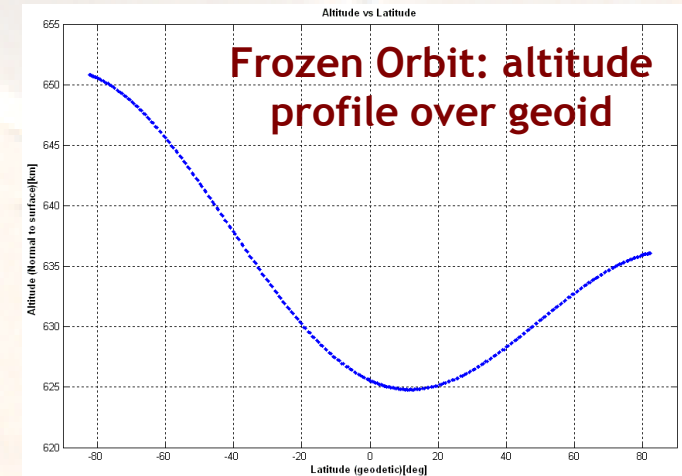


• Orbit Characteristics

- **Sun-synchronous** and **frozen** orbit
 - ✓ Constant altitude profile vs. arg. of latitude

• Eclipse and Beta-Angle Analysis

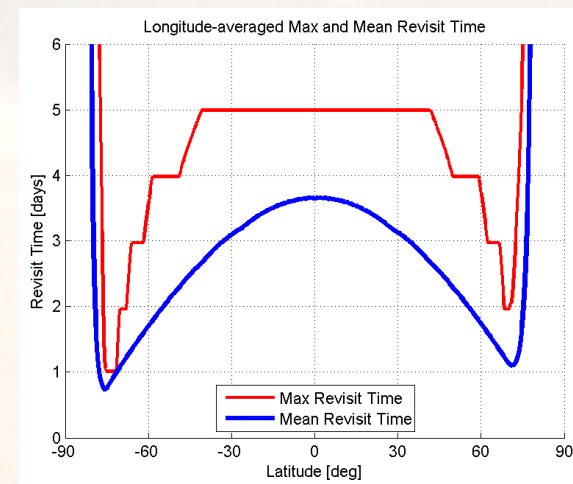
- Evolution of eclipse duration & beta-angle along the year
- Eclipse duration and beta-angle statistics



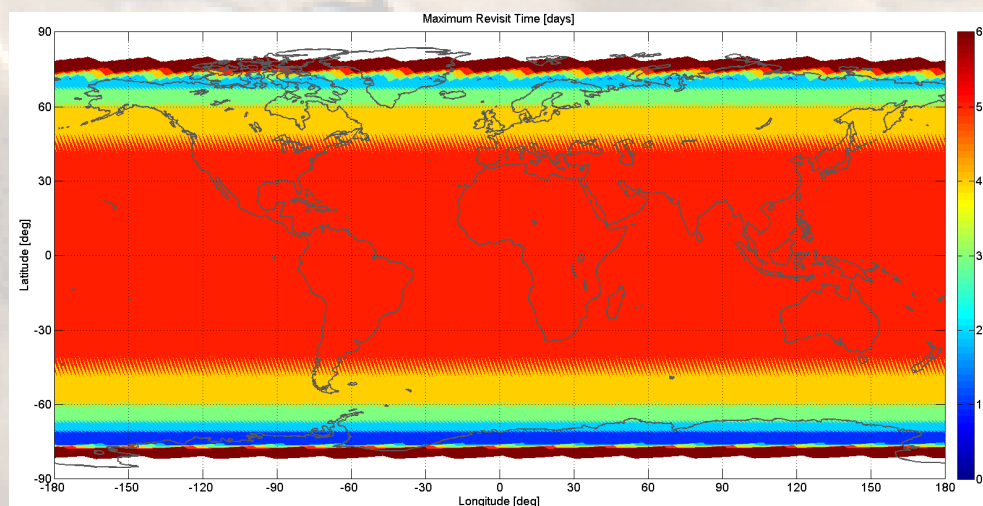
• Coverage Analysis

- Mission scenario
 - ✓ SSO 14+13/16 (~620 km), LTAN = 10:30
 - ✓ Nadir-centred FoR: $\pm 30^\circ$ (nominal), $\pm 45^\circ$ (extended), $\pm 15^\circ$ (background mapping)
 - ✓ Observation constraints: $SZA < 80^\circ$
- Coverage performance figures of merit
 - ✓ **Revisit time:** max, average (maps, statistics)

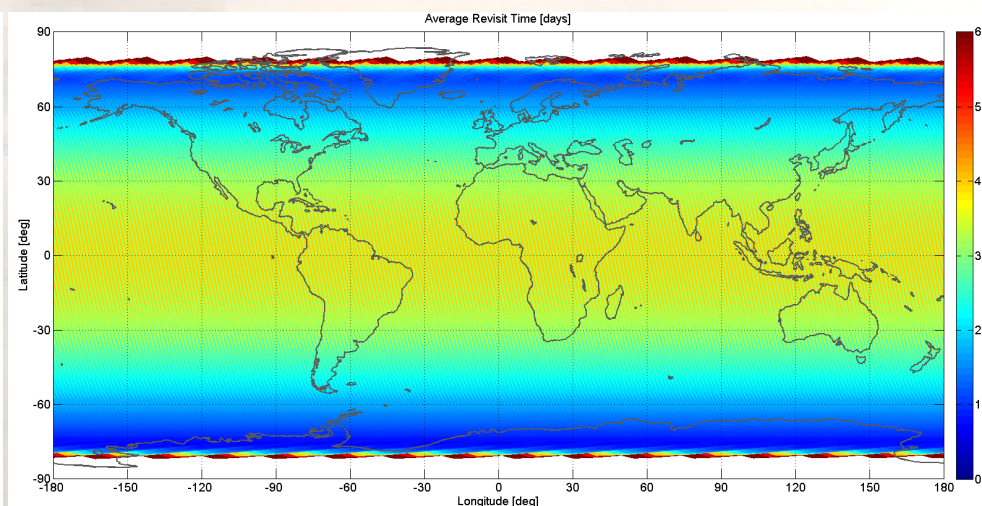
Longitude-Averaged Max and Mean RT vs. Latitude (FoR = $\pm 30^\circ$)



Max Revisit Time Map (FoR = $\pm 30^\circ$)

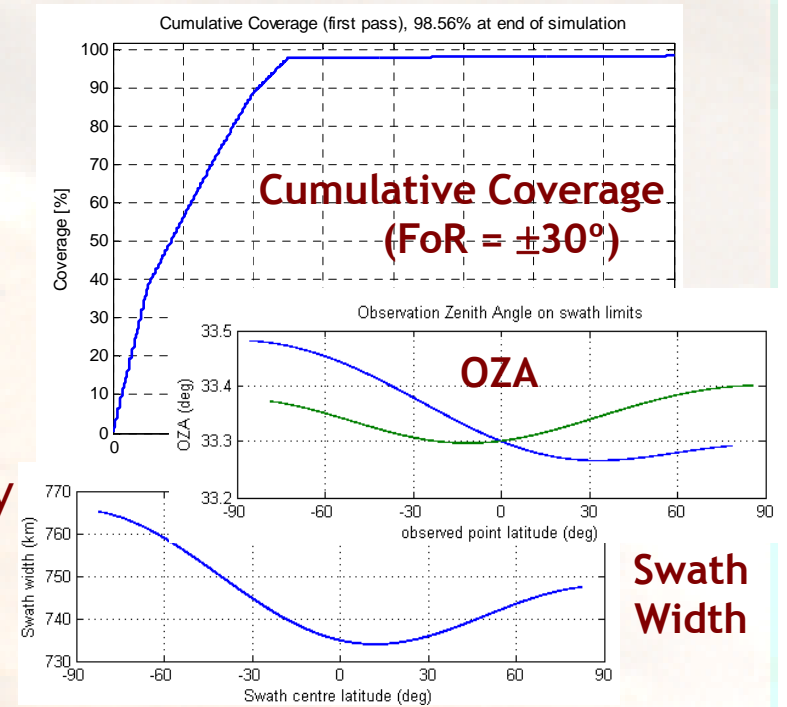


Mean Revisit Time Map (FoR = $\pm 30^\circ$)

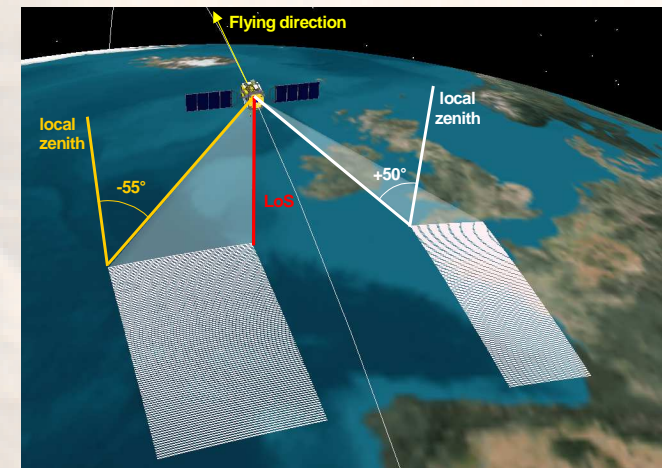
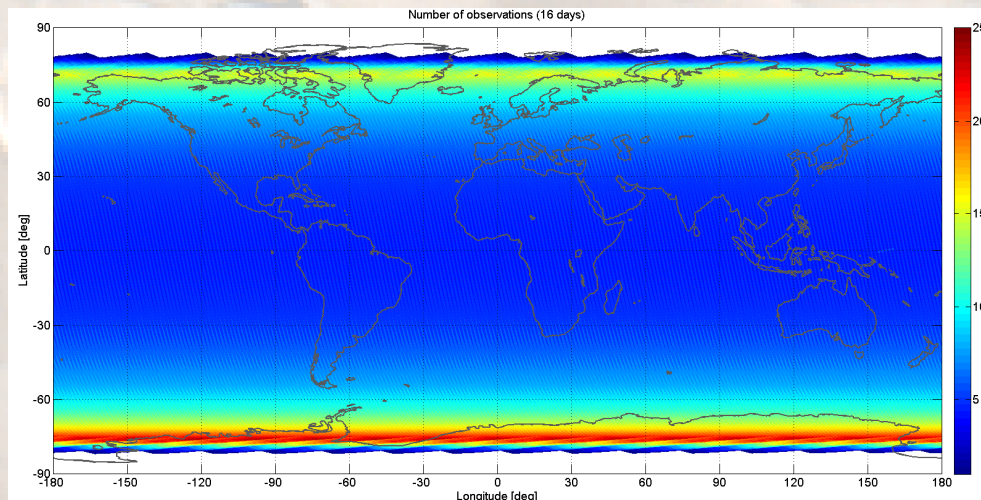


• Coverage Analysis

- Coverage performance figures of merit
 - ✓ **Revisit time:** max, average (maps, statistics)
 - ✓ **Coverage:** % of area of interest, cumulative vs. time
 - ✓ **Observations:** entire simulation, per day
 - ✓ **Observation geometry:** OZA, swath



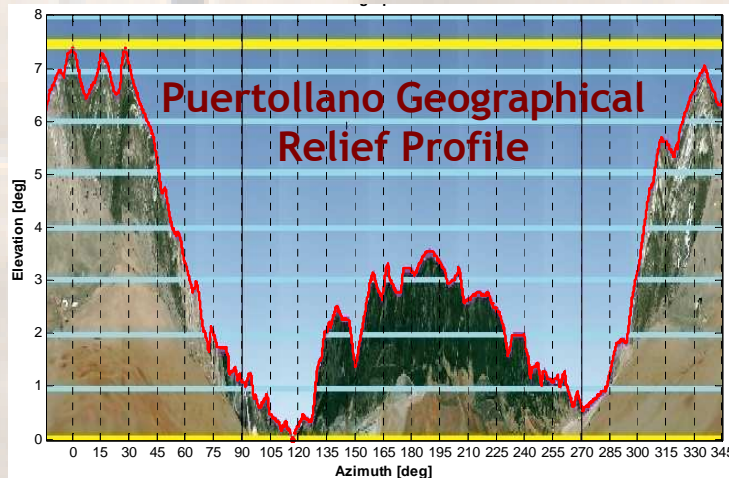
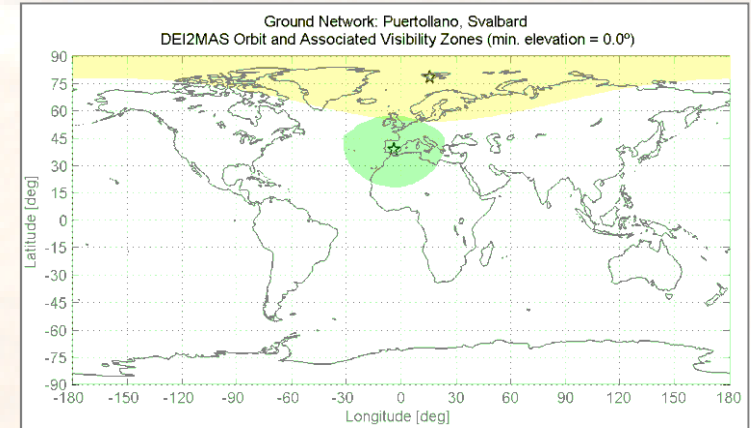
Observations Map (FoR = $\pm 30^\circ$)



• Ground Station Contact Analysis

- Mission and ground station scenario
 - ✓ SSO 14+13/16 (~620 km), LTAN = 10:30
 - ✓ GS: **Puertollano** (TTC in S-band, data in X-band), **Svalbard** (data in X-band)
 - ✓ Visibility constraints: az-el mask, min elevation, min contact time
- GS contact performance
 - ✓ GS visibility maps, GS contacts statistics
 - ✓ O/B & O/G visibility geometry

Svalbard and Puertollano Coverage Zones



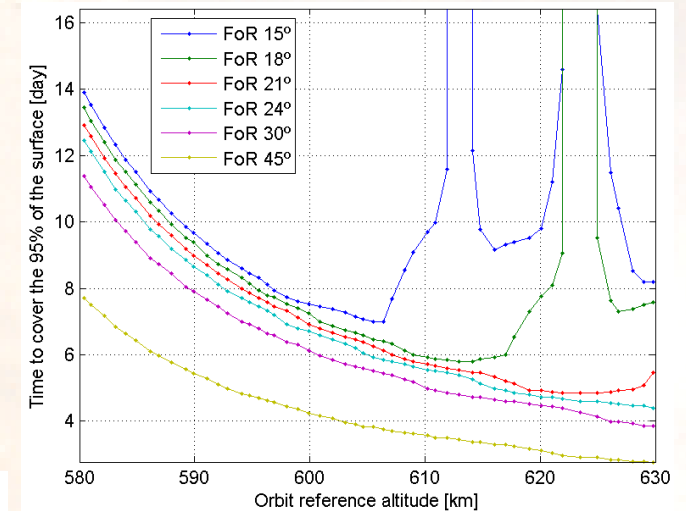
GS Contact Statistics

GS	Min Elev [deg]	Contacts / Max Blind Orbits	Average Contact Time			Max gap [hrs]
			[min/day]	[s/orbit]	[s]	
Svalbard	0°	237 / 0	172.5	698.6	696	1.5
	2°	237 / 0	153.7	622.4	624	1.5
Puertollano (with relief profile)	3°	74 / 6	38.7	156.8	504	11.7
	5°	69 / 6	34.9	141.3	486	11.7
Svalbard + Puertollano (with relief profile)	0°, 3°	311 / 0	210.1	851.0	648	1.5
	0°, 5°	306 / 0	206.3	835.5	648	1.5
	2°, 3°	311 / 0	192.2	778.6	594	1.5
	2°, 5°	306 / 0	188.4	763.2	594	1.5

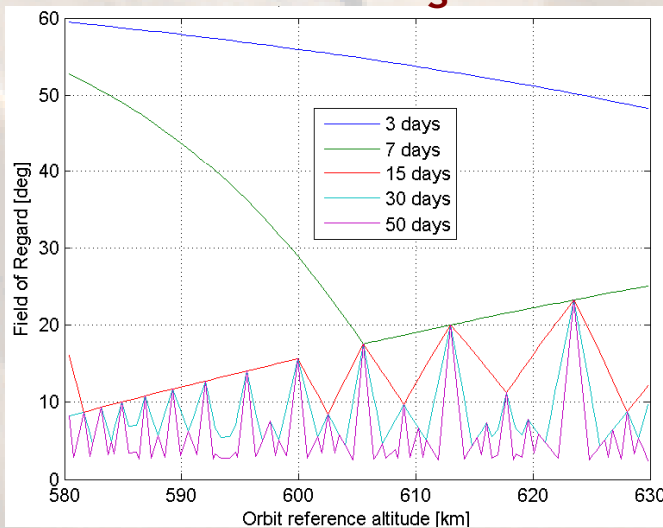
• Impact of Free Orbit Decay on Mission Performance

- **No orbit altitude control** implemented during mission lifetime
- **Coverage analysis vs. altitude**
 - ✓ Time to cover 95% of Earth surface
 - ✓ Percentage of area covered
 - ✓ Max and mean revisit time
 - ✓ FoR to achieve a given revisit time
 - ✓ Max OZA at swath edges

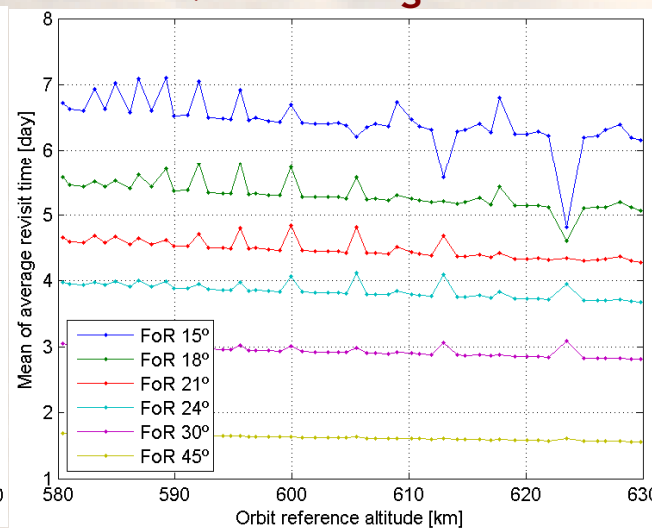
Time to cover 95% of the Earth



FoR needed for a given RT



Mean Average RT



- ✓ RT increases when altitude gets lower
- ✓ Large-enough FoRs, enabled by S/C agility, lead to **smooth effect of altitude on RT** in 590-620 km range

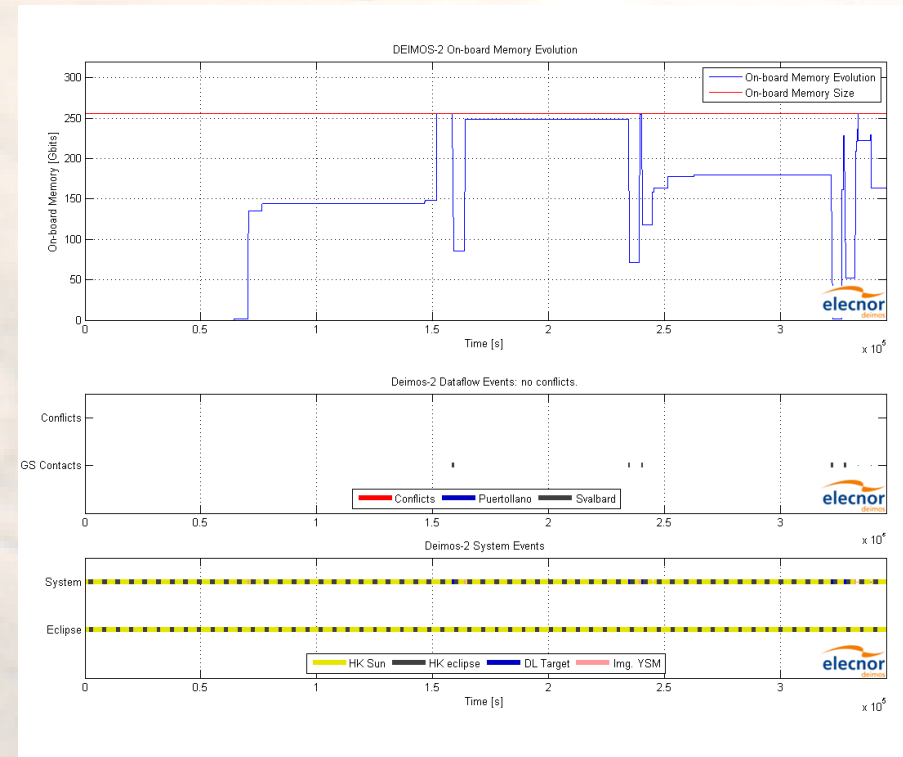
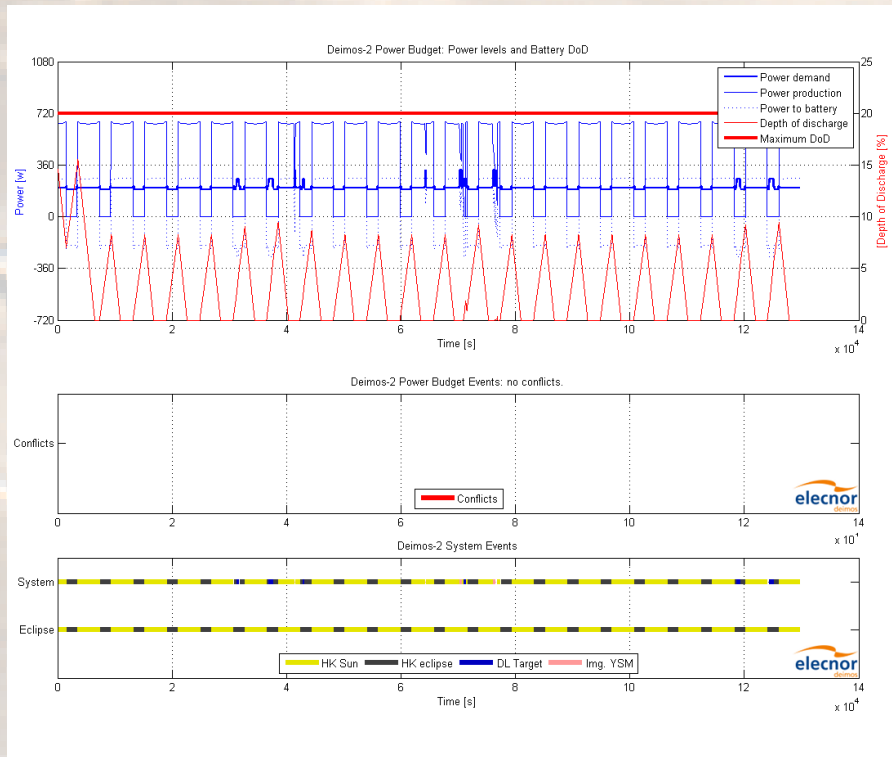
• ΔV and Fuel Budgets

- ΔV , fuel: per manoeuvre and total, no margins
- Technically feasible manoeuvre scenario w.r.t. S/C capacity (power, battery DoD) & thruster performance

Manoeuvre	ΔV [m/s]	Fuel Mass [kg]
Initial Orbit Acquisition		
Injection Error Correction	18.6	0.6
In-plane manoeuvres	5.4	0.2
Out-of-plane manoeuvres	13.2	0.4
Orbit Maintenance		
Tot in-plane orbit control	5.6	0.2
Average in-plane orbit control manoeuvre	0.002	0.0001
Collision Avoidance		
Tot in-plane collision avoidance	3.8	0.1
Average in-plane collision avoidance manoeuvre	0.54	0.02
EOL Disposal		
In-plane de-orbiting manoeuvre	19.7	0.6
Total Budget		
Total results	47.6	1.5

• Power and Mass Memory Budgets

- Agile mission with very complex power and mass memory management



- **DESEO is a powerful toolkit, designed to**
 - supply an exhaustive set of functionalities, covering the most common and frequent mission/system analysis needs
 - obtain fast quantitative results to support trade-offs and internal analyses
 - manage large analysis campaigns
 - a complete EO mission Phase 0 or Phase A study
 - minimize future developments implementation effort
 - modular and generic architecture

- **DESEO has a wide margin of extendibility covering**
 - the increase of computation capabilities
 - other system and more specific mission analyses modules
 - the consolidation of GUI for improving the user experience

- **DESEO has been proven successful in different missions**
 - EE Phases A/B1 and DEIMOS-2 Phases A/B/C/D/E