

DIPARTIMENTO DI SCIENZE E TECNOLOGIE AEROSPAZIALI 6th International Conference on Astrodynamics Tools and Techniques 17th March 2016 – Darmstadt, Germany

GNC TECHNIQUES FOR PROXIMITY MANOEUVRING WITH UNCOOPERATIVE SPACE OBJECTS

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Background – ACTIVE DEBRIS REMOVAL



Credits ESA

- Active servicer s/c
 - Non-cooperative tumbling target
 - Approach & Docking
 - De/Re-Orbiting
- GNC challenges autonomy
- Preferred strategies for ADR:
 - **ROBOTIC ARM** rigid connection
 - THERED-NET flexible connection

- 6000 satellites into orbit, <1000 still operational
- LEO and GEO orbits
- ADR focused on massive elements, effective if 10 objects removed per year



Figure 1. LEGEND-simulated LEO debris populations (objects 10 cm and larger) between 1957 and 2006 (historical), and between 2007 and 2206 (future projection). Each curve represented the average of 100 Monte Carlo runs.

Credits Liou, NASA, IAC 2007



Background – ON ORBIT SERVICING



DEOS: credits DLR

- Active servicer s/c
 - Partially cooperative target
 - Approach & Docking
 - Refuelling
 - Repairing/Maintenance
- GNC challenges autonomy
- Strategy for OOS:
 - **ROBOTIC ARM** with endeffector
 - Grippers/Clamping mechanisms

RESEARCH OBJECTIVES

- 1. Develop fast & reliable dynamics models for ADR and OOS system and GNC design
- 2. Implement GNC laws for proximity manoeuvring to increase level of autonomy
- 3. Validating dynamics and test control experimental activities and HIL



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ADR/OOS Simulation Environment

- Fast simulation environment to support GNC design
 - Born for ESA's study MUST [Benvenuto, 2014]
 - Further upgrades after activity
- Fully integrated in Matlab/Simulink
 - Simulink libraries
 - SimMechanics
- Describe 6-DOF orbital and attitude bodies dynamics with:
 - Flexible tethers/nets
 - Robotic manipulators
 - Flexible appendages and sloshing
 - Environment and perturbations
 - GNC blocks
- Tool validated by means of benchmarking and experimental activities







Tethers & Nets



Flexible Appendages

DOF & Joints



Environment & Perturbations



GNC Blocks



Utilities & Settings

ADR/OOS Simulation Environment





Flexible models based on lumped parameters methods

- Model higher order modes
- Approximate flexing/whipping arbitrary number of discretizing elements
- Fast computations
- Accounting for different viscoelastic laws
- **Parametric**, different configurations





- Original concept for **SPACE TRANSPORTATION**:
 - Exploitable for Active Debris Removal
 - Passive target connected to an active chaser, its thrusters exciting stack
 - Fixed-length flexible tethers
 - V-bar alignment configuration unstable
 - Fast dynamics
 - Target mass > chaser mass



Critical modes and instabilities

- Whiplash effects: leading to difficult control recovery, (necessary tether tensioning and stabilization)
- Bounce-Back effects: post burn phaseslack tether and possible collisions due to residual tension
- Tether entanglement and breakage: avoid free motion and slack tether
- Atmospheric re-entry: chaser control authority holds till a minimum altitude where drag on target prevails

Stack G&C

Closed loop:		
Stabilization + Bang-o post-burn recovery [<i>Benvenuto, 2014</i>]	off +	
Tension feedback to	reduce	
DOUNCE-DACK	Open loop: Feed-forward shaping of commanded thrust profile [Jasper, 2014] • Input shaping • Command smoothing/filtering	
Closed loc control	op attitu	de



Input Shaping

- Signal convolved with series of impulses
- Thrust modified by steps
- Achievable with thrusters' cluster (on-off)
- Increased time to obtain same ΔV

Command smoothing

- Signal filtered
- Thrust modified continually
- Need continuous thrust modulation throttling
- Increased time > shapers
- **Robustness to uncertainties** (ω, ξ) > shapers





ZVD Shaper





Frequency sensitivity

ZVD

5%

Smoother

100



Chaser Mass [kg]	1300
Target Mass [kg]	5000
Initial orbit altitude [km]	600
Thrust [N]	800
Tether Young's modulus [GPa]	32
Tether damping factor [-]	0.1
Tether length [m]	60
Tether diameter [m]	0.003
ΔV [m/s]	160
Flight angle at 120 Km [deg]	-1.6

y [m]

Bang-Off

ZVD Shaper

3000

C - Smoother







Experimental Microgravity Validation





- PoliTethers Team selected for ESA's Education Fly Your Thesis! 2016
- Parabolic flight campaign October 2016
- Release and retrieval of tethered floating module
- Actuator to simulate thrust profiles
- Stereo-vision reconstruction + acceleration/tension sensors





Benefits

- Larger capture distance
- Isotropic loads
- Partially independent from target features (physical and dynamic)
- Centre of mass alignment with thrust axis not a constraint
- Scalable device, light payload

Criticalities

- Passive device (except if controlled bullets)
- Non repeatability one shot chance
- Difficulty in detecting capture
- Possible slippages: closing device





Enhanced modelling

- Collision detection
 - Multi-step refinement computational time
 - Hierarchical bounding boxes
 - Avoid interpenetration
 - Auto-collisions

Contact dynamics

- Non-linear contact law (H-C)
 - Avoid shock loads
 - Avoid sticking effects
- Regularized friction law
- Coefficients determined through lab tests – characterization of mechanical properties









G&C

- RDV along track
- No reaction compensation at shooting occurrence (limited bandwidth)
- Chaser attitude controlled (eigenaxis)

$$u = \omega_C \times I_C \,\omega_C - D \,\omega_C - K \,q_e$$

$$K = \omega_{BW}^2 I_C \qquad D = 2\xi \omega_{BW} I_C$$

 Tumbling target passive angular momentum damping



Friction damping effect during towing [*Benvenuto, 2015*]



Parabolic flight campaign to validate net flexible model and contact model

ESA's sponsored activity **PATENDER** [*Medina, 2015*]: PoliMi in consortium with GMV Spain and Prodintec Spain

Experiment design:

- Nets shot at target mock-up
- Reconstruction of net 3D trajectory thought high resolution- high speed cameras
- Reconstruction process: based on net colourcoding, stereo-matching and knots tracking
- Target position and attitude reconstruction based on mock-up markers
- Simulator validation based on trajectory comparison



Novespace, Bordeaux, June 2015







Parabola 17, Frame N=283, Time=0.13347s



Parabola 17, Frame N=285, Time=0.16683s











Parabola 17, Frame N=304, Time=0.48382s



Robotic Arm Scenario

GNC

- Different guidance phases:
 - Motion synchronization at safe distance
 - Final approach along Haxis and arm deployment
 - Capture & rigidization
 - De-tumbling and disposal
- Coordinated control strategy (decoupled):
 - Chaser:
 - LQR for relative position control
 - Eigenaxis attitude control
 - Arm:
 - Joints coordinates & Cartesian coordinates (IK)
- Noisy measurements + Kalman Filter





Robotic Arm Scenario

Chaser Mass [kg]	1450
Target	Envisat (simplified model)
Target angular velocity [deg/s]	3.5 H-axis
Arm Mass [kg]	30
Arm links [#]	2 + Gripper
Arm DOF [#]	4
Forward reach [m]	3.1
Control set-point rate [Hz]	10
Joints angular accuracy [deg]	0.3

Capture simulation results













POLITECNICO MILANO 1863

Conclusions

- Autonomous servicing/removal missions:
 - complex, multidisciplinary and challenging task
 - capturing tumbling objects not equipped with dedicated docking ports
 - sophisticated technologies and reliable lightweight manipulators
- Developed dynamics simulator allowed
 - to assess the capture technique feasibility
 - to drive system design
 - to support GNC design
 - selected mathematical model ensures
 - fast and numerically stable simulations
 - flexibility in different scenarios analysis
 - Experimental activities on-going to validate simulation models and test G&C
- Throw-nets and towing-tethers VS. robotic manipulators
 - different scenarios analysed
 - preliminary G&C laws tested in simulation
 - Open loop preferred for tether
 - Closed loop preferred for arm





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