



Benchmarking ionizing space environment models

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ONERA

THE FRENCH AEROSPACE LAB



Approach

In-flight data and comparison to proton model predictions

- TNID from OSL at 660 km altitude, 82° inclination
- SEU from EDAC counter at 719 km altitude, 82° inclination
- DCNU from Star Tracker at 960-1160 km altitude, 29.7° inclination
- TNID from OSL at 1336 km altitude, 63° inclination
- SEU from EDAC counter at 1336 km altitude, 63° inclination
- DCNU from Star Tracker at 1336 km altitude, 63° inclination
- DCNU from Star Tracker at 265-5000 km altitude, 49° inclination

Synthesis (Trapped protons)

In-flight data and comparison to electron model predictions

- Solar array power loss at GEO

Approach

Orbit is from NORAD TLE with a 20s time step throughout the entire mission.

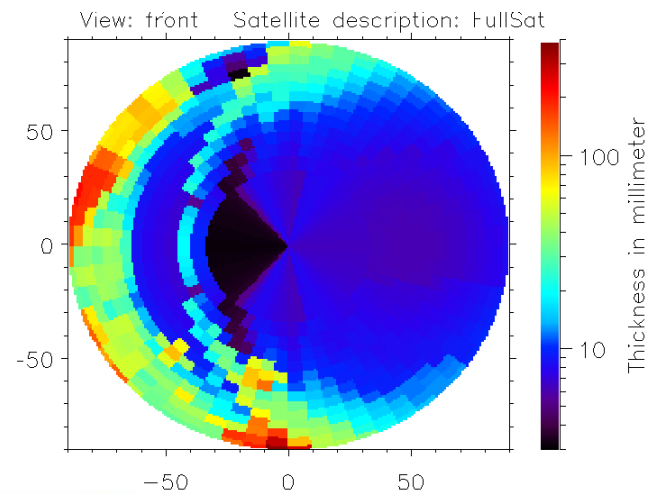
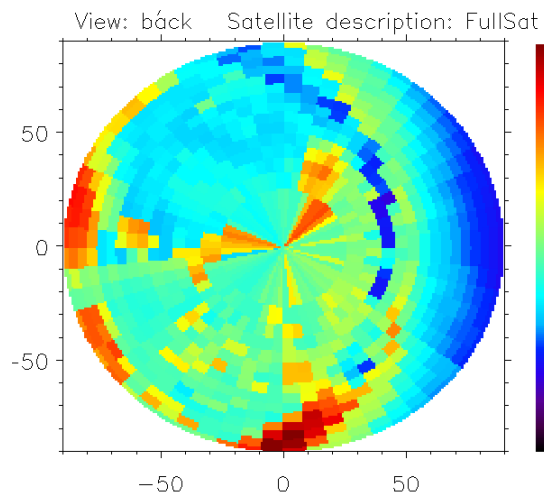
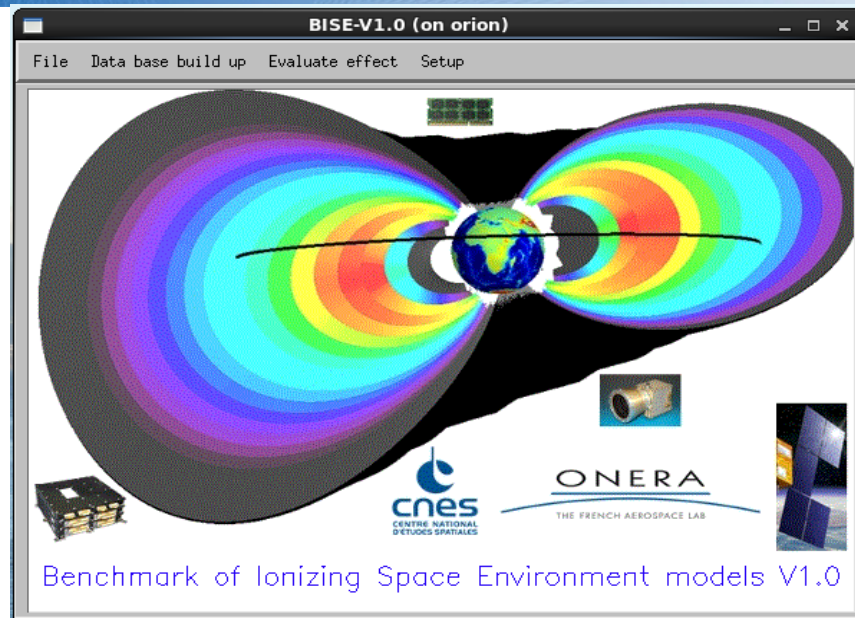
Trapped radiation environment is computed every 20s along the spacecraft orbit.

Account for 3D shielding around component of interest → sectoring analysis from FASTRAD

Response function or transmitted flux are computed from Monte-Carlo run using GEANT-4 or MCNPx

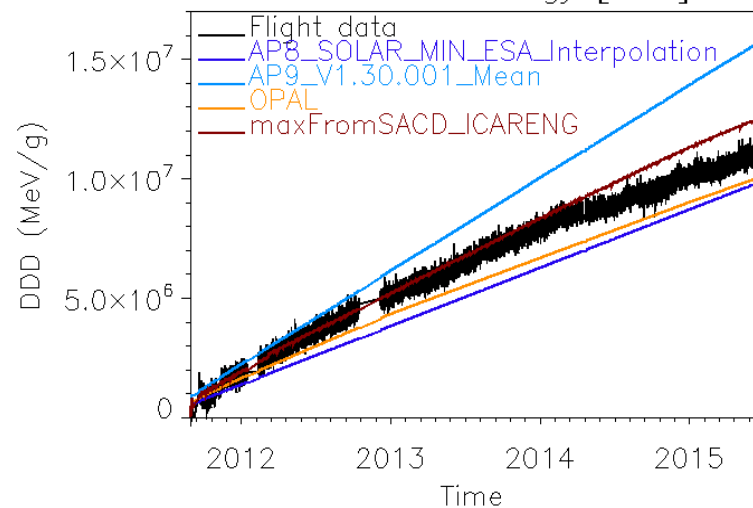
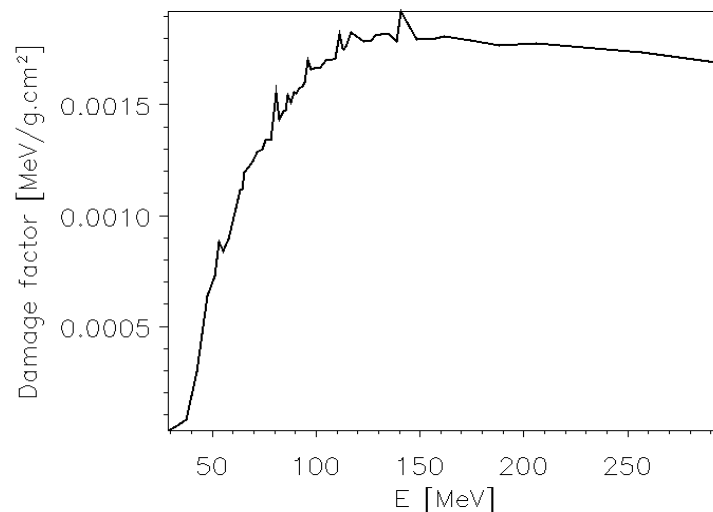
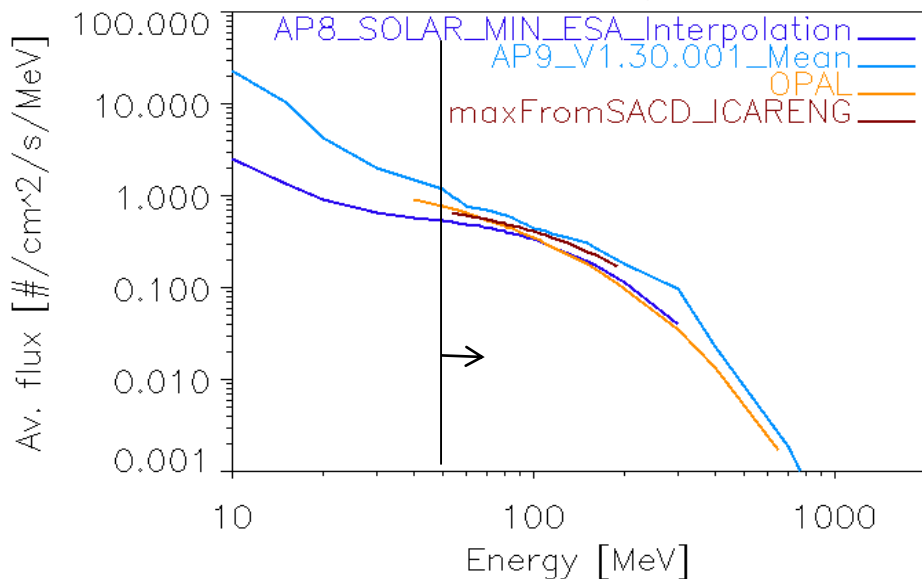
Comparison of predicted degradation with in-situ measurements (TID, TNID, SEU-EDAC, DCNU, Solar array power).

FASTRAD sectoring analysis from SAC-D/ICARE-NG/OSL/LED



In flight data and comparison to proton model predictions

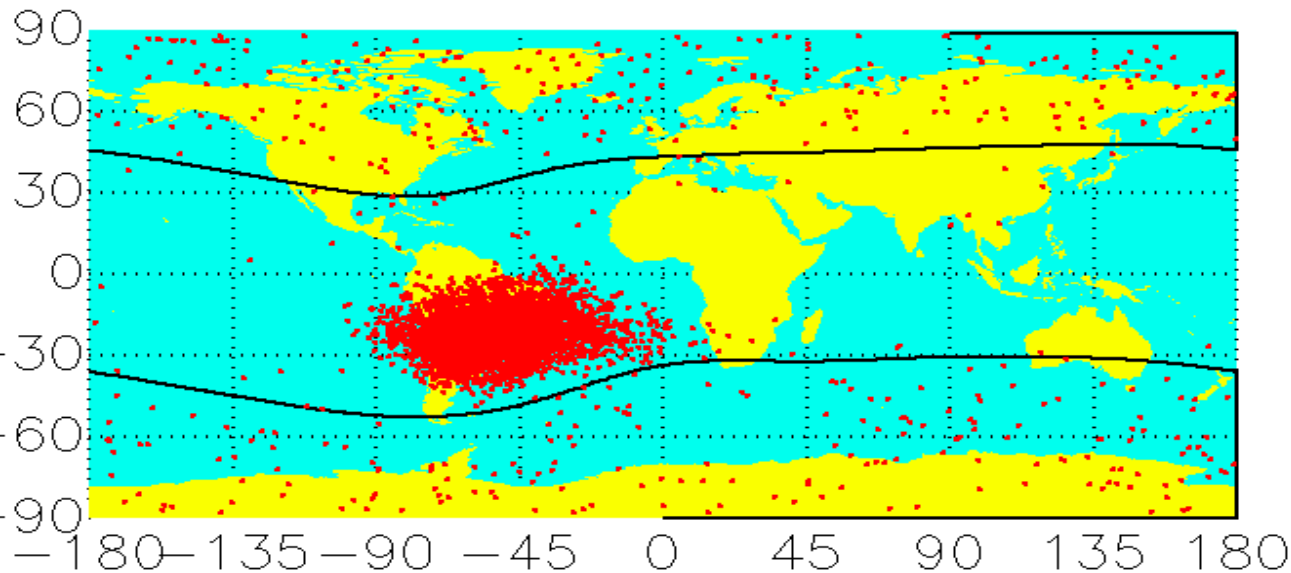
Total displacement damage (DDD) at 660 km altitude (SAC-D, 660 km, 98°)



Model / data	Deviation (ratio prediction / flight data)
AP8 min	0.89
AP9 V1.30.001 Mean	1.42
OPAL	0.91
ICARE_NG	1.07

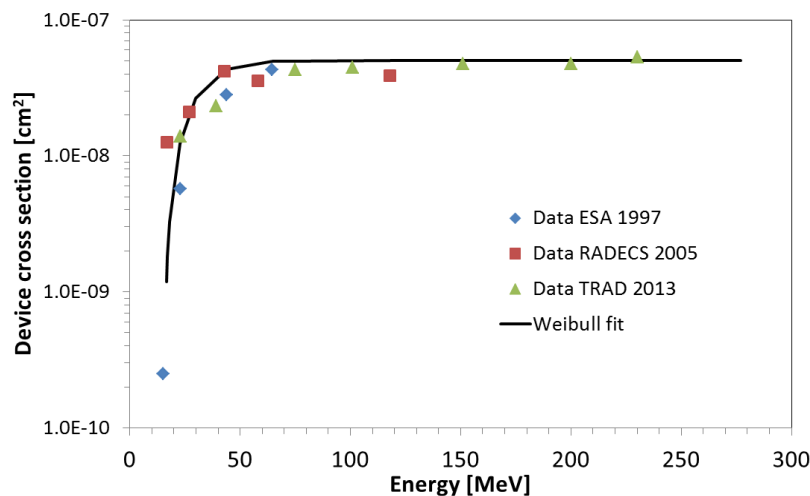
In flight data and comparison to proton model predictions

SEU rate from EDAC counter at 719 km altitude (CRYOSAT-2, 719 km, 98°)



$L^* < 1.9 \Rightarrow 88.4\%$

$L^* > 1.9 \Rightarrow 11.6\%$



10 SRAM (1MB SRAM M65608, 0.5 μ m CMOS process, developed by ATMEL)

$W=18$

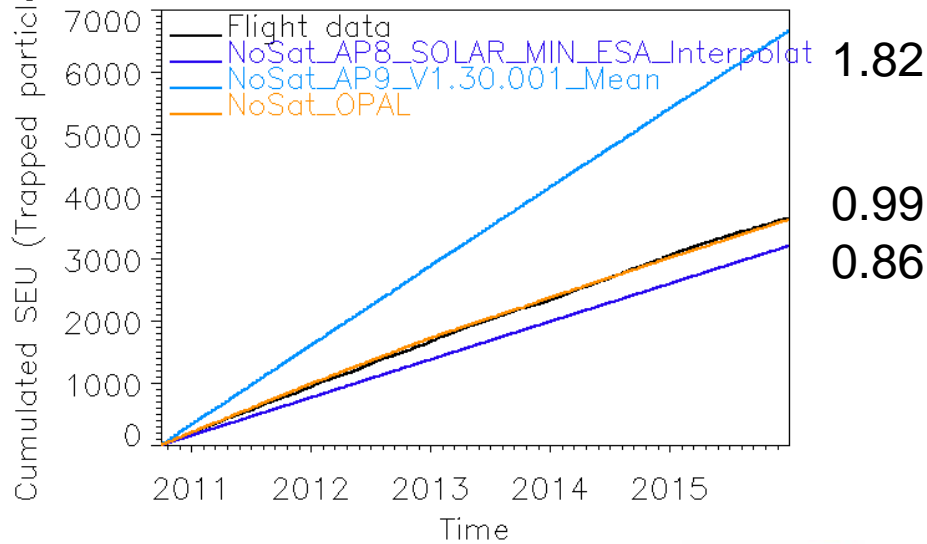
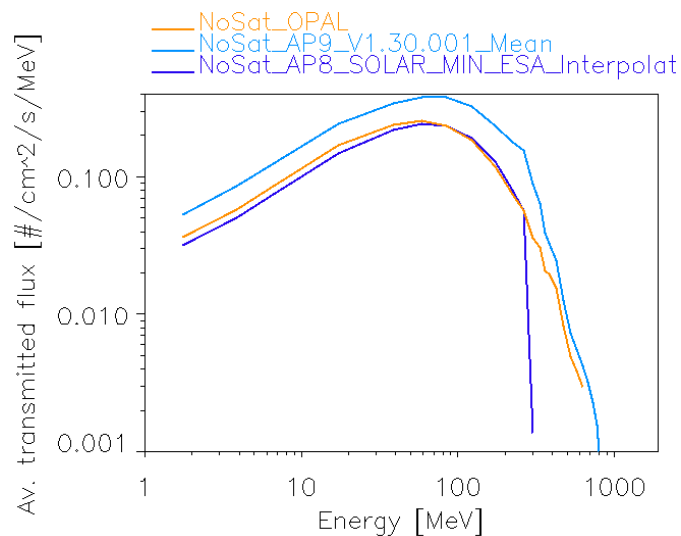
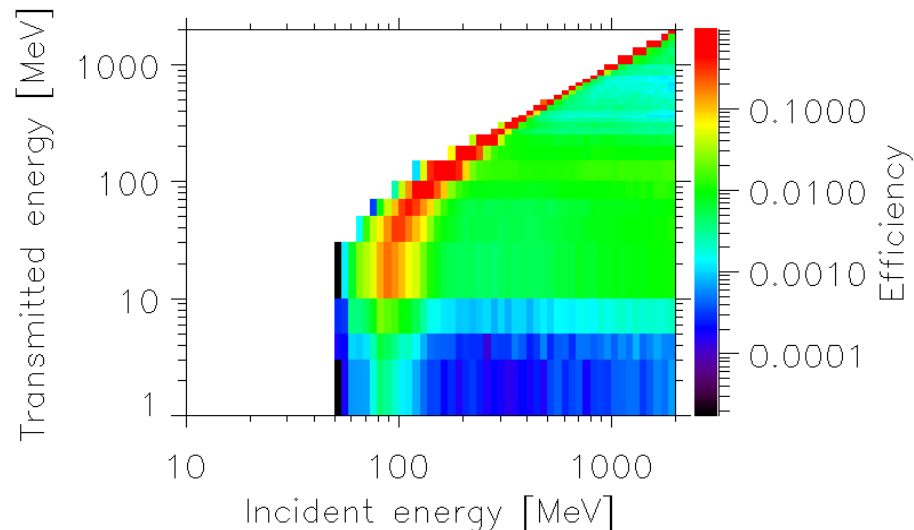
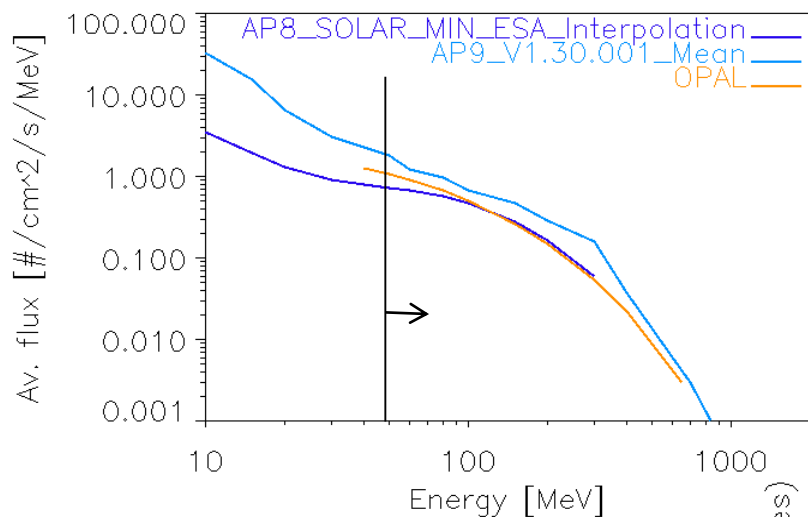
$S=1.5$

$E_{th}=15$ MeV

$Sat=5.28 \cdot 10^{-8}$ cm²/device

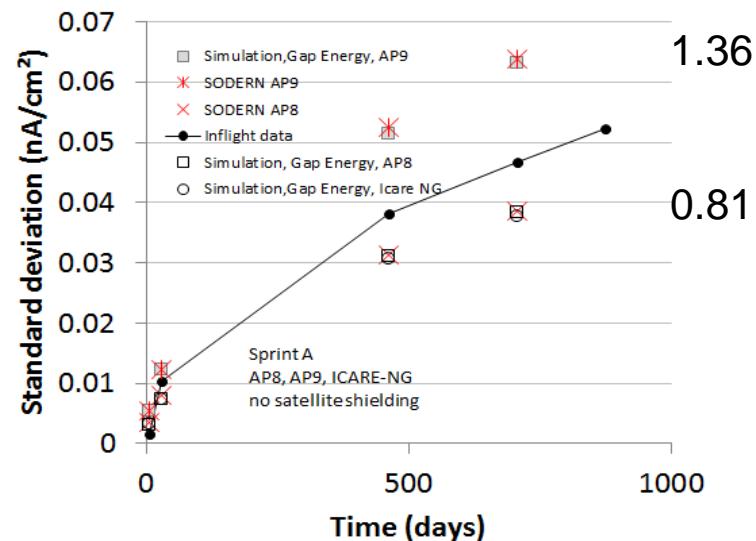
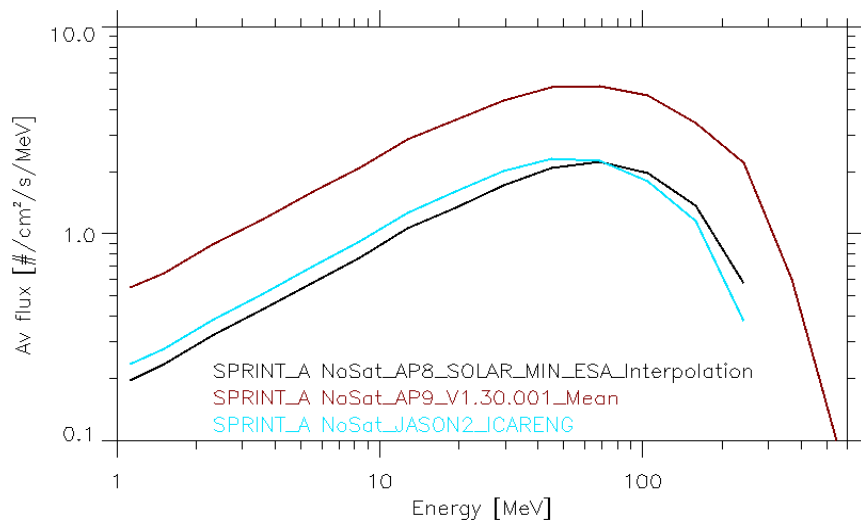
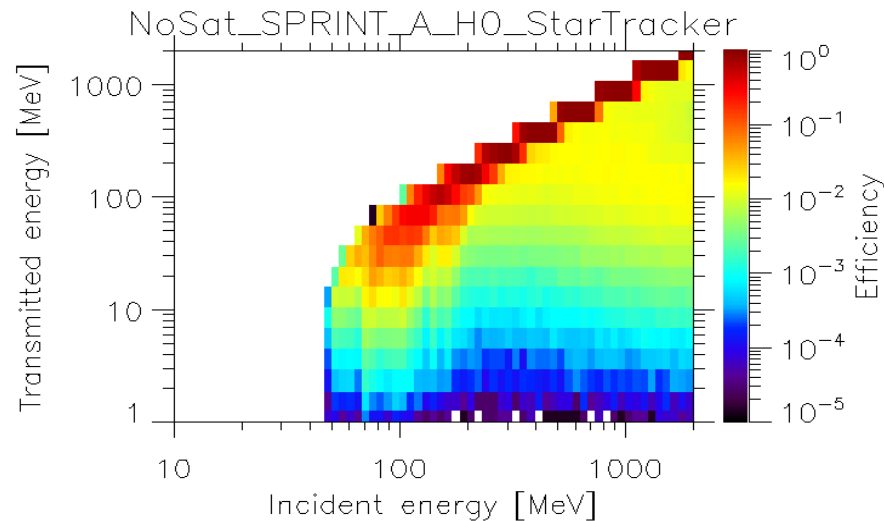
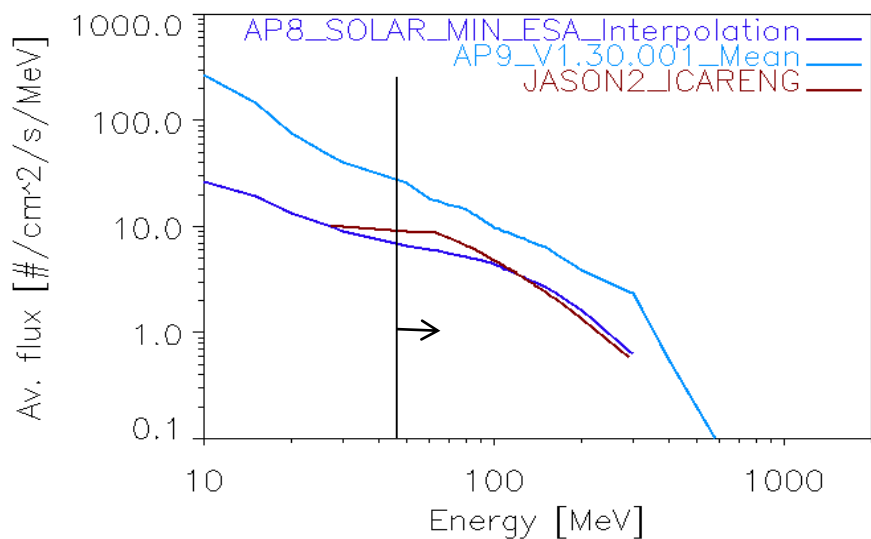
In flight data and comparison to proton model predictions

SEU rate from EDAC counter at 719 km altitude (CRYOSAT-2, 719 km, 98°)



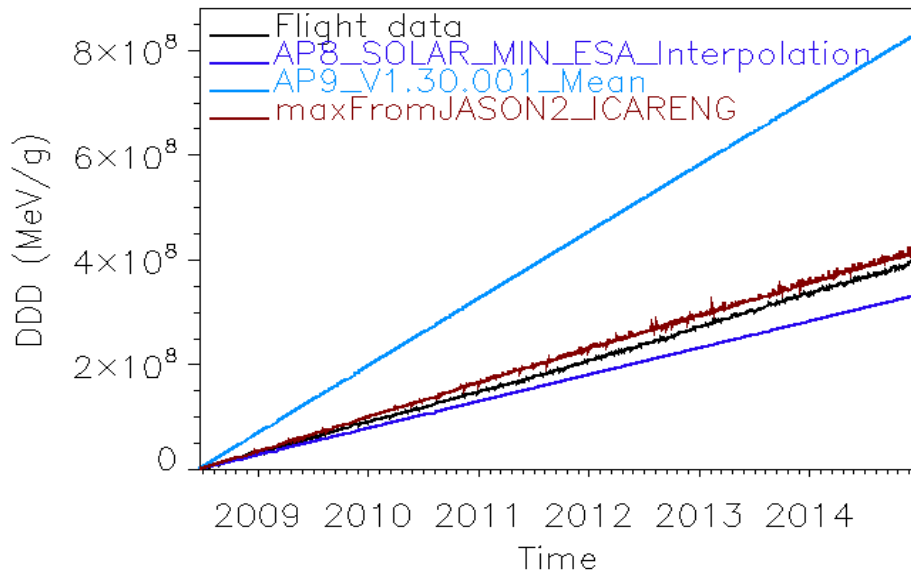
In flight data and comparison to proton model predictions

DCNU from Star Tracker at 900-1100 km altitude (SPRINT-A, 960-1160 km, 31°)



In flight data and comparison to proton model predictions

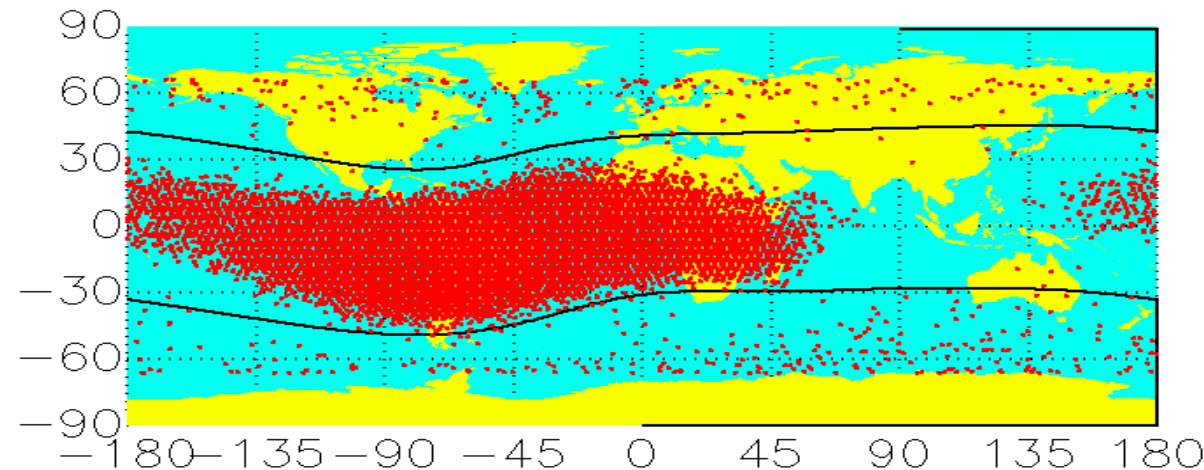
Total displacement damage (DDD) at 1336 km altitude (JASON-2, 1336 km, 63°)



Model / data	Deviation (ratio prediction / flight data)
AP8 min	0.84
AP9 V1.30.001 Mean	2.13
ICARE_NG	1.05

In flight data and comparison to proton model predictions

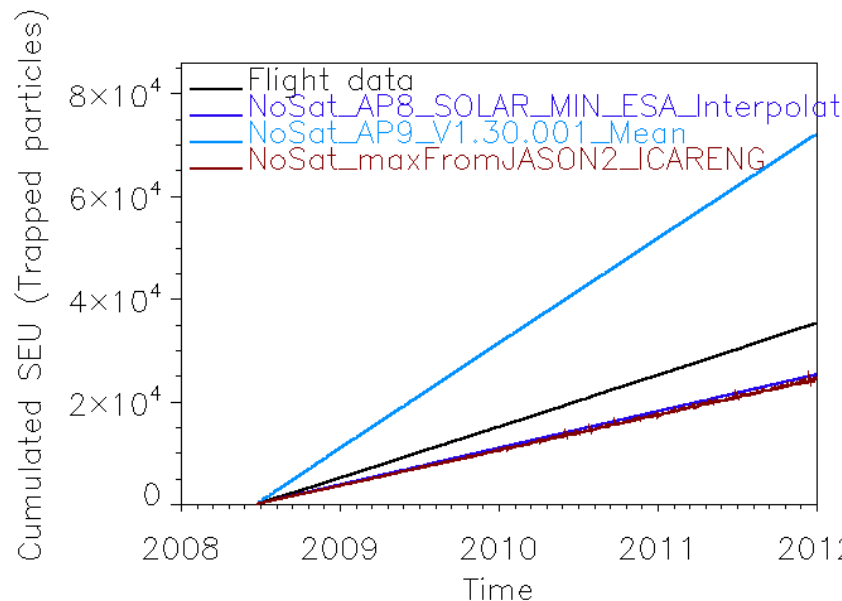
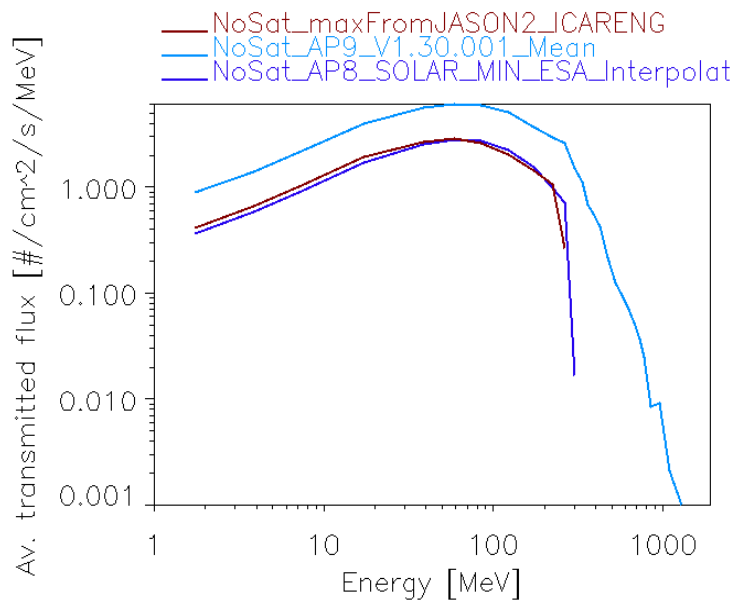
SEU rate from EDAC counter at 1336 km altitude (JASON-2, 1336 km, 63°)



$L^* < 1.9 \Rightarrow 98.8\%$

$L^* > 1.9 \Rightarrow 1.2\%$

10 SRAM (1MB SRAM M65608,
0.5µm CMOS process, developed
by ATMEL)



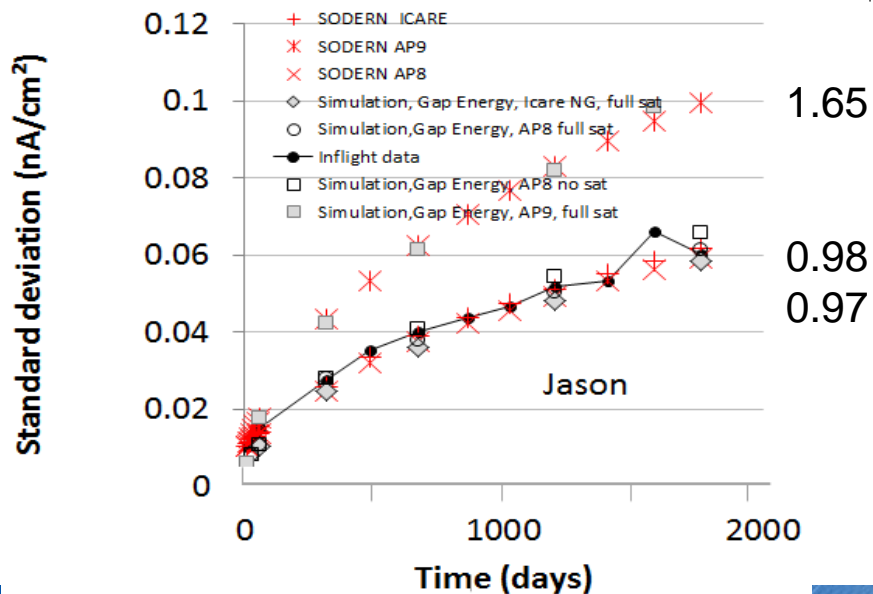
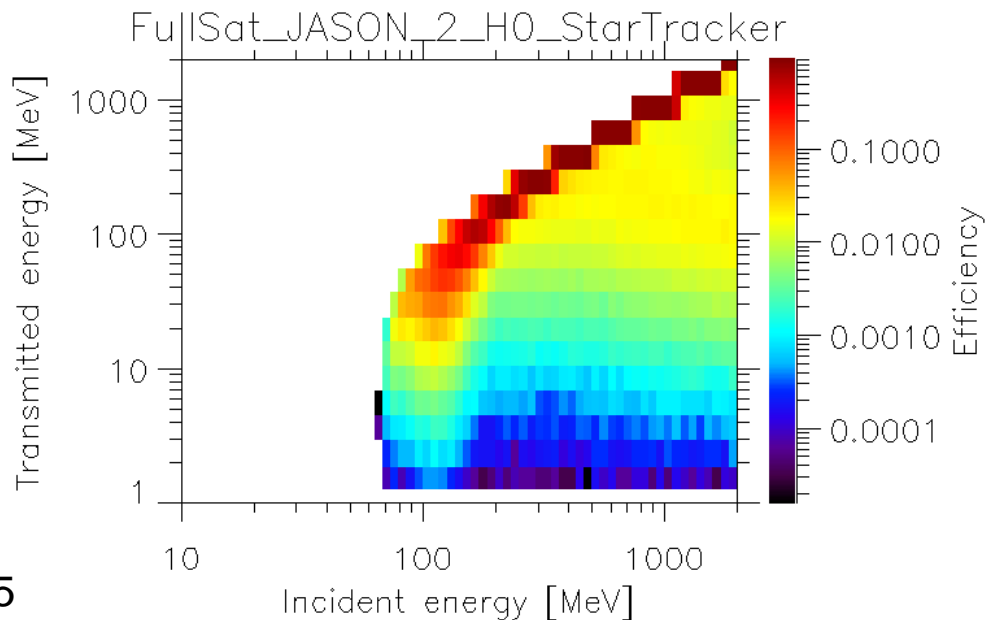
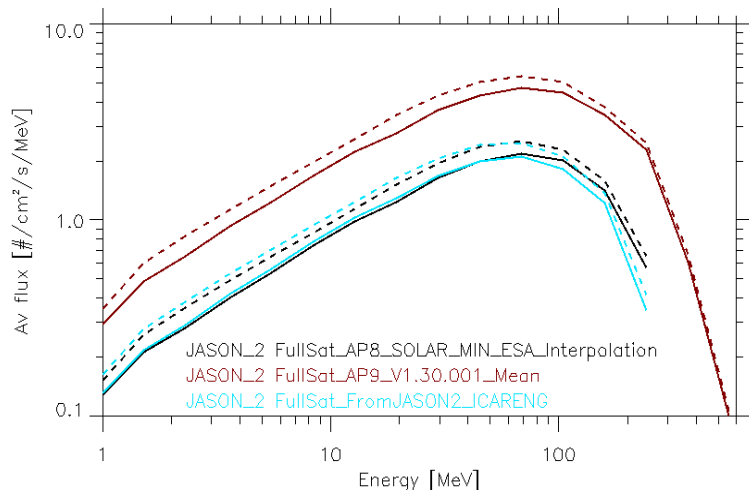
2.10

0.72

0.70

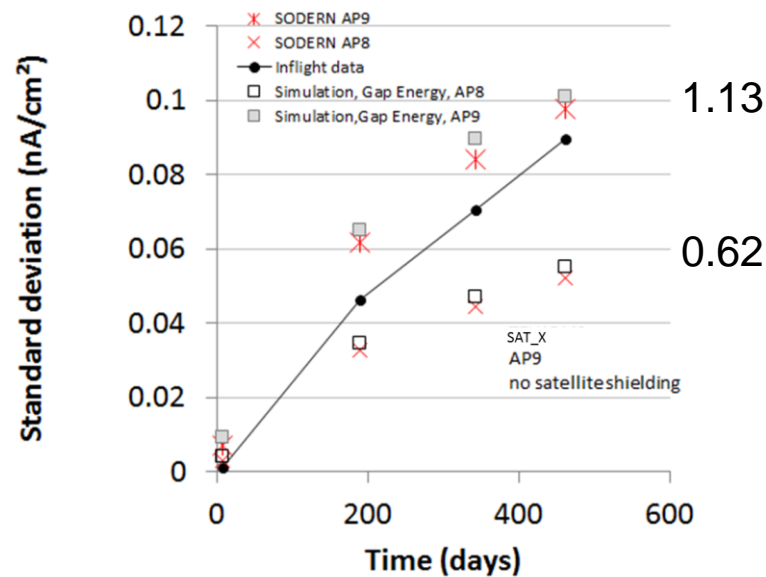
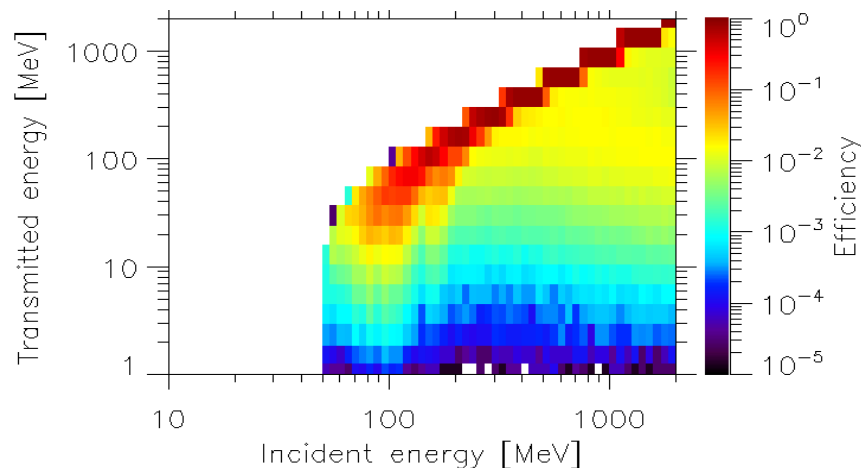
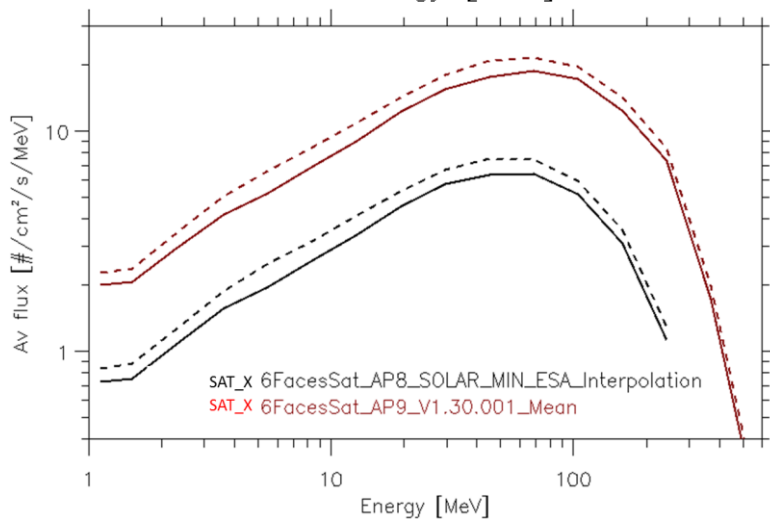
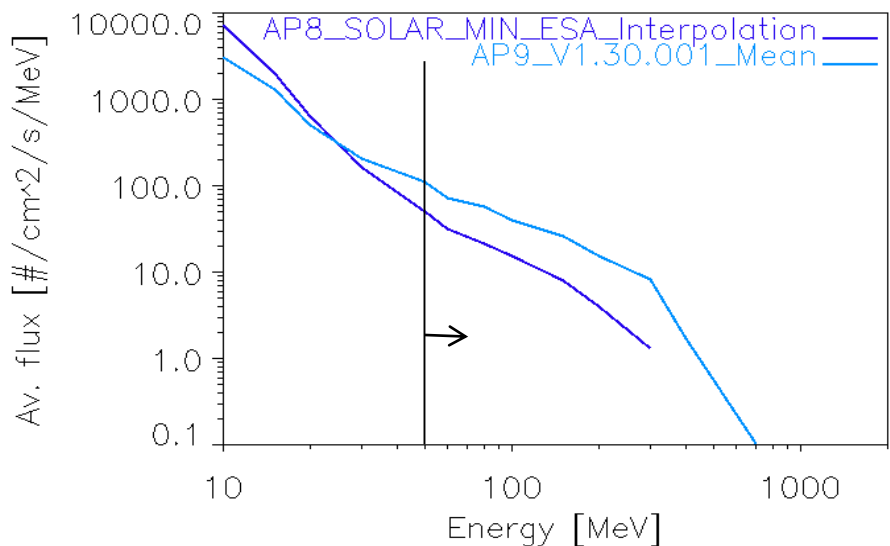
In flight data and comparison to proton model predictions

DCNU from Star Tracker at 1336 km altitude (JASON-2, 1336 km, 63°)

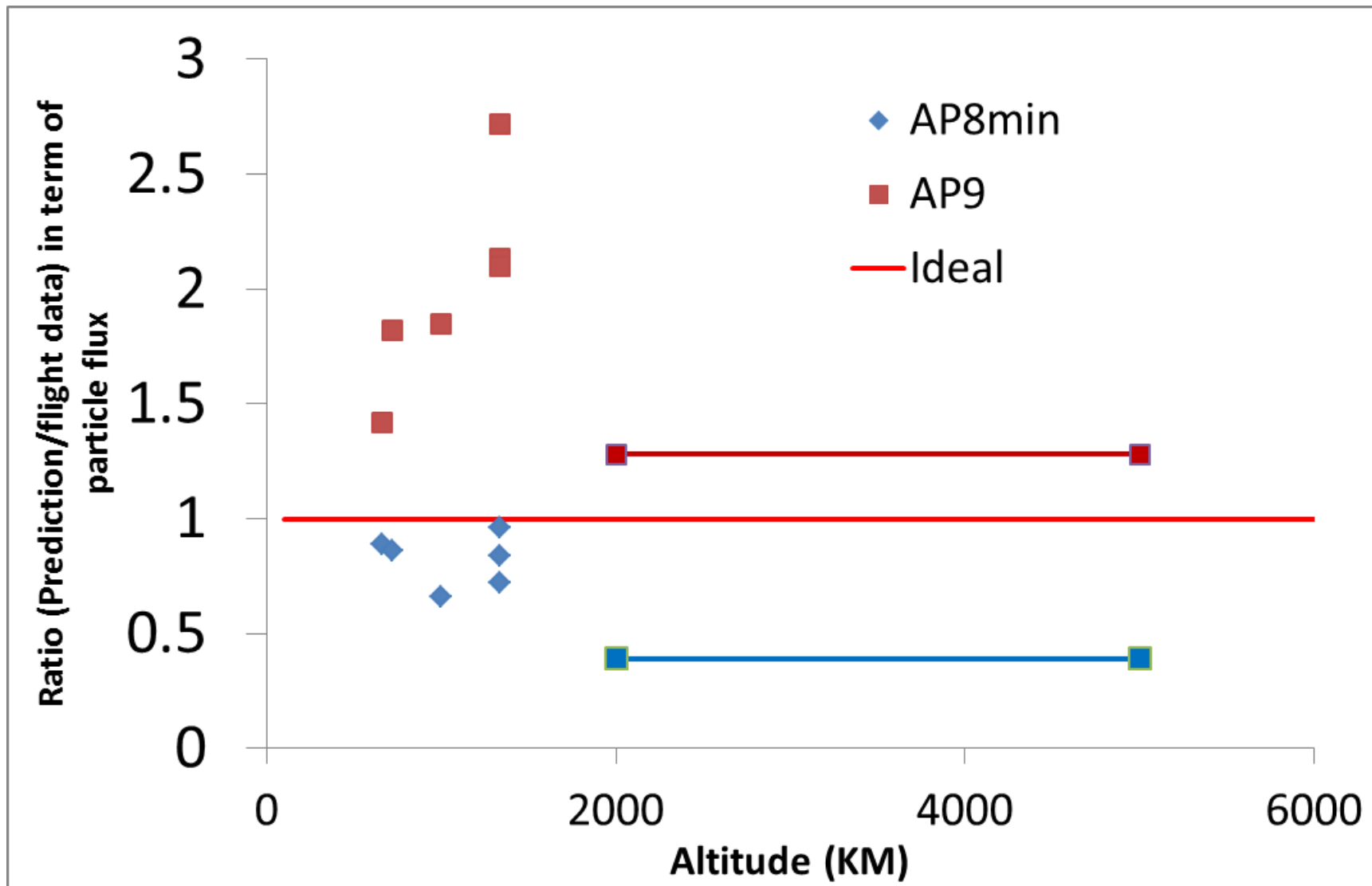


In flight data and comparison to proton model predictions

DCNU from Star Tracker at >1000 km altitude (Sat-X, 265-5000 km, 49°)

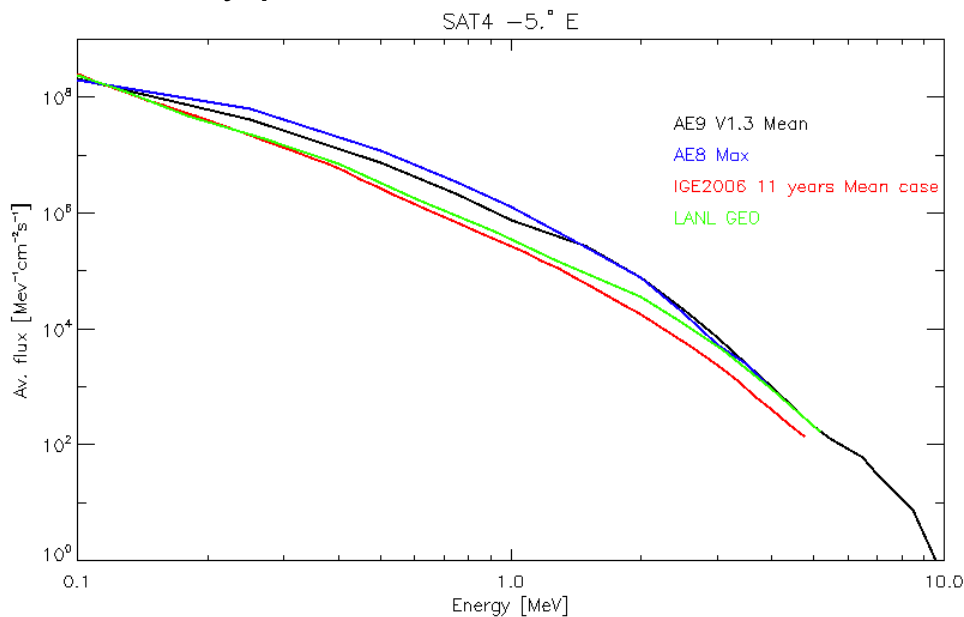


Synthesis (Trapped protons > 40 MeV)



In flight data and comparison to electron model predictions

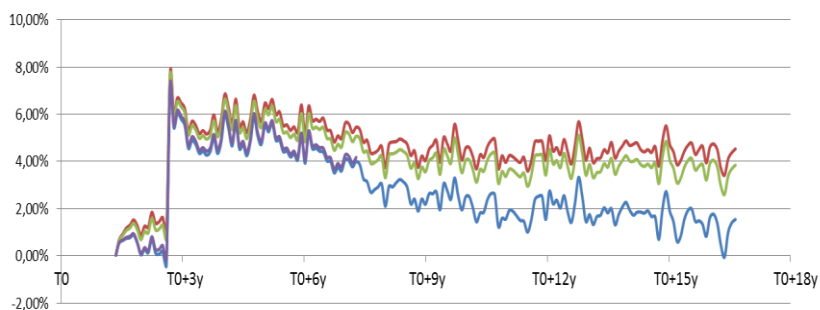
Solar array power loss at GEO



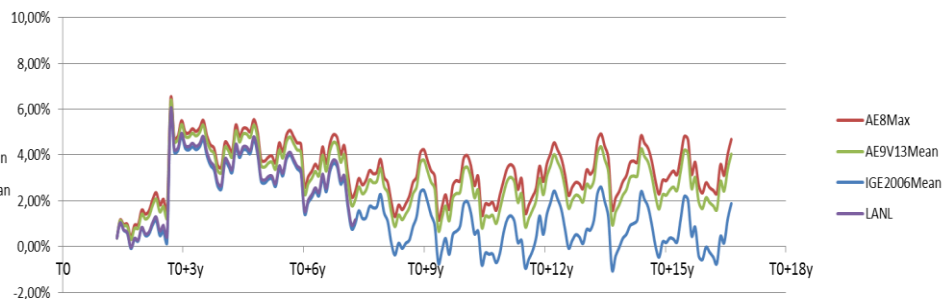
Solar protons deduced from GOES data available in IPODE (consistent results were found using SEPTEM V2.0 data set).

Solar cell: Si with 100 μm & 150 μm coverglass

SAT 4 Difference between TM and prediction for North wing



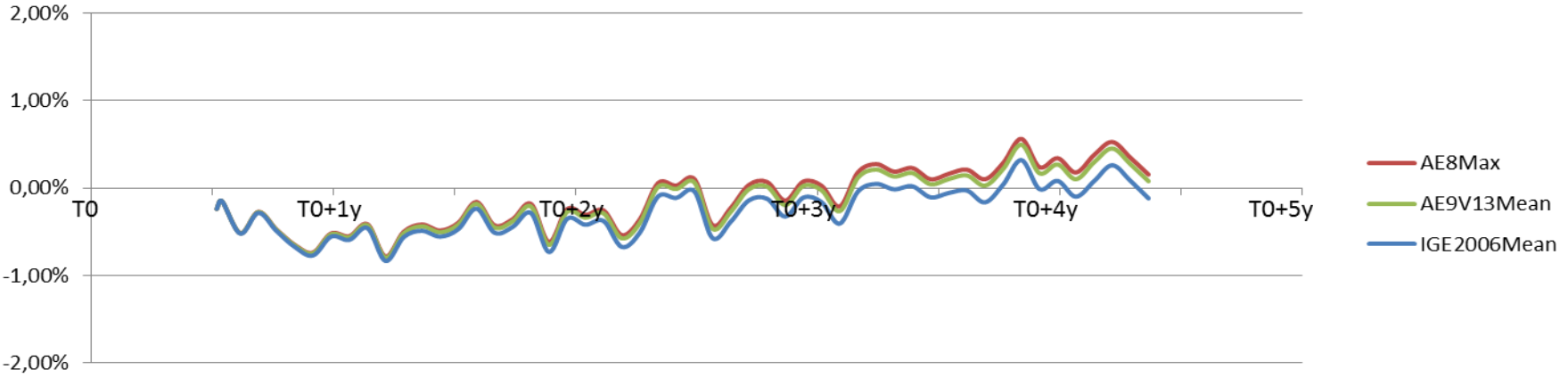
SAT 4 Difference between TM and prediction for South wing



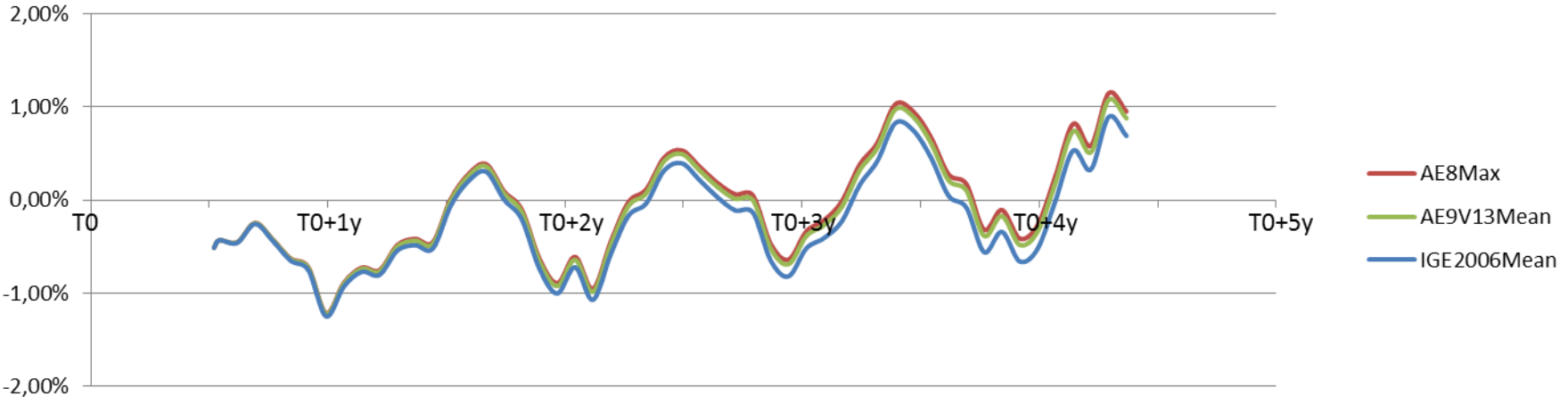
In flight data and comparison to electron model predictions

Solar cell: GaAs with 100 μm coverglass

SAT 5 Difference between TM and prediction for North wing



SAT 5 Difference between TM and prediction for South wing



Conclusions

1. Cumulative effects (TNID, Cumulative SEU from EDAC and DCNU) have been used to investigate uncertainties in trapped proton models
2. Investigating different types of radiation effect makes it possible to avoid any biases
3. AP8 allows for closer predictions than AP9 1.30.001 Mean (and Perturbed) except in the 2000-5000km altitude range
4. Solar arrays power loss has been used to investigate uncertainties in trapped electron models
5. Predictions from IGE2006 (+1.6%) are closer to observations than those from AE8/AE9 (+2.2%)