

# e.Deorbit and the Space Servicing Vehicle

Clean Space

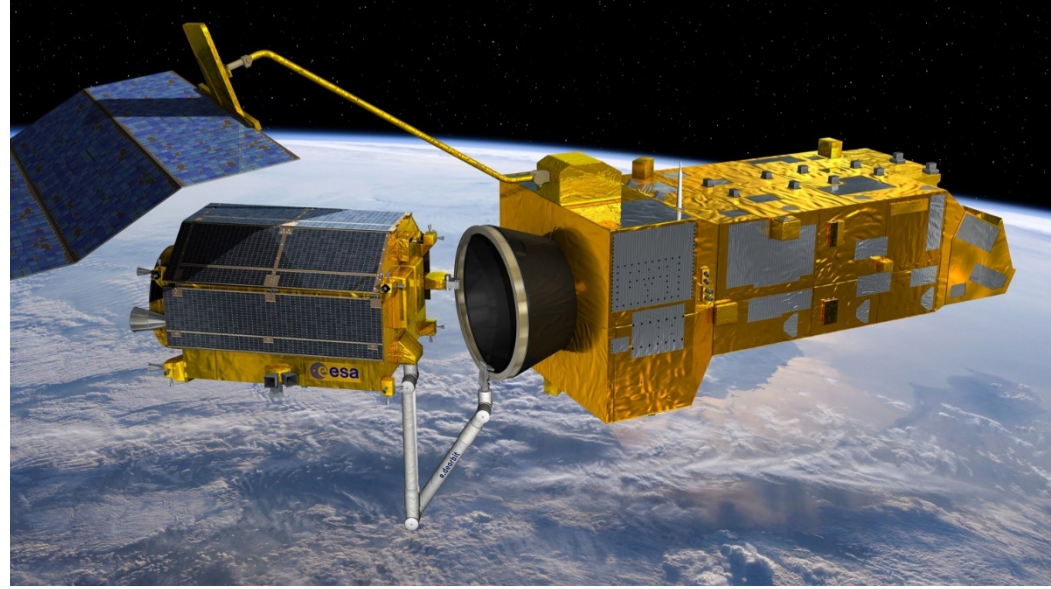
23/10/2018

# Active Debris Removal



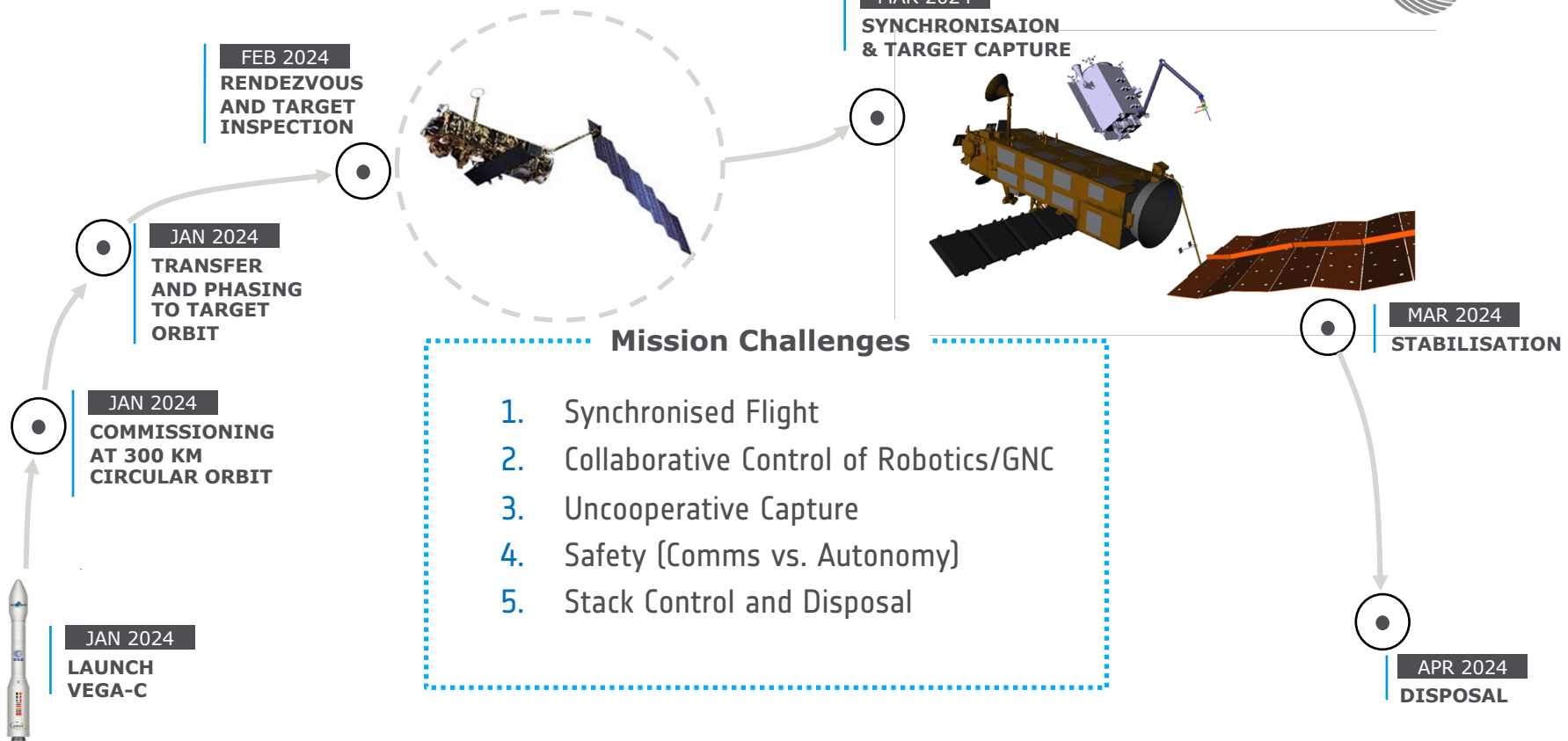
**e.deorbit**

→ MISSION GOAL

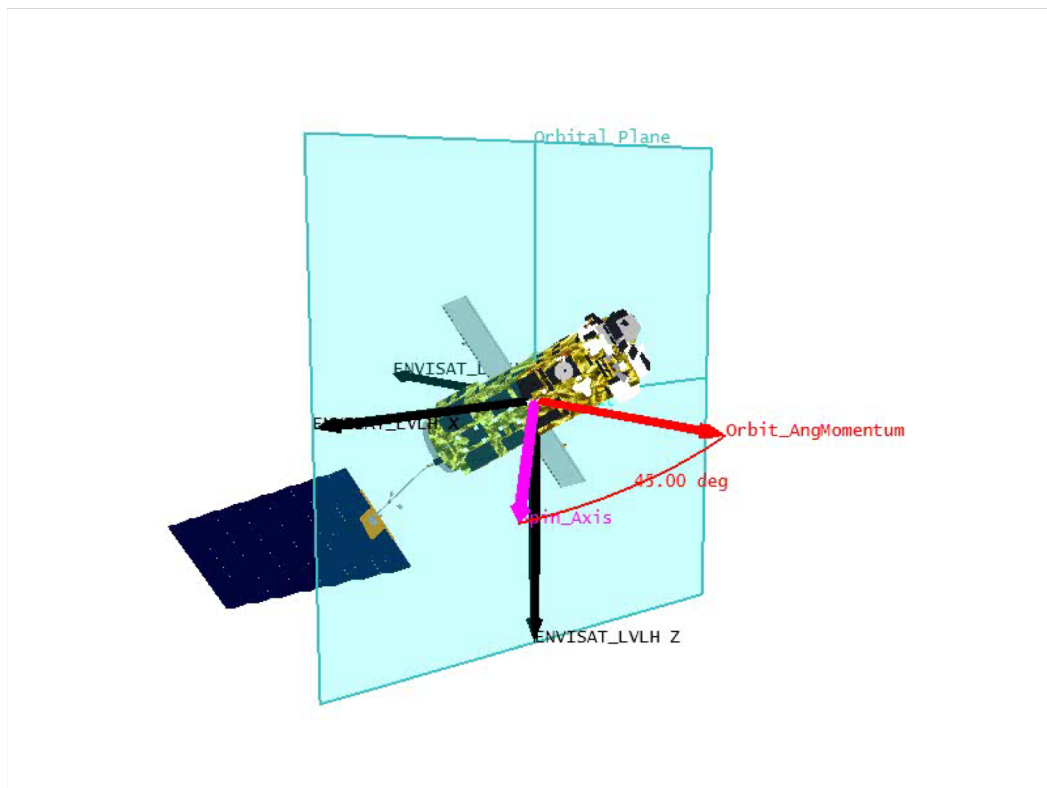


To remove an ESA-owned heavy debris from 800-1000 km (near polar region).

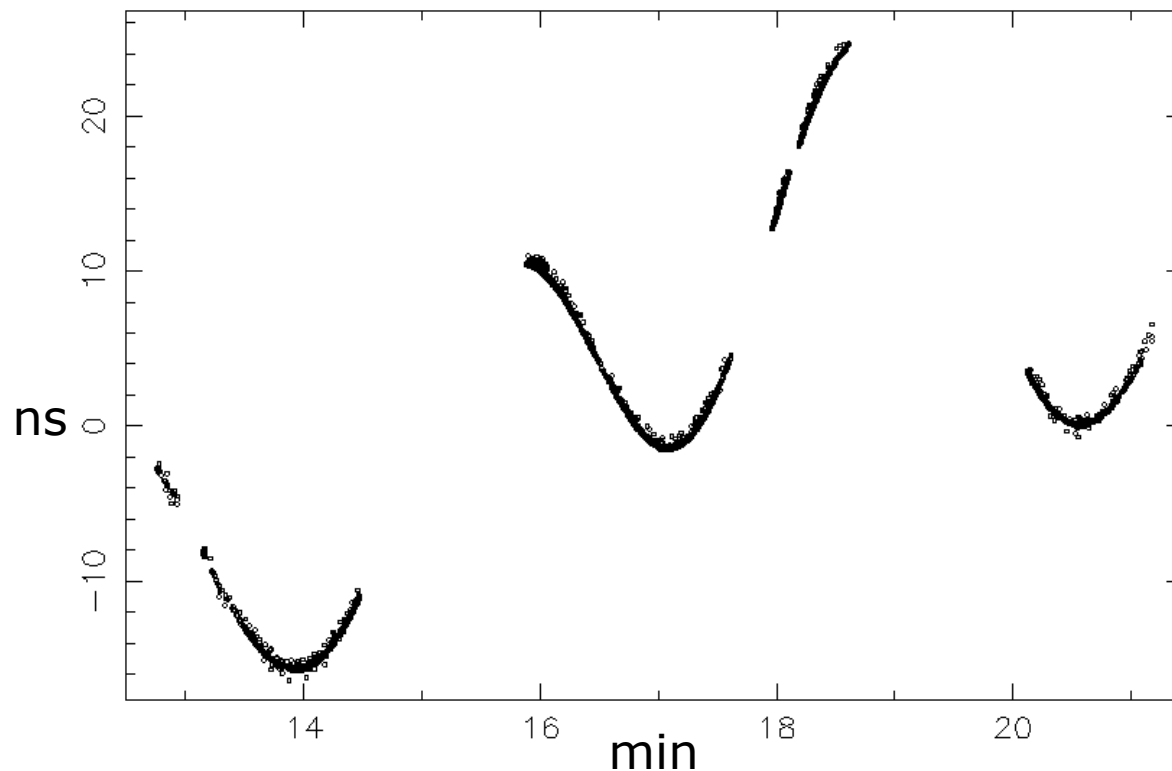
# e.Deorbit Mission Scenario



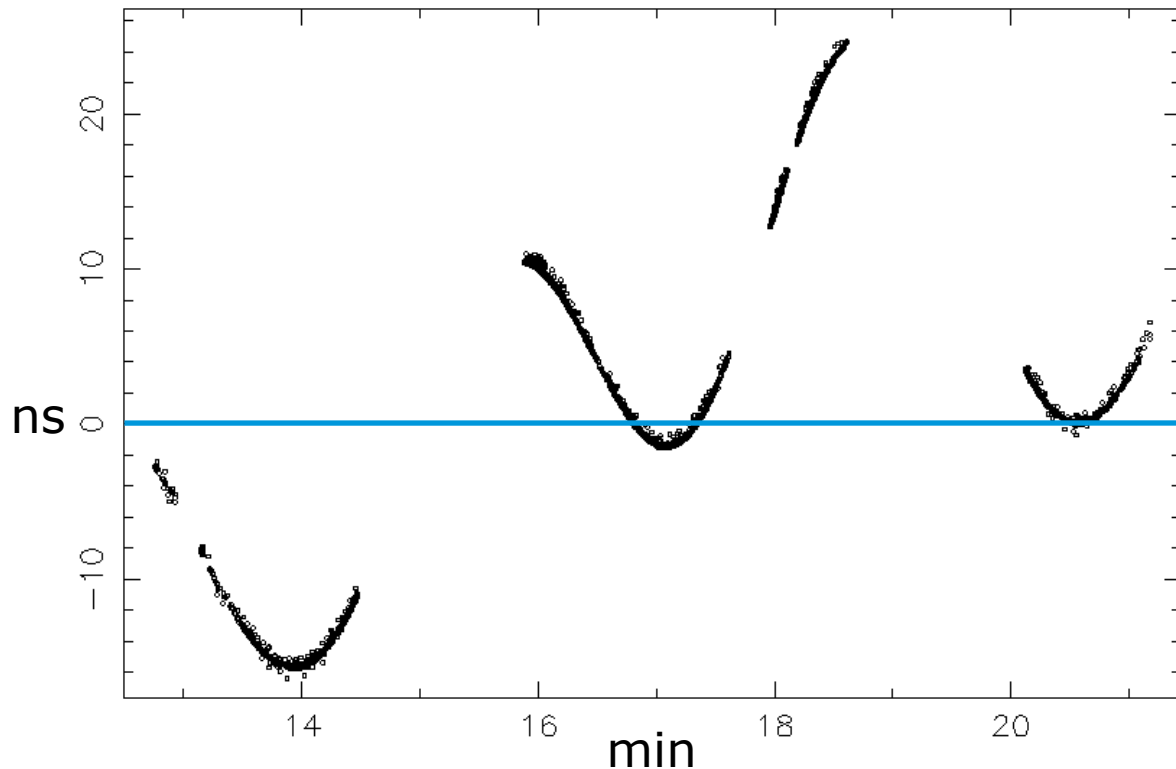
# ENVISAT Attitude Update



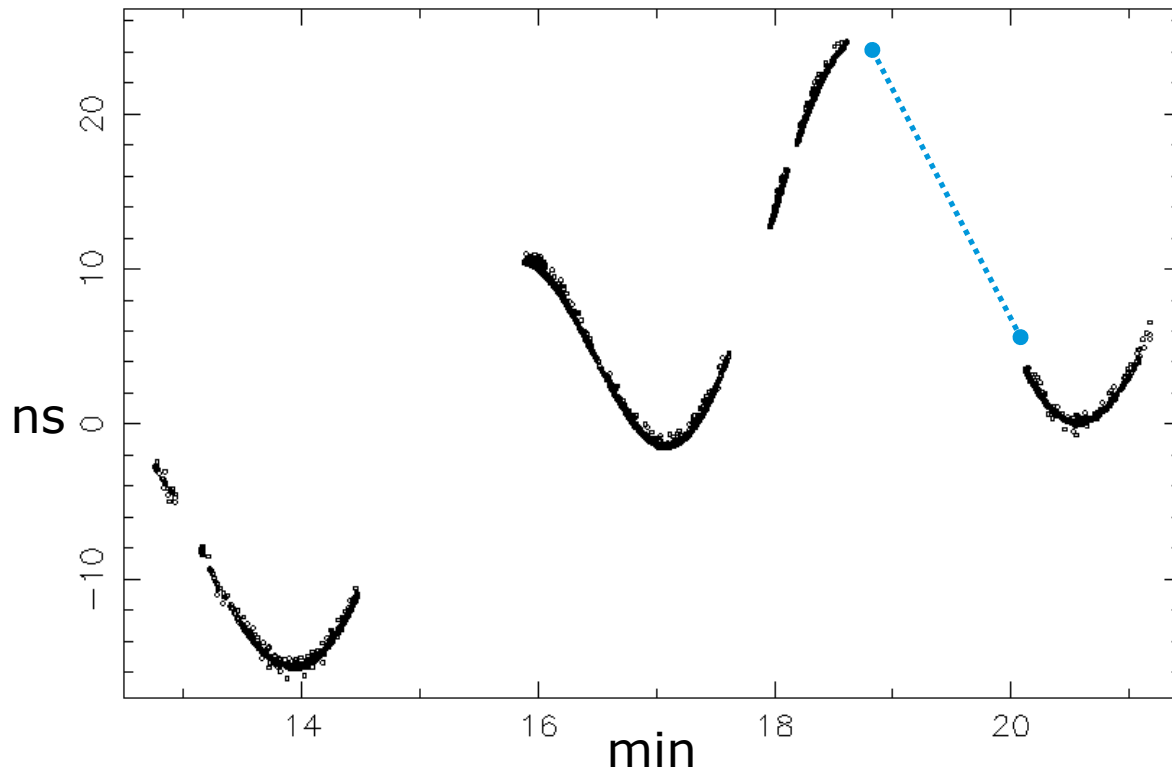
EV20SE18H OBSERVED-COMPUTED [NS] (2nd Counter)



**Credits: AIUB 20<sup>th</sup> of September**

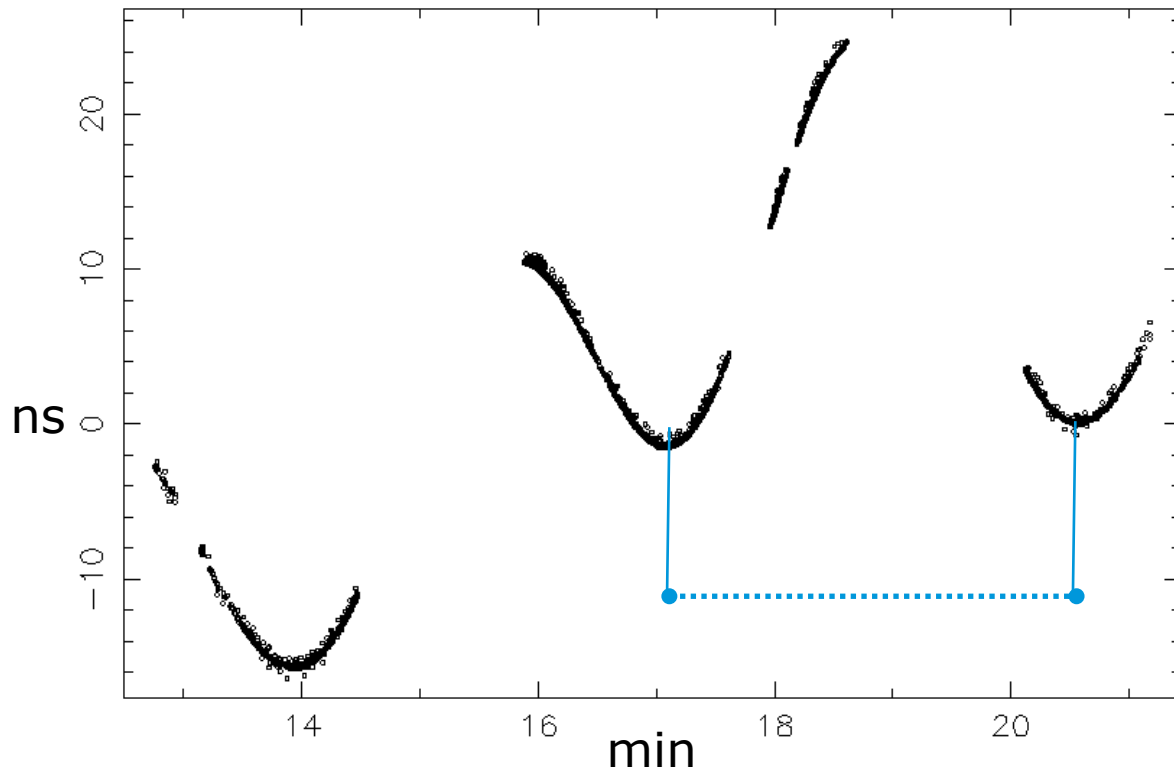


**Expected Position**



**Gaps in Observation  
→ Possibly due to tumbling motion**

EV20SE18H OBSERVED-COMPUTED [NS] (2nd Counter)

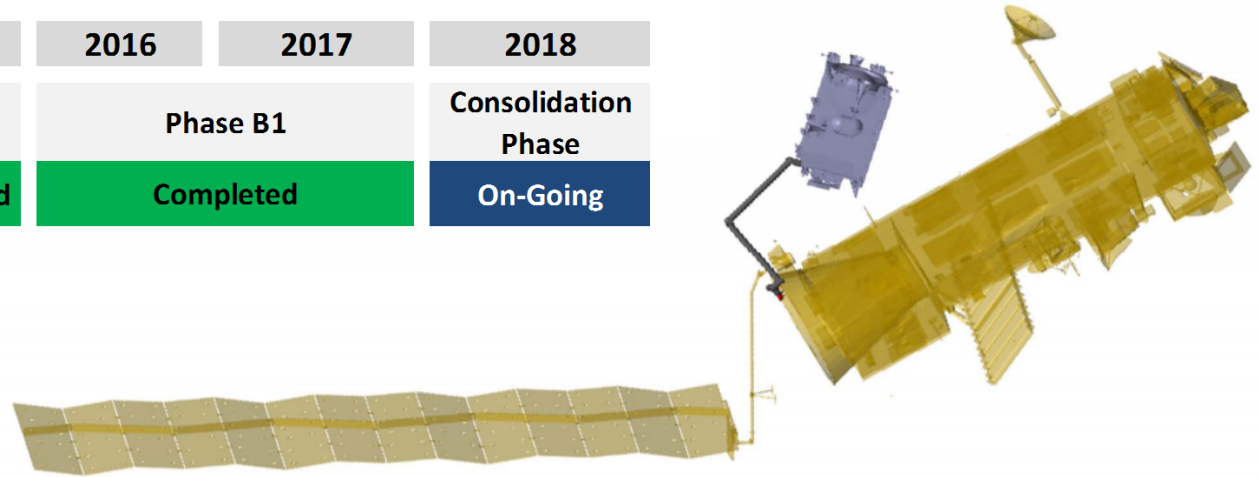


**Period of 190s**



# Active Debris Removal - e.Deorbit

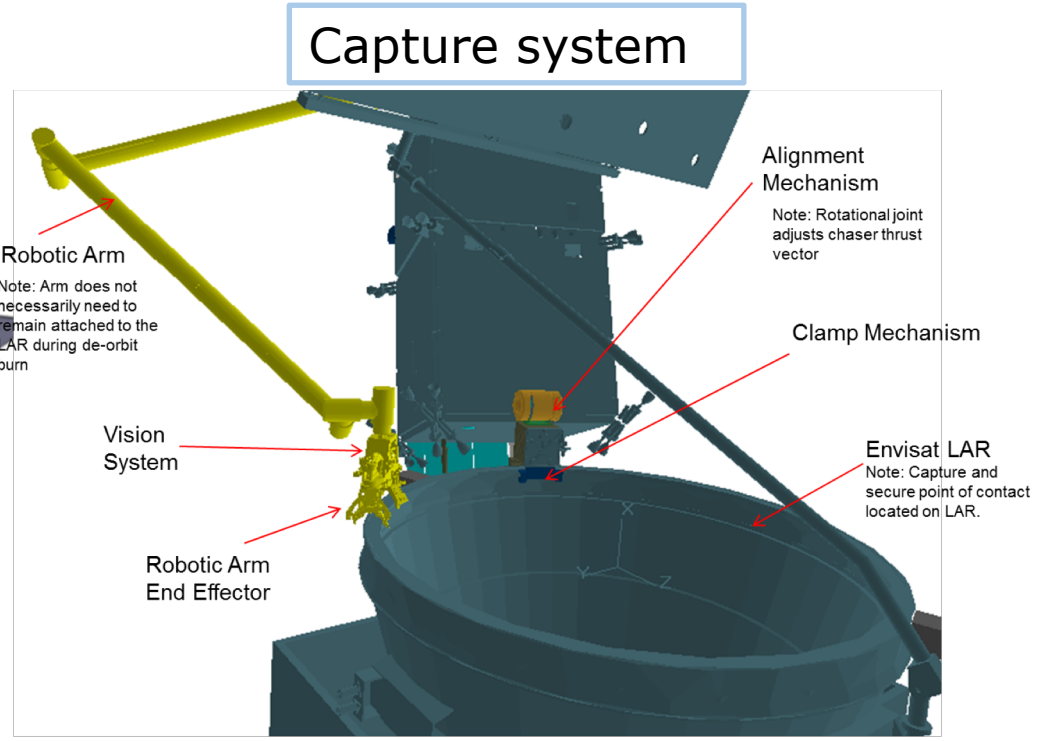
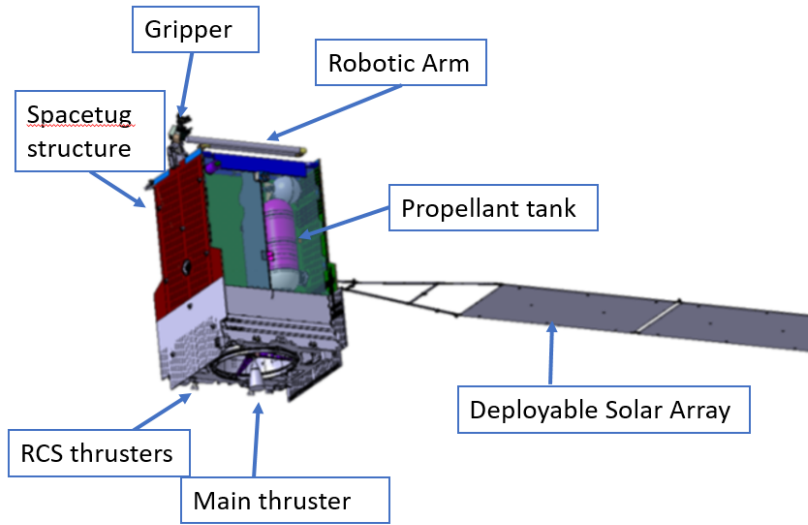
|                  | 2014                 | 2015                 | 2016                  | 2017 | 2018                            |
|------------------|----------------------|----------------------|-----------------------|------|---------------------------------|
| e.Deorbit System | Phase 0<br>Completed | Phase A<br>Completed | Phase B1<br>Completed |      | Consolidation Phase<br>On-Going |



## → e.Deorbit Consolidation Phase

A bridge between Phase B1 and Phase B2, with the objective to study the synergies with e.Deorbit and the Space Servicing Vehicle / Space Tug

# e.Deorbit Consolidation Phase



|                       |                |
|-----------------------|----------------|
| <b>Total Wet Mass</b> | <b>2977 kg</b> |
| Dry Mass              | 2025 kg        |
| Propellant            | 952 kg         |



# Clean Space – Active Debris Removal and Servicing



|                  | 2014                 | 2015                 | 2016                  | 2017 | 2018                            |
|------------------|----------------------|----------------------|-----------------------|------|---------------------------------|
| e.Deorbit System | Phase 0<br>Completed | Phase A<br>Completed | Phase B1<br>Completed |      | Consolidation Phase<br>On-Going |

2018 Space Servicing Vehicle CDF Study - Complete

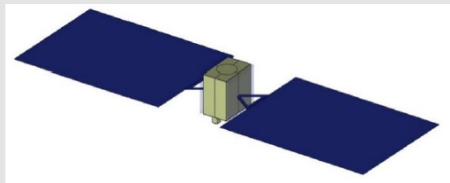
2018 Request for Information



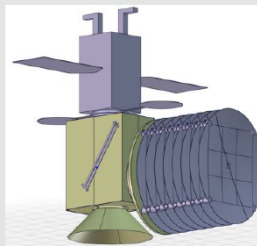
# Space Servicing Vehicle – CDF Pre-Phase A



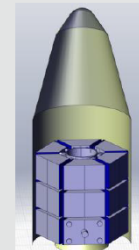
**Mission Scenario 1**  
Deep Space Gateway Tug



**Mission Scenario 2**  
Antenna Assembly  
for GEO S/C



**Mission Scenario 3**  
Megaconstellation  
ADR

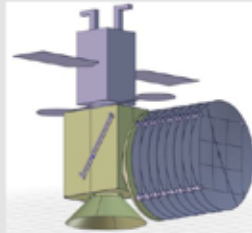


## Conclusions:

1. Synergies in Technologies
2. Different System Designs Required

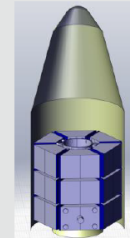
## → Follow-On Activities

### **Mission Scenario 2** Antenna Assembly for GEO S/C



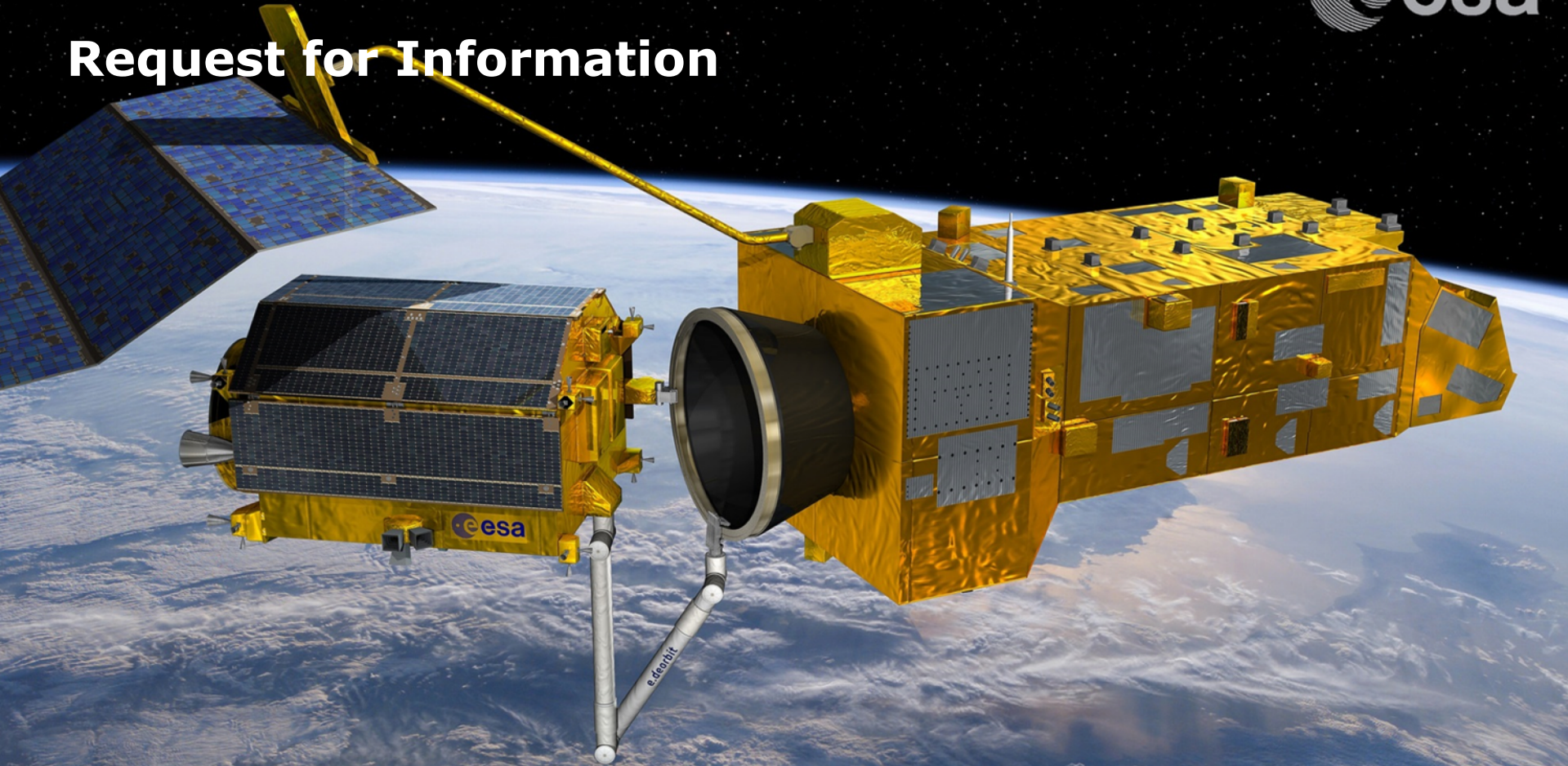
- On-Orbit Antenna Assembly
- Servicing Platforms in GEO

### **Mission Scenario 3** Megaconstellation ADR



- Megaconstellation Phase A

# Request for Information



This initiative addresses three objectives:

- to perform the removal of ESA satellite(s) as a precursor of in-orbit servicing;
- to demonstrate technologies, functions and operational know-how to perform other in-orbit services; and
- to achieve the above by means of service contract(s) to provide an opportunity to space industry to enter into this new space market.

## Mission Statement

**Perform the removal from orbit of an ESA-owned satellite(s) in execution of service contract(s) placed by ESA and demonstrate capabilities and technologies for in-orbit servicing**



**MIS-01** Remove from orbit ESA-owned satellite(s) with a total mass larger than 100 kg at least 5 years prior to its/their natural re-entry\*

**MIS-02** Demonstrate feasibility of critical technologies enabling other in-orbit TBD servicing opportunities

**MIS-03** Provide a robust business model for in-orbit servicing activities beyond the service provided to ESA

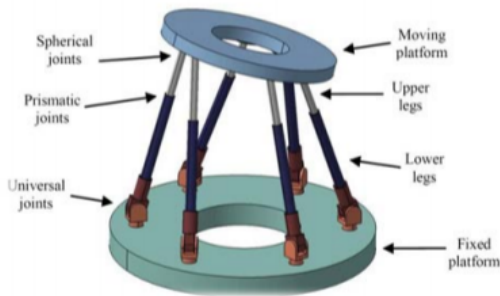
**MIS-04** Comply to space debris mitigation requirements stated in RD.1

\* list of satellites published



# Servicing and ADR Technologies

## 3.3 M€ in TRP / GSTP Technology Developments on-going

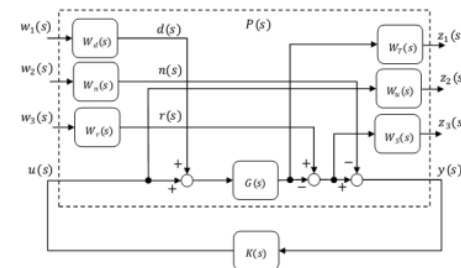


**6-Degree of Freedom Gripper Platform**

Credits: Hellenic Technology Robotics

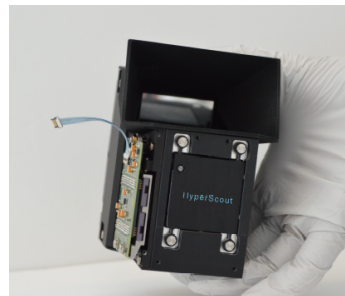


**Tether**



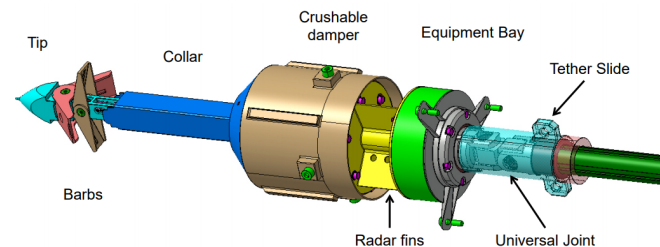
**Combined Control**

Credits: GMV



**Multi-Spectral Camera**

Credits: Cosine



**Harpoon Development**

Credits: Airbus

| Activity Title  | Target Prog. | Total Budget | Year | Status   |
|---|--------------|--------------|------|----------|
| Space Servicing Vehicle Robotics Subsystem - Phase 1 (RENEGADE) | GSTP         | €7,000,000   | 2018 | Approved |
| e.Inspector Phase A   | GSTP         | €250,000     | 2019 | Planned  |
| Space Servicing Vehicle Robotics Subsystem - Phase 2            | GSTP         | €5,500,000   | 2020 | Proposed |
| Space Servicing Vehicle Pre-Development of Algorithms           | GSTP         | €400,000     | 2019 | Proposed |
| Space Servicing Vehicle Multispectral Camera                    | GSTP         | €800,000     | 2019 | Proposed |
| Space Servicing Vehicle Stereo Camera                           | GSTP         | €800,000     | 2019 | Proposed |

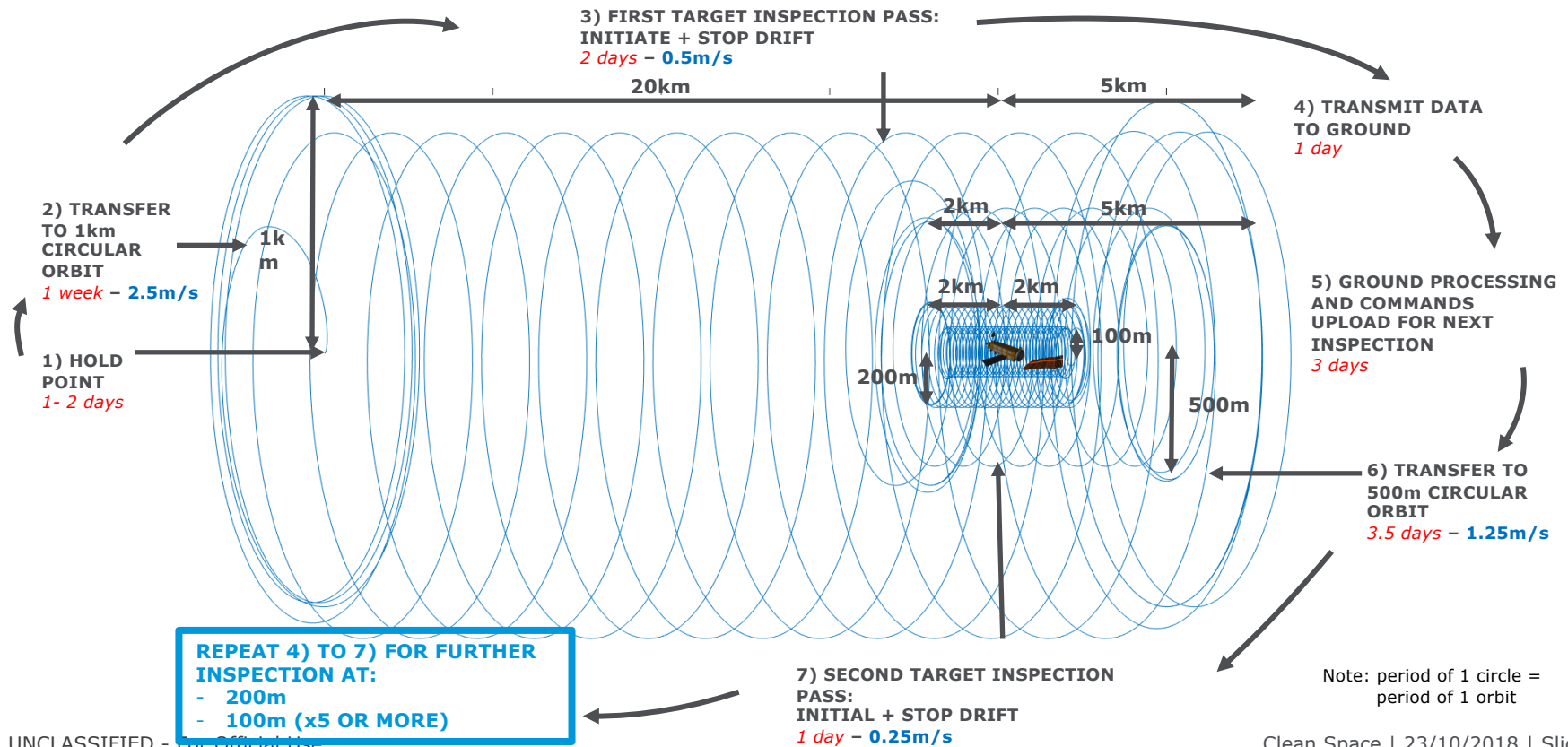


## → Mission Objectives

R-MIS-1: Image ENVISAT in its current status

R-MIS-2: Use obtained images for the verification and validation of a Space Servicing Vehicle GNC sensors

# Inspection Timeline – EP to ENVISAT

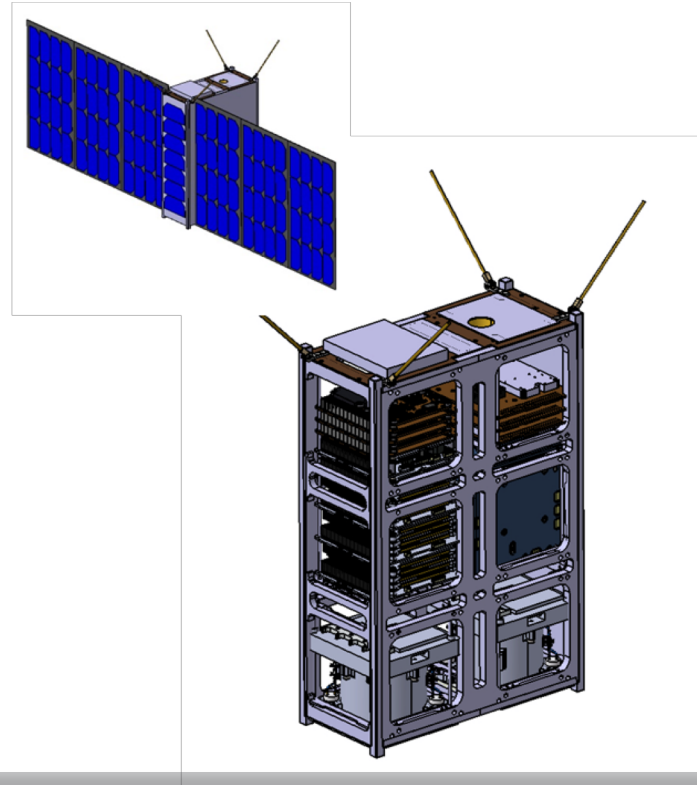


# e.Inspector Summary

## Subsystems:

- **Communications:** UHF (housekeeping) and S-band (payload data) (1 antenna, 1 transceiver of each)
- **Data-handling:** 1 CubeSat computer, 1 FPGA, 1 power board
- **Power:** 1 PCDU, Battery, 1 Body Mounted 3U SA, 2 SA wings of 3x6U
- **Mechanisms:** 2 HDRM, 12 Hinges
- **AOCS:** GPS rcvr, 6 SAS, IMU, 3 MT, 1 Rel Nav Imager, 3 RW, 1 STR
- **Thermal Control:** 45 Temp sensors, 15 Heaters, BP, MLI, OSR
- **Propulsion:** 2 FEPP systems

**Cubesat Dispenser Type: ISIS – 6U**



Proposed as a 250 k€ Phase A in GSTP

# Design for Removal

## Following the failure of a satellite...

- 1 Track the satellite orbit, and characterize the attitude – *implement retroreflectors, radar corner reflectors*
- 2 Launch an ADR satellite & perform the rendezvous and approach – *implement RF tags, 2D-3D markers*
- 3 Stabilise the satellite – *shortcircuit the magnetorquer*
- 4 Capture the satellite – *integrate a capture interface*



Planned for the future Copernicus Programme.



Design for Removal - Passive Mechanical and Rendezvous Interface for Capture after End-of-Life (PRINCE)

PRINCE shall include the following elements developed to TRL 3:

- Passive interface on the target satellite including the mechanical interface to facilitate capture and the navigation supports (e.g. 2D / 3D markers)
- Mechanical interface on the space servicing vehicle (e.g. the gripper at the end of a robotic arm)

The output: Interface Control Document to be provided to EO

150 k€ Open Competition (ITT Closure - 7 weeks)

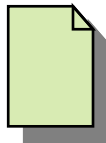
# Scope

## Objective:

- Provide technical input for a future discussions on design principles / requirements for safe close proximity operations

## Benefits:

- Support industry through technical guidance and identification of potential licensing methods
- Capture of knowledge
- Protection to the orbital environment and other assets
- Enable international engagement



**Output:**  
Technical input for a future design principles document



Large Space Integrators



## Working Group

|          |            |                |
|----------|------------|----------------|
| Robotics | Propulsion | Safety         |
| Legal    | Quality    | ATV Operations |
| GNC      | Systems    | Operations     |

2019 Industrial Activity ≈ 400 k€

# Activity Summary



## System:

- RFI – Q4/18 or Q1/19
- TBC - Servicing Provisions for GEO Satellites (ARTES) – '19 [600k€]
- TBC - On-Orbit Assembly (ARTES) – Q4/18 [250 k€]
- TBC - Megaconstellation Phase A (ARTES) – Q1/19 [350 k€]

## Technology:

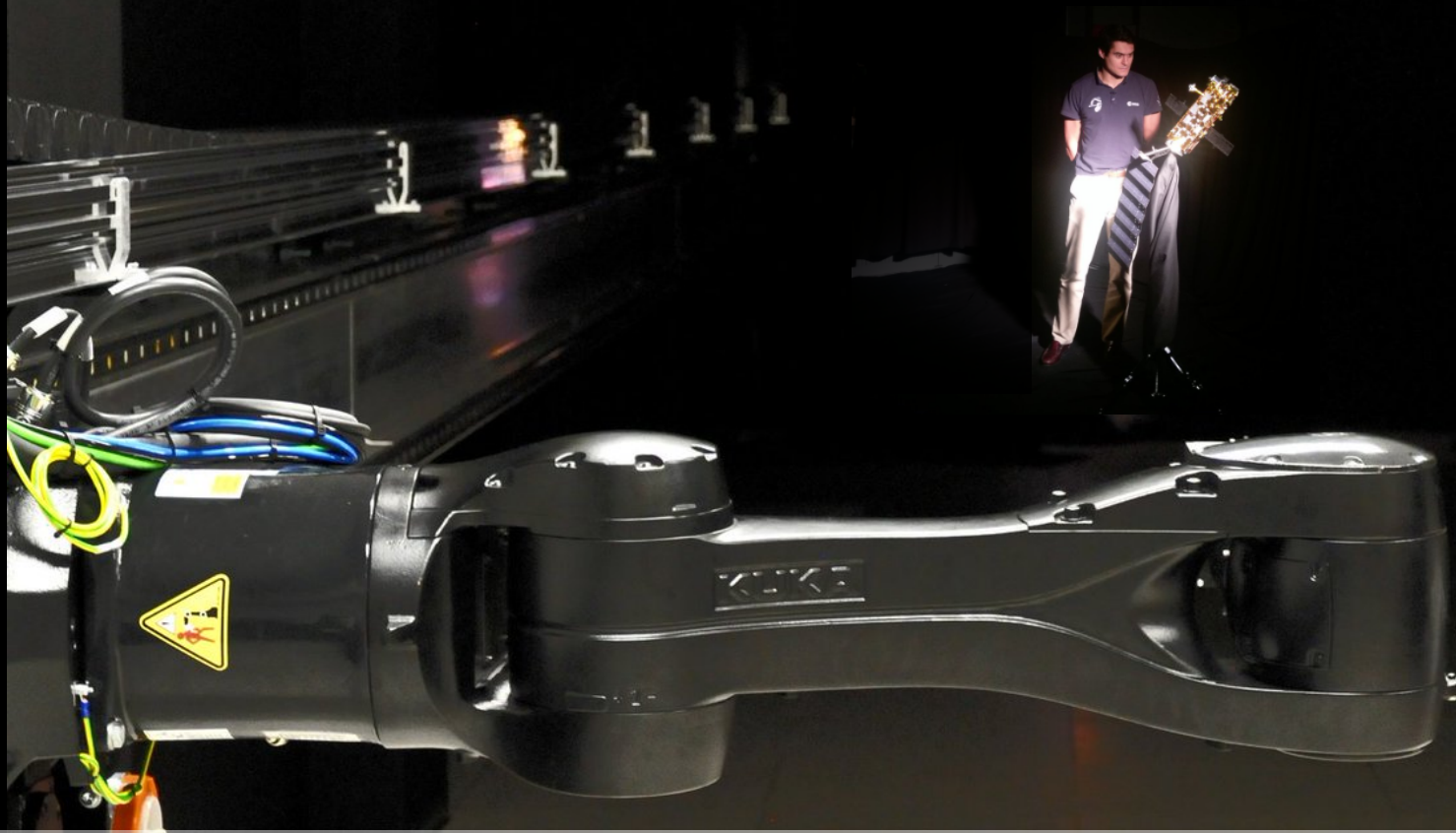
**Robotics for Space Servicing Vehicles:** Renegade ITT Closed [7M€]

**GNC Space Servicing Vehicles:**

- 4 x activities in GSTP Workplan [10 M€]
- e.Inspector Phase A Proposed [250 k€]

**Design for Removal:** 1 x mechanical interface/robotic (open) [150 k€]

**Close Proximity Operations:** Q1/Q2 '19 [400 k€]



**Don't Forget the Lab Tours**