

Using Geant4 to interpret data from MAVEN's Solar Energetic Particle (SEP) detector

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- Presented material includes contributions from the entire SEP team:
 - Davin Larson, Instrument Lead
 - Robert Lillis, Scientist
 - Christina Lee, Scientist
 - Ali Rahmati, Scientist
- All located at Space Sciences Lab, University of California, Berkeley*

1. Mars Atmosphere and Volatile Evolution (MAVEN) Mission
2. Solar Energetic Particle (SEP) Instrument
3. Geant4 Simulations for SEP

Mars Atmosphere and Volatile Evolution

Three major science objectives:

- 1) determine the ***current state of the upper atmosphere*** and the processes that control it
- 2) measure the ***current escape rate*** of gases and to space and determine how the escape rate depends on the controlling processes
- 3) extrapolate to the ***total atmospheric loss to space over Mars' history***



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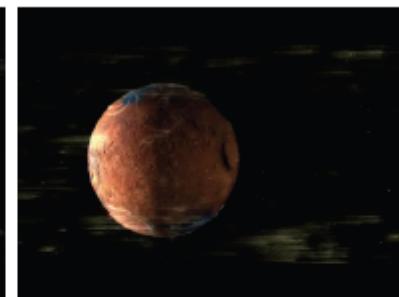
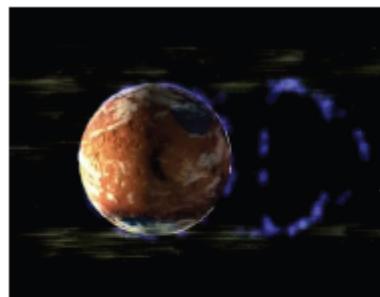
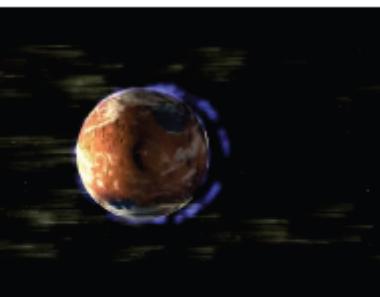
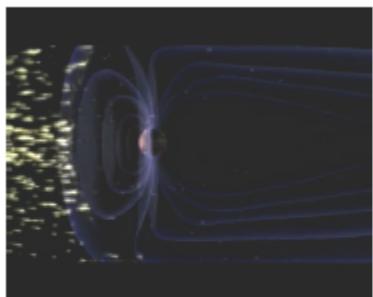


Ancient Valleys

Mars' atmosphere is cold and dry today, but there was once liquid water flowing over the surface.

Where did the water and early atmosphere go?

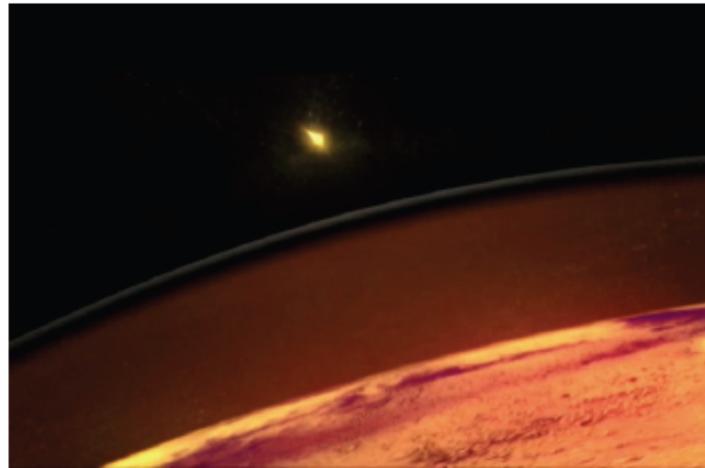
- H_2O and CO_2 can go into the crust or be lost to space.
- MAVEN will focus on volatile loss to space.



Turn-off of the Martian magnetic field allowed turn-on of solar-wind stripping of the atmosphere ~ 3.7 billion years ago; combined with solar-EUV-driven loss, resulted in the present thin, cold atmosphere.



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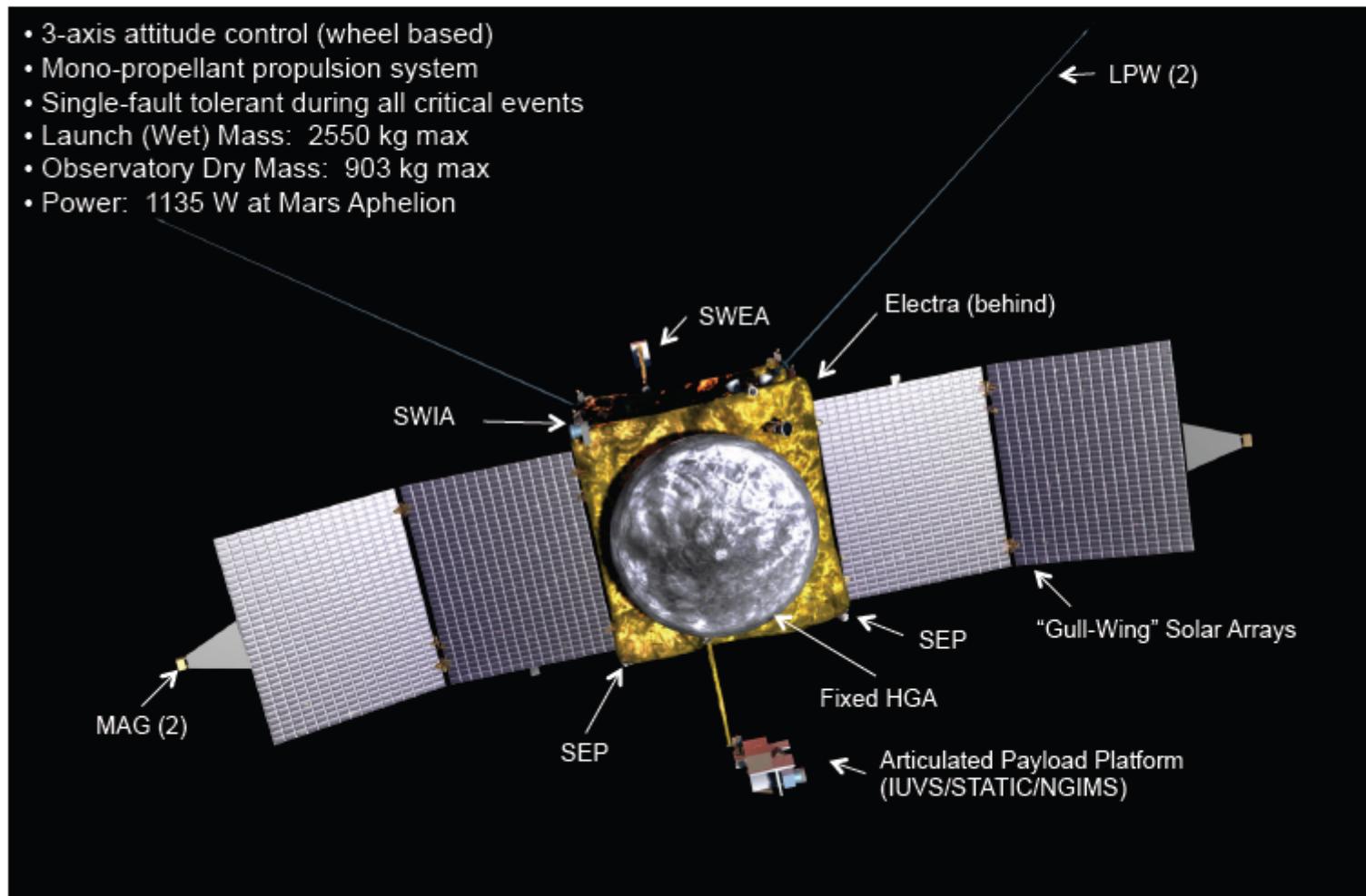
- Determine the structure and composition of the Martian upper atmosphere today
- Determine rates of loss of gas to space today
- Measure properties and processes that will allow us to determine the integrated loss to space through time

MAVEN will answer questions about the history of Martian volatiles and atmosphere and help us to understand the nature of planetary habitability.



The MAVEN Spacecraft

- 3-axis attitude control (wheel based)
- Mono-propellant propulsion system
- Single-fault tolerant during all critical events
- Launch (Wet) Mass: 2550 kg max
- Observatory Dry Mass: 903 kg max
- Power: 1135 W at Mars Aphelion





The MAVEN Science Instruments

Mass Spectrometry Instrument



*Neutral Gas and Ion Mass Spectrometer;
Paul Mahaffy, GSFC*

Particles and Fields Package



*Solar Energetic Particles;
Davin Larson, SSL*

*SupraThermal and Thermal Ion Composition; Jim McFadden,
SSL*

Remote-Sensing Package



Imaging Ultraviolet Spectrometer; Nick Schneider, LASP



*Solar Wind Electron Analyzer;
David Mitchell, SSL*

*Solar Wind Ion Analyzer;
Jasper Halekas, SSL*



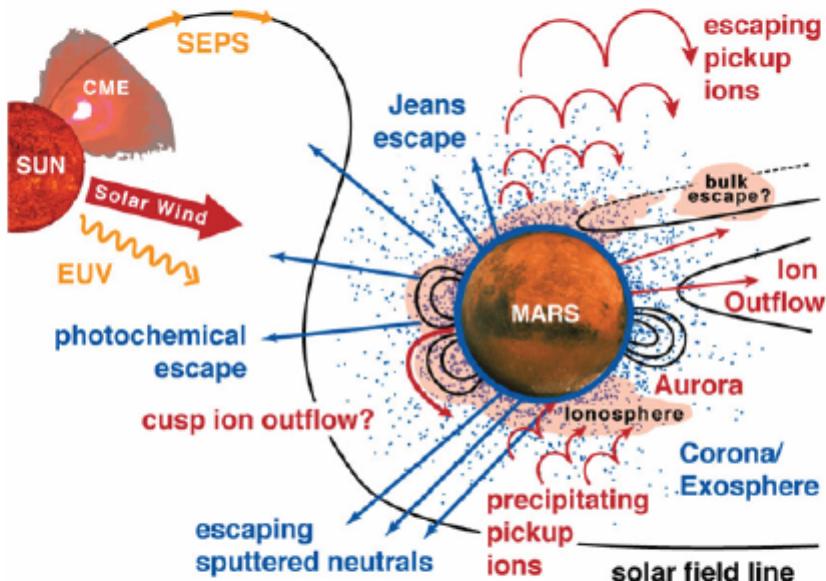
*Langmuir Probe and Waves;
Bob Ergun, LASP*

*Magnetometer;
Jack Connerney, GSFC*

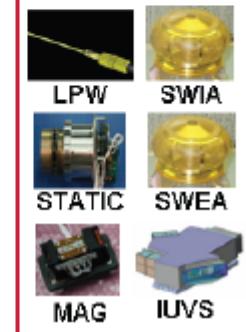


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MAVEN IS MEASURING THE DRIVERS, RESERVOIRS, AND ESCAPE RATES



Plasma Processes



Neutral Processes



- MAVEN will determine the present state of the upper atmosphere and today's rates of loss to space.
- Measurements will allow determination of the net integrated loss to space through time.

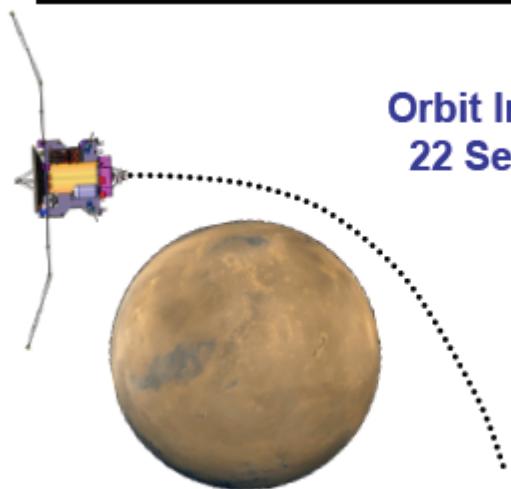
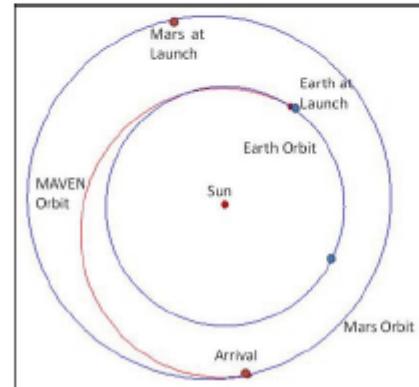


MAVEN Mission Architecture



20-Day Launch Period:
November 18 –
December 7, 2013

Ten Month Ballistic Cruise to Mars

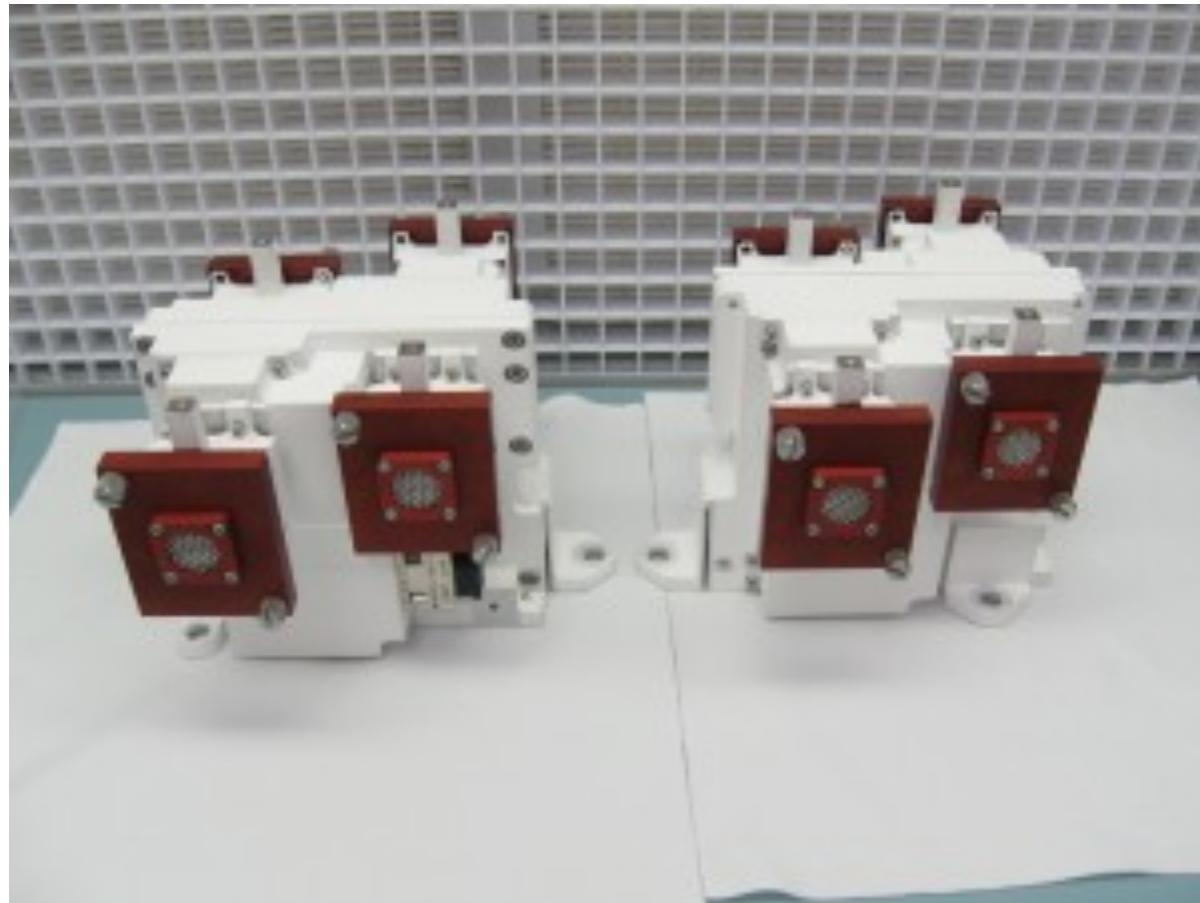


Orbit Insertion:
22 Sept 2014

Just completed 4th Earth year since MOI



Solar Energetic Particle (SEP) Instrument



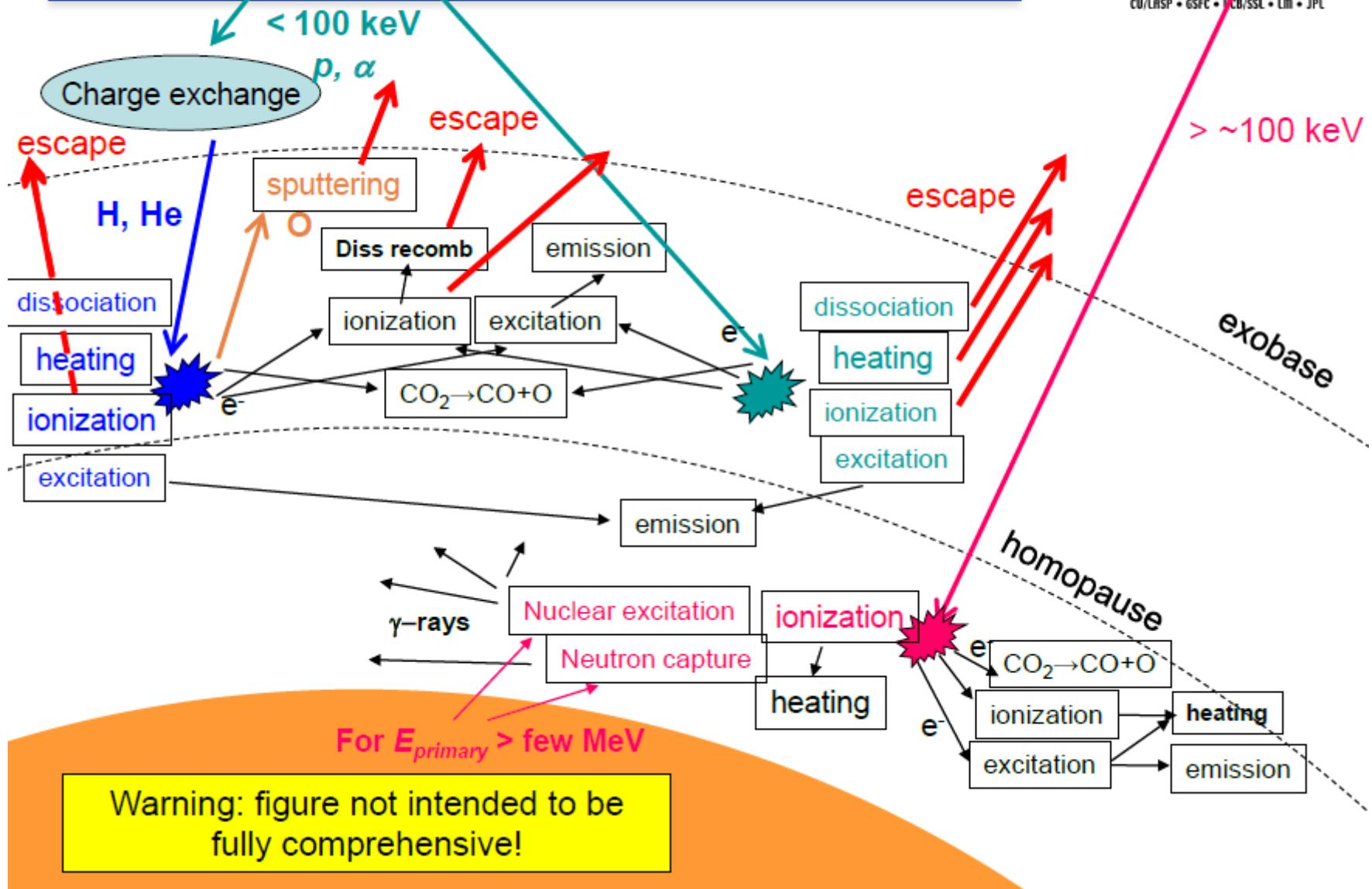
Measurements will help address important SEP-related questions



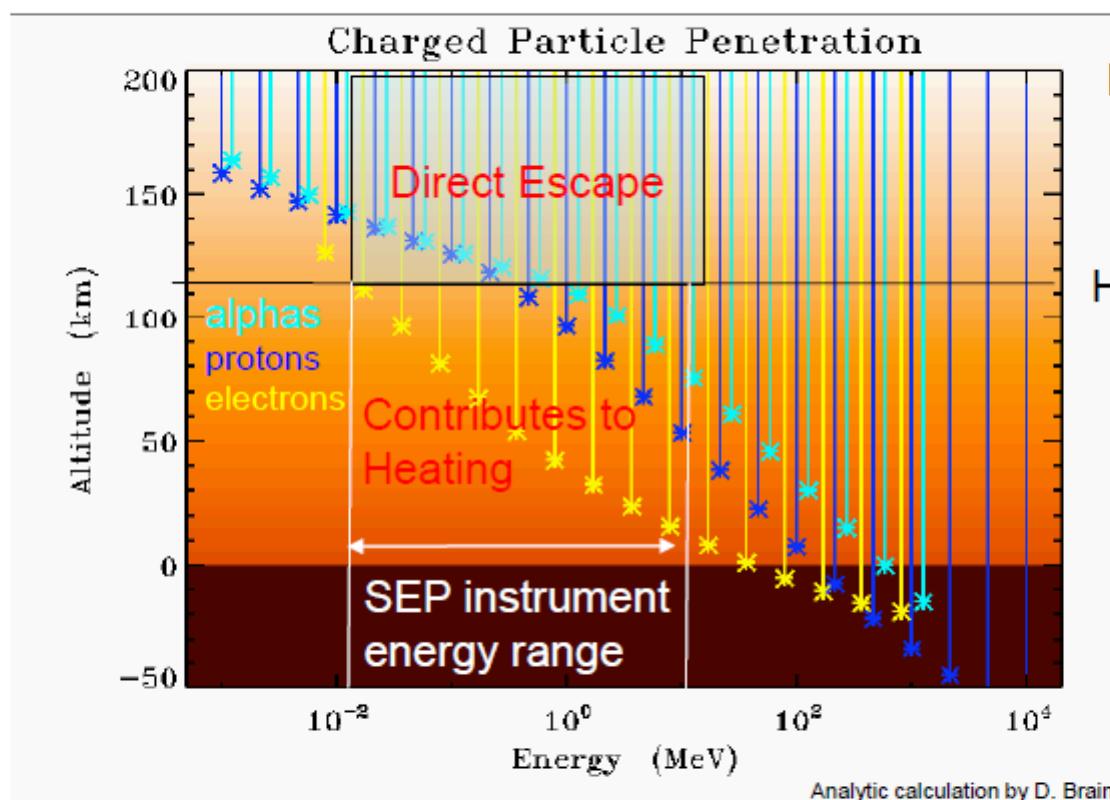
- 1) How & to what degree is the Mars atmosphere shielded
 - by the planetary-scale magnetosphere?
 - by the crustal magnetic fields?

- 2) What are the ways in which incident SEPs of various energies affect the atmosphere?
 - SEP spectrum degradation with altitude?
 - Sputtering by energetic neutrals created via charge exchange?
 - Bulk heating from SEP collisions with neutrals ?
 - SEP impact ionization → dissociative recombination → escape
 - Electronic excitation → auroral emission?
 - Molecular dissociation → effect on atmospheric chemistry?
 - Ionospheric currents?

SEPs in the atmosphere: energy partitioning is complicated and energy dependent



SEP instrument will measure particles that penetrate to altitudes important for escape processes.

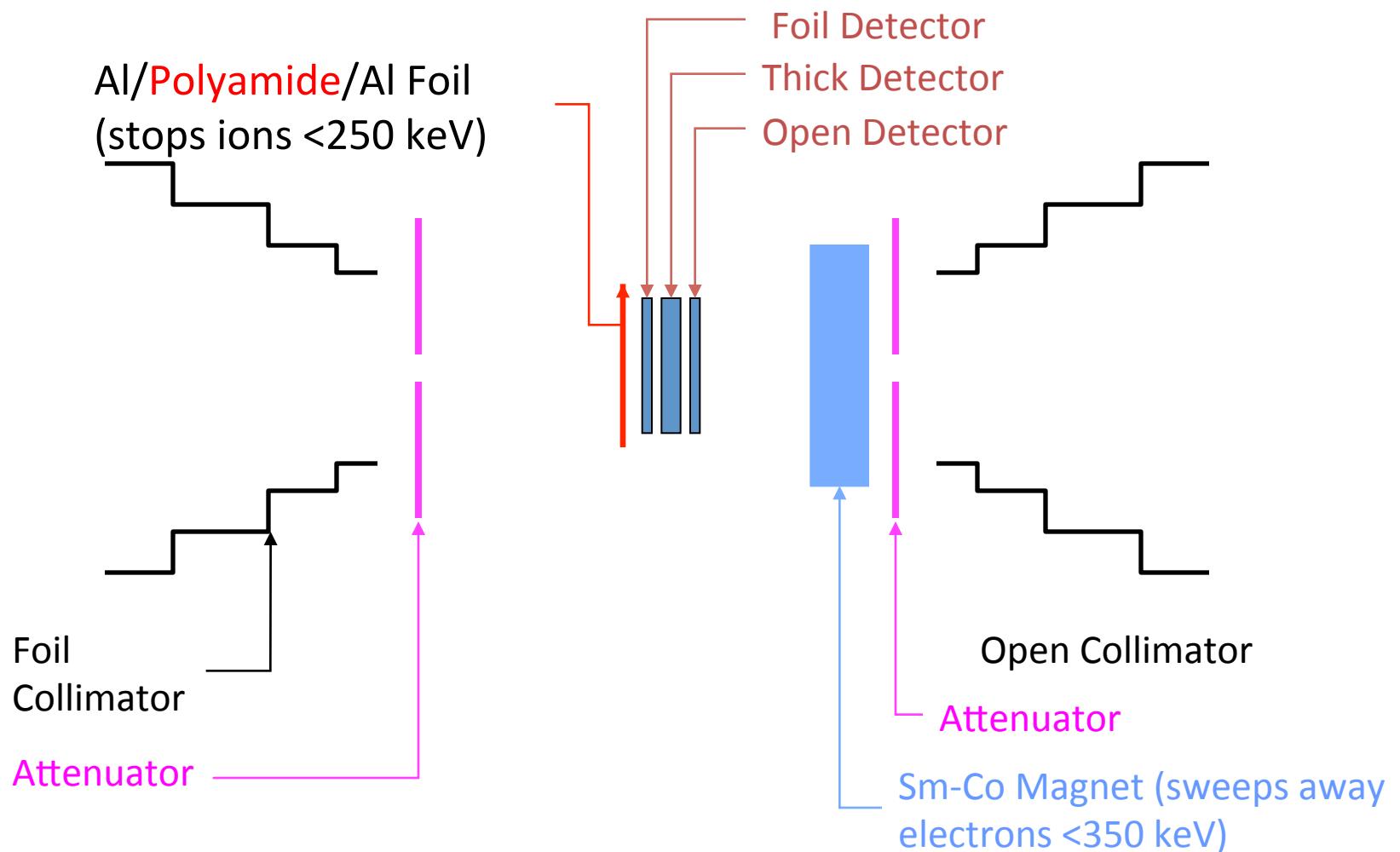


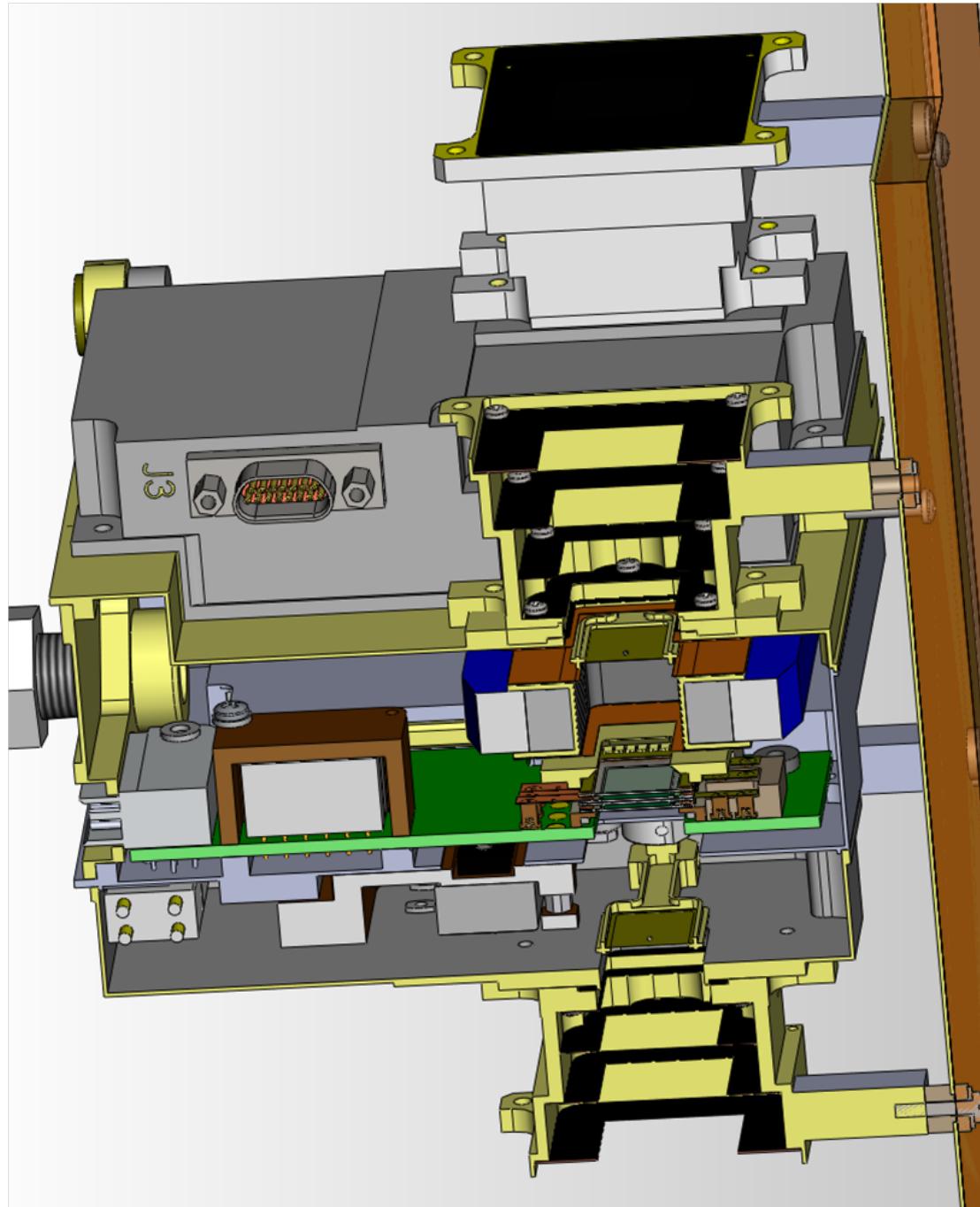
Exobase

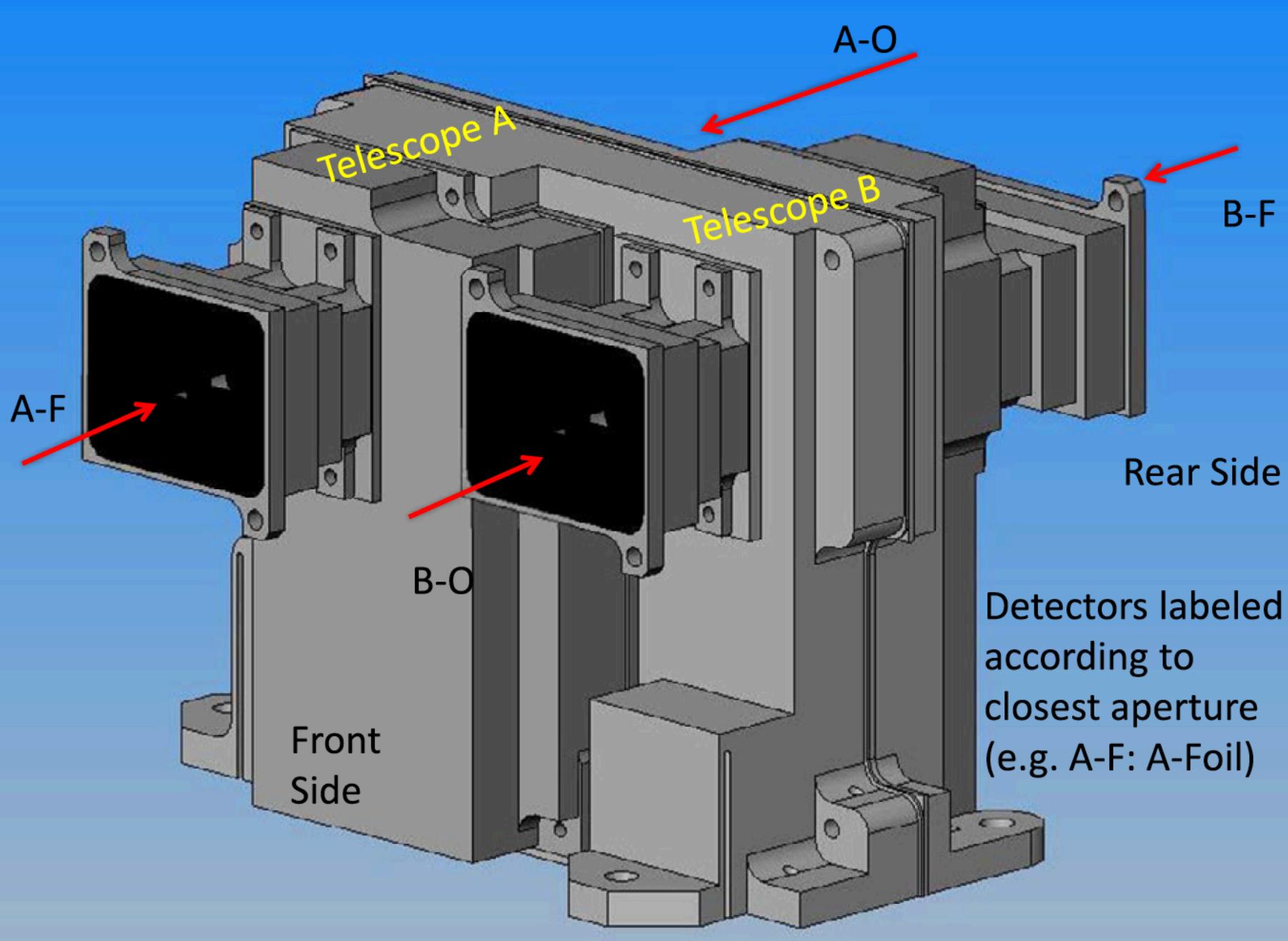
Homopause

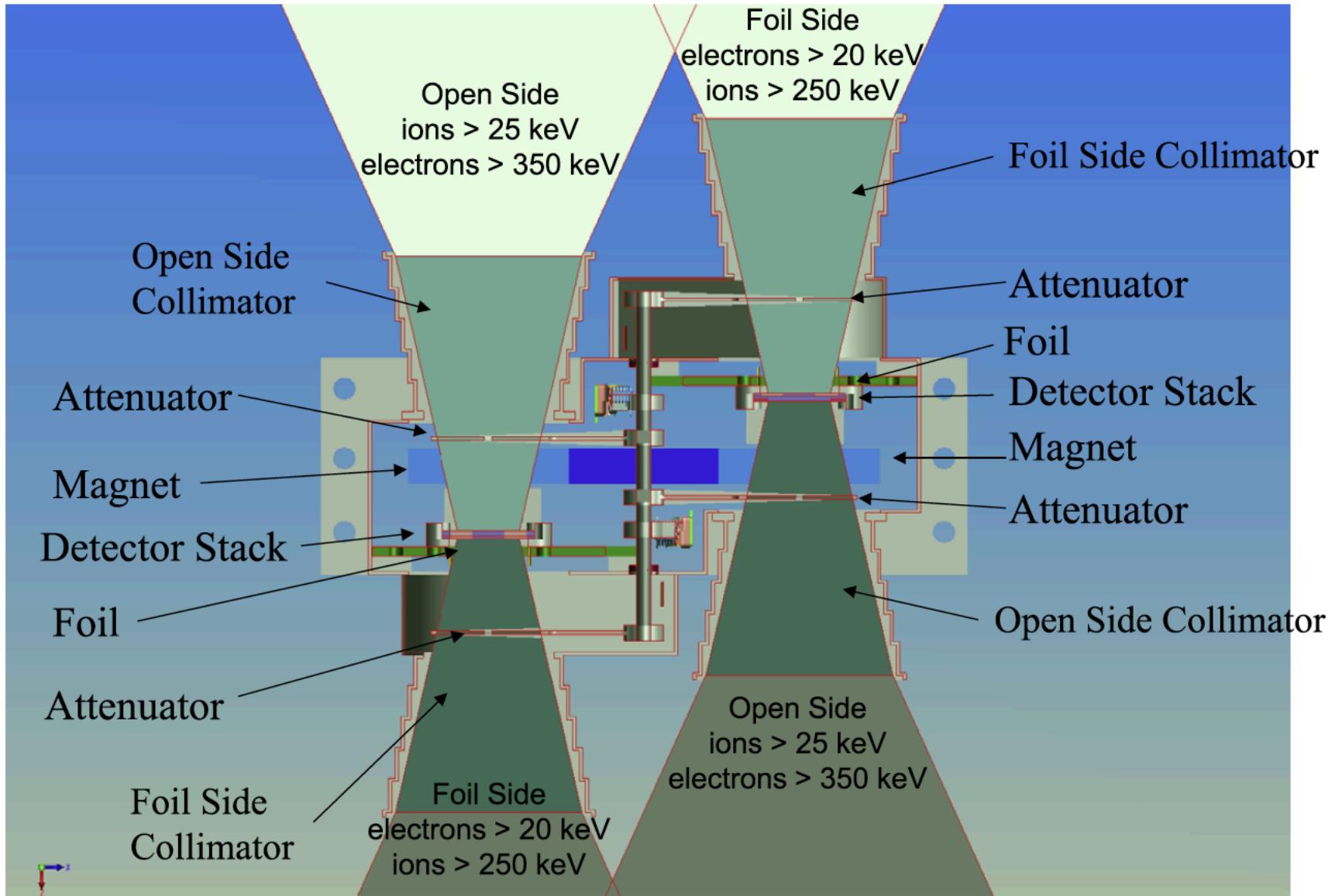
- The bulk of SEP event total energy is generally below 50 keV, deposited mostly between 100 km and 130 km [LeBlanc et al., 2002], though events widely vary.
- We will measure particles that penetrate to 50 km-150 km, providing important constraints on modeling of atmosphere/ionosphere dynamics.

Basic design of SEP instrument









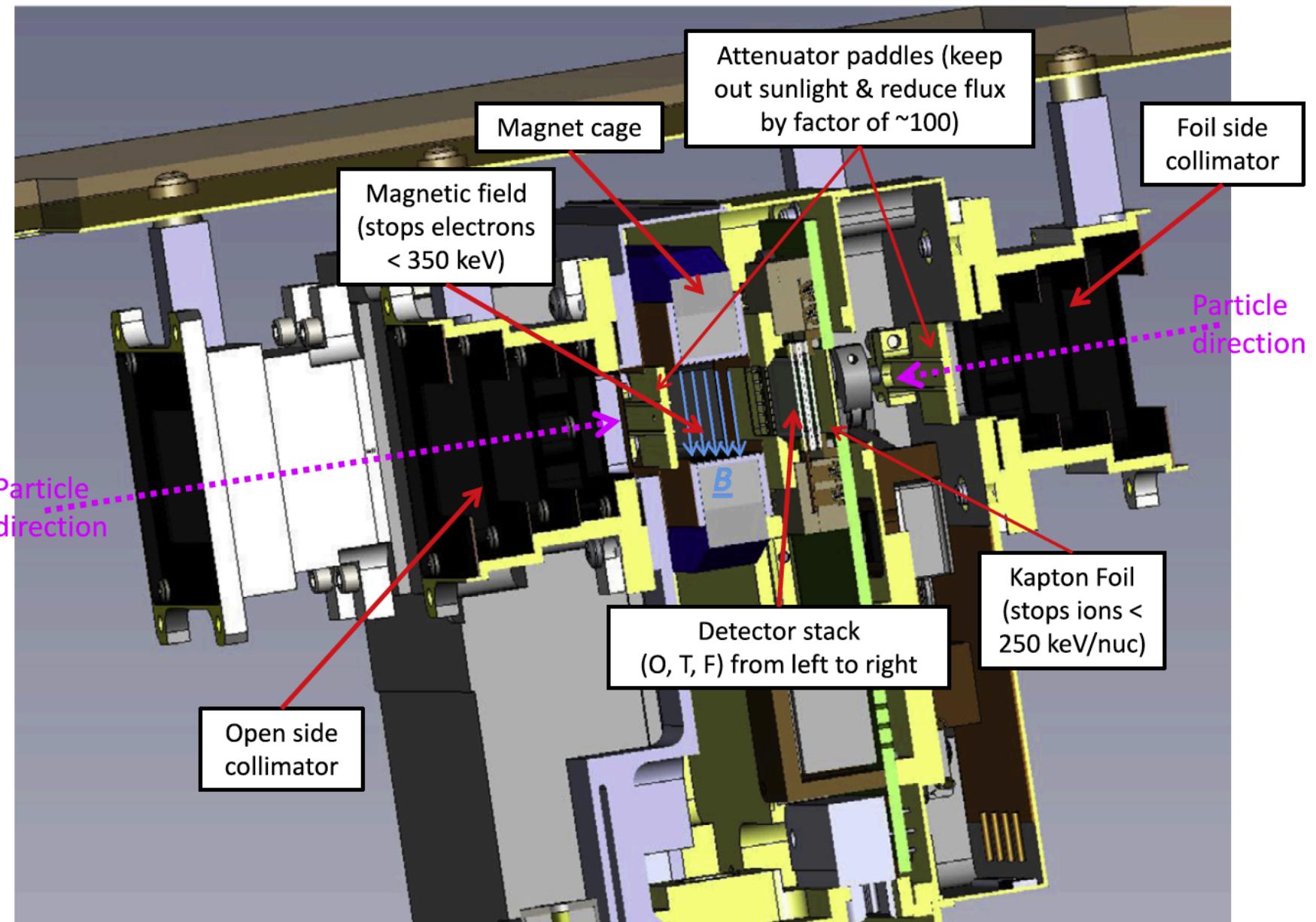
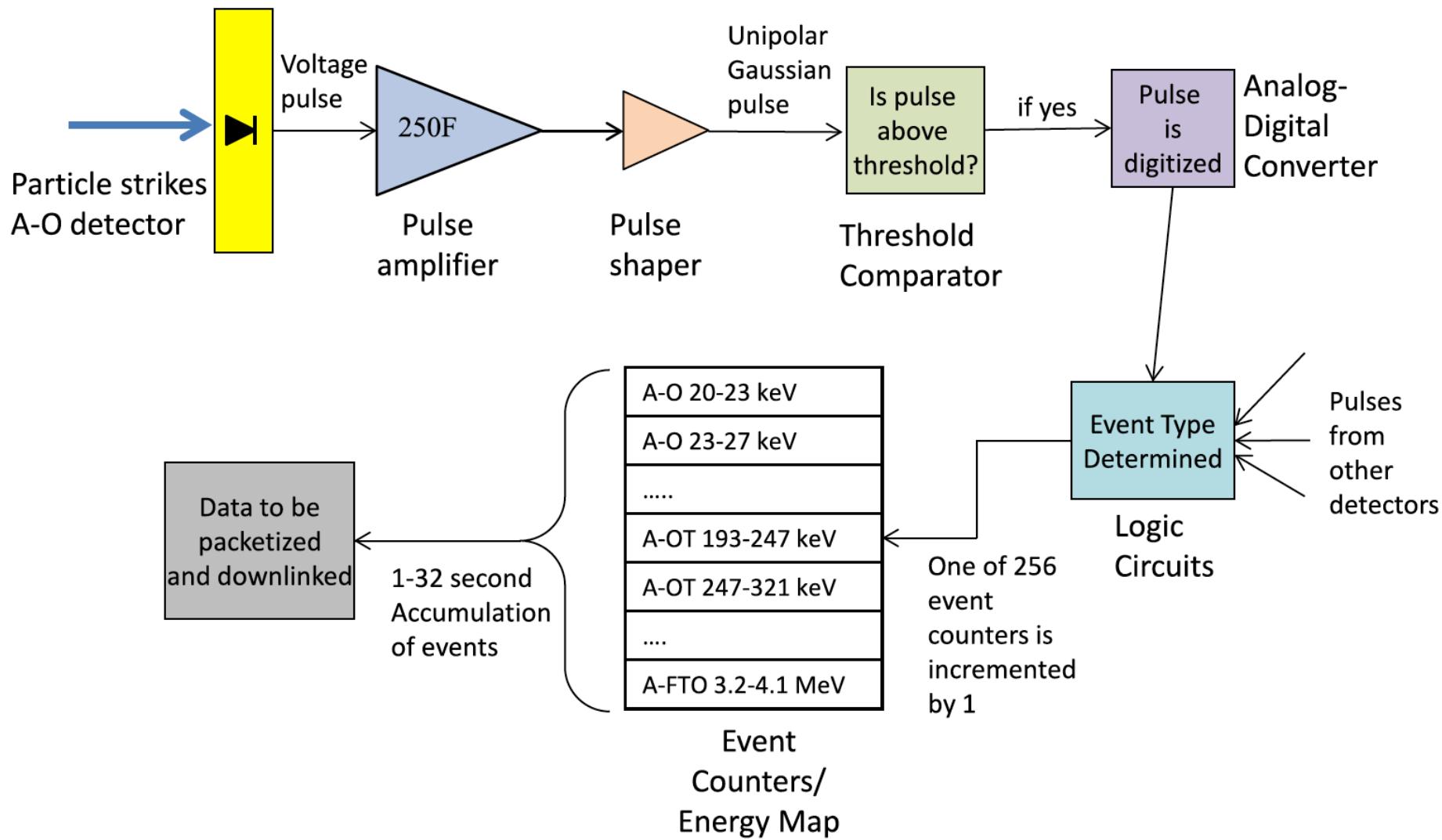


Table 1 Comparison of the SEP Level 1 requirements versus the instrument performance

Requirement (ions only)	MAVEN Level 1 requirement	SEP instrument performance
Energy range	50 keV to 5 MeV	20 keV to 6 MeV
Energy resolution $\Delta E/E$	50 %	< 25 % (better at lowest energy)
Energy flux range	10 to 10^6 eV/[cm ² s sr eV]	3 to 3×10^6 eV/[cm ² s sr eV]
Energy flux precision	30 %	< 10 % (based on modelling)
Time cadence	1 hour	1–32 seconds (mode-dependent)

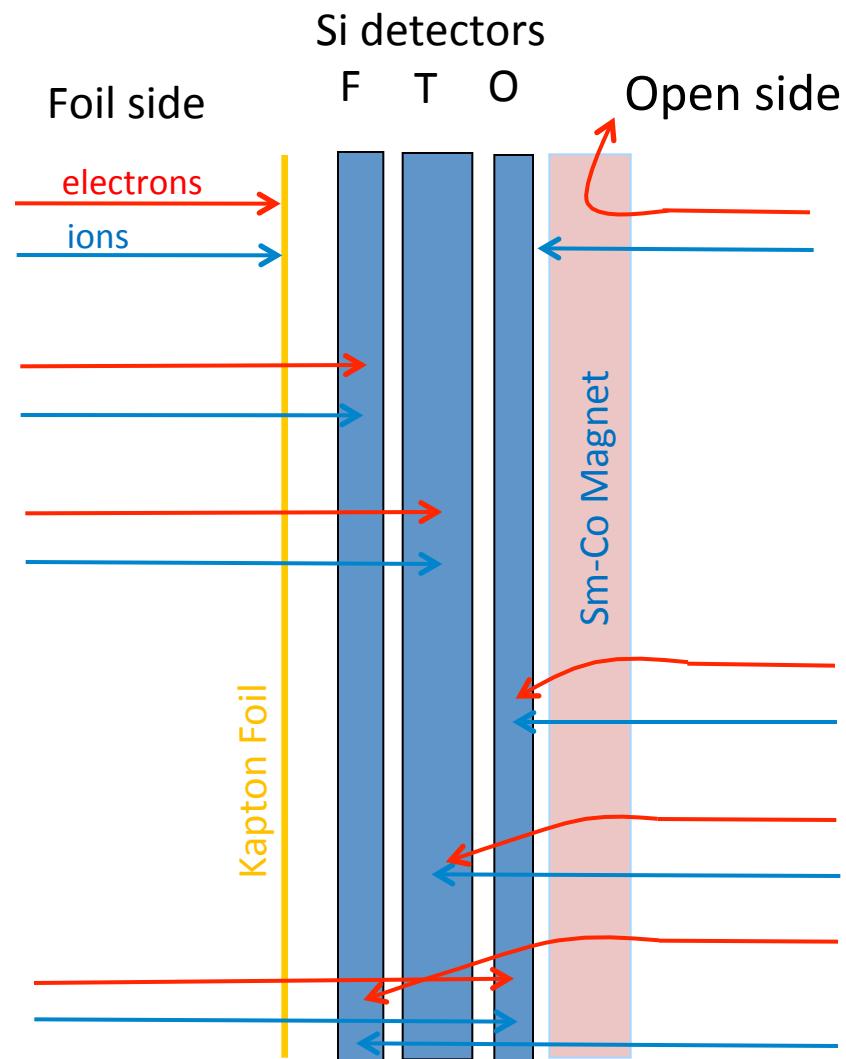
SCHEMATIC DESCRIPTION OF SEP SIGNAL PROCESSING



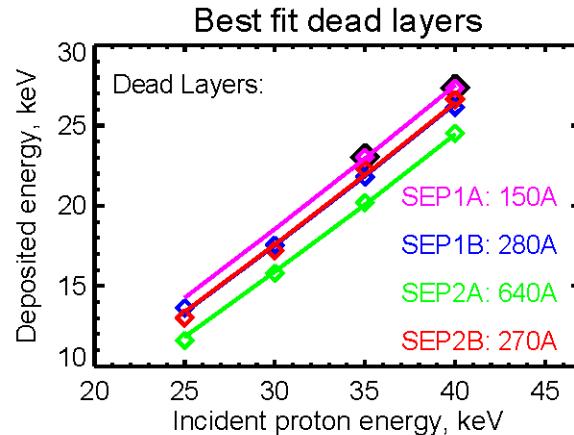
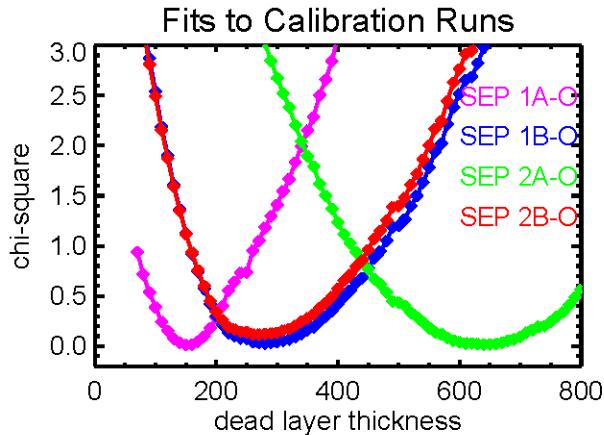
Basic separation strategy: 3 detectors, 2 filters

Energy ranges for counted events

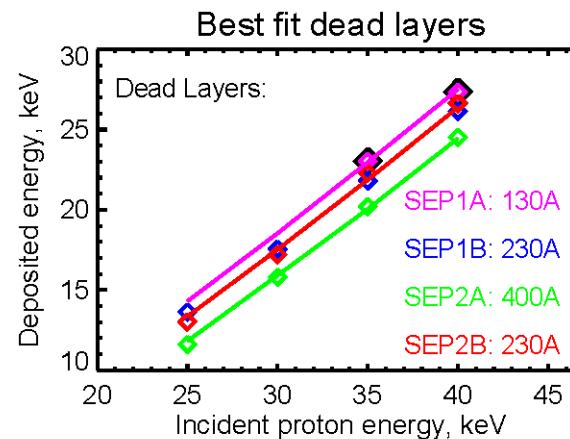
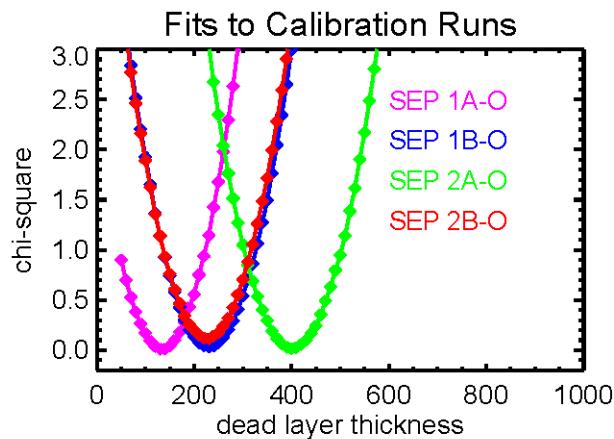
	Electrons		Ions	
keV	Foil side	Open side	Foil side	Open side
No count	<20	<350	<250	<15
F	20-700	X	250-6000	X
FT	350-1300	X	6000-11,00 0	X
O	X	350-700	X	15-6000
OT	X	350-1300	X	6000-11,00 0
FTO	300-2000		>11,000	



DETERMINING THICKNESS OF DETECTOR “DEAD” LAYERS

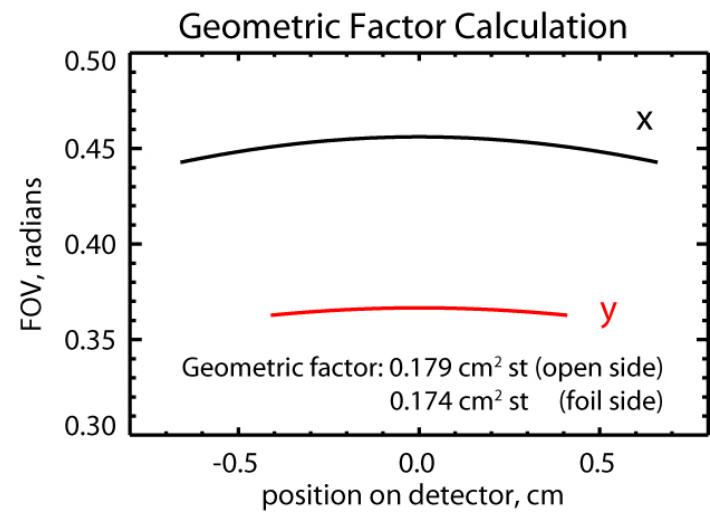
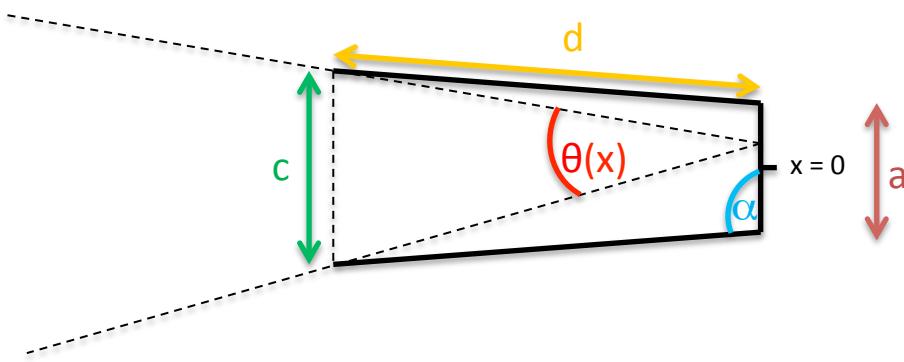


*Oops!
Misplaced
volume.*



*OK, much
better! ☺*

GEOMETRIC FACTOR CALCULATED GEOMETRICALLY



- Rectangular field of view: consider x,y separately
- Determine opening angles $\theta(x),\phi(y)$ as a function of position on the detector.
- Integrate across the detector in x and y .
- Geometric factors:
 - $0.179 \text{ cm}^2 \text{ st}$ (open side)
 - $0.174 \text{ cm}^2 \text{ st}$ (foil side)
- GEANT4 simulations on following slide.

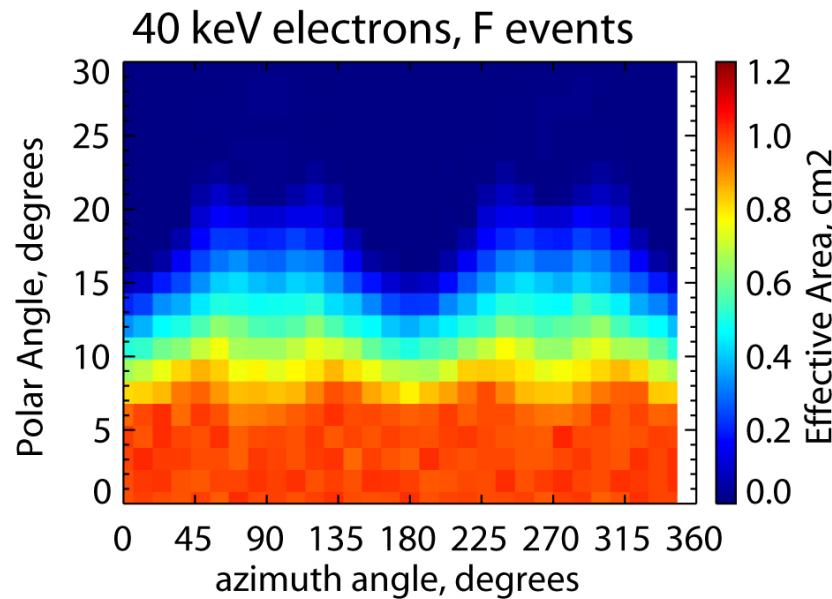
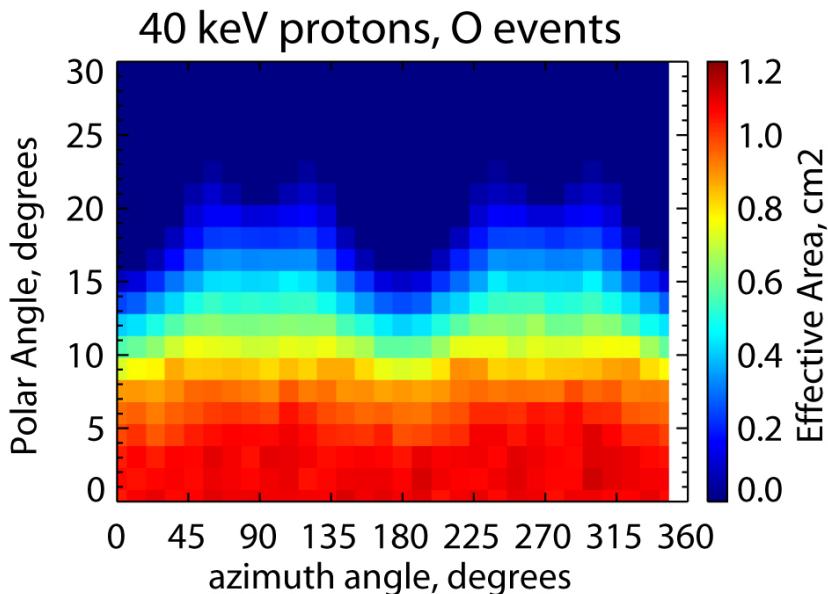
$$p_1^2 = d^2 + \left(\frac{a}{2} + x \right)^2 - 2d\left(\frac{a}{2} + x \right)\cos\alpha$$

$$p_2^2 = d^2 + \left(\frac{a}{2} - x \right)^2 - 2d\left(\frac{a}{2} - x \right)\cos\alpha$$

$$\theta(x) = \cos^{-1}\left(\frac{p_1^2 + p_2^2 - c^2}{2p_1p_2} \right)$$

$$GF = \int_{-b/2}^{b/2} \int_{-a/2}^{a/2} \theta(x)\phi(y)dxdy$$

GEOMETRIC FACTOR CALCULATED WITH GEANT4 SIMULATIONS

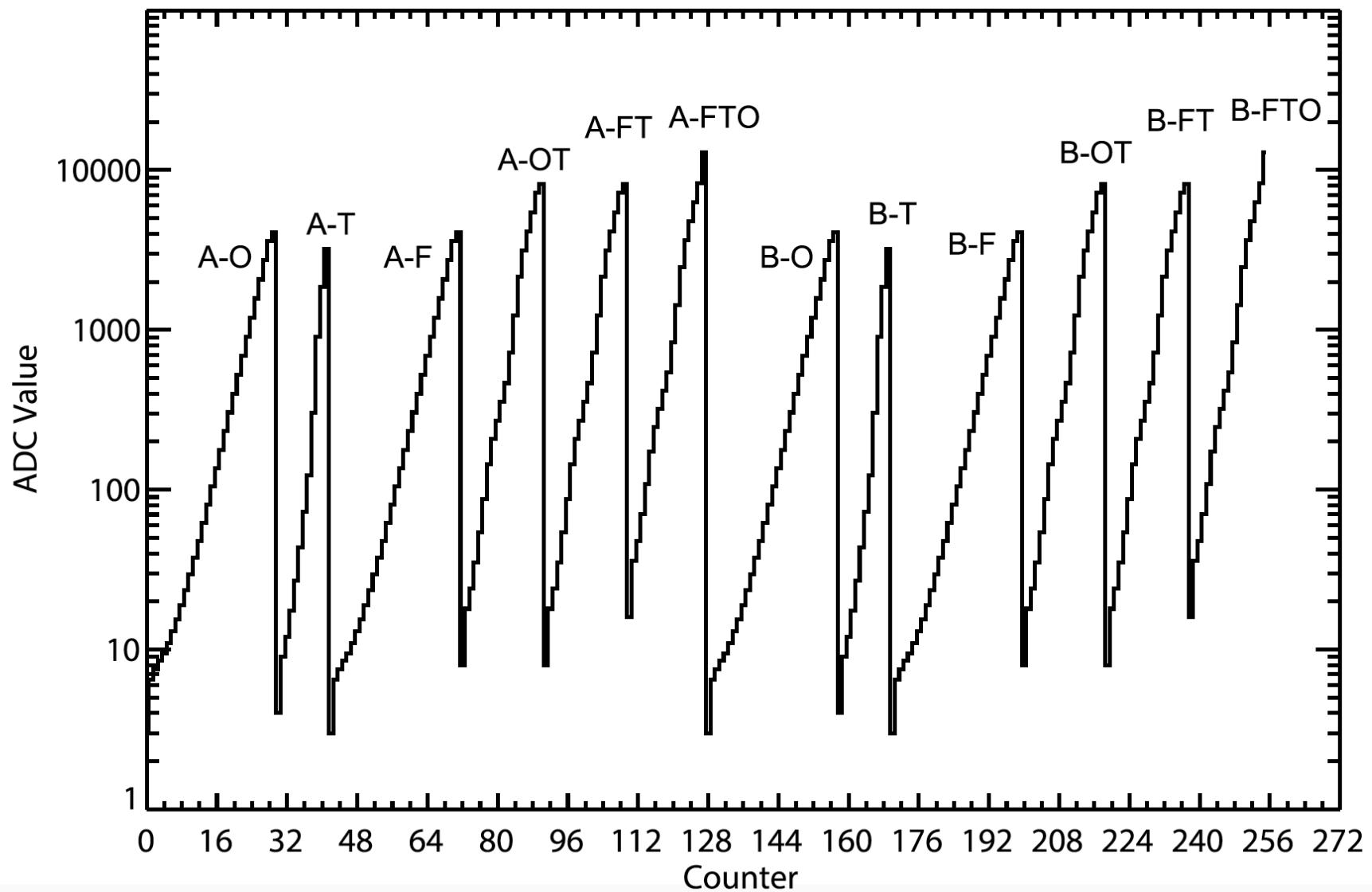


- 40 keV electrons. Foil side.
- Beam Area : 7.07 cm^2
- # particles per angle: 20,000
- $GF = 0.1689 \pm .0025 \text{ cm}^2 \text{ sr}$
- 1.47% error from counting statistics
- 2.87% lower than simple geometric value of $0.174 \text{ cm}^2 \text{ sr}$ (see previous slide)

- 40 keV protons. Open side.
- Beam Area : 7.07 cm^2
- # particles per angle: 20,000
- $GF = 0.1769 \pm .0027 \text{ cm}^2 \text{ st}$
- 1.54% error from counting statistics
- 1.17 % lower than simple geometric value of $0.179 \text{ cm}^2 \text{ sr}$ (see previous slide).

$$GF = \frac{A_{beam}}{N_{incident}} \int_0^{2\pi} d\phi \int_0^{\pi/2} N_{detected}(\phi, \theta) \sin \theta d\theta$$

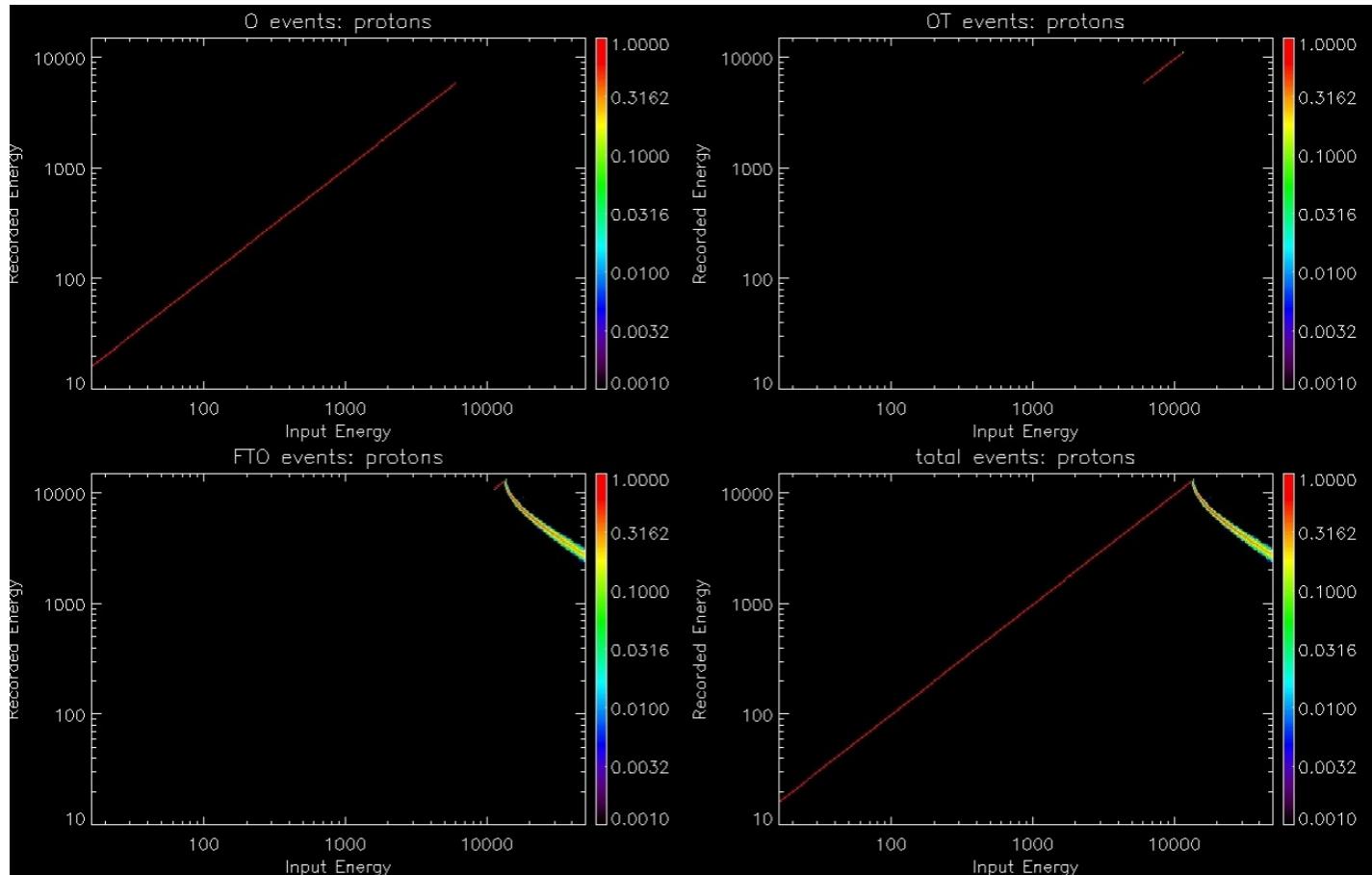
SEP ENERGY MAP



DETECTOR RESPONSE TO NORMALLY INCIDENT BEAM JUST ABOVE APERTURES

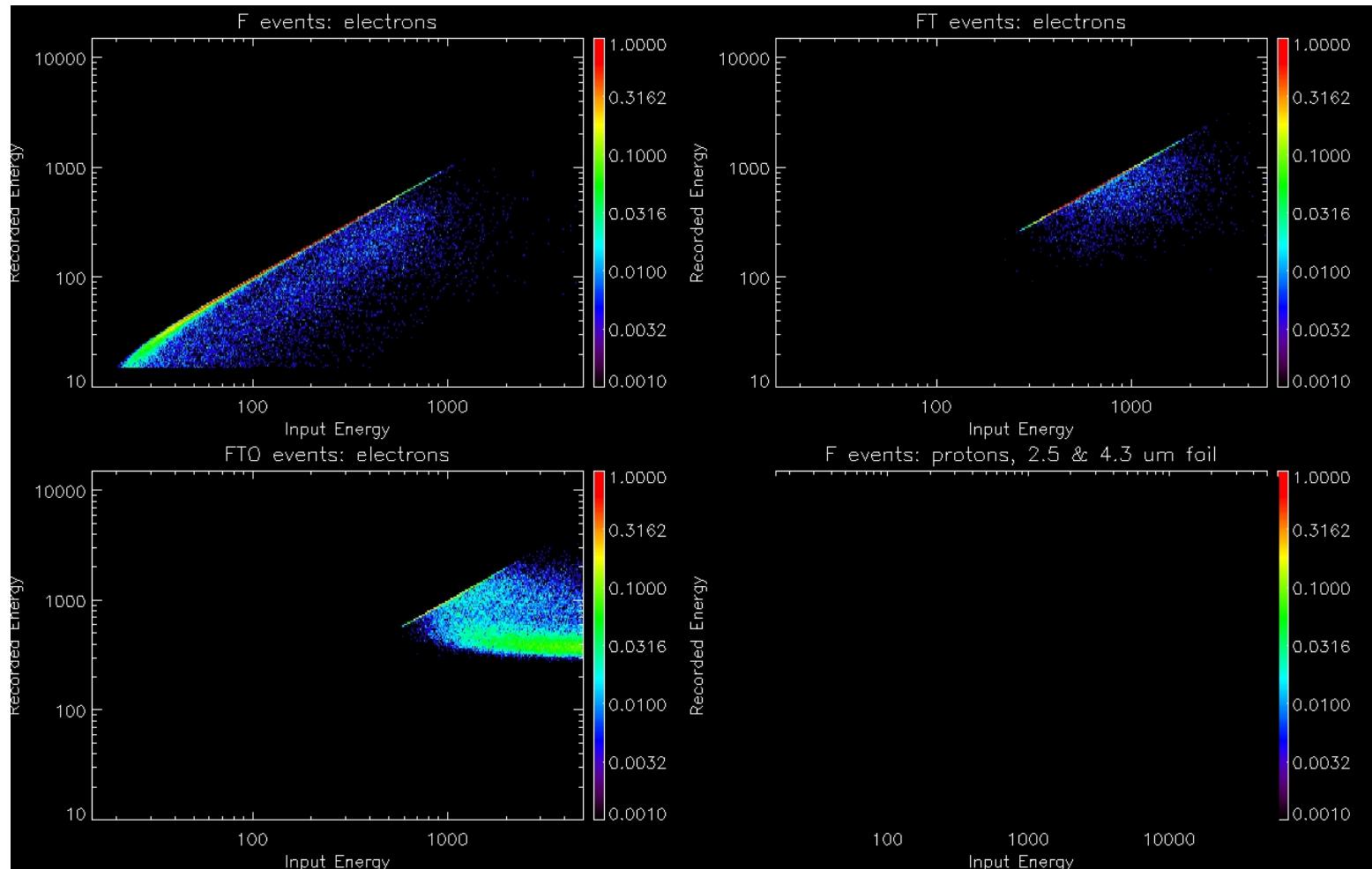
ION RESPONSE MATRICES

- Since ions deposit most of their energy at the very end of their trajectory, proton response matrix for O, OT and FTO events is quite clean.

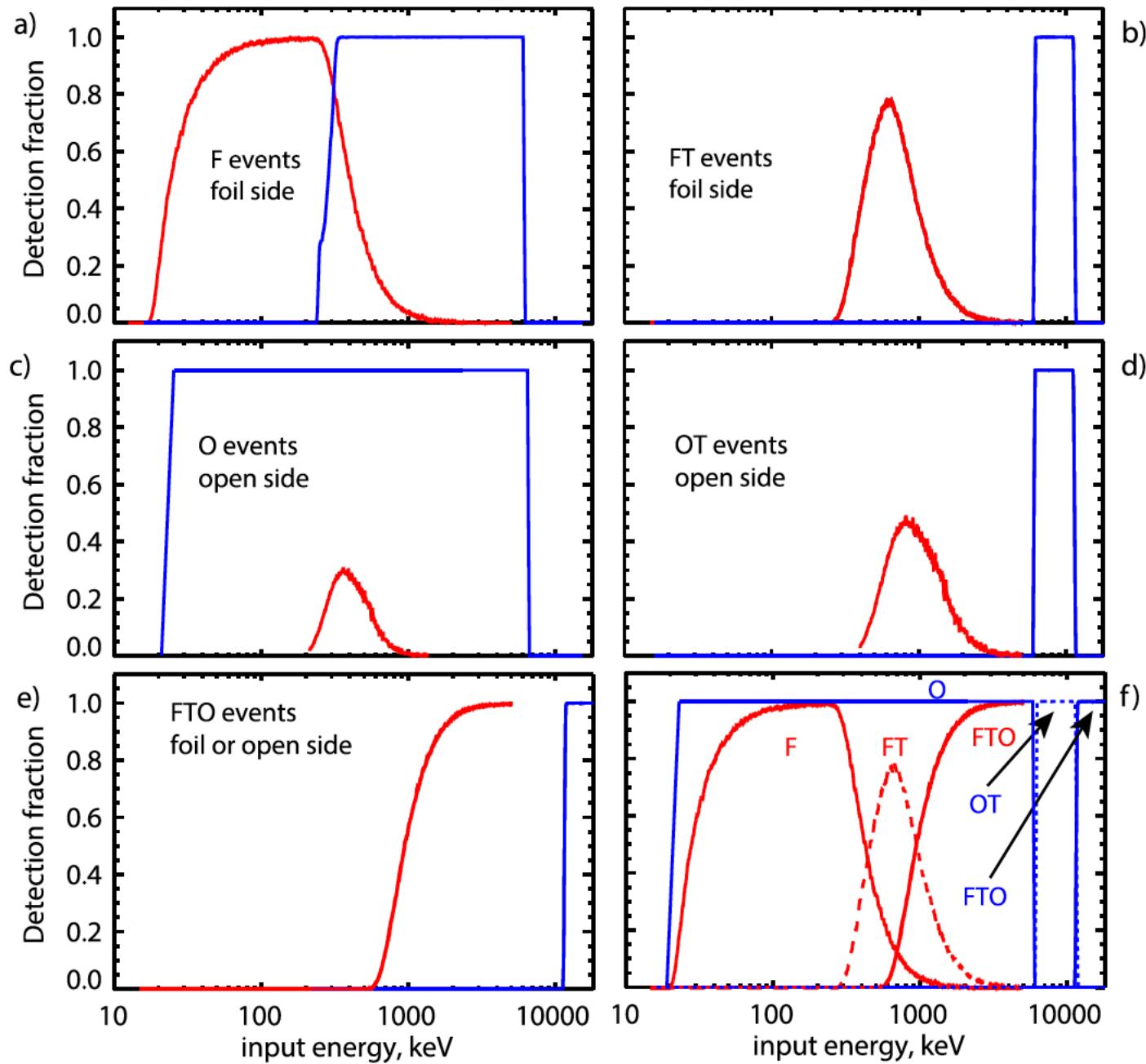


ELECTRON RESPONSE MATRICES

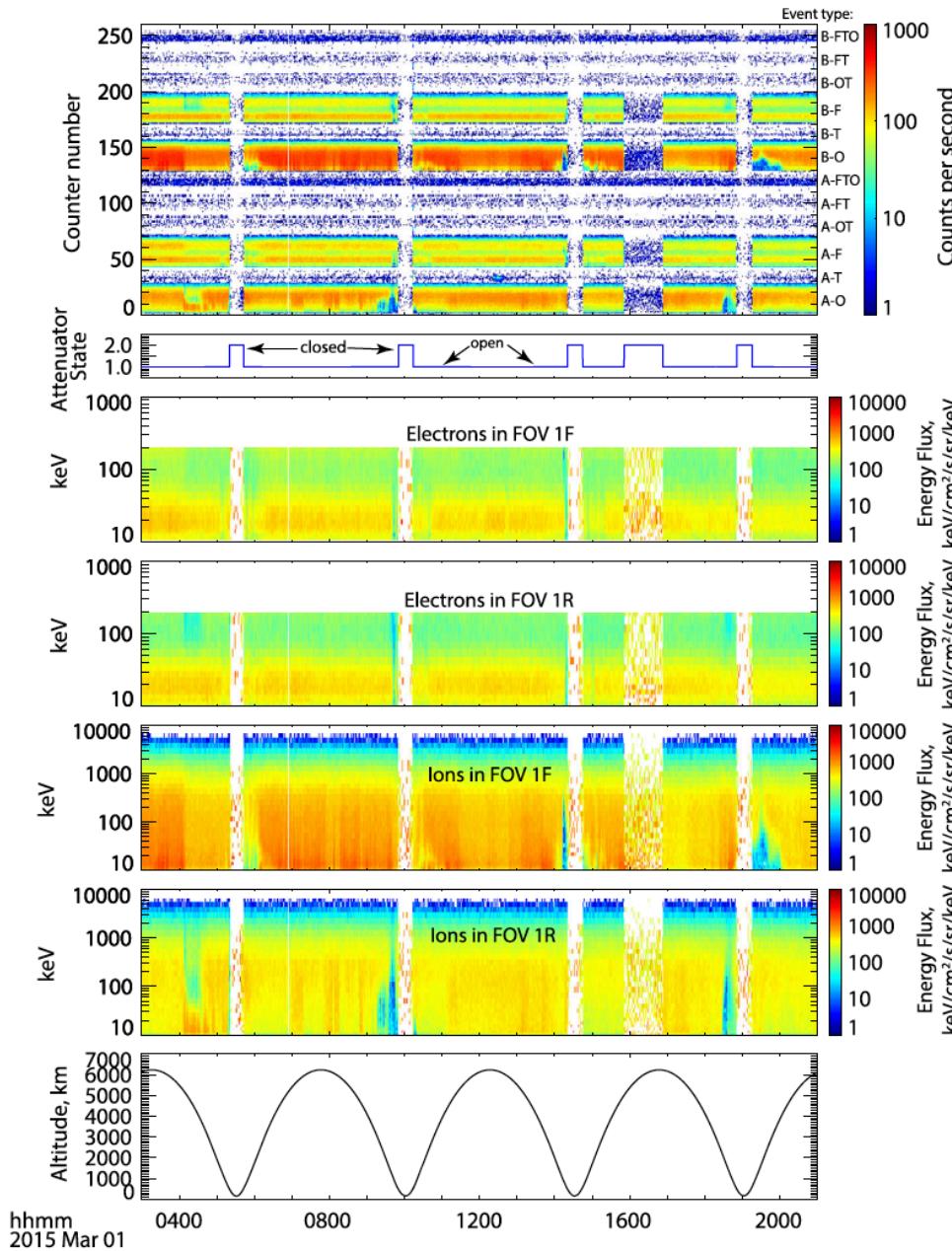
- Electrons deposit their energy over a much longer distance than ions.
Response matrices are much less diagonal.



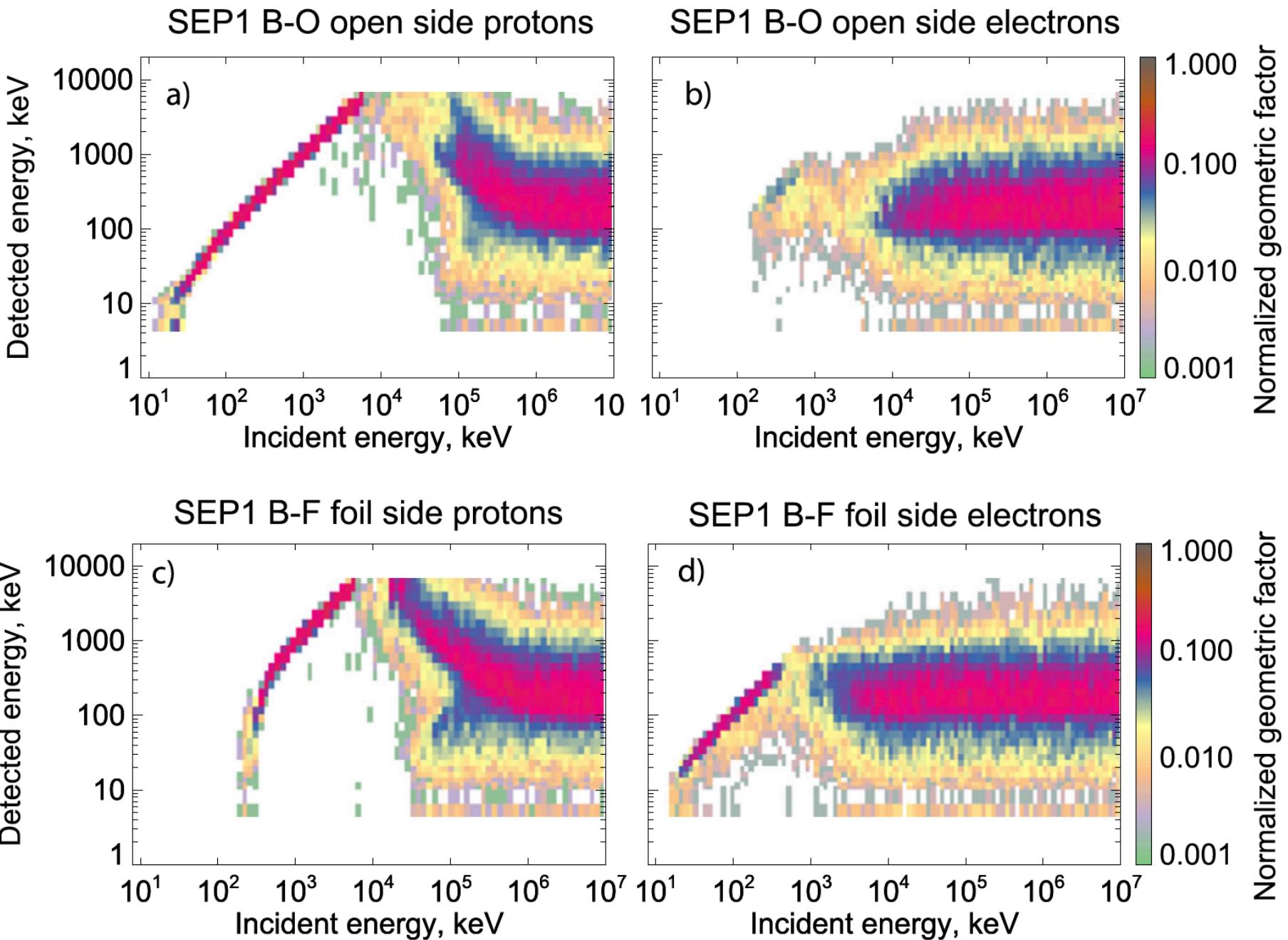
Detection Fraction for Electrons and Protons



EXAMPLE OF SEP DATA USING BEAM RESPONSE

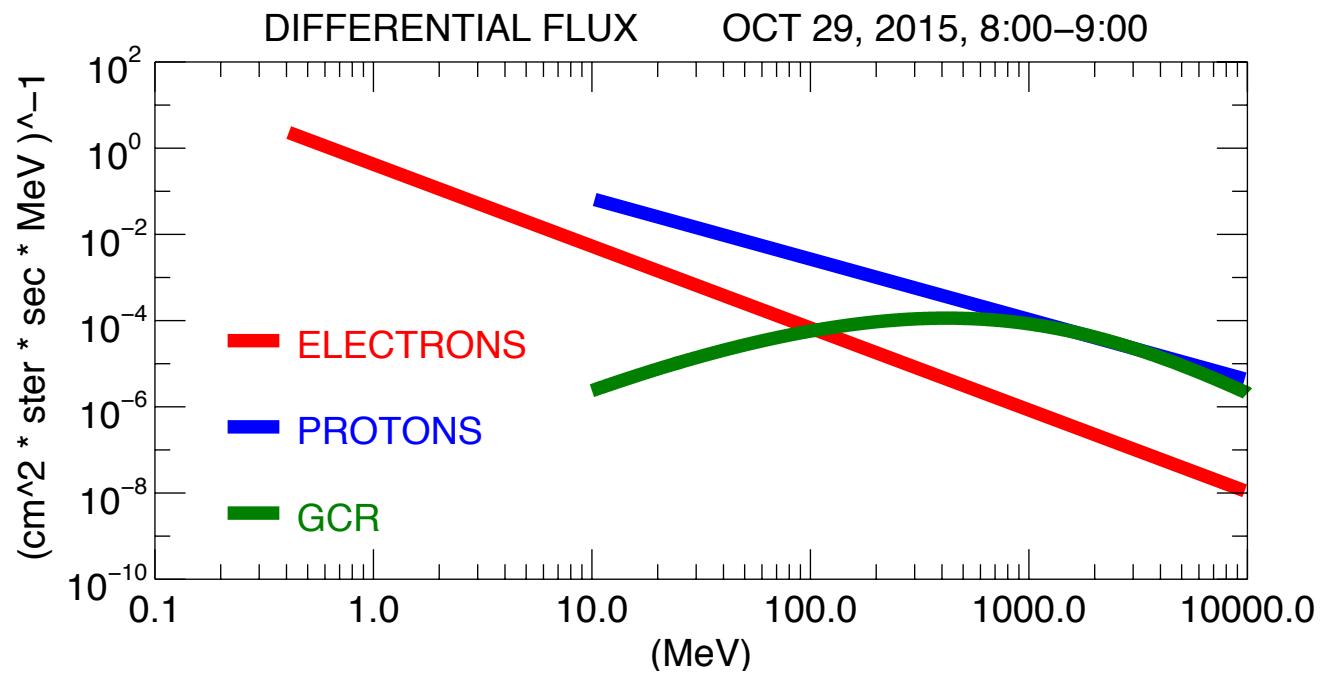
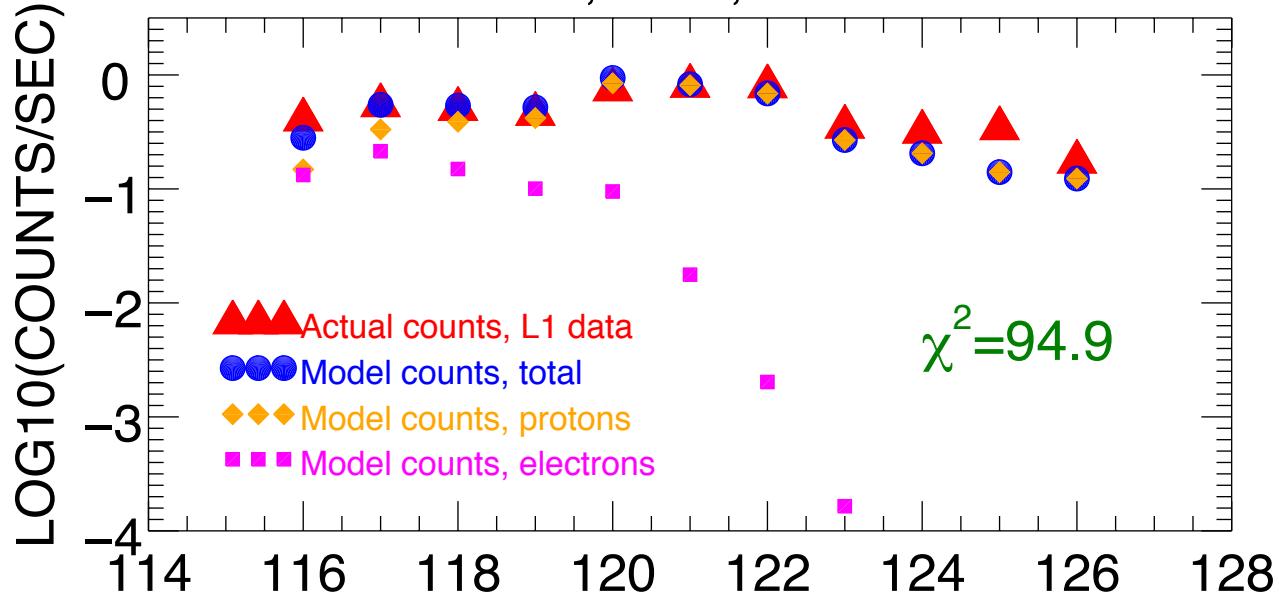


DETECTOR RESPONSE TO ISOTROPIC SOURCE SURROUNDING THE INSTRUMENT

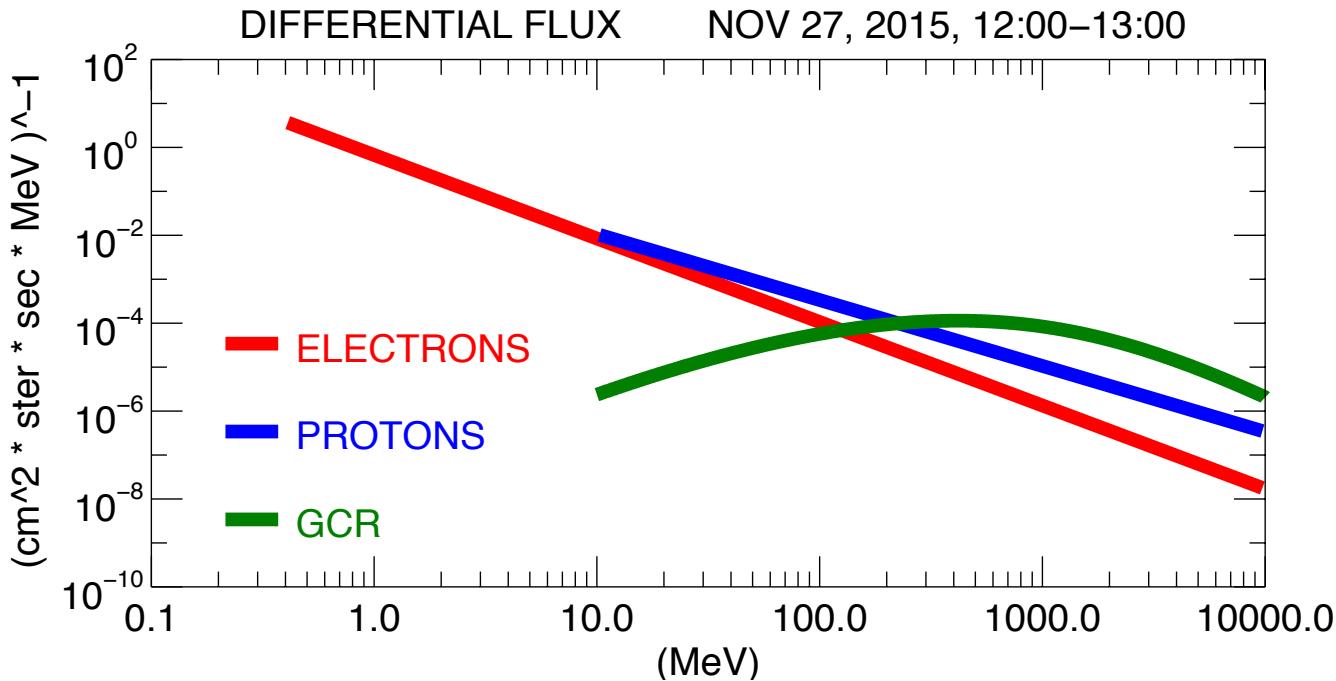
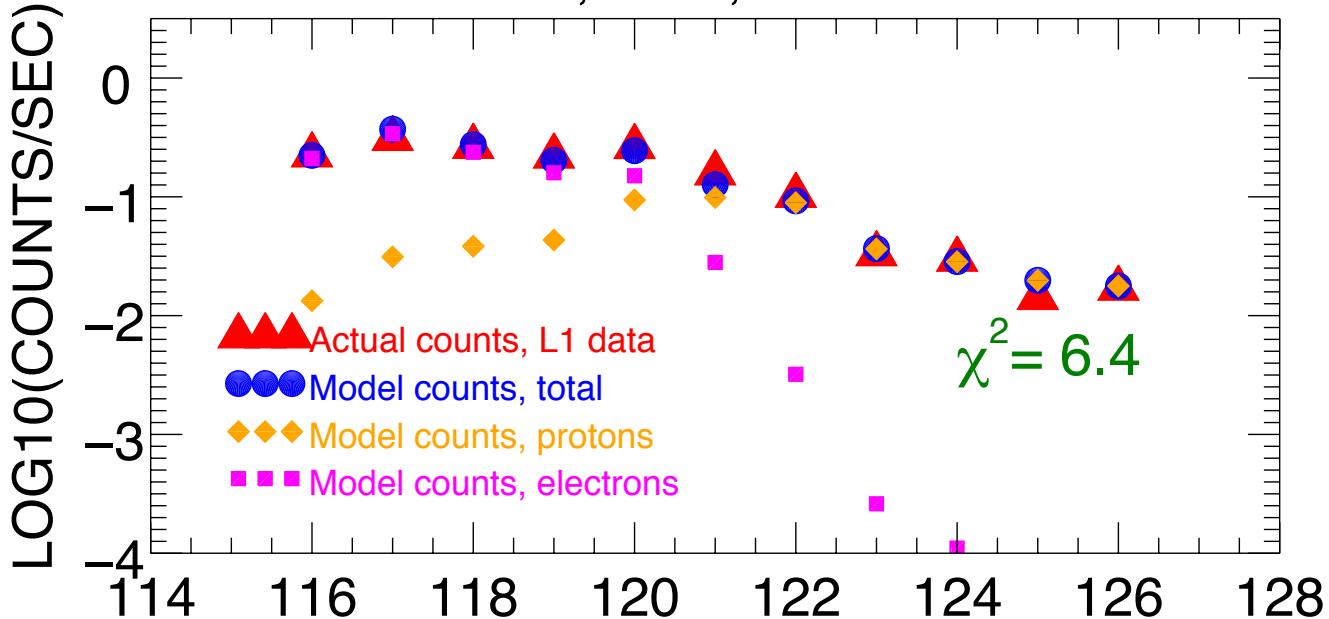


ATTEMPTS TO EXTEND ENERGY SPECTRUM

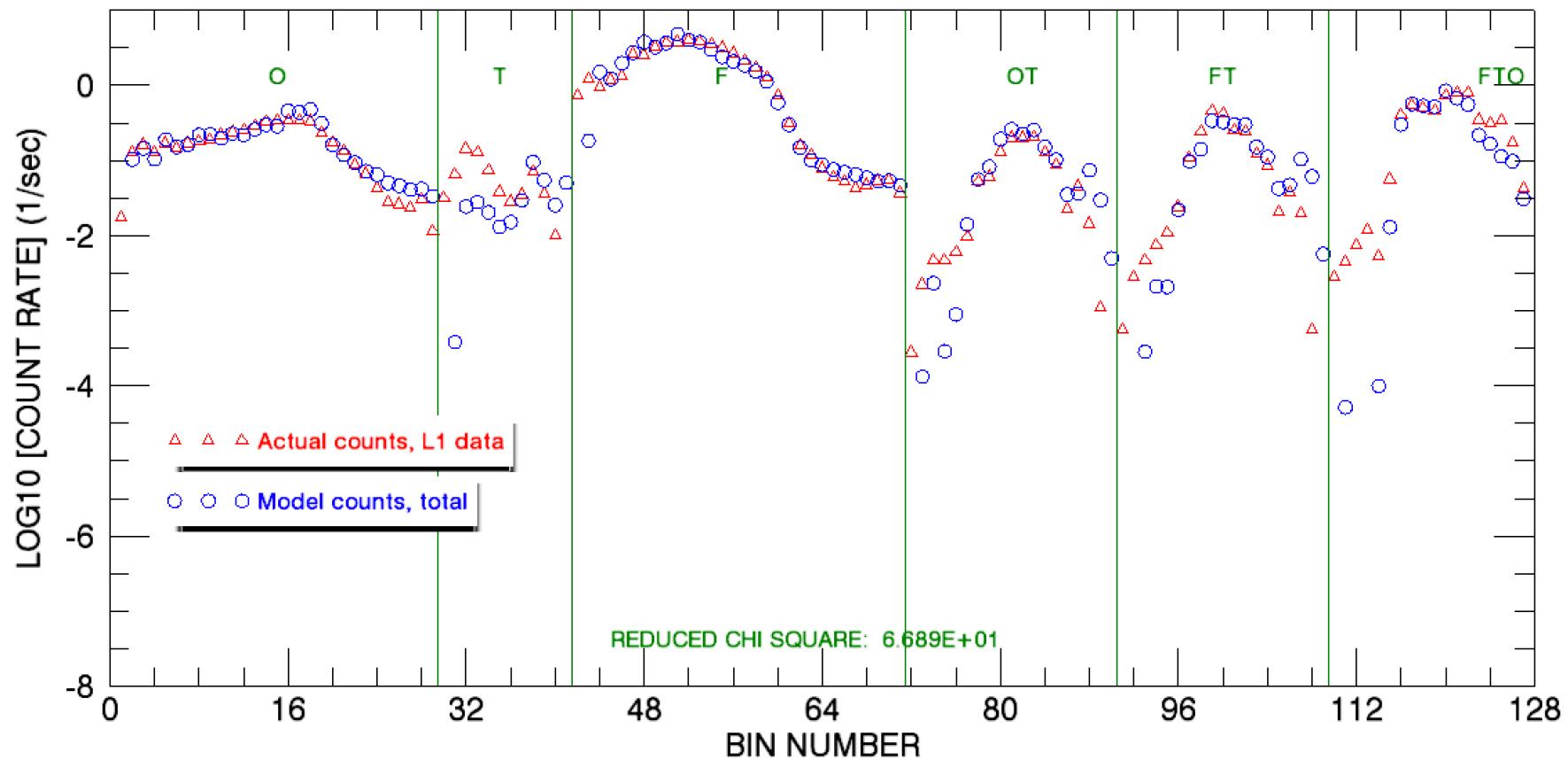
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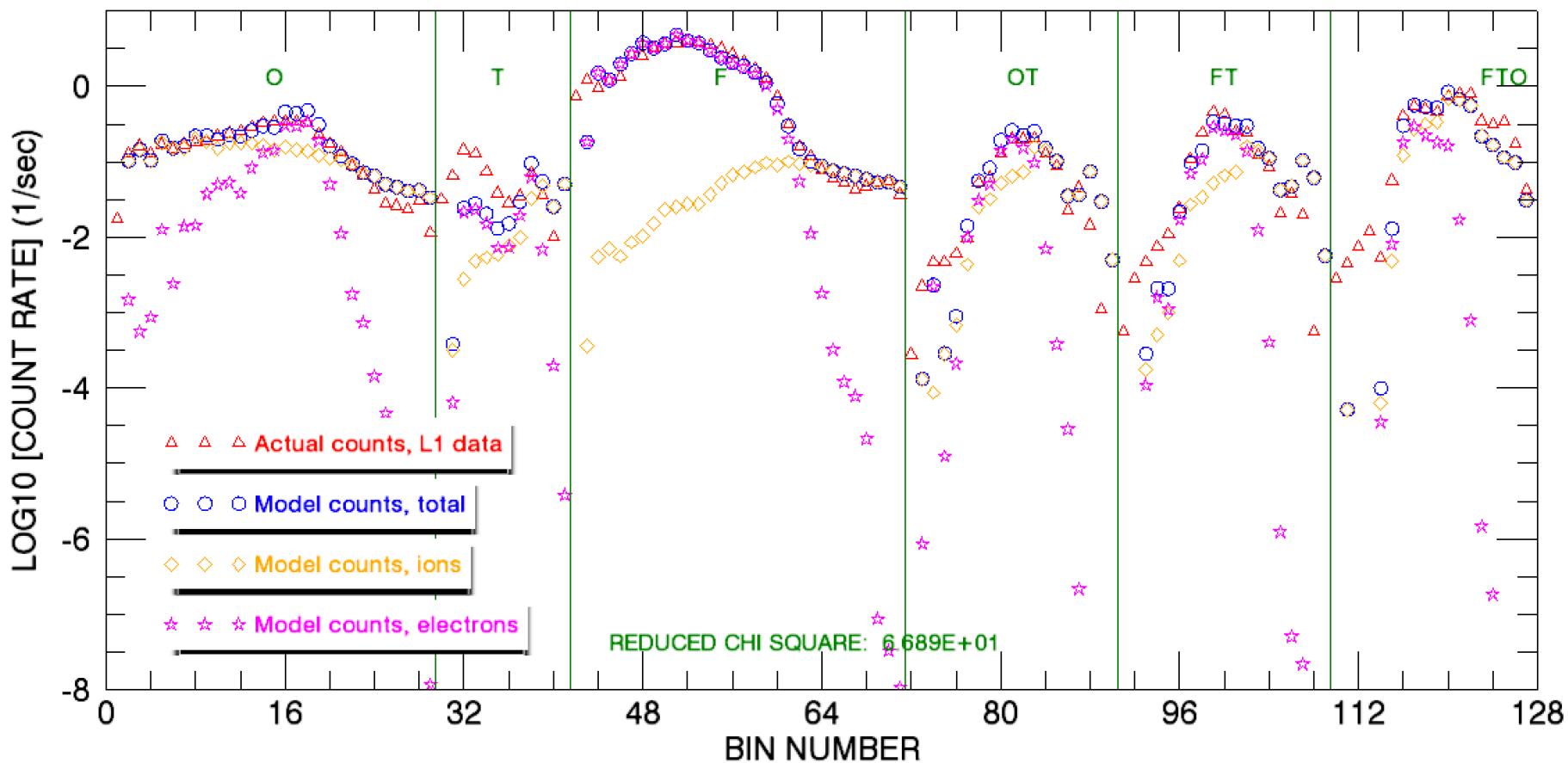
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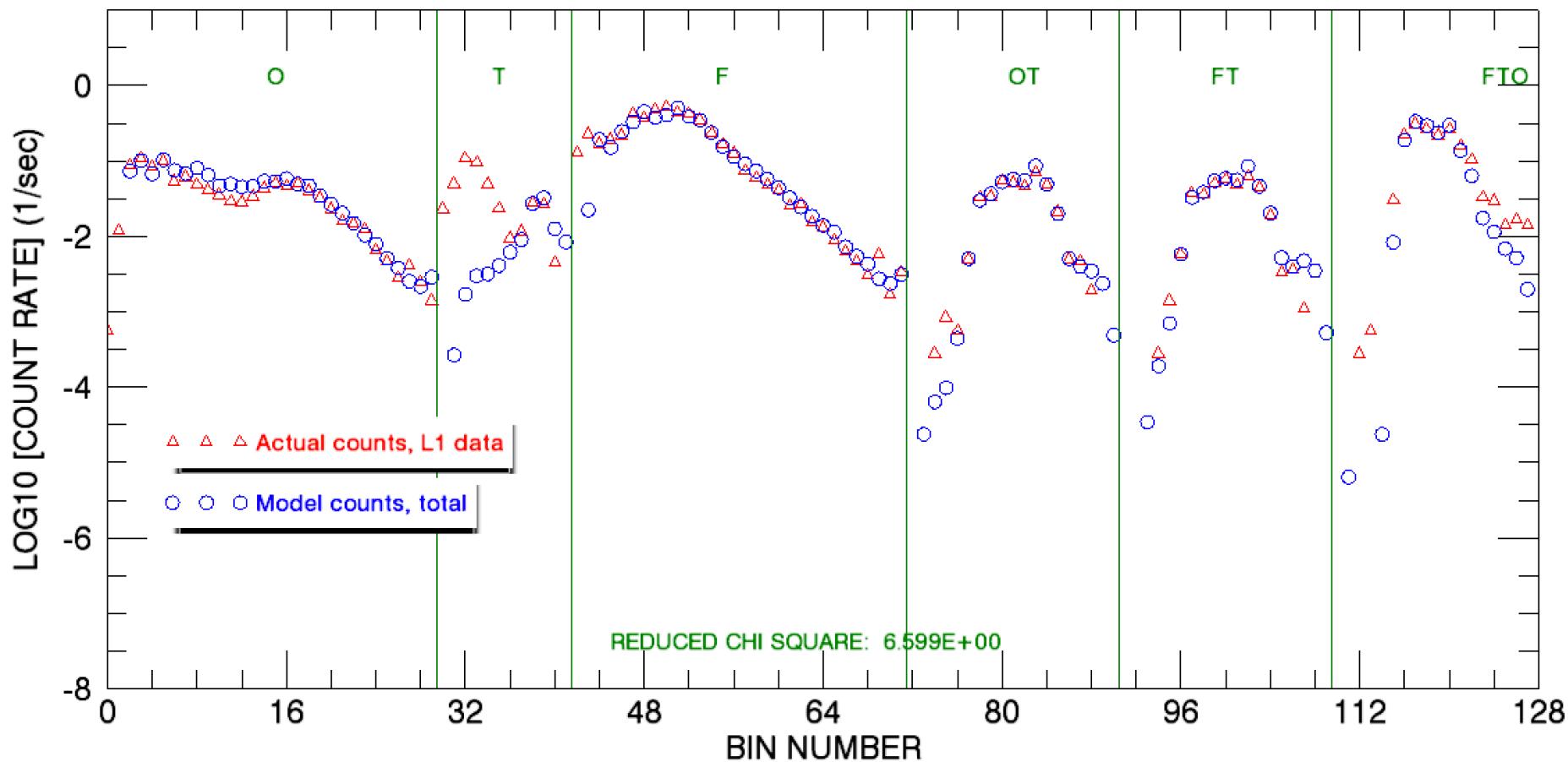
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