

### CAN in Space Applications Small Satellite Platforms

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## SSTL X - series

- A new range of platforms with common core features
  20kg to 500kg
- A single system architecture and technology designed to meet all mission applications and requirements across the entire range.
- Common, modular and expandable specifcation available on all platforms and missions:
  - Dual redundant systems

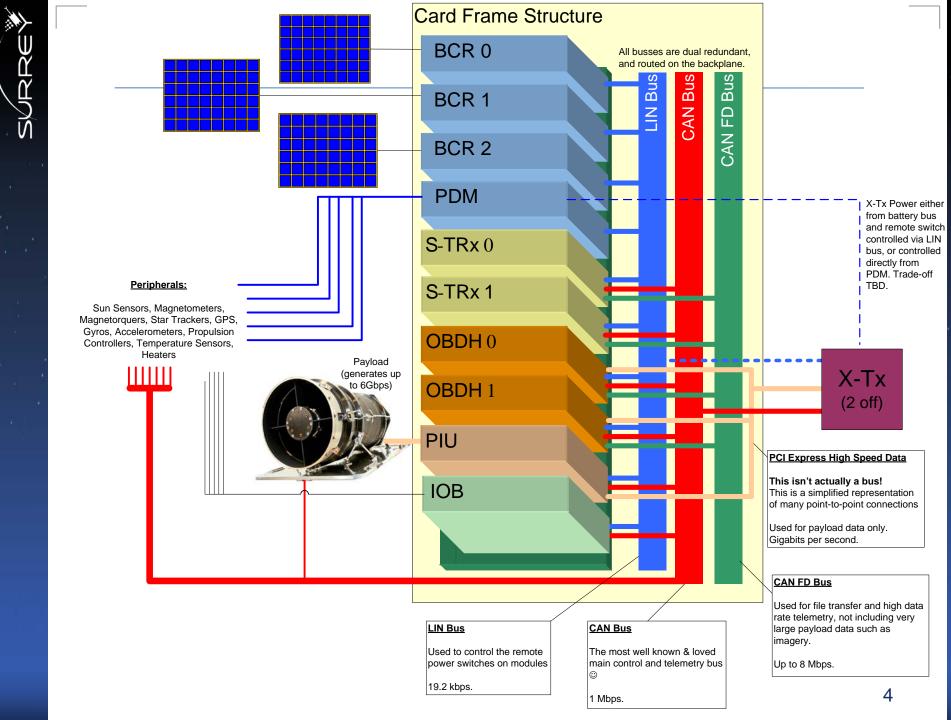
- Data storage up to Terabytes capacity
- Data bus transfer rates of Gigabits per second
- Onboard real-time processing of Gigainstructions per second
- High speed X-Band downlink



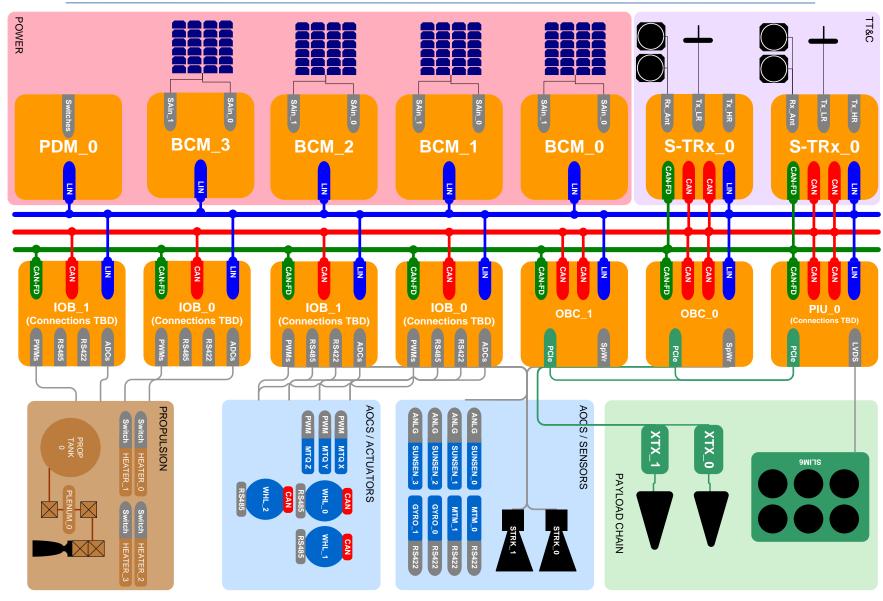
## **Internal Comms Architecture**

- Evolve from stacked trays with harness to card chassis with common backplane
- All avionics interconnects bussed for growth, flexible redundancy and testability
- All digital interconnects

- Backplane implemented via single common connector
- Second connector on each board passed through backplane to provide board specific external I/O
- CAN FD bus replaces existing mesh network connecting Tx, RX and OBCs
- LIN bus for power control to distributed power switches/LCLs and power conditioners
- CAN bus for connection to spacecraft subsystems (including legacy items)
- High speed payload interfaces sRIO up to approx 6 Gbps to adjacent three card slots
- Retain existing CONOPS with ground segment simplification

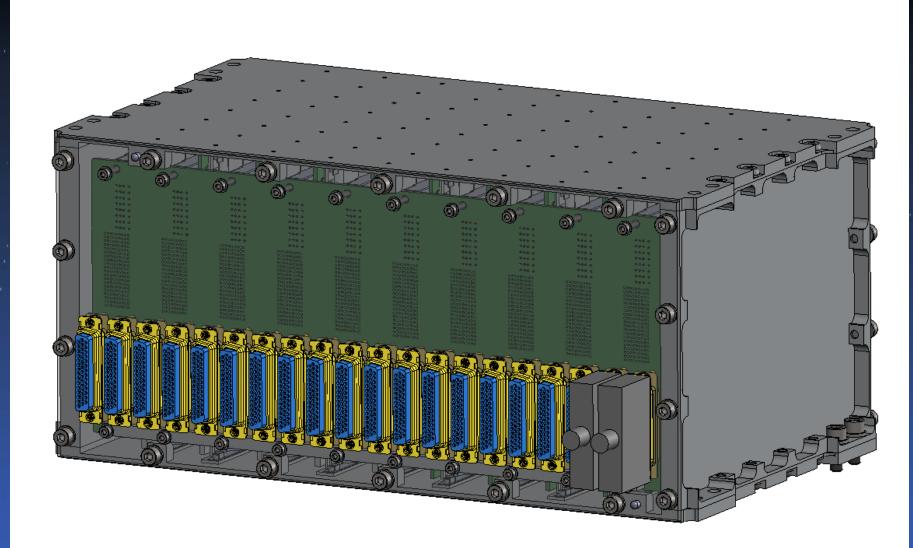


### **Avionics Block Diagram**



### **Avionics Card Frame**

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## CAN-FD

- CAN Flexible Data
- Payload field has flexibility in clock rate and data encoding and also allows up to 64 octet payload
- Throughput up to about 8Mbps possible
- Developed by Bosch and supported by all major European car manufacturers
- Field bus with guaranteed longevity, long term support, no threatening technologies
- But:

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- Devices only just starting sampling. IP cores currently expensive

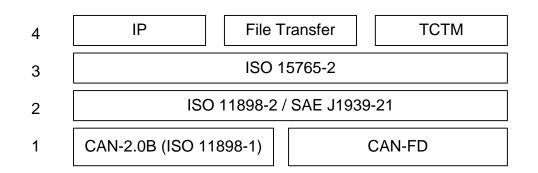
## **CAN Protocol**

#### CANopen studied but discounted

- + Most widely supported industry protocol
- + Will (eventually) support CAN-FD
- Very hungry stack (memory, CPU load, bus performance)
- Costly we would need to buy in a stack and have it customised to our platforms
- Doesn't match our service (peer to peer) model
- ISO15765-2 (ISO-TP) Adopted
  - + Offers everything we need and nothing we don't
  - + Lightweight and simple to implement
  - + SAE J1939 addressing scheme
  - + Automotive grade solution
  - Doesn't natively support transfers larger than 4kb, however use of a reserved field overcomes this without violating the specification

# ISO15765-2 (ISO-TP)

- Layer 3 protocol, natively supporting
  - Addressing (both Physical Device and Functional Groups)
  - Message Prioritisation
  - Robust multi-packet transfer and reconstruction
  - Layer 4 identification fields
- Works natively with CAN-2.0 or CAN-FD without adjustment
- Layer 4 is either:
  - IP Frames
  - Legacy File Transmission (<1Mb)</li>
  - TC/TM



### **Space-Ground Comms**

 Direct access to CAN-FD, CAN and LIN bus via link packet payload

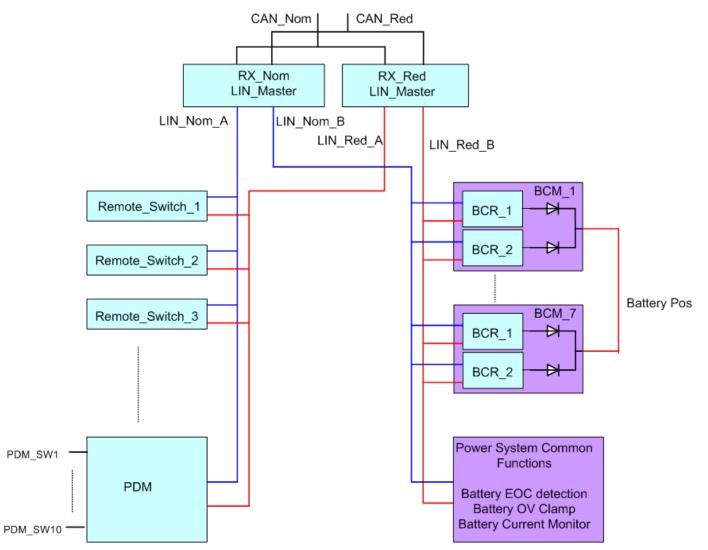
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 CCSDS compatibility via VHDL change

Backdoor		TT&C	File Transfer
CAN/ CAN- FD	LIN	Bespoke	Saratoga
AES-128		ТСР	UDP
		IP/IPSEC	
HDLC, 16bit CC-ITT CRC			
NRZI, G3RUH Scrambling (James Miller)			
BPSK			
Uplink 2.025GHz – 2.110GHz Downlink 2.200GHz – 2.29GHz			

### LIN Bus

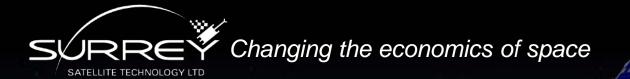
Single ended, low rate (19200 bps), half duplex, bi-directional bus Power consumption < 100  $\mu$ s



# Rapid IO

- High speed up to 6 Gbps
- Same PHY as PCIe

- Simple peer to peer operation doesn't need root complex
- Adopted by Next Generation Spacecraft Interconnect Standard working group



### Thank You

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