



# cleansat

BBO2

Thermoplastic propellant tank  
AIRBUS DEFENCE AND SPACE

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# Description of proposed technology Building Block

## Baseline

- Liner material: rotomolded thermoplastic liner
- Shell: thermoplastic composite overwrap

## Trade off variable

- Liner material (thermoplastic grade)
- Composite shell material (carbon / kevlar)
- Propellant management system

## Trade-off criterion

- Compatibility with propellant
- Demisability
- Mass
- RC/NRC
- Development time



# Description of proposed technology Building Block



- Applicability range (satellite class and target orbits in LEO)
  - LEO platform with blowdown chemical propulsion
  - Propellant tank size about 200L and less
  
- Discussion of the system level impacts (risk, mass budget, power budget and link budget)
  - Main risk identified during trade-offs is a versatility of thermoplastic liner grade to futur green propellant
  - Main impact at system level is the limitation to bladder type propellant management of this concept



## — Main technical challenges during development

- Final choice of liner material grade vs futur green propellant
- Bladder interface to tank as to be challenged
- CFRP composite overwrap is still baselined but additional work has to be performed to show compliance with demisability requirements:
  - Behavior of hollow composite tank after matrix pyrolysis
  - Ground impact of an almost dry fiber tank