

RadMag combined cosmic ray and magnetic field measuring space weather instrument development for CubeSat/SmallSat applications

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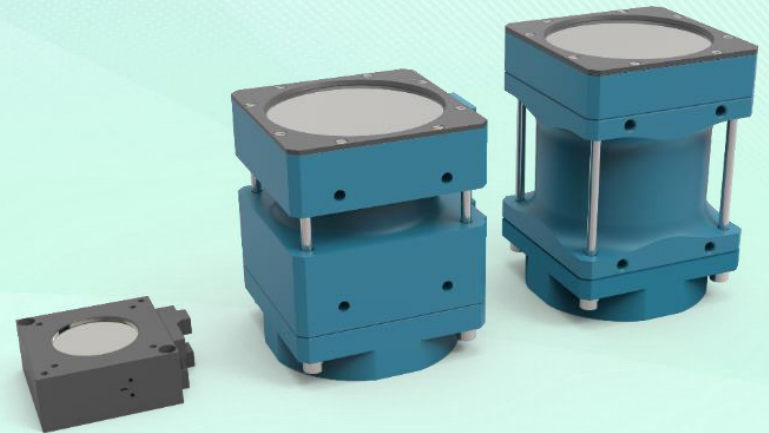
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- Future concept: fleet of monitoring CubeSats/SmallSats
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Instrument Concept & Development Team



RadMag instrument concept

- ***RadMag = Cosmic Radiation and Magnetic Field Instrument Package***
- **Space weather monitoring by combination of cosmic ray and magnetic field measurements into one instrument**
- Specification is reflecting to ESA SSA SWE product requirements
- Cosmic ray measurements by silicon based telescopes: proton, electron, HZE spectra separately
- In-board and outboard 3-axis magnetoresistive sensor
- Built-in boom system to support the magnetic field measurements
- Built-in dose rate monitoring and Radiation Hardness Assurance (RHA) capabilities
- Small size to fit for CubeSat/SmallSat missions (fitting ~1.2U CubeSat standard)
- **Low-cost alternative in future space weather studies and forecast services and in general radiation damage monitoring for commercial use**
- Instrument development within ESA GSTP programme

RadMag Development Team

Abbrev.	Team Member	Responsibility
	Centre for Energy Research HUNGARY	Instrument development coordinator Radiation sensor system development
Imperial College London	Imperial College London UNITED KINGDOM	Magnetometer development
	Astronika POLAND	Boom system development
	European Space Agency ESA ESTEC	RHA development Instrument development reviewer

Instrument specifications



Radiation Sensor System specification

Parameter	Values, ranges
Particle types	electrons, protons, heavy ions
Minimum electron energy	100 keV
Electron energy range	0.3 MeV – 8.0 MeV Channel number: 2-5 Contamination: <10 %
Minimum proton energy	1 MeV
Proton energy range	4 MeV – 1 GeV Channel number: 11-18 Contamination: <10 %
Heavy ion energy range (He&C&N&O&Fe)	100 MeV/n – 1 GeV/n Channel number: 4 Contamination: <10 %
Field of view for electron and proton measurement (half-angle)	31°
Field of view for heavy ion measurement (half-angle)	46°

Magnetometer specification

Parameters	Mode	
	Nominal	High resolution
Range	$\pm 60,000$ nT	
Sampling rate	1.0 Hz	10 Hz
Orthogonal directions	3	
Orthogonality error	$\leq 0.1^\circ$	
Noise limit	≤ 500 pTrms/ $\sqrt{\text{Hz}}$ (at 1 Hz at 25°C)	
Temperature coefficient	$\leq \pm 1$ nT/°C	
No. of sensors	2 (1: inboard, 1: outboard)	

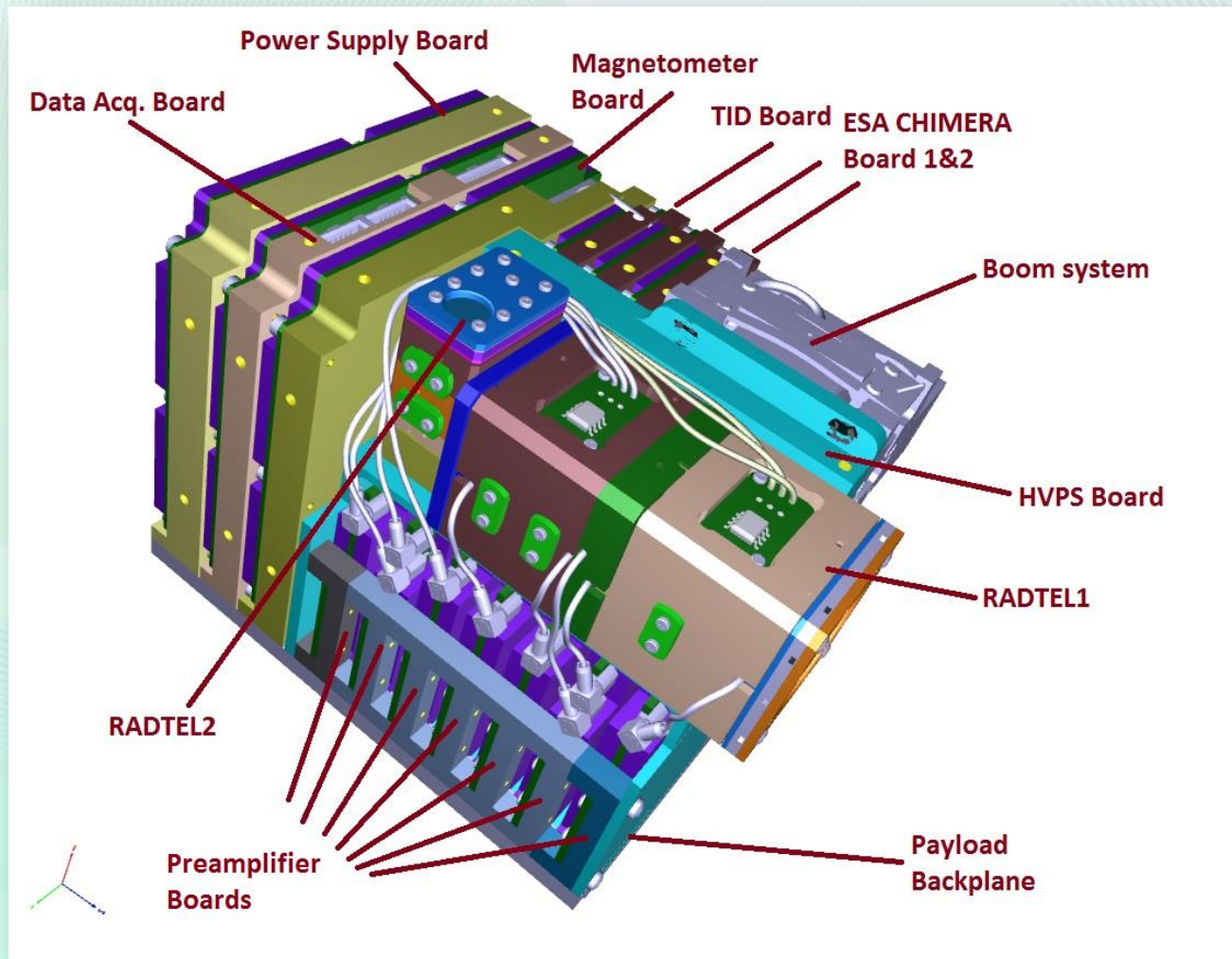
Boom System Specification

Parameters	Values
Maximum deployed length	80 cm
Retraction capability	No
Carried mass (with harness)	30 g
Carried volume	2.0x2.0x1.2 cm ³
Deployed position accuracy	$\leq \pm 3$ cm
Deployed angular accuracy	$\leq \pm 1.5$ deg/m
Deployment encoder	Yes
Available volume for the boom system (with harnessing)	4x4x10 cm ³
Maximum deployment velocity	20 mm/s
Peak power consumption	4 W

Instrument Design Status Overview

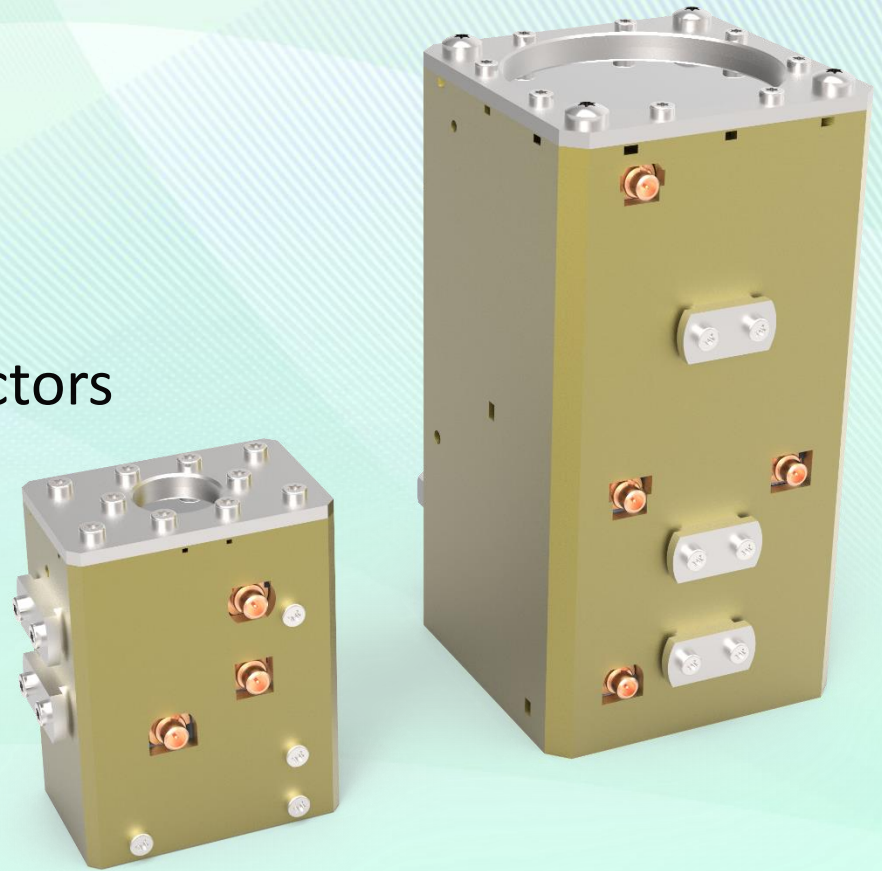


Instrument overall design



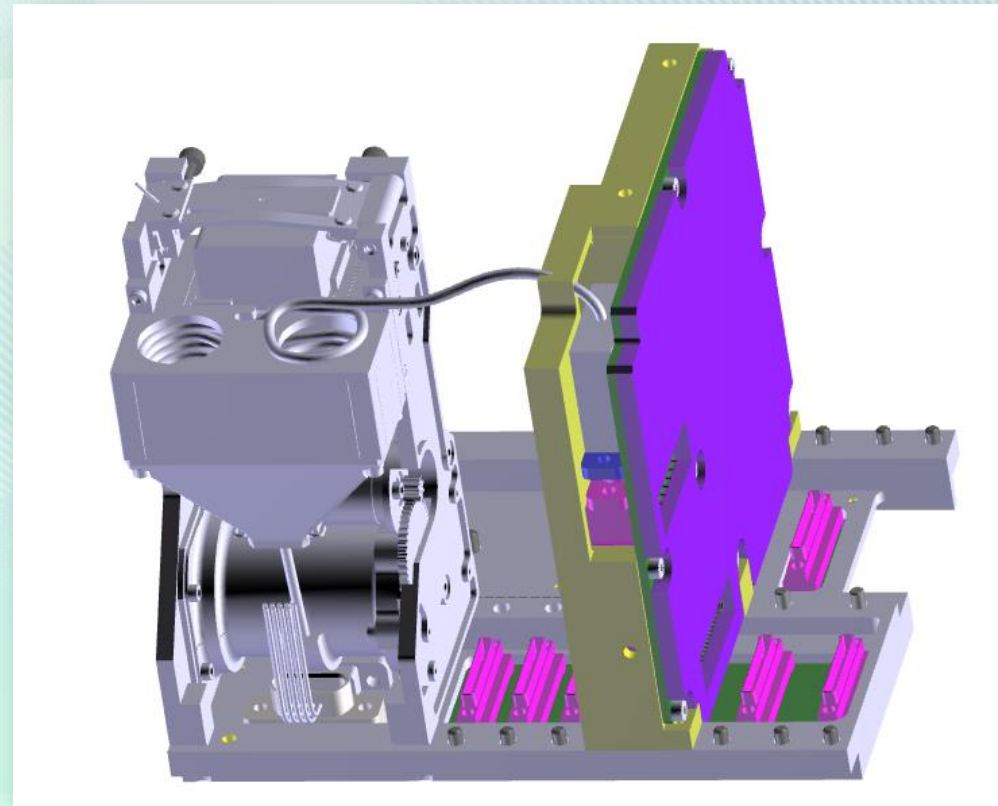
Radiation telescopes

- Two different telescopes
 - different sensitive area
- Perpendicular arrangement
 - directionality assessment
- Utilising Micron silicon detectors
- 10- μm thick titanium entrance windows



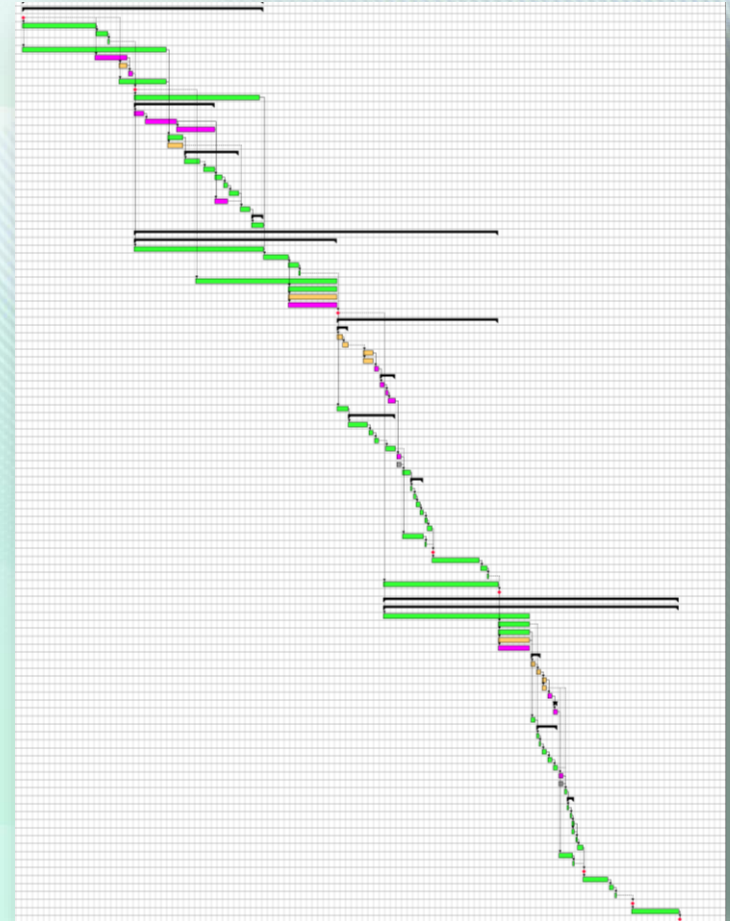
Magnetometer and boom system

- Magnetoresistive sensors
 - 3-axis sensors
 - inboard and outboard
- Tape spring based boom
 - up to 80cm long
 - opening in definitive steps



Development status

- Kick-Off Meeting – June 2016
- IRR – December 2016
- PDR – December 2017
- CDR – August 2018
- QR – February 2019
- FAR – August 2019
- 1st flight – Q3 2019



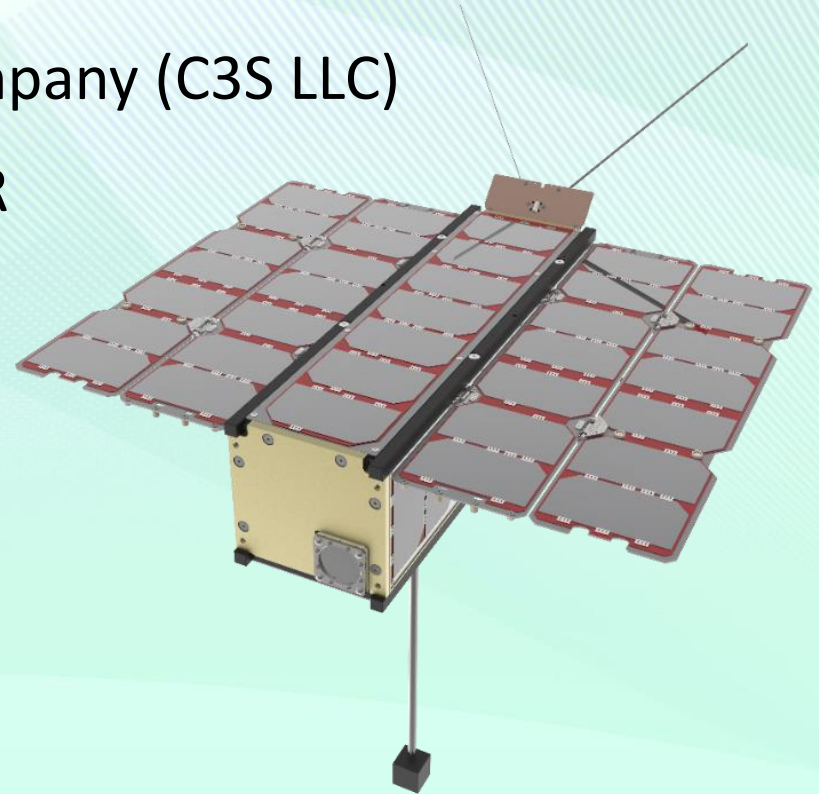
1st IOD Mission: RADCUBE

Future Concept: CROSS Network



RADCUBE Mission

- In-orbit demonstration 3U CubeSat mission within ESA GSTP 6.3 programme
- Lead by a Hungarian CubeSat Company (C3S LLC)
- Project is just now passed the PDR
- Expected launch: Q3 of 2019



COMPLEX SYSTEMS &
SMALL SATELLITES

RADCUBE Mission

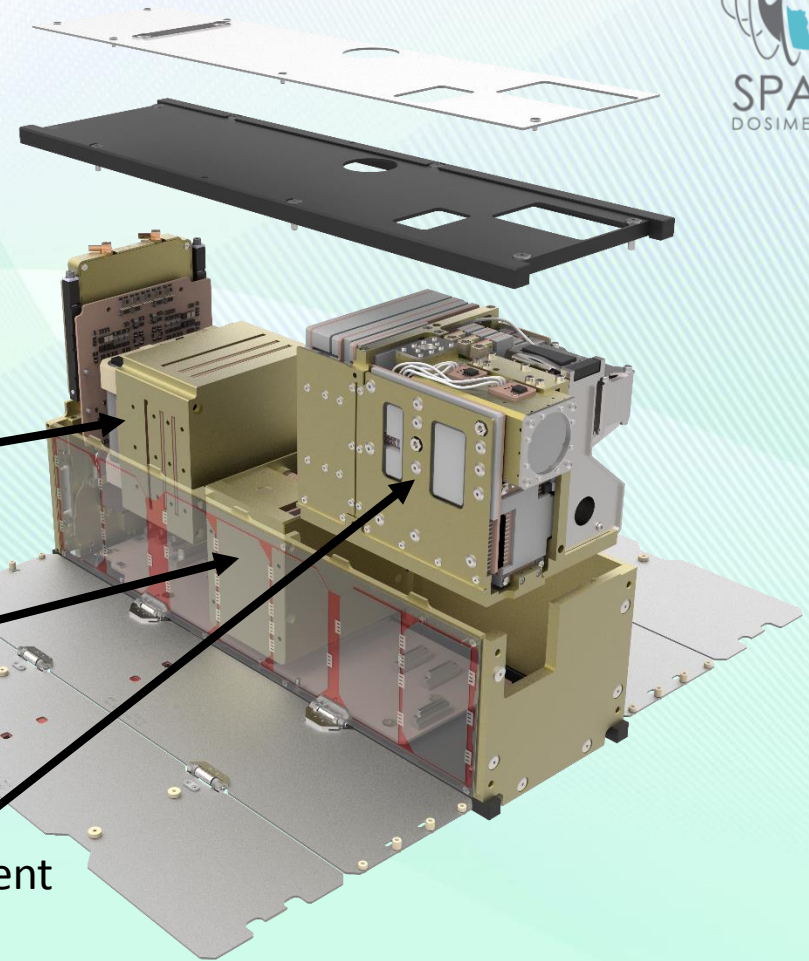


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Platform systems

ADCS

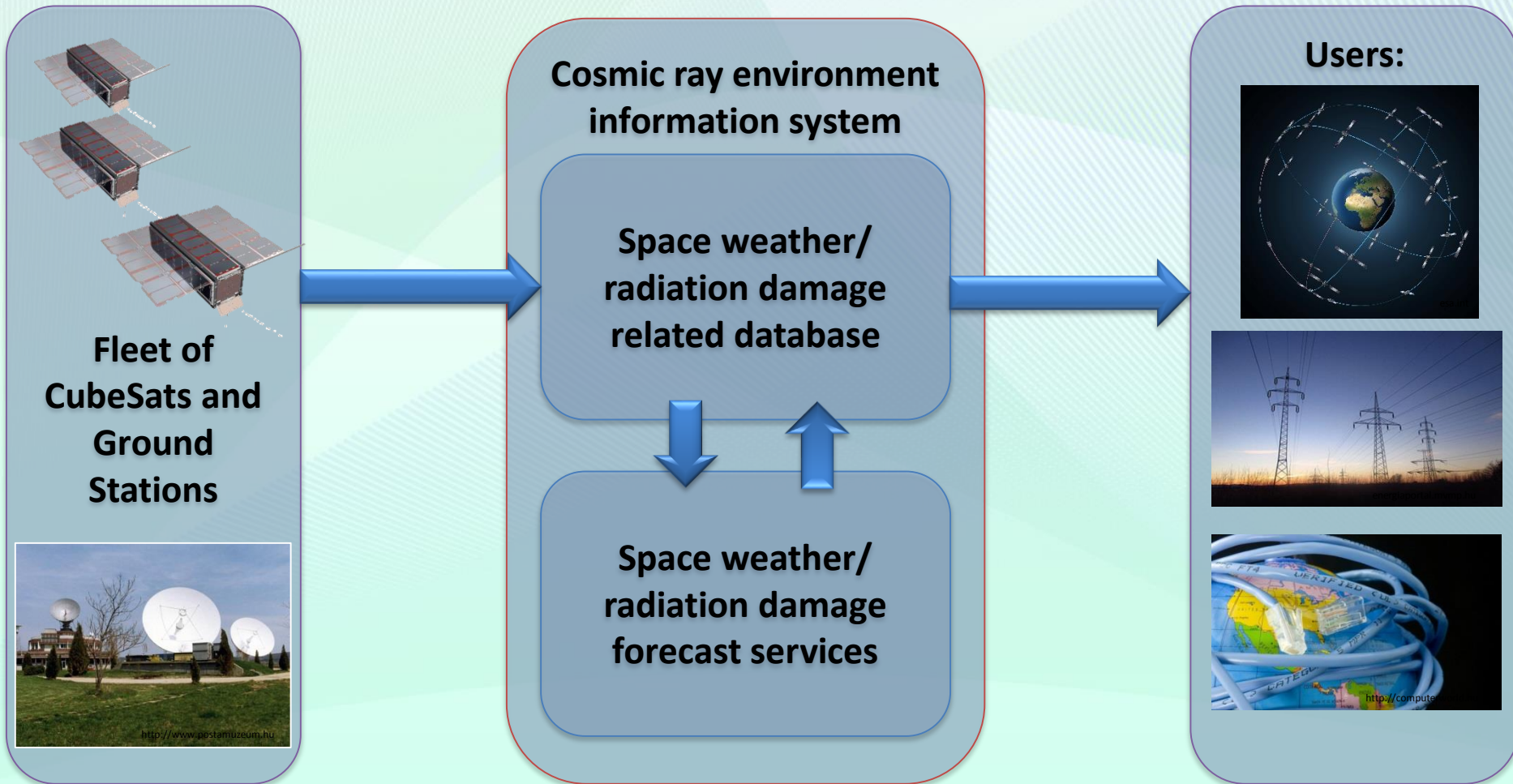
RadMag instrument



- **Future vision: CROSS Network = Cosmic Ray Observatory Satellite System**

Future vision

- ***CROSS Network = Cosmic Ray Observatory Satellite System***



Conclusions



Conclusions

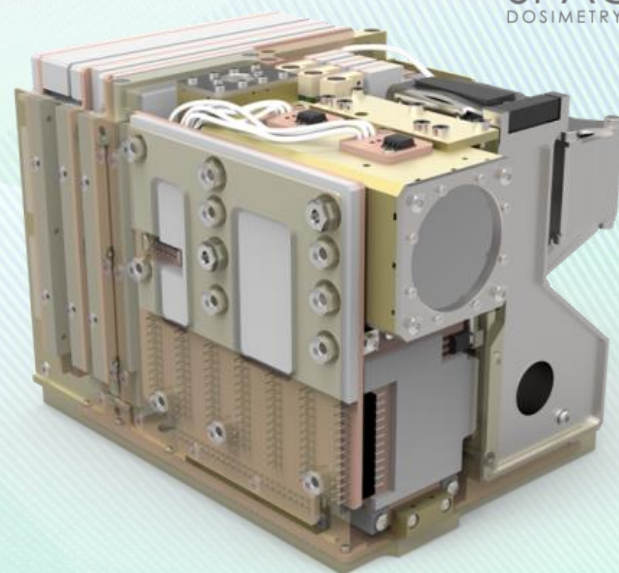
- ***New space weather monitoring instrument development: RadMag***
- **Unique combination of cosmic ray and magnetic field measurements into one instrument**
 - Very small size to fit for CubeSat/SmallSat missions (fitting ~1.2U CubeSat standard)
 - Built-in boom system to support the magnetic field measurements
 - Relatively low-cost alternative for future space weather studies and in general radiation damage monitoring for commercial use
- Instrument development just now passed PDR and moving into phase C
- 1st IOD Mission: **RADCUBE 3U CubeSat**
 - Expected launch is Q3 2019
- **Future vision (CROSS Network): CubeSat/SmallSat constellation for space weather and radiation damage monitoring services**



**Imperial College
London**



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Thank you for your attention!

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CENTRE FOR ENERGY RESEARCH
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