

Space Missions in IHEP & Geant4 Applications

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OUTLINE

Space Missions in IHEP

 HXMT, POLAR, SVOM/GRM, eXTP, ...

- Application of Geant4 simulation on these instruments
 - background estimation、calibration(arf&rmf)、 data analysis、design optimization、shielding etc.

Hard X-Ray Modulation Telescope

- HXMT:
 - Orbit: ~550km, ~43°
 - Collimated X-ray telescope
 - 1~250keV
 - Launch time 2017 June
- Payloads:
 - HE(Nal/Csl, 20-250keV)
 - ME(Si-Pin,5-30keV)
 - LE(SCD,1-15keV)
 - SEM(space environment monitor)
 - Particle Monitors



In-orbit Background Simulation for HXMT





FIG. 6.—HEXTE background for a single detector plotted versus energy. Background lines due to activation of the iodine by cosmic and trapped radiation are evident at 30, 55, 66, and ~190 keV, as are the $K_{\alpha\beta}$ lines at 74 and 85 keV from fluorescence of the lead collimator. The data are from a 7.5 hr observation of blank sky fields.



Geant4 Simulation Helps to Determine the Inorbit Background Estimation Methods





Difference of the particle background between different FOV/position detectors can be used to determine the background analysis method and its uncertainties. Using in-orbit observation data to determine these bias



component	rate, cnts/s	rate after veto between pixels, cnts/s
схb	15.2	14.9
crp	39.9	17.1

Cosmic ray protons background can be reduced to ~50% after veto between pixels.

HXMT calibration

- Experiment + MC
 - E-channel relation
 - Energy resolution
 - Effective area
 - Response matrix
 - Etc...

Ground Calibration of HXMT/HE

Double-crystal monochromator for HE



Nonlinearity response of X-ray to Nal:



- For each energy point, the X-ray beam was sequentially poured into the 27 points sampled on the phoswich plane.
- The accumulated spectrum from these 27 points was considered as the response of the detector plane to a parallel X-ray beam uniformly illuminating the surface of HE.

Visualization of HE main detector



Position of incident X-ray:





D P

0.5

0.4 LL

40

60

80

Energγ(keV)

100

120

140

160

10⁻⁹

100

0

50 100 150 200 250 300 350 400 450 500

E(keV)

MC vs Data

10² Energy(keV)

1000 F

500

Ground Calibration and MC of ME



Schematic view of the calibration vacuum chamber

ME also used the double crystal monochromator for its calibration.





LE: Swept Charge Devices



Only take the interaction of X-ray and materials:

Readout mechanism of LE:





- The process of charge transport in Swept Charge Devices is very important to understand the nature of the spectral distribution function, especially to explain the *shoulder on the left side of the main peak*. The events on the shoulder are almost split events.
- Simulation method of LE:
 - Interaction of X-ray and materials in LE (primary position of deposit energy, by Geant4) + charge transport (Analytical model)+ Readout mechanism of LE

LE: Charge transport process

The method similar to Chandra and XRT (Townsley et al. 2002, Godet et al. 2009) Reference: NIM A 428(1999)348-366

Depletion Region:

$$r_{i} = 0.0044 \left(\frac{E_{f}}{1 \text{ keV}}\right)^{1.75} \mu m \qquad r_{d} = 10^{4} \sqrt{\frac{2kT\epsilon_{s}}{e^{2}N_{a}}} \ln \frac{d_{d} + \epsilon}{d_{d} - z_{0} + \epsilon} \mu m$$

$$Q_{ij}(x_{0}, y_{0}) = \frac{Q_{0}}{4} \left[erf\left(\frac{a_{i+1} - x_{0}}{\sqrt{r_{i}^{2} + r_{d}^{2}}}\right) - erf\left(\frac{a_{i} - x_{0}}{\sqrt{r_{i}^{2} + r_{d}^{2}}}\right) \right] \times \left[erf\left(\frac{b_{i+1} - y_{0}}{\sqrt{r_{i}^{2} + r_{d}^{2}}}\right) - erf\left(\frac{b_{i} - y_{0}}{\sqrt{r_{i}^{2} + r_{d}^{2}}}\right) \right]$$

Field Free Region:

The method similar to Chandrayaan-2 (Athiray et al. 2012)

$$\begin{aligned} Q_{ij}(x_0, y_0) &= \frac{Q_0}{4\pi^{1/2}} \int_0^\infty \frac{d\tau}{\tau^{3/2}} G_i(\alpha, \xi_0, \tau) G_j(\beta, \eta_0, \tau) S(\zeta, \tau) \exp\left(-\frac{\tau}{4\Lambda^2}\right) \\ G_i(\alpha, \xi_0, \tau) &= \operatorname{erf}\left(\frac{\alpha_{i+1} - \xi_0}{\sqrt{\tau + \tau_d}}\right) - \operatorname{erf}\left(\frac{\alpha_i - \xi_0}{\sqrt{\tau + \tau_d}}\right) \\ S(\zeta, \tau) &= \sum_{m=-\infty}^\infty C_m \left(\zeta - 2m\right) \exp\left[-\frac{(\zeta - 2m)^2}{\tau}\right] \\ Q_{ij}(x_0, y_0) &= \frac{Q_0}{4} \left[\operatorname{erf}\left(\frac{\alpha_{i+1} - x_0}{r}\right) - \operatorname{erf}\left(\frac{\alpha_i - x_0}{r}\right)\right] \left[\operatorname{erf}\left(\frac{b_{i+1} - y_0}{r}\right) - \operatorname{erf}\left(\frac{b_i - y_0}{r}\right)\right] \end{aligned}$$

Method 1 -> time consuming!

Method 2

Some results of LE

Geant4 simulation+ Charge Transport+ Readout Mechanism



Effective areas of LE detectors:



PSF:



POLAR: Gamma-ray Burst Polarimeter

- Determine polarization by direction distribution of photon Compton scattering
- 1600 channels (5×5 modules, 8×8 bars per module)
- Energy range 50 500 keV



ESRF on-Ground data vs MC





Plots are from M. Kole's paper, preparing

Geant4 simulation + digitization of electronics

Reproduce the on-Ground data

Use the simulation + digitization software to generate the in-orbit geometry modulation curve and the μ_{100} of GRBs

SVOM/GBM

- Space-based multiband astronomical Variable Objects Monitor
- Orbit: ~635km, 30°
- GBM: 15-5000 keV
 - 3 GRDs
 - PS/Nal detector



Background simulation of SVOM

Component	Count rate(count/s)	
CXB (prompt)	795	
Albedo Gamma (prompt)	61	
Cosmic Proton (prompt/delayed)	87/22	
SAA (delayed)	116	
Cosmic Electron (prompt/delayed)	5.97/0.048	
Albedo Neutron (prompt/delayed)	1.08/0.67	
Total	~ 1088	





Bkg variation with zenith angle

space missions: instruments and detectors

3D equivalent aluminum thickness based on simplified mass model used for Total Ionization Dose



enhanced X-ray Timing and Polarimetry (eXTP)



Payload	Configuration	Optics	Detectors
Spectroscopic Focusing Array (SFA)	9 telescopes	Nickel/SGO	SDD
Large Area Detector (LAD)	40 modules	MCP collimator	SDD
Polarimetry Focusing Array (PFA)	4 telescopes	Nickel	GPD
Wide Field Monitor (WFM)	6 cameras	1.5D coded mask	SDD
Gamma Ray Burst Monitor (GRM) (optional)	3 units	-	Scintillator

Preliminary configuration of SFA Focal Plane Detector used for simulation



A simple configuration of the FPD unit

Particle Background Level Estimate using Geant4



106

10⁴

Observaion Time (s)

10⁵

10⁻¹⁶

summary

- Simulations of the bkg level, spectrum, components ratio, variation modulated by orbit are useful to determine the performance and bkg reconstruction on orbit.
- The Response Matrix benefit from simulations of detector response to x-ray when using limited energy points especially in high energy range.
- Detailed and accurate simulation and verification experiment are needed to achieve optimized design of FPD with lower bkg. For example, grade-Z shielding or low energy secondary charged particle shielding.