

# **Clean Space Industrial Days 2018**

# **COMRADE - FULLY COMBINED**

# **CONTROL FOR ROBOTIC**

# **SERVICING SPACECRAFT**

# COMRADE: CONTROL & MANAGEMENT OF ROBOTICS ACTIVE DEBRIS REMOVAL

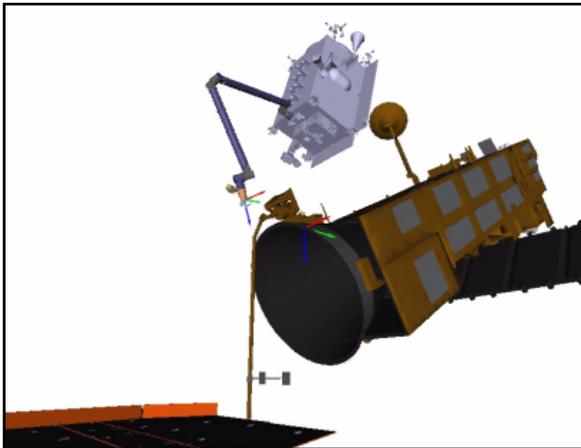
- **ESA** funded TRP
- Furthering technology for ADR/Servicing
  - Reducing the number of debris objects in space.
  - Extending life or repairing damaged on-orbit assets attractive economic option for satellite operators
- **Technical challenges Robotic Servicer:**
  - Control of **uncertain coupled dynamics** (spacecraft platform + robotic manipulator + and end-effector)
  - Synchronization with **fast tumbling** targets
  - Limitation of structural **loads** on arm



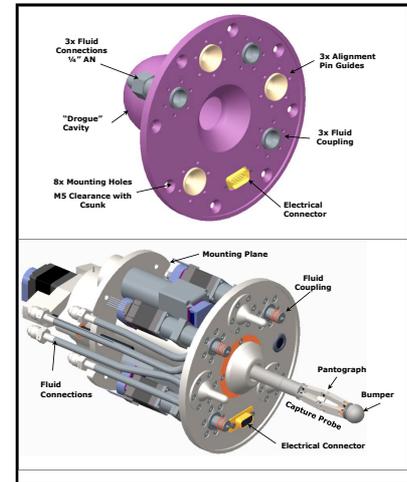
# FULLY COMBINED CONTROL

- **Fully combined control** ( alternative to decoupled, tele-op, *collaborative*)
  - overcome the problem of arbitrary, case-tailored control authority, improve performance, savings
- Two control design approaches:
  - **H $\infty$**  and
  - nonlinear **compliant Lyapunov-based**).

## ADR eDeorbit



## Servicing (ASSIST)

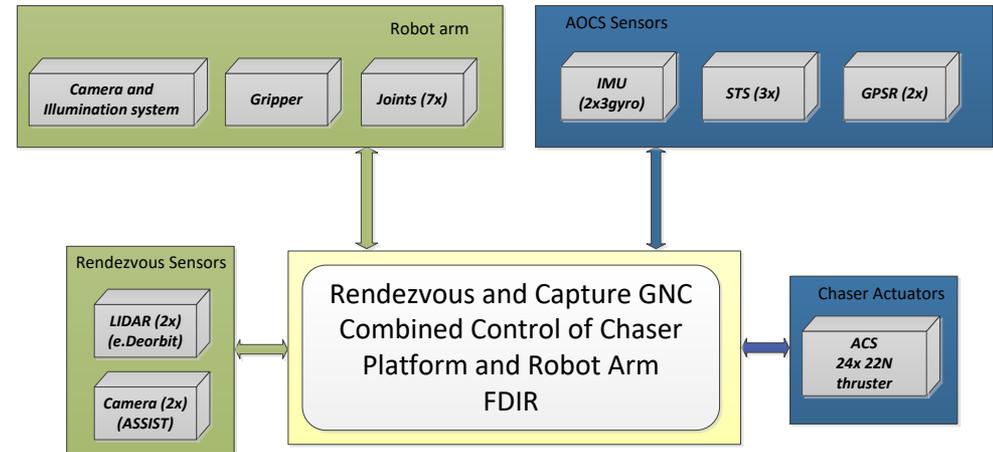


# SYSTEM SPECIFICATION

## Robotic Servicer Elements:

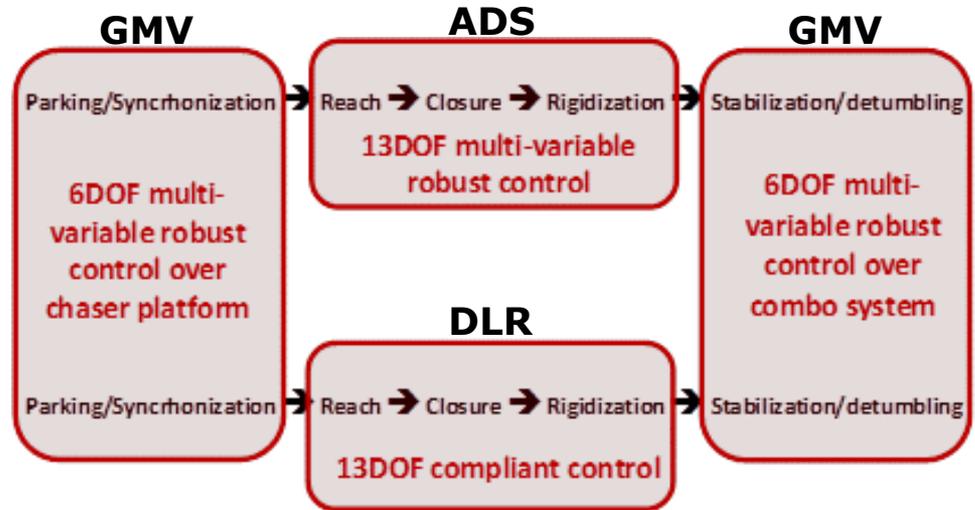
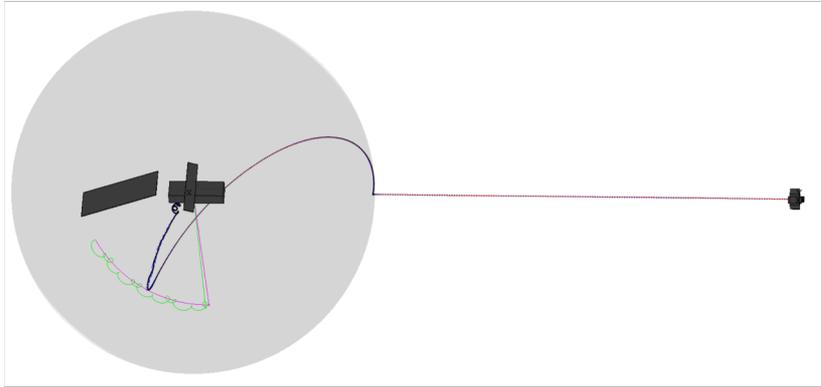
- AOCs sensors: **IMU, star tracker, GPS**
- Propulsion: **24x22 N** thrusters (eDeorbit-based)
- Relative navigation:
  - **LIDAR** for eDeorbit scenario
  - **Vision-based** camera for re-fueling scenario
- Robotic manipulator: **7 DOF** with joint encoders
- End-effector:
  - PIAP developed **gripper** for eDeorbit scenario
  - **ASSIST re-fueling** device developed by a team led by GMV
- Control analysis and synthesis considers:
  - Fuel **sloshing**
  - **Flexible modes** (solar arrays and robotic manipulator)
  - **Arm dynamics**

## Architecture



# MISSION PHASES

- Angular/linear **Synchronisation** of Chaser wrt target rotation
- **Reach and Capture** (Robotic arm deployment and target grasping)
- **Rigidisation** of robotic arm joints
- **Stabilisation/detumbling** of target rotation (for ADR case only)



# GNC SYSTEM REQUIREMENTS

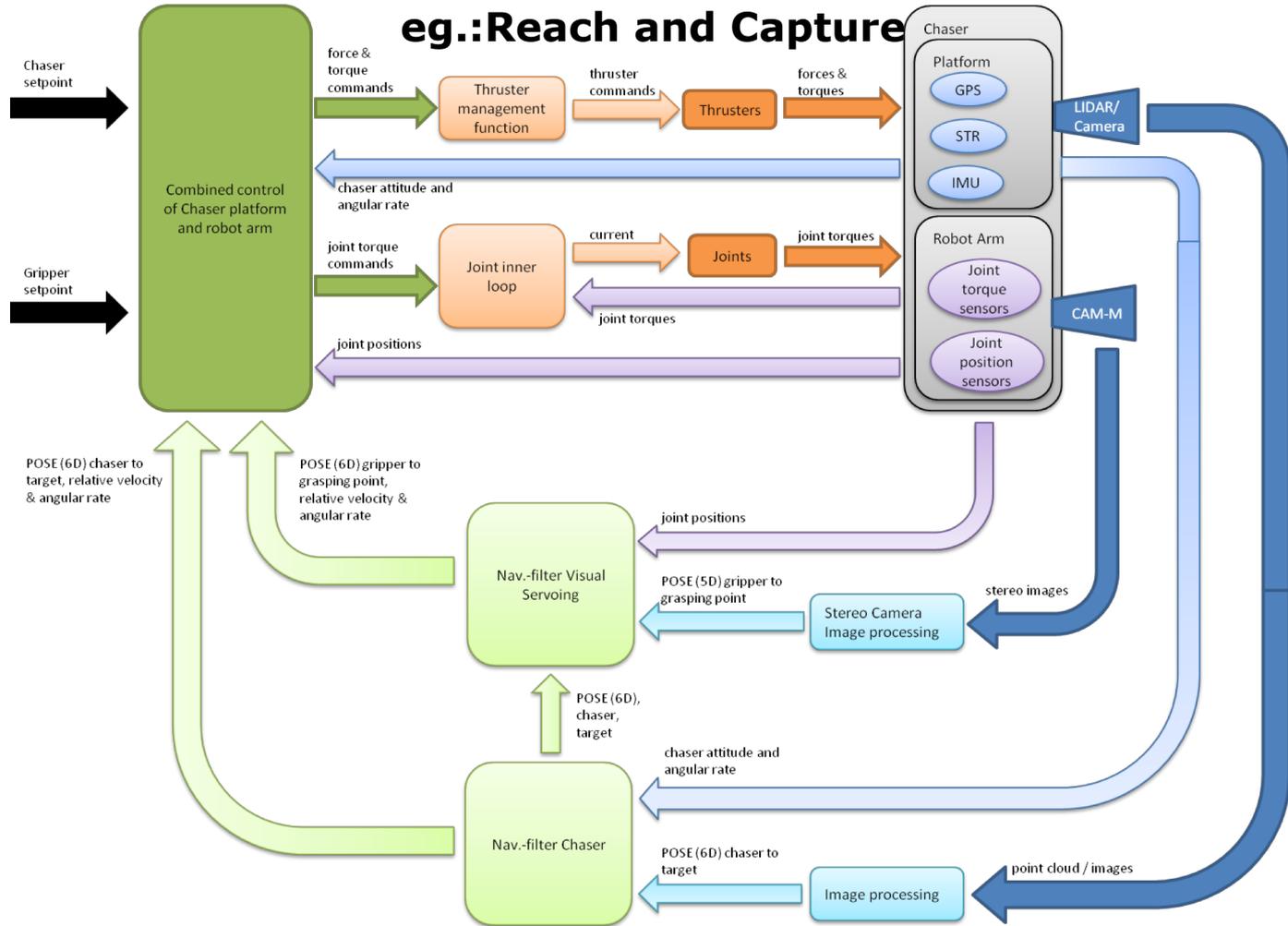
## Performance requirements:

- **Synchronization** control performance requirements (95%) for relative state (CoM to CoM):
  - [100 mm 10 mm/s 2° 0.5°/s] (6DOF control)
- **Reach & Capture** control performance requirements (95%) for relative to TGFF:
  - [50 mm 5mm/s 2° 0.5°/s] (CoM wrt to point in TGFF)
  - [10mm 5mm/s 2° 0.1°/s] (end-effector wrt to grasping point)
- **Stabilisation** control performance requirements ( $2\sigma$ ):
  - Angular rate of chaser+target damped to **0.5°/s**.

## Safety requirements:

- **Synchronization/Reach/Capture:**
  - Distance between chaser platform and target surfaces larger than **0.5m**.

# ARCHITECTURE



# MISSION PHASES

## ■ **Synchronisation**

- Robot arms is not active
- Forced motion to profile computed by guidance

## ■ **Reach and Capture**

- Station keeping at capture point
- Gripper moves towards LAR
- Control issues: Force and Torque to RCS , Joint Torque Commands
- Capture = Reach + Gripper control (closure command)

\*compliant control : stiffness and damping for end effector, chaser at set point

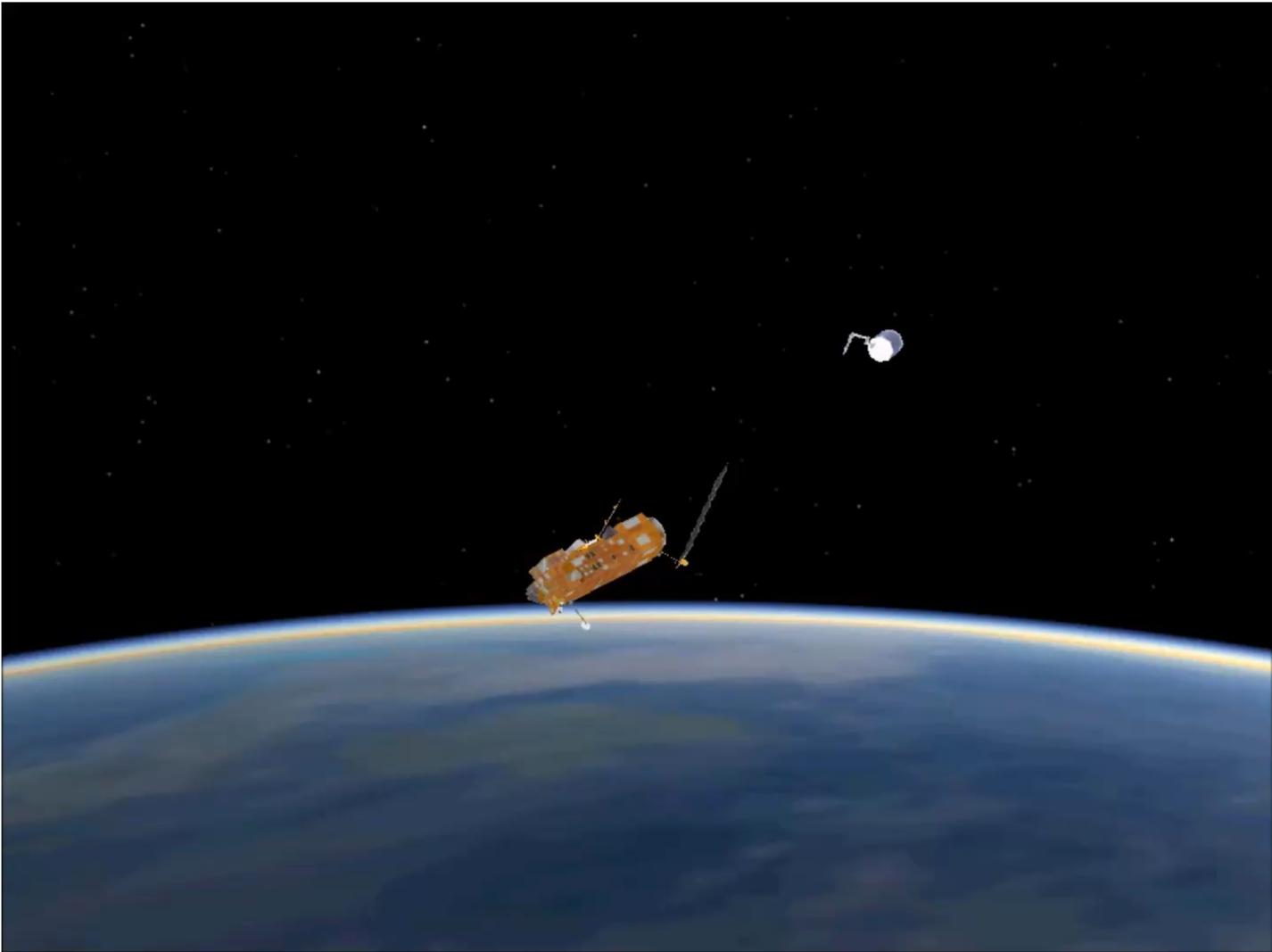
## ■ **Rigidisation** of robotic arm joints:

- Thruster commands are inhibited
- Angular rates of joints controlled to zero
- Locking brakes are engaged

## ■ **Stabilisation/detumbling** of target rotation (for ADR case only) – limit for joint torquers

## ■ **Escape** : combined control that tracks a collision-avoidant guidance

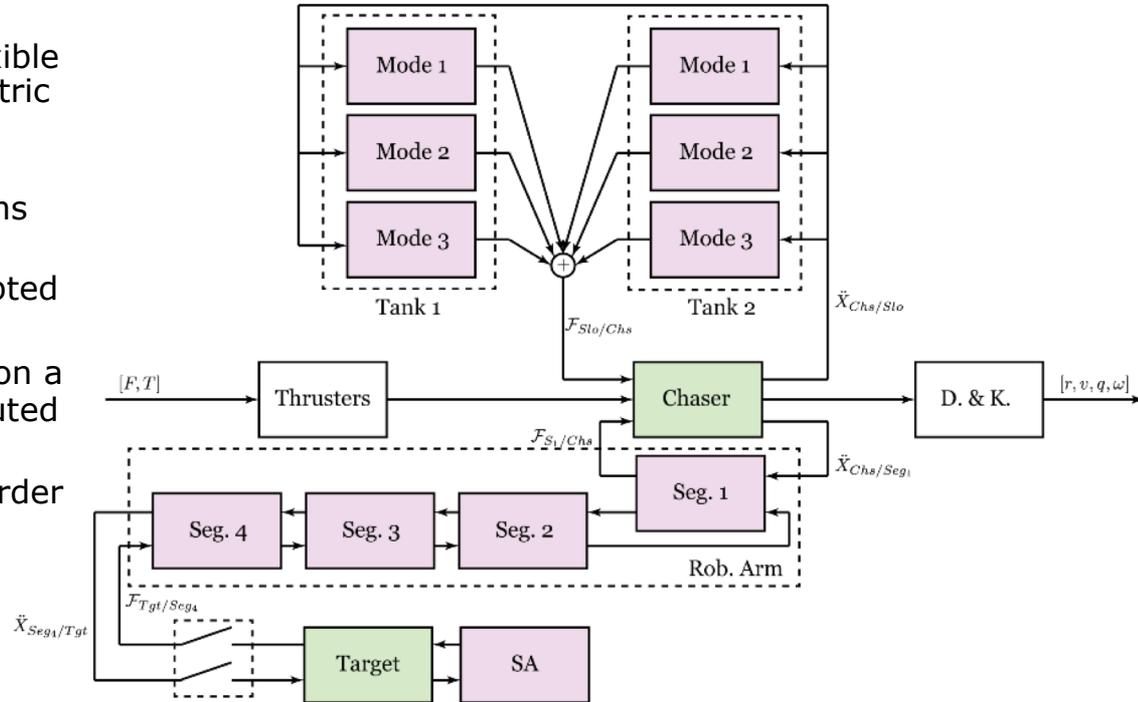
# SYNCHRONISATION



# CONTROL SYNTHESIS & ANALYSIS

## $H_\infty$ synthesis / $\mu$ -analysis:

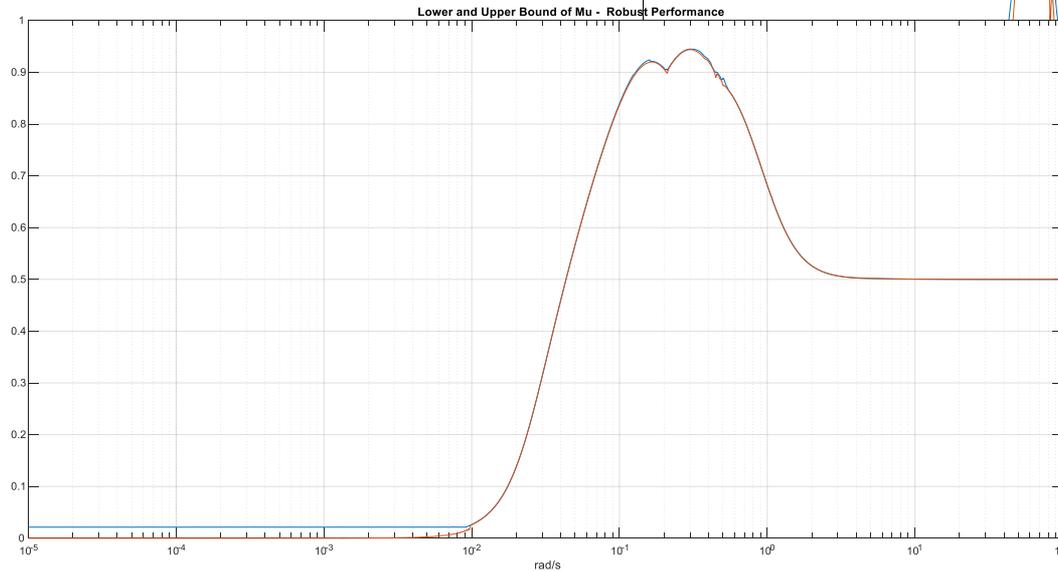
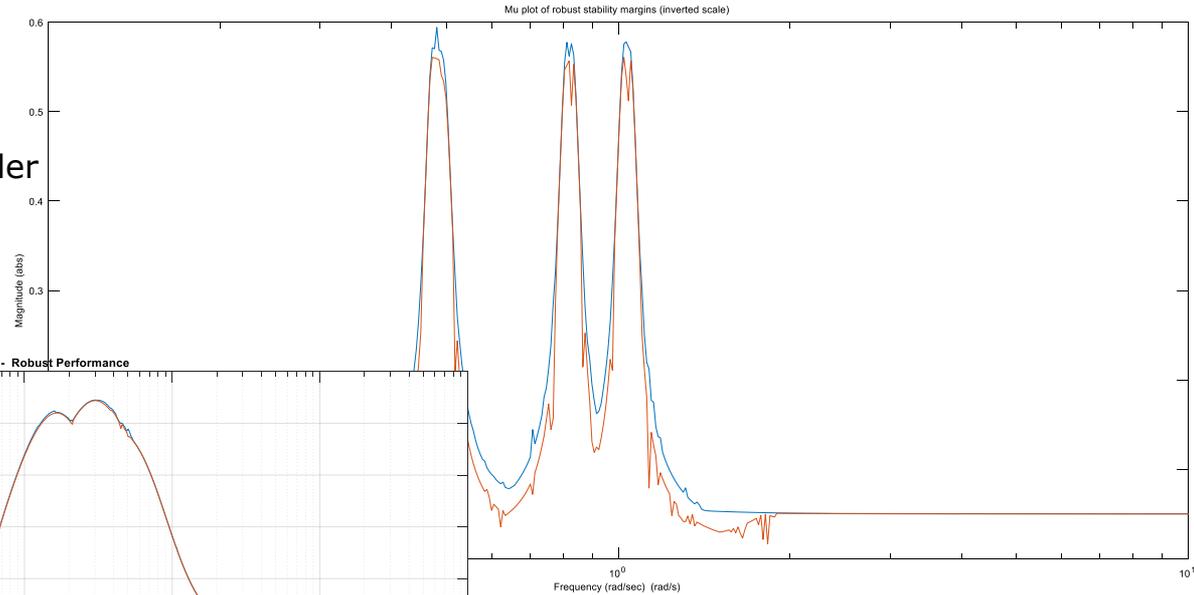
- **LFT** representation of the linearized flexible spacecraft model to account for parametric **uncertainty**.
- **Two-Input Two-Output** Port (TITOP) modelling paradigm for multi-body chains [D. Alazard, J. Alvaro Perez et al.]
- The control synthesis methodology adopted is  $H_\infty$  **Mixed Sensitivity** Design.
  - The  $H_\infty$  control approach is added upon a nonlinear precompensation by computed torque control (feed-forward)
  - Shaping the sensitivity functions in order to achieve robust stability and performance.
  - Requirements are translated into frequency domain weights (of MIMO nature).



# CONTROL SYNTHESIS AND ANALYSIS

## Synchronization phase

**$\mu$ -analysis results for Synchronization  $H_\infty$  controller Robust stability**

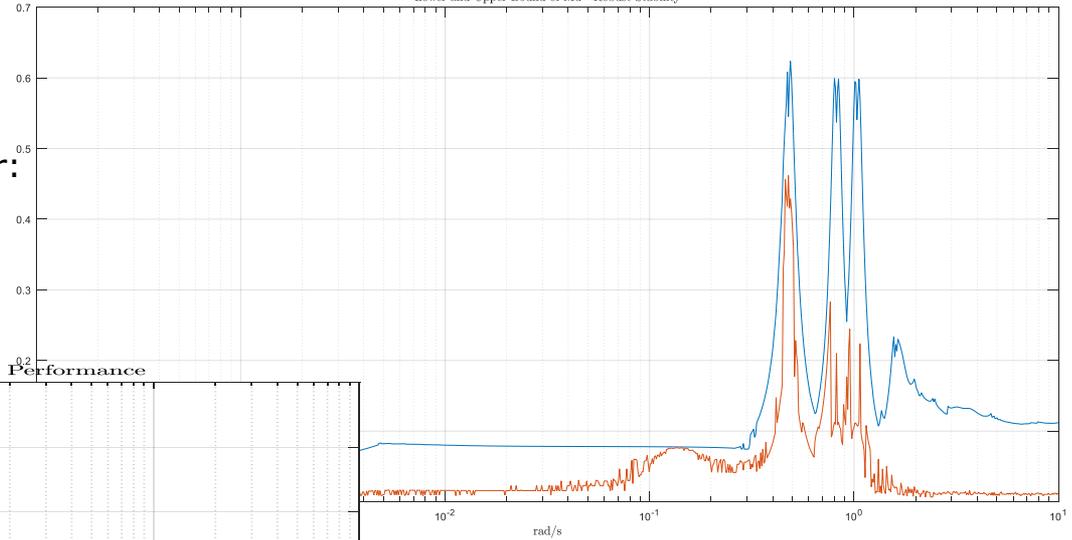


**$\mu$ -analysis results for Synchronization  $H_\infty$  controller Robust performance**

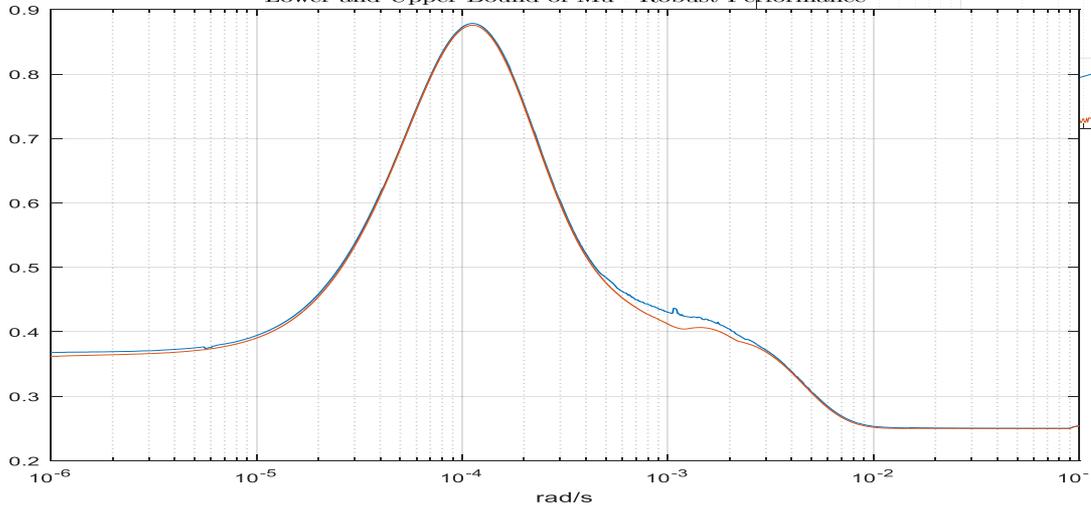
# CONTROL SYNTHESIS AND ANALYSIS

**$\mu$ -analysis** results for  
Stabilization  $H_\infty$  controller:  
Robust stability

Lower and Upper Bound of Mu - Robust Stability



Lower and Upper Bound of Mu - Robust Performance



**$\mu$ -analysis** results for  
Stabilization  $H_\infty$  controller:  
Robust performance

# CONTROL SYNTHESIS AND ANALYSIS

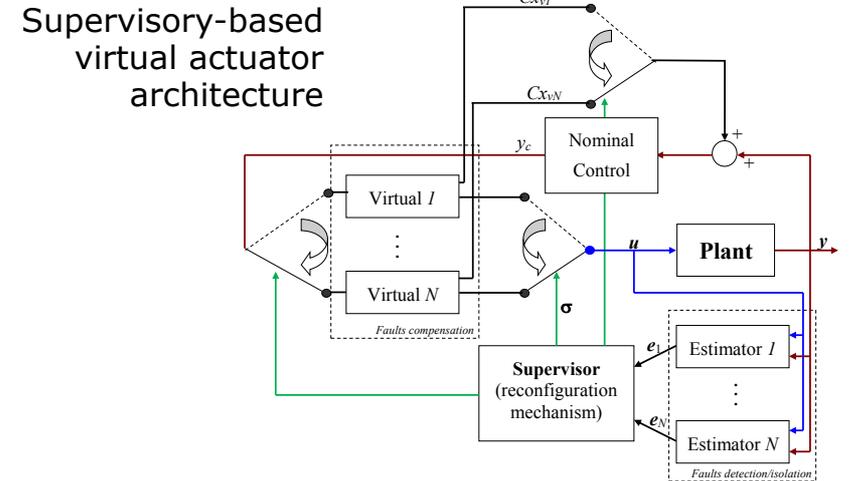
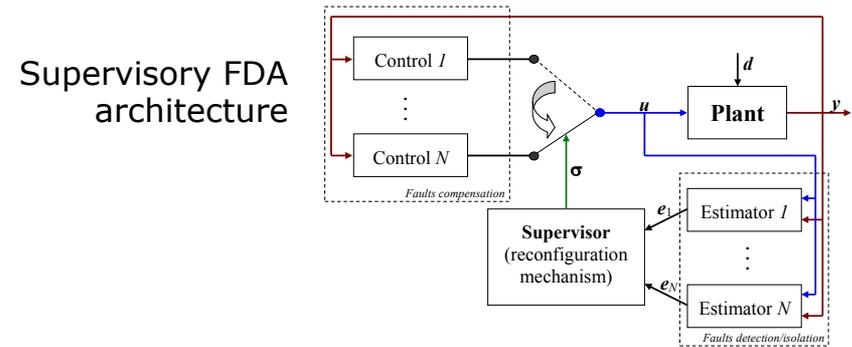
## Non-linear compliance control method:

- Impedance controlled arm is able to follow a **given trajectory** in free motion, and at the same time exhibits a **desired disturbance response** (i.e. impedance) when in contact with the environment.
- Shaping only the **stiffness and damping**, while keeping the inertial behaviour unchanged.
  
- **Reach/Capture** phase:
  - Generalization of **passivity based** compliance.
  - Aiming at a closed loop structure as the one resulting from **PD+ control** in case of fixed base manipulators.
  - **Stability analysis** (an **strict Lyapunov function** for the PD+ control is available in literature, proving asymptotic stability).
  
- **Rigidization phase**
  - Damping of the remaining relative velocity.
  - A PD control with bounded input (saturation effect) has been used.
  - Stability is proved in literature under the condition that the **saturation function** for the PD torque controller must be *a strictly increasing linear saturation function*.

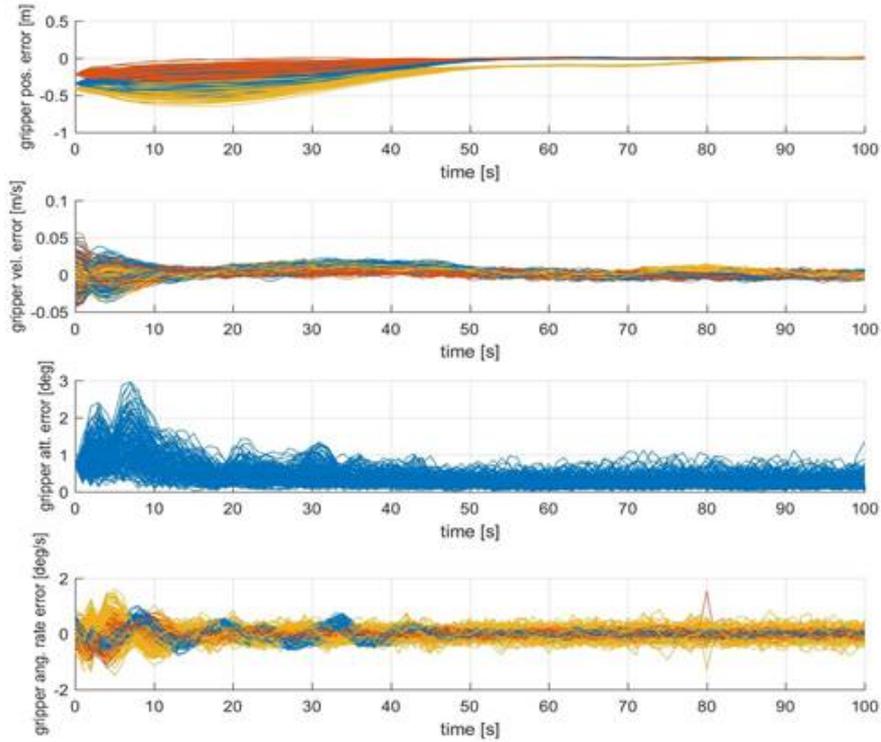
# HEALTH MONITORING SYSTEM

## Failure Tolerant Control (FTC) system:

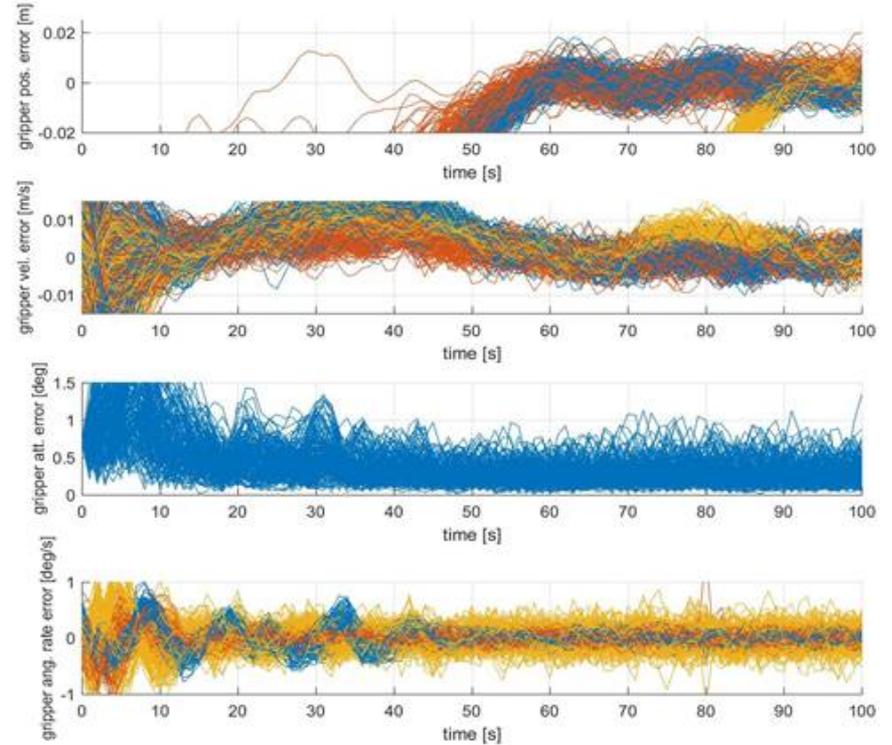
- Failure **Detection and Isolation** (FDI)
  - 4 of the 24 thrusters have been identified as the most problematic from a FDI viewpoint (stuck-open or stuck-closed failures)
  - Bank of 4 dedicated  $H_\infty$  UIOs
- The **Accommodation** of the failure (after isolation) through the use of the system total or partial redundancies
  - The dwell-time supervisory-based FDA solution (recently extended, by IMS Laboratory to the virtual actuator paradigm).
  - Goal: select timely the suitable FTC controller from a bank of virtual actuators.



# RESULTS (MONTECARLO)

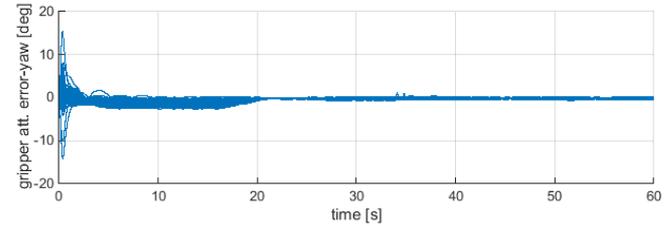
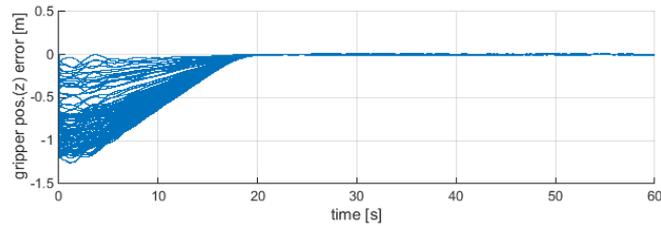
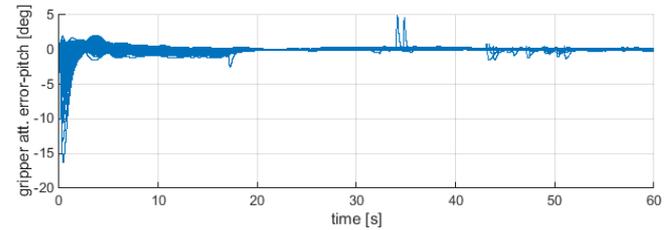
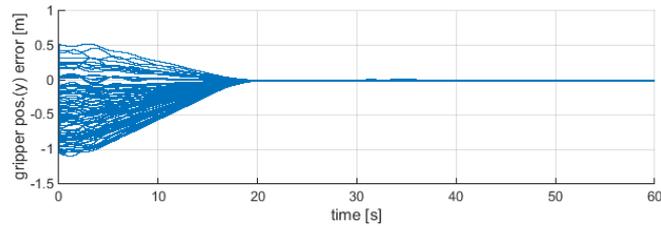
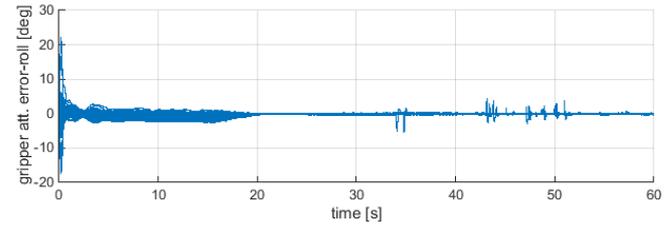
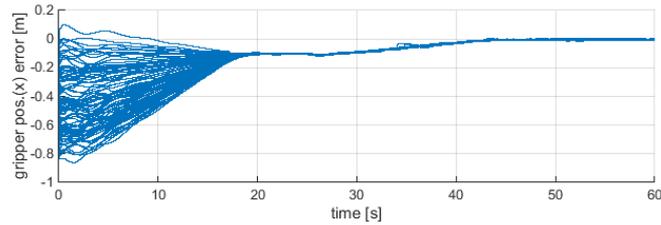


Reach Phase MC, gripper performance (Hinf controller).



Reach Phase MC, gripper performance zoomed (Hinf controller).

# RESULTS (MONTECARLO)



Reach Phase MC, gripper performance. (non-linear compliant controller)

# RESULTS AND CONTROLLERS COMPARISON

**Monte Carlo** test campaign:

- **Synchronisation phase** (only  $H_\infty$  robust control):
  - The obtained Control errors are **within specifications**
- **Reach phase** (both  $H_\infty$  robust controller and non-linear compliant controller):
  - Both controllers behave **similarly**, with the nonlinear compliant controller having tighter tracking in pointing accuracy
- **End-effector performance:**
  - Position and pointing accuracy requirements **met** by both controllers
  - Velocity and angular rate accuracy requirements are **overpassed** by both controllers, with lower error mean value for the  $H_\infty$  control (over Monte Carlo test campaign).

|                      | Synch. phase | Reach phase   |                     |
|----------------------|--------------|---------------|---------------------|
|                      | $H_\infty$   | $H_\infty$    | Nonlinear Compliant |
| Position [m]         | 0.067±0.028  | 0.023± 0.008  | 0.026± 0.011        |
| Velocity [m]         | 0.006±0.002  | 0.002± <0.001 | 0.001± <0.001       |
| Pointing [deg]       | 1.544±0.619  | 0.318± 0.153  | 0.105± 0.077        |
| Angular rate [deg/s] | 0.155±0.089  | 0.075± 0.041  | 0.114± 0.047        |

| End-Effector Performances |   | $H_\infty$   | Nonlinear Compliant |
|---------------------------|---|--------------|---------------------|
| Position [m]              | X | -0.001±0.004 | 0.004±0.003         |
|                           | Y | 0.002±0.005  | 0.005±0.003         |
|                           | Z | -0.001±0.004 | 0.003±0.002         |
| Velocity [m]              | X | 0.001±0.003  | 0.004±0.003         |
|                           | Y | 0.001±0.003  | 0.005±0.003         |
|                           | Z | -0.001±0.002 | 0.005±0.003         |
| Pointing [deg]            | X | -0.016±0.105 | 0.120±0.084         |
|                           | Y | -0.033±0.140 | 0.086±0.054         |
|                           | Z | -0.015±0.293 | 0.182±0.113         |
| Angular rate [deg/s]      | X | 0.011±0.073  | 0.520±0.335         |
|                           | Y | -0.004±0.098 | 0.286±0.217         |
|                           | Z | 0.029±0.191  | 0.434±0.314         |

# RESULTS AND CONTROLLERS COMPARISON

## ■ Rigidization phase:

- The achieved **joint position error** for the  $H_\infty$  robust controller is 60% of the one for the compliant controller case.
- For **joint velocity**, the error achieved for the  $H_\infty$  robust controller is 21%, of the one for the compliant controller case
- Probably, better results can be obtained (future work) for the non-linear compliant control by a more adjusted tuning of the position gains in case a specific requirement for the joint positions is given.

| Rigidization  | $H_\infty$                                     | Nonlinear Compliant                               |
|---|--|---|
| Angle [deg]   | 1.618±0.809                                    | 2.710±1.169                                       |
| Angular rate [deg/s]  | 0.016±0.013                                    | 0.078±0.030                                       |
| Initial angular velocity [deg/s]  | 0.14±0.20(1 $\sigma$ )<br>(Max=3.02)           | (1,1,1,1,1,1,1)                                   |
| Maximum torques around the actuation axis (z) for all the 7 joints [Nm] | (3.87, 12.23, 3.42, 29.80, 12.28, 8.11, 28.10) | (10.94, 12.89, 19.83, 50.10, 17.16, 23.32, 41,45) |
| Simulation time[s]  | 120 s  | 120 s   |

## ■ Stabilization phase:

- Requirements (ENVISAT case) are comfortably met

| Stabilization        |   | $H_\infty$ |
|----------------------|---|------------|
| Angular rate [deg/s] | X | -0.023     |
|                      | Y | 0.006      |
|                      | Z | 0.013      |

# CONCLUSIONS

COMRADE project has currently finalized the Model-in-the-Loop (MIL) level validation phase with successful results and will now enter into the Processor-in-the-Loop (PIL) and HW-in-the-Loop (HIL) validation level.

From MIL-based design/validation phase:

- **Approach/synchronization phase** has considered robust **H $\infty$  6DOF** controller over a rigid body with sloshing and flexibility (solar arrays, stored robotic manipulator) effects as main perturbations.
- **Reach, capture and rigidization phase** has considered a dual approach and implementation (both controllers have demonstrated to be valid options with some better performance results obtained for the first one):
  - **Robust H $\infty$  13DOF** controller over a multi-body system composed by the spacecraft platform plus a robotic manipulator with 7DOF (and grasping/re-fueling end-effect at the end).
  - A **compliance/impedance 13DOF** controller over the same multi-body system as for the robust H $\infty$  controller.
- **Stabilization/detumbling** phase has considered robust **H $\infty$  3DOF attitude controller over the full composite** (chaser S/C + target S/C + rigidized robotic manipulator joining both vehicles) with sloshing and flexibility effects as main perturbations.
- Advanced **FDA/FTC techniques** have been also considered as an additional Failure Detection and Accommodation layer on top of the nominal control design.



THANK YOU