# Solar Wind Analyzer The Solar Orbiter milestone towards on-board intelligent decision making systems



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## Solar Orbiter SWA

#### Solar Wind Analyser

In-Situ Instrument suite (one out of the four on-board) composed of 4 sensors:

- EAS 1 & 2
- PAS
- HIS

and a

Data Processing Unit





#### **SWA Scientific Data**

Each sensor collects (and counts) particles carried by the solar wind,

sampling a portion of the full sky per

- azimuth angles,
- elevation angles,
- energy levels





#### **On-Board Scientific Data Processing**



## **SWA Operations**

Overview

anetek

italia

- The SWA sensors are operated in different "modes" (designed to meet the science goals), producing a wide variety of data products;
- Modes control 'raw' telemetry rates defining:
  - time resolutions,
  - on-board processing and/or
  - data compression

(i.e. specific data products);

 Duty cycle designed to comply to SWA suite telemetry budget limit (14,5 KiB/s).



# **Computational Load Analysis**

**Test Conditions** 

#### Hardware

- LEON-2 ASIC board (AT697F)
  - 32KB instruction, 16KB data caches
  - Clock: 100 MHz
    - ~86 Mips & ~23 Mflops
  - SRAM: 8 MB
- Software
  - C code implementations
    +
    RTOS (VxWorks) & ASW





Compression Ratio

 $CR = \frac{size \ of \ the \ input \ data \ stream \ [bytes]}{size \ of \ the \ output \ data \ stream \ [bytes]}$ 

Computational Load, estimated by means of

 $CPU \ Load = \frac{actual \ task \ computing \ time \ [ms]}{task \ cadence \ [ms]} \%$ 

### **Data Compression**

Overview

- Analysis
  - Select data and test cases
  - Entropy evaluation
  - "Discover" any structure in the data...



- Exploit the 3-D structure of the data [Energy, Azimuth, Elevation] groups
  - Correlation in the data
  - "preferential directions" in measurements similarity



## **Algorithms Definition**

Approach



#### **Computational Load**

#### **Estimated Worst-Case**



\* NORMAL MODE: the estimated load includes three components:

- overall DPU task's worst case load
- EAS additional load due to trigger event (assuming an arbitrary task cadence of 1200 sec)
- PAS additional load due to trigger event (assuming an arbitrary task cadence of 1200 sec)



\*\* BURST MODE: the estimated load for a given algorithm includes two components:

- EAS load (average over a time slot of 1 sec)
- PAS load (average over time slot of 8 sec)

#### Data Structure: re-ordering counts

- EAS sensor's acquisition sequence:
  - From slower to faster variation: Elevation-Energy-Azimuth, hereafter shortly denoted as El-En-Az
- Possible permutations of the acquisition sequence have been considered





#### Re-ordering counts CCSDS 121



Target Compression Ratio: 4.3

#### «Complex» Re-ordering

" <i>Simple</i> " re-ordering (slow to fast variations)	CCSDS 121
Az-En-El	109%
Az-El-En	83%
En-Az-El	111%
En-El-Az	101%
El-Az-En	83%

Once Elevation as been detected as the direction showing more correlation between measured counts, it is possible to apply a *«complex» reordering* able to discover more (removeable) redundancy by avoiding *«jumps»* in data continuity





#### Moments of a distribution

Given a velocity distribution function f, one can define the n-th order moment, in the satellite reference frame as

$$M_n = \int f(\boldsymbol{v}) \boldsymbol{v}^n d\boldsymbol{v}$$

The moment of order zero is called **number density** and it is defined as:

$$n=\int f(\boldsymbol{v})d\boldsymbol{v}$$

The first order moment is called **number flux density vector**:

$$n\boldsymbol{V} = \int \boldsymbol{v}f(\boldsymbol{v})d\boldsymbol{v}$$

-> from which one can compute the *flux velocity* by dividing for *n*.

The second order moment is the **moment flux density tensor**:

$$\boldsymbol{\Pi} = m \int \boldsymbol{v} \boldsymbol{v} f(\boldsymbol{v}) d\boldsymbol{v}$$

-> from which one can compute the *temperature* 

Finally, the third order moment is the **energy flux density vector**:

$$\boldsymbol{Q} = \frac{m}{2} \int v^2 \boldsymbol{v} f(\boldsymbol{v}) d\boldsymbol{v}$$



## **Operations Telemetry Corridor (OTC)**

An input to instrument planning at *medium-/short-term* planning cycles.





From EGSE SW to Continuous Integration in Space

> SpacePTS Payload Test System





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#### Conclusions

- 1. The increased level of *autonomous scientific data assessment* presents a possible solution to classical issues like bandwidth limitations, which can be conflated into the problem of data modeling
- 2. Onboard science data analysis will improve the capabilities of existing sensors and enable *transformative new operational modes* to address novel science issues thus relieving constraints on time, bandwidth and power, and by *responding automatically to events on short time scales*
- 3. Unprecedented opportunities to *downstream* data from Space to Earth
- 4. The technologies/methods designed for the SWA's on-board science data processing chain, in line with the *ESA OBPDP roadmap*





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