System Verification Through the Life Cycle Study Overview & next steps in Avionics domain ADCSS 2017

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Agenda

□ SVTLC ESA Study (Airbus Defence & Space, Novabase , Scopeset)

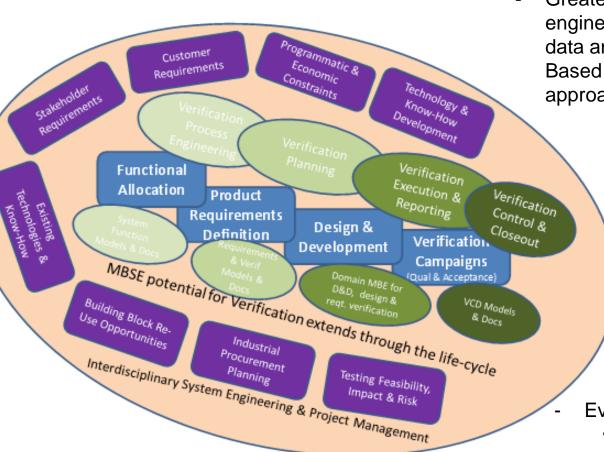
- Study Objectives
- Model based potential to validate re-use approaches
- Towards a Dynamic and Suitable Review Logic
- Advanced Models Philosophy
- Study outcomes : today models for Electrical & Functional domain in Airbus DS

□ Next steps : Airbus Defence & Space view on MBSE approach on Avionics



SVTLC study focus : Towards Improved System Verification Practice





- Impact of Elements Re-Use
 Definition of Suitable Review Logic
 - Demonstration

Best practice comparison with automotive sector

- Greater integration of domain engineering and system engineering data and processes through Model Based System Engineering approaches
 - Improved deployment of correct, efficient and timely efforts to achieve system verification
 - Sound early planning
 - Robust implementation
 - Avoid over and under specification
 - Eliminate wasteful / low-value activities without adding risk
 - Innovate new methods and tools
 - Focus on virtual models in this study
 - Evaluation of Suitability of Models for Verification
 - Nomenclature linking model class with lifecycle verification objectives, aligned with ECSS TM-10-21 A
 - Concrete ideas and propositions in avionics & system domains, based on projects return of experience
 - Rigorous distinction between Qualification and Acceptance objectives





Status quo

- Top-down V approach and bottom-up Product Line approach often meet together in a less well defined landscape of ad-hoc adaptations of model philosophy and review approach for re-use of design artefacts and equipment / subsystems
- These requirements may drive away from the overall programmatic optimisation target if not sufficiently validated up-front against the most open acceptable scenario of user needs (over-specification)
- Validation of bottom-up re-use opportunities are often very costly to achieve against top-down requirements especially across contractual boundaries

WHAT?

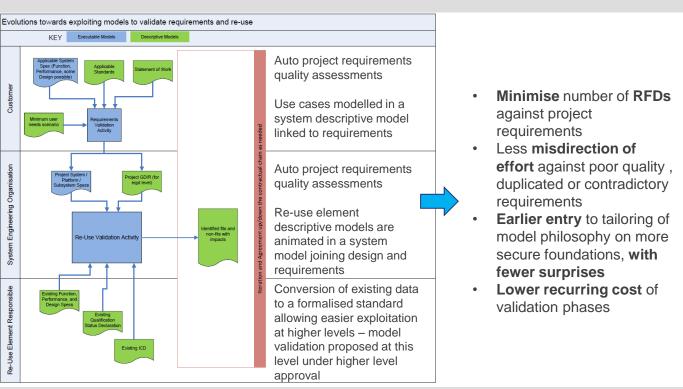
- Focus on requirement validation to avoid over-specification, followed by re-use validation
- **Develop and exploit** potential of **models** of both requirements & design characteristics, and their interactions throughout the system
- Define model validation responsibilities and tailor the model philosophy

WHY?

- To **prevent to lose** some re-use opportunities through over-specification
- To earlier reveal fits / no fits of the proposed re-use to validated requirements
- To reduce cost and duration of the re-use validation phase

HOW?

- 1. Executable models for requirements validation against minimum defined set of user needs
 - Formalized modelling of requirements categorised as per ECSS-E-ST-10-06C, and needs as use cases, with auditing of relationships to reveal un-needed reqts
- 2. Executable models for re-use validation against the previously validated requirements
 - Function, Performance, Interface, Qualification Status – *tech reqt. related*
 - Verification Content, PA, Industrialisation, Management – SOW related
- 3. Model based compatible data exchange across contractual boundaries
 - Standardisation of formalised data models and exchange protocols









Towards a Dynamic Review Logic through systematic Design & Verification Maturity Assessment and Management

Status quo

- Whilst technology readiness and assessment is generally well treated on a formalised TRL scale with associated thresholds for entry to implementation phase. the emerging system design maturity is subject to fewer categories and considered via the classic system reviews PRR, SRR, PDR, CDR, QR, AR.
- These milestones impose a major programmatic environment that drive project activities, and not always in direct synergy with the technical and industrial maturity, including non-ideal phasing with unit and software level review cycles.
- Reactive adaptation of the review logic already takes place e.g. delta-reviews, splitting reviews to part 1 and part 2, also renegotiated payment milestones...

WHAT?

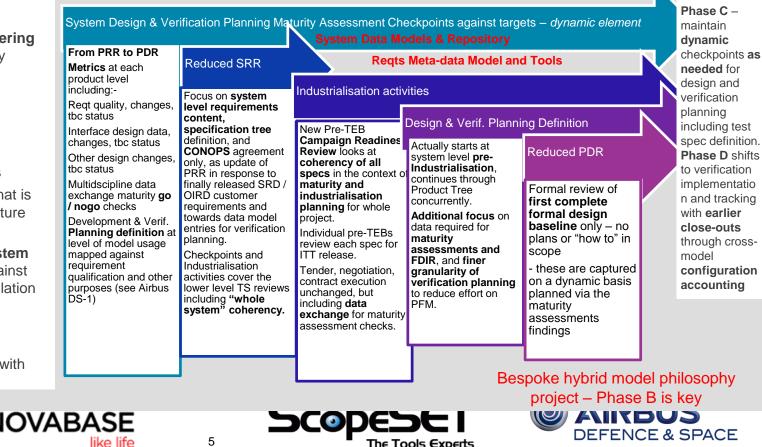
- Turn the reactive review logic adaptation into a proactive one with the optimised technical and industrial maturity evolution planning in the driving seat, within overall programmatic constraints
- Formulate the B2CD business agreement on the basis of this agreed evolution planning with light systematic maturity assessment points, and a leaner content and implementation of the classic review cycle

WHY?

- To achieve much greater alignment of the programmatic, technical, and industrial realities based upon greater visibility of the real maturities and risks
- To allow decision-makers to more systematically take an informed holistic view on concrete facts and recognition of unknowns
- To reduce consequences of incorrect maturity assessment e.g. redesign / rework / retrofit, and improve the value added of the overall review cycle

HOW?

- 1. Common team access to a System Engineering environment built to facilitate rapid and highly accurate multi-discipline data exchange, plus discipline specific views, supporting design, verification and models configuration (to identify regression and change impact)
 - reduce iteration and cycle times
 - rapid metrics for maturity assessments
- tbc is your friend allows to make visible what is not really fully mature, and plan to make it mature taking into account all interactions
- Phase B1 outcome includes definition of system 3. design & verification maturity planning against which the checkpoint plan is made for formulation of business agreement in Phase B2CD.
- Model sharing across contractual chain to 4. facilitate requirement, design, and verification reviews, focussed on key questions aligned with the above planning



The Tools Experts



Mapping Model classes with verification objectives to derive Fidelity Requirements on the lifecycle

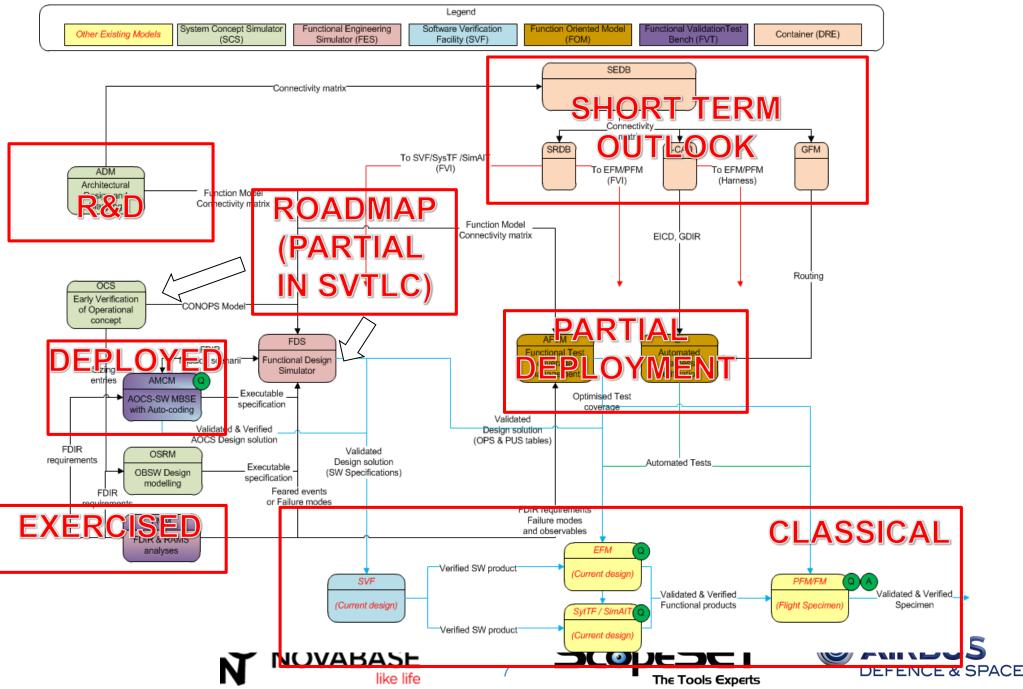
Facility	SCS	MPS	FES	FVT	SVF	MU	FOM	DRE
Name (ECSS-TM-10-21 A)	System Concept Simulator	Mission Performance Simulator	Functional Engineering Simulator	Functional Validation Test bench	Software Validation Facility	Mock-Up	Function Oriented Model	Data repository
Scope	Functional architecture of the system	Mission product quality	Spotted functional design item(s)	Spotted final design solution	Software Validation	Spotted design item(s) solution	Spotted final design item(s) solution	Spotted final design item(s) solution
System Milestone(s)	SRR, PDR	SRR, PDR, CDR	SRR, PDR, CDR	CDR, FAR	CDR, QR/AR	SRR, PDR, CDR	CDR, FAR	Whole lifecycle
Models Validated Against	Mainly ad-hoc tailored generic models against specifications	PRR Specifications, Design solution at System PDR / CDR	System Specifications and Design solution at System PDR / CDR	System Specifications and Design at System PDR / CDR / FAR	Equipment PDR specifications and Design, Equipment CDR design	PRR Specifications, Design solution at System PDR / CDR	System Specifications and Design at System PDR / CDR / FAR	System Specifications and Design at System PDR / CDR / FAR
Facility Validated Against	Consistency with output from the Concurrent Design Process (if any)	System Specifications (SRR, PDR, CDR)	Real Data/Other Systems (All) System requirements (SRR, PDR, CDR)	Product Under Test (e.g. Breadboard Hardware and Software)	Product Under Test (e.g. Software function) and overall Design solution	Real Data/Other Systems (All) System Specifications (SRR, PDR, CDR)	Product Under Test (e.g. Breadboard Hardware and Software)	As designed / As built
Verified Products	Mission Concept compliance to Requirements Design consistency System performance	Performance of the Mission Product(s)	System functional design & performance validation in the targeted area	Compliance of Product Under Test with system interfaces and design and mission requirements	OBSW Product function Under Test against SW and mission requirements Associated SRDB elements	Pending use case : Architecture/ Configuration / interfaces / operational procedures	Compliance of Product Under Test with system interfaces and design and mission requirements	N/A Feeds ad configures As designed / As built through life cycle
Verification class	Proof of Architecture (POA)	Design or I/F freeze – proof of concept (POC) Requirement closure – Verification (REQ)	Design or I/F freeze – proof of concept (POC)	Detailed design consolidation – bread boarding for risk mitigation (DDC) Overall Design validation (VAL) Requirement closure – Verification (REQ)	Requirement closure – Verification (REQ) (for S/W)	Design or I/F freeze – proof of concept (POC) , Detailed design consolidation – bread boarding for risk mitigation (DDC) AIT or OPS preparation (PREP)	Detailed design consolidation – bread boarding for risk mitigation (DDC) Overall Design validation (VAL) Requirement closure – Verification (REQ) AIT or OPS preparation (PREP)	N/A







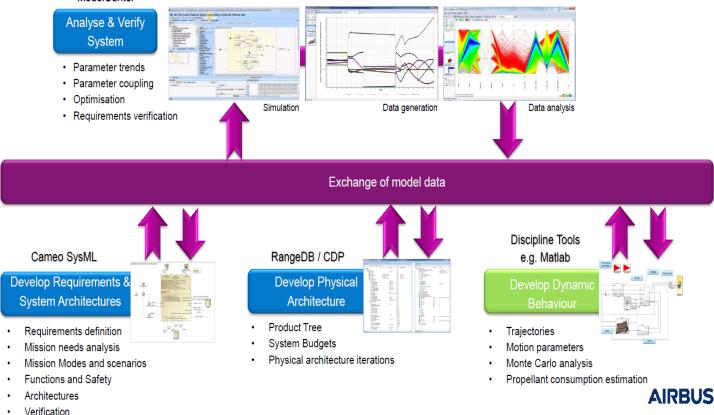
Resulting Outlook of models for Electrical & Functional world : vision, continuities, and state of practice in Airbus DS



DEFENCE AND SPACE Next steps : ADS view on MBSE approach on Avionics

Model Based System <u>Verification & Validation</u> relies on Model Based System <u>Engineering</u>
 <u>Model based Avionics Engineering</u>, <u>Validation & Verification</u> has to be consistent and integrated with:

- **System Engineering level** (Missions phases and operational scenario, modes and states, physical architecture and equipments)
- Avionics functional Engineering on CONOPS, FDIR, AOCS/GNC, OBSW, Functional Validation & Infrastructure (Simulators and test benches)
- A strong management of Data consistency and continuity in a shared and common repository through the different phases of life-cycle (Phase B/CB/E), different levels of the system (large ground/board system, spacecraft, avionics, equipment & SW), project organization including equipment suppliers - if possible extended to Models inheritance and reuse

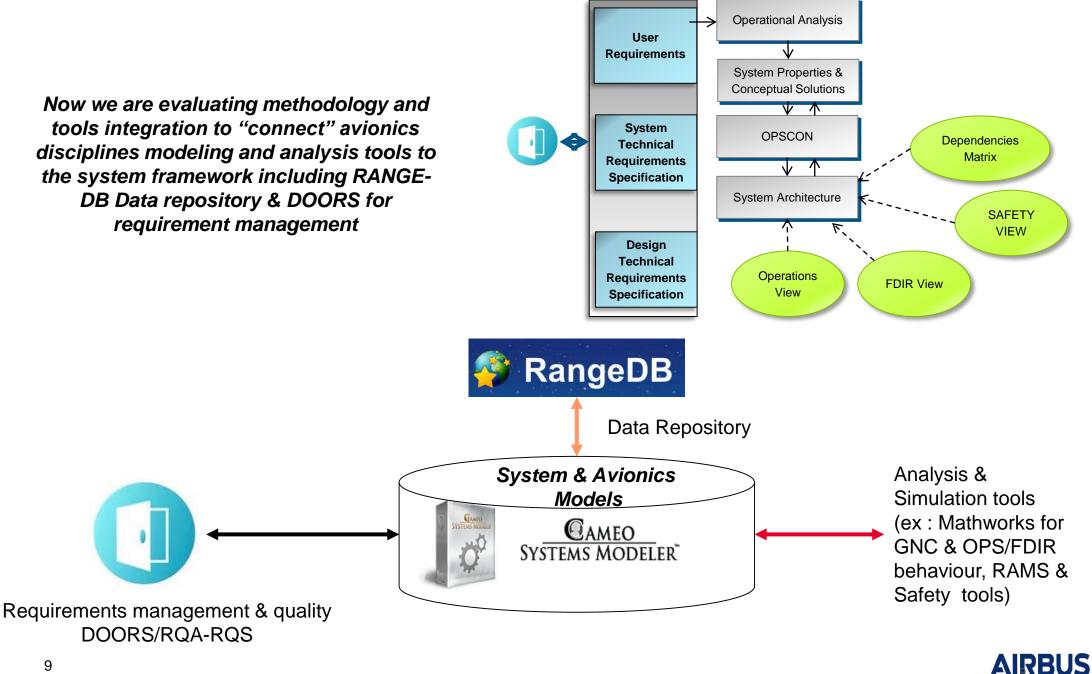


On ESA e.Deorbit Phase B1 project, an MBSE approach based on federated and executable models has been implemented for supporting the generation of requirements and system architectures

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Next steps : ADS view on MBSE approach on Avionics Modelling engineering approach from Systems to Avionics FES



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Next steps : ADS view on future Mission needs and advanced Avionics DDV Process

□ Space systems verification through models will face two major stakes in next decade especially on Avionics and Functional systems

□ New mission needs and new avionics architectures are more demanding in early validation through simulation & model-based approach supporting increasing complexity

- Mini or mega constellations
 → several spacecraft to be controlled and maintained at same time, sharing of C&C on ground /on-board systems
- More autonomous spacecraft → reducing ground system effort and/or mission constraints,
- Several spacecraft/systems in close cooperation → composite spacecraft, in-orbit servicing or manufacturing with robotic sub-systems

□ New Avionics Design , Development & Validation step on efficiency (cost, schedule & risks)

- More reuse → better product family formalisation with early maturity assessment & product line management along the lifecycle
- Reduction of HWIL models and intensive use of numerical models

 more virtualization
 and
 igitalization
- Less documents and more shared models with collaborative environments
- More life-cycle flexibility with agility and dynamic approach in reviews and verification activities
- Data continuity, baseline management and models are key

Thank you for your attention



