SpaceWire Physical Layer Testing





- ~ Context & objectives
- Test procedure !
- Test results (extracts)
- Recommendations & ways forward



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SpaceWire context & study objectives (1)

SpaceWire: network or point-to-point technology enabling 2-200Mbps communications
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SpaceWire is a key technology deployed over all new avionics & baselined in all new CPU SoC developments

- ECSS standards exist since 2003 (ECSS-E-50-12A originally)
- Deployed over many missions (and baselined on all new TAS missions)
- ~ Active working group with worldwide community (EU, Japan, Russia, US & more)
- Many updates scheduled (standard upgrade, new services around SpaceWire, SpaceFibre)





SpaceWire context & study objectives (2)

But: there is still no standard procedure to validate SpaceWire

Main issues:

- Fast data link (need to take care about the eye diagram)
- ➤ Variable link speeds
- Low voltage technology & sensitive to common modes
- No clock line: data & strobe technology
 - TX: STROBE = DATA XOR CLOCK (strobe encoding)
 - **RX:** CLOCK = DATA XOR STROBE (clock recovery)

SpaceWire characteristics defined by std, but board? System?

Main goals of this activity are to:

- >> Define a test procedure for the SpaceWire physical layer
- >> Develop a test board to challenge the procedure and propose a reference implementation & test mean
- Provide guidelines for testing & designing test means

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SpaceWire characteristics \rightarrow System?

How to check electrical detailed characteristics in case a link is misbehaving?





Define a thorough validation approach enabling SpaceWire physical layer testing for any SpaceWire equipment:

DC-CM tests (Derived from TIA-644-A limits – LVDS standard)

AC-CM tests (Related to the condition of usage of the interface within the spacecraft & electrical mass global management policy; needs to be tuned as per specific implementation)

>> Eye-pattern tests & signal quality measurements

➤ Cross-talk tests

🛰 BER tests

- EMC/EMI tests

Thermal tests

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>>> DC-CM tests

The SPW DC common mode test purpose is to measure the overall Vos of the LVDS drivers and the overall Vos of the SpW Board: required for drivers operation

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DC-CM test set-up of Vos_tot measurement on FPGA PCB Boards



>> DC-CM test set-up of Vos_tot_Board measurement on FPGA PCB Boards



>>> DC-CM tests

>> With a test-aid to reach the relevant measurement points on the hardware to be tested



Success criterion (ECSS-E-ST-50-12C standard)

| Parameter | Min | Мах | |
|--|--|---------------------------------|--|
| Vos (steady-state value of the driver offset voltage) | 1125mV | 1375mV (1450mV in rev1) | |
| $\Delta Vos = Vosh-Vosl$ | 50mV | | |
| ∆Vos Dynamic-State | 150mV(peak-peak) | | |
| Vod (output/driver differential voltage) to be measured in both logical states | 247mV | 454mV | |
| ∆Vod in Steady-State | 50 mV | | |
| Vos_tot | 1050mV | 1450mV (1525mV) | |
| Vcm_driv_FPGA | 50 mV (due to the number of terminals) | | |
| Vcm_driv_10X | 50 mV (due to the number of terminals) | | |
| Vos_tot_board_FPGA | 1050mV | 1500mV (1575mV) | |
| Vos_tot_board_10X | 1050mV | 1500mV (1575mV) | |
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- AC-CM tests

Measurementof the impedance of the LVDS driver/10X router secondary ground connection to the box chassis.



∼ The impedance not to exceed is $Z = \frac{Vcm_{driv_{FPGA}}}{100 \, \mu A * SQRT(2)}$ so Z ≤ 250 Ω.

| Main Frequency(MHz) | Third Harmonic (MHz) | Fifth Harmonic (MHz) | A CARLES |
|---------------------|----------------------|----------------------|---------------------|
| 2 | 3 | 5 | |
| 10 | 15 | 25 | |
| 100 | 150 | 250 | |
| 150 | 225 | 375 | |
| 200 | 300 | 500 | |
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~ Eye pattern tests

The eye pattern test aims at verifying the signal integrity at receiver level with combined effect of all the communication chain



T1: combination of rise-time, skew and jitter, triggered on the latest signal transition (Data XOR Strobe)

- T2: minimum edge separation allowed by the receiver
- T3: combination of fall-time, skew and jitter
- For a good communication: T1+T2+T3 <= Bit period</p>

Measured minimum edge separation: minimum duration on which the signal is valid

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~ Eye pattern tests

The eye pattern shall be measured at receiver termination resistor level by :

7 Either using the reconstructed clock of the signal (logic: Data XOR Strobe) pending that the timing degradations of the scope are negligible with respect to the bit period and measured the Data and Strobe signal while triggering on the reconstructed clock signal.

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- Or If the timing degradations of the scope are not negligible or this feature is not available in **~**2. the scope's functionalities, the following measurement set up is predefined:
 - Measurement of the Data and Strobe signal while triggering on the Strobe signal positive and negative edge at +100mV for rising edge and -100mV for falling edge (or preferably the Rx threshold if known).
 - Measurement of the Data and Strobe signal while triggering on the Data signal positive and negative edge at +100mV for rising edge and -100mV for falling edge (or preferably the Rx threshold if known).



Eye pattern tests: which pattern?

There are worst-case patterns that affect specific tranceivers and needs to be checked:

- >> Default: random transitions through the transmission of PRBS patterns
- ~ TS+S- : Strobe 0: {Long chains of [0x55], [0x57, EoP]}
- TS-S+ : Strobe 1: Long chains of single 0xAA N-Chars each ended with EOP
- TD+D- : Data 0: {Long chains of {[0x00], [0x01, 0x00, EoP]), (long chain of [0x00], [0x01, 0x00, 0x00], EoP)}
- TD-D+ : Data 1: {Long chains of [0xFF], [0xFE], ESC(ape), 0xFF,0xFF], EOP}

Tests have been performed over 2 Mbps, 10Mbps, 100 Mbps, 150 Mbps and 200 Mbps:



The Success criteria of the EPT is at receiver termination resistor level : Measured min edge separation > allowed min. edge separation = [skew + jitter] contribution of the LVDS receiver (value coming from the datasheet) + FPGA routing contributions (value coming from the FPGA timing analysis)

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Eye pattern tests: a test defined in ECSS-E-ST-50-12Crev1:

➤a. When a test signal is sent by the line driver through the SpaceWire connectors and cable assembly, the received signal measured at the receiver termination resistor or within 20 mm of the line receiver inputs shall have:

1. a minimum edge separation which is greater than the minimum edge separation required by the SpaceWire line receiver, see clause 5.3.6,

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> 2. a differential amplitude of more than +100 mV or less than -100 mV in the region between the edges, and

- ➤ 3. a differential amplitude of less than 600 mV.
- ∼ NOTE […]

➤b. The test signal shall comprise a repeating sequence of the following set of SpaceWire packets following one another sent without any Nulls between the data characters:

- 1. SpaceWire packet of length greater than 1000 bytes containing a pseudorandom bit sequence;
- 2. SpaceWire packet of length greater than 1000 bytes all set to 0x55.
- ∼ NOTE [...]

➤c. The receive signal shall be measured across the line receiver termination resistor using an oscilloscope and differential probes which have bandwidths of at least 1,05 times the reciprocal of the signal rise time.

~[…]

>d. Where the termination resistors are internal to an integrated circuit, the receive signal may be measured within 3 cm of the line receiver pins on the integrated circuit.

>e. When access to the termination resistors is not possible, the receive signal may be measured at the connector adjusting for the transfer impedance between the connector and the termination resistor or other equivalent method.



~ Cross-talk tests

The cross talk test will identify any perturbations on the receiver side coming from the interferences between LVCMOS signals and the LVDS signals generated on the same board



The Success and Failure criterions of the Cross-talk test shall be based on an EPT test: no degradation shall affect the EPT signal shape/timing results, validated by the minimum edge separation

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$$P\left(\varepsilon < N\right) = \sum_{k=0}^{N} P_n\left(k\right) = \sum_{k=0}^{N} C_{n,k} \cdot p^k \cdot q^{n-k} \quad \mathbf{CL} = \mathbf{1} - \mathbf{e}^{-\mathbf{N} \times \mathbf{BER}_{\mathbf{S}}} \times \sum_{\mathbf{k}=0}^{\mathbf{E}} \frac{\left(\mathbf{N} \times \mathbf{BER}_{\mathbf{S}}\right)^k}{\mathbf{k}!}$$

million BER tests

➤ Evaluation of the Bit Error Rate over a given duration. For such a link: BER figure expected < 10⁻¹²

The BER test durations shall be as minimum as follows for 2 Mbps, 10Mbps, 100Mbps, 150Mbps and 200Mbps:

| Dara Rate (Mbps) | Number of observed errors | Minimal duration for | Minimal duration for | Minimal duration for | |
|------------------|------------------------------|----------------------|----------------------|----------------------|--|
| | | 1 sigma accuracy | 2 sigma accuracy | 3 sigma accuracy | |
| | | (67% confidence) | (95% confidence) | (99.7% confidence) | |
| 2 | 0 | 153.98 | 416.07 | 806.83 | |
| 2 | 1 | 320.83 | 659.72 | 1112.50 | |
| 2 | 2 | 480.56 | 875.00 | 1376.39 | |
| 10 | 0 | 30.80 | 83.21 | 161.37 | |
| 10 | 1 | 64.17 | 131.94 | 222.50 | |
| 10 | 2 | 96.11 | 175.00 | 275.28 | |
| 100 | 0 | 3.08 | 8.32 | <mark>16.14</mark> | |
| 100 | 1 | 6.42 | 13.19 | 22.25 | |
| 100 | 2 | 9.61 | 17.50 | <mark>27.53</mark> | |
| 150 | 0 | 2.05 | 5.55 | <mark>10.76</mark> | |
| 150 | 1 | 4.28 | 8.80 | <mark>14.83</mark> | |
| 150 | 2 | 6.41 | 11.67 | <mark>18.35</mark> | |
| 200 | 0 | 1.54 | 4.16 | <mark>8.07</mark> | |
| 200 | 1 | 3.21 | 6.60 | <mark>11.13</mark> | |
| 200 | 2 | 4.81 | 8.75 | <mark>13.76</mark> | |

The BER test can be considered as successful or failed by checking the following table:

| CL | 1sigma | | 2sigma | | 3sigma | |
|----|--------|--------|--------|--------|--------|--------|
| - | REJECT | ACCEPT | REJECT | ACCEPT | REJECT | ACCEPT |
| E | (>= E) | (<=E) | (>= E) | (<=E) | (>= E) | (<=E) |
| 0 | NA | 1.11 | NA | 3.00 | NA | 5.81 |
| 1 | 0.41 | 2.31 | 0.06 | 4.75 | 0.01 | 8.01 |
| 2 | 1.18 | 3.46 | 0.36 | 6.30 | 0.08 | 9.91 |
| 3 | 2.03 | 4.58 | 0.82 | 7.76 | 0.29 | 11.65 |
| 4 | 2.90 | 5.69 | 1.37 | 9.16 | 0.59 | 13.31 |

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~ EMC/EMI tests

SpW EMC performance characterization of the 3 SpW Boards with respect to radiated emission and susceptibility





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~ CE: Conducted Emission Test

>> The narrowband electric field radiated emissions from the specimen including interconnecting cables shall not exceed the specified limits (range 10 kHz and 400 MHz) defined in the Figure here-below:



+ CS: Conducted susceptibility Test + RE-E: Radiated Emission, Electric field + RS-E: Radiated Susceptibility, Electric field **ThalesAlenia**

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🛰 Thermal tests

>> Verification of compliance status over the operational thermal temperature range

The Success and Failure criterions of the Thermal test shall be based on a BER test.

~ For the activity: tested at -30° C & $+80^{\circ}$ C

| | Performance electrical Tests | | | | |
|----------------|------------------------------|------|------|------|------|
| | BER | | | | |
| Data- | 2 | 10 | 100 | 150 | 200 |
| rate | Mbps | Mbps | Mbps | Mbps | Mbps |
| Ambient T°C | | Х | Х | Х | Х |
| - 30 °C | | | Х | | Х |
| + 80 °C | | | Х | Х | Х |



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Specific test board were developed by TAS-I to undergo the test procedure, verify the¹⁸ implementation compliance and cross-check the test procedure relevance

~ All done expect for EMC tests (scheduled week 50)

 \sim AC-CM tests modified: computed initially according to a specific impedance/inductance pattern which is not an actual fail/success criterion \rightarrow a mask has been defined based on common mode conducted emission level instead

Edge Separation recorder for data rate of 200 Mbits/s, at Receiver Input (Soldered-Transceivers Board left, Socketed-Transceivers Board right)

| Time | Edge Separation Recorded (Ambient Temperature) | Edge Separation Recorded (+3°C) | Edge Separation Recorded (Ambient Temperature) | Edge Separation Recorded (+3°C) |
|-----------------------|---|---------------------------------------|---|---------------------------------------|
| T _{S+D-} | 4.02ns | 2.96ns | 2.62ns | 3.01ns |
| T _{S+D+} | 3.18ns | 3.04ns | 2.55ns | 2.94ns |
| T _{S-D-} | 3.33ns | 3.23ns | 2.63ns | 3.06ns |
| T _{S-D+} | 3.34ns | 3.23ns | 2.61ns | 3.12ns |
| T _{D+S-} | 3.06ns | 3.06ns | 2.82ns | 3.06ns |
| T _{D+S+} | 3.06ns | 3.06ns | 3.68ns | 2.13ns |
| T _{D-S-} | 2.13ns | 2.16ns | 2.72ns | 2.01ns |
| T _{D-S+} | 2.03ns | 2.09ns | 2.73ns | 2.05ns |
| T _{S+S@100} | 3.69ns | 3.68ns | 3.59ns | 2.92ns |
| T _{S+S@1000} | 3.76ns | 3.71ns | 3.59ns | 2.89ns |
| T _{S+S@4096} | 3.64ns | 3.74ns | 3.59ns | 2.91ns |
| T _{S-S+} | 3.52ns | 2.74ns | 3.64ns | 2.85ns |
| T _{D-D+} | 3.61ns | 3.54ns | 2.71ns | 2.69ns |
| T _{D+D-} | 3.72ns | 3.64ns | 3.68ns | 3.68ns |





>> SpaceWire is a high-speed, low-voltage, network technology

>> It needs to be designed & handled with care: not as straight-forward as a bus

>> Each link can have its own characteristics (speed, transceiver, boards design, routing within the spacecraft, cable shielding policy, cable length, etc...)

As for a RF link, it needs its own "link budget": skew/jitter apportionment & specific check-ups

This activity enabled defining a test procedure and characterized a test setup using representative elements: ESA CODEC & Gaisler Cobham transceivers (socketed or soldered and 2 separate boards)

A key point: BER test can provide a confidence in the communication establishment in a specific configuration & environment but does not provide any margin figure nor provides an explanation in case of failure

Hence the need for thorough testing !

Test procedure & report will be available to the community



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Thanks for your attention !

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