

JAXA status report 2018

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Brief Summary



- Not used frequently for spacecraft or manned-structure designing by engineers
- Used for detector design/calibration and astrophysical simulations by scientists
 - Payload design/calibration: Hitomi/XRISM, Suzaku, Arase, Cubesat, ...
 - Detector development
 - Astrophysical simulations

Spacecraft or Manned-Structure Designing?



- Geant4-use looks not so active:
 - Spacecraft design matured enough for near-earth mission?
 - Lunar, Martian or Venus mission can be treated as an extension of geostationary orbit—they have no magnetosphere.
- Next frontier:
 - Jovian or Saturn mission (but no active study yet)
 - Lunar manned-base? (related to post-ISS)

Payload design/calibration

- Several Geant4-utilized missions: on-orbit or terminated
 - Hitomi (a.k.a., ASTRO-H: X-ray observatory) terminated due to on-orbit accident
<http://www.isas.jaxa.jp/en/missions/spacecraft/past/hitomi.html>
 - XRISM—successor of Hitomi—(X-Ray Imaging and Spectroscopy Mission) under development
 - Arase (a.k.a., ERG: geospace probe satellite) on orbit
<http://www.isas.jaxa.jp/en/missions/spacecraft/current/erg.html>
 - Suzaku terminated due to big solar flare in 2015
<http://www.isas.jaxa.jp/en/missions/spacecraft/others/suzaku.html>
- Geant4 usage?
 - Detector's shielding design ← already reported in previous workshops
 - Precise detector calibration
 - Hitomi/XRISM ← will be presented 3rd day
 - Arase
 - Suzaku

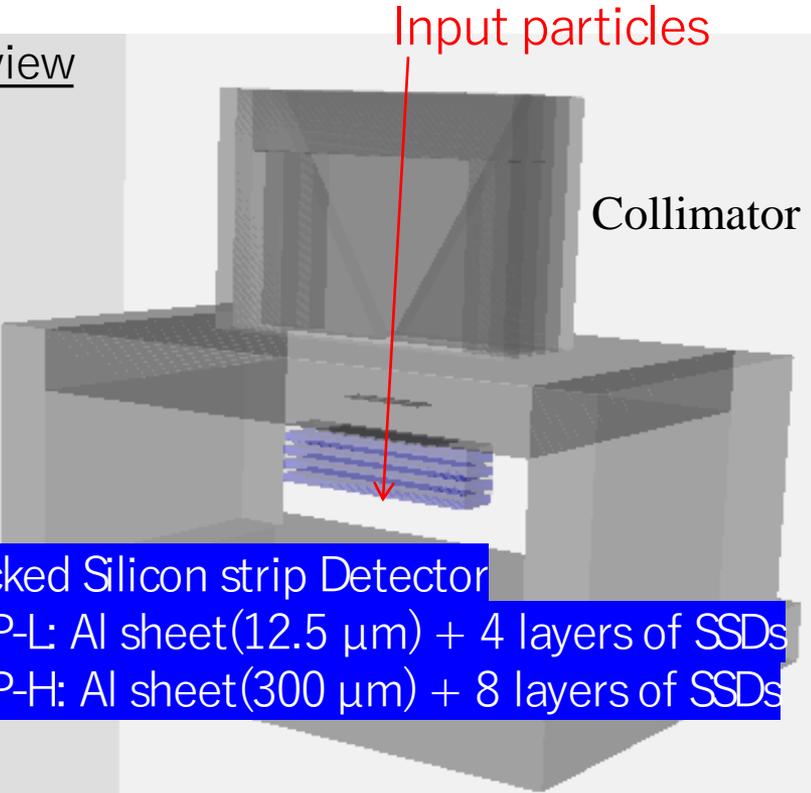


High energy electron sensor onboard Arase

High-energy Electron exPeriments (HEP) instrument on board Arase observes 70 keV– 2 MeV. The HEP consists of three HEP-L modules (70 keV – 1MeV) and three HEP-H modules (700 keV - 2MeV).

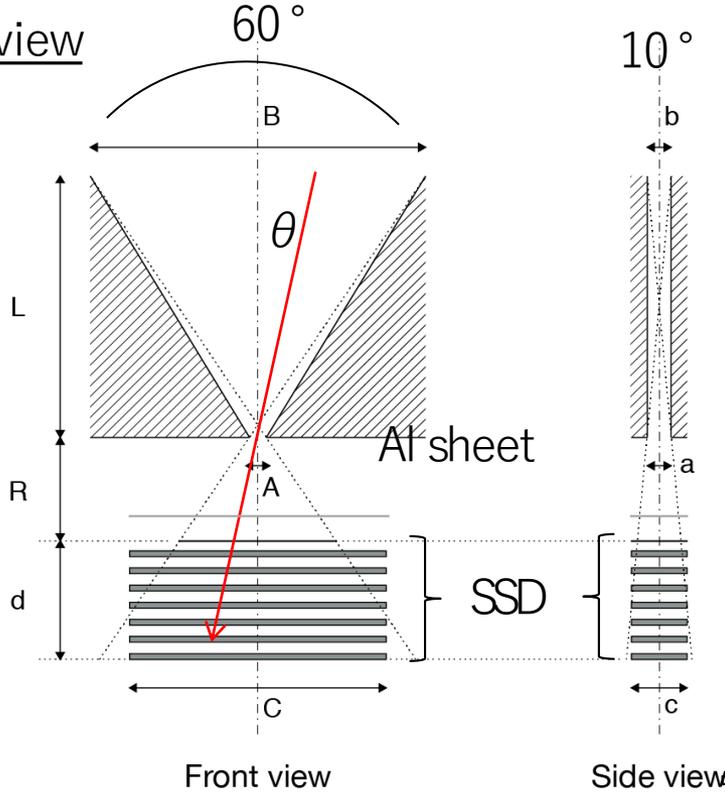
A module consists of a mechanical collimator and stacked silicon strip detectors (SSDs). The Opening angle of the collimator: ~60 degree x 10 degree. To determine elevation angle, SSD has one dimensional position resolution.

3D view



Stacked Silicon strip Detector
HEP-L: Al sheet(12.5 μm) + 4 layers of SSDs
HEP-H: Al sheet(300 μm) + 8 layers of SSDs

Plain view



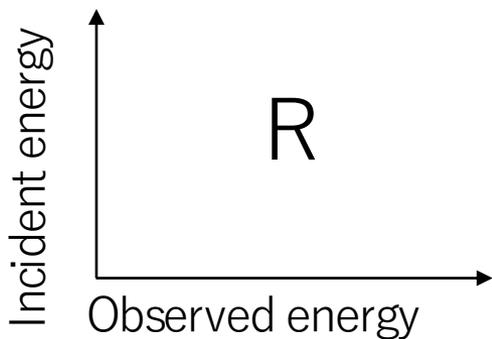
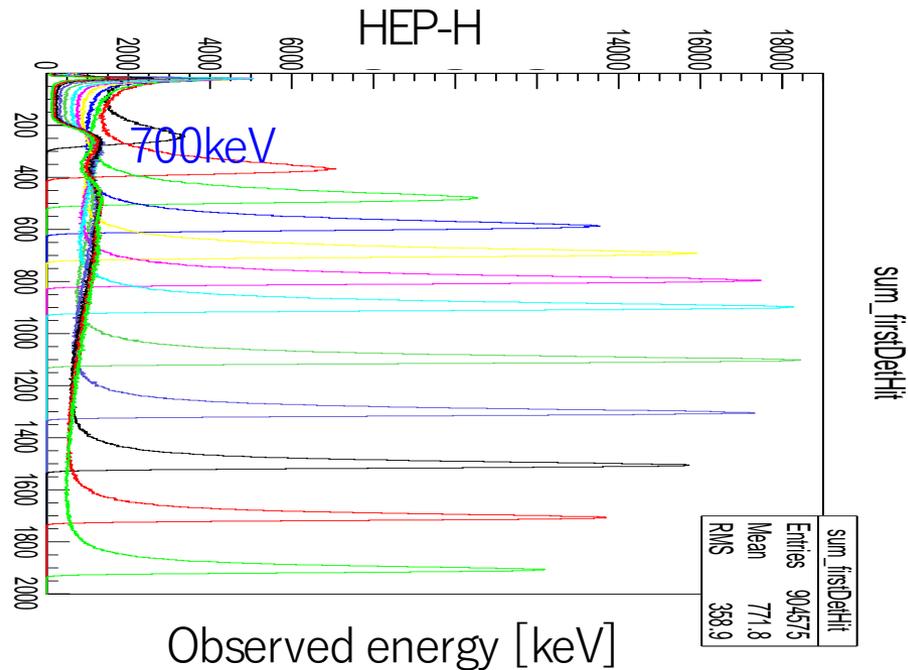
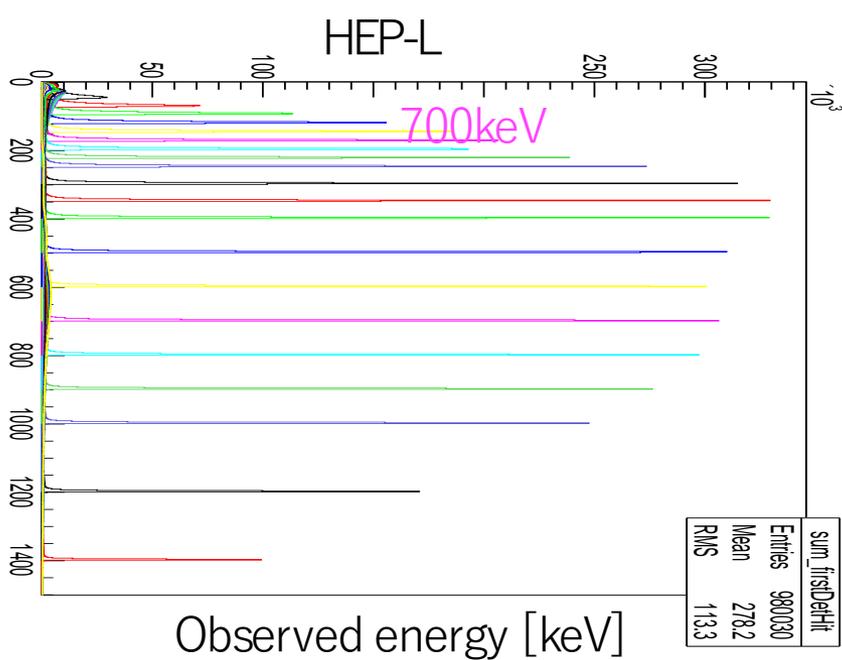
Front view

Side view



Energy response of HEP detector

To calculate detector response to electrons, we use Geant 4 toolkit. Figures below show detected spectra when monoenergetic electrons with various energies are irradiated to the HEP SSD module.



To deduce incident energy distribution from observed histogram, we calculate a response matrix R.

$$\text{Incident flux} = R^{-1} * \text{Observed count}$$

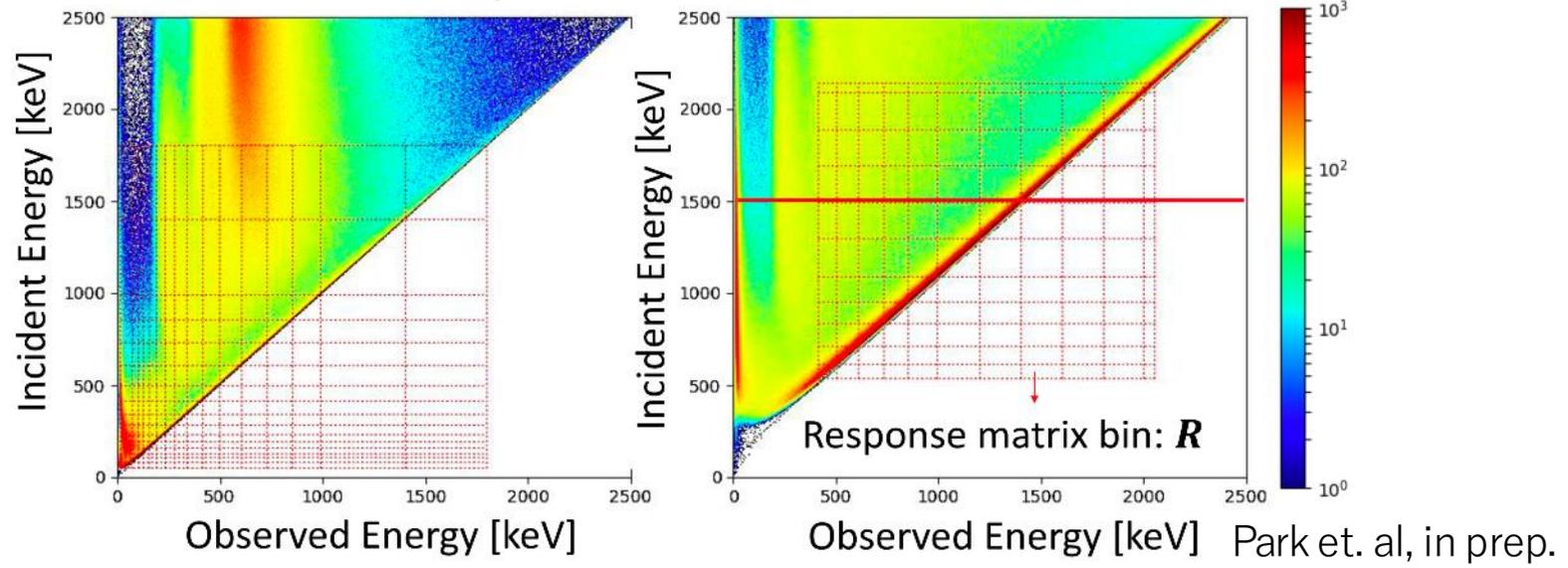


Response Matrix

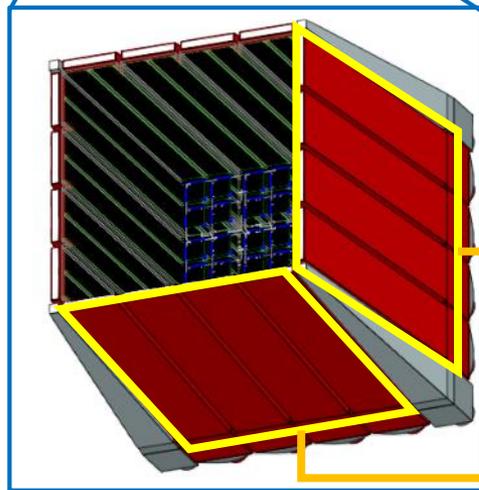
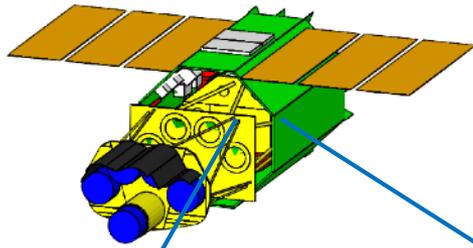
- Based on Geant-4 simulations, we have constructed response matrices for HEP-L and HEP-H. By inverting the the matrices, we derive incident electron differential fluxes from an observed count histogram. The resultant flux data are open to the public as Level-2 science data product.

$$[\text{Observed histogram}] (1D) = [\text{Response Matrix}] (2D) \times [\text{Differential flux}] (1D)$$
$$\rightarrow [\text{Differential flux}] = [\text{Response Matrix}]^{-1} \times [\text{Observed histogram}]$$

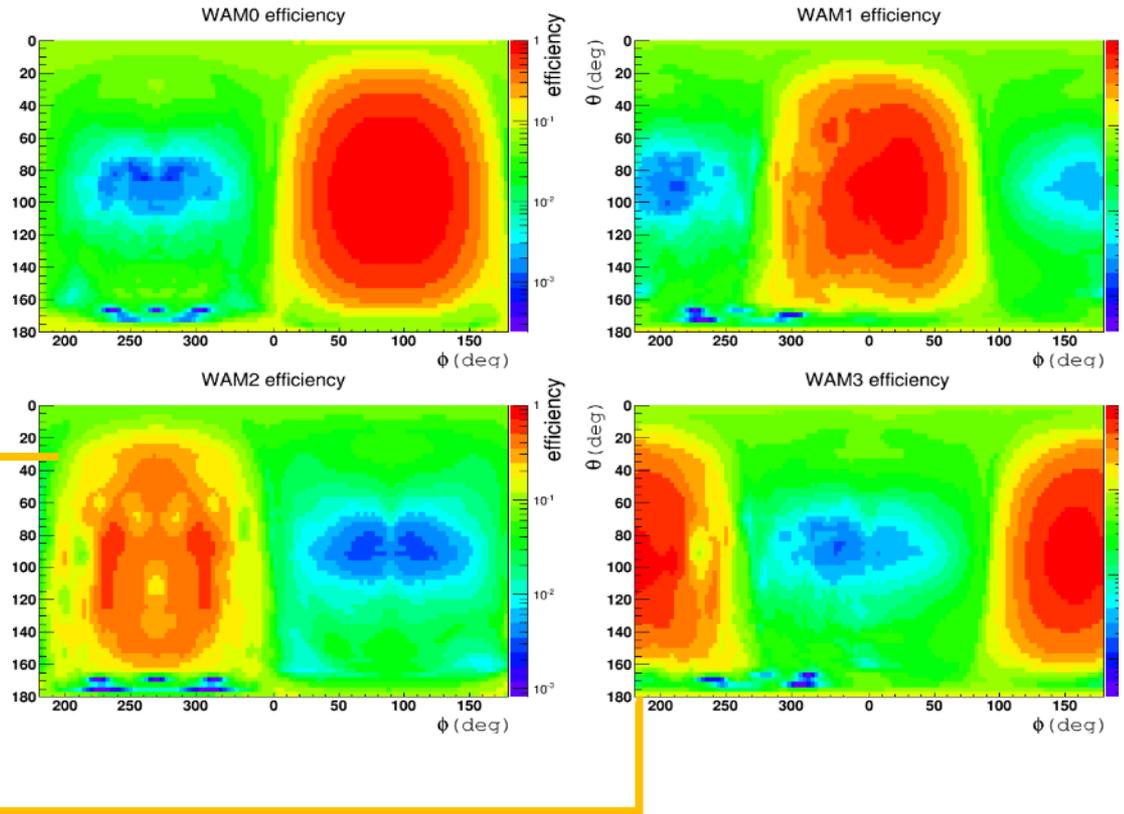
Electron count histograms over the incident-observed energy space simulated by the ERG-HEP simulator using Geant 4 toolkit



Suzaku HXD detector precise calibration



Geant4 mass models

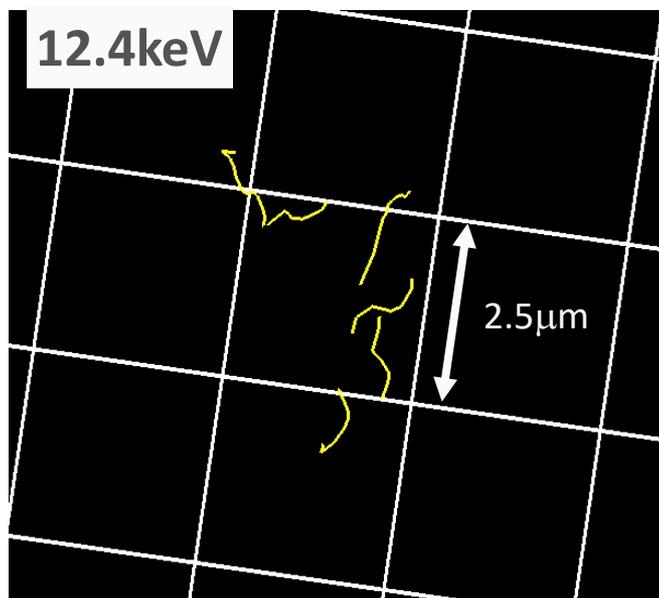


Gamma-ray detection efficiency for each panel

Detector Development

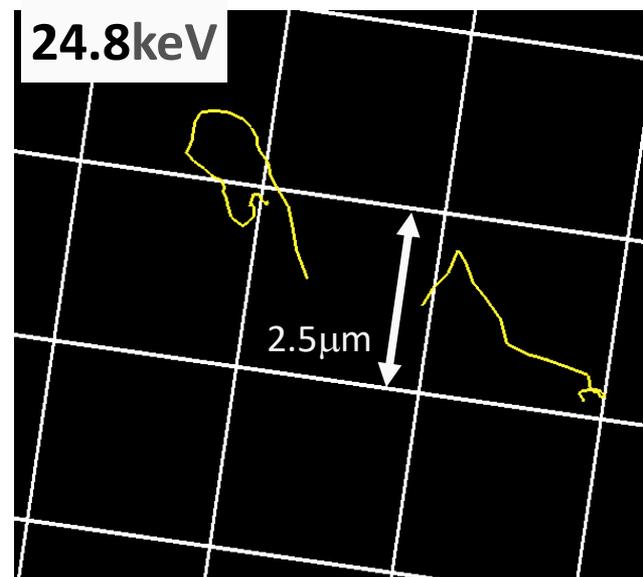
- Geant4 is used to simulate the detector behavior
 - X-ray polarization detection with precise CMOS image sensor
 - Optimal scintillator design using G4 optical photon simulation

By courtesy of K.Asakura (Osaka-U)



MF=16.6±1.2%@12.4keV

$$MF = \frac{C_{max} - C_{min}}{C_{max} + C_{min}}$$



MF=18.7±0.7%@24.8keV

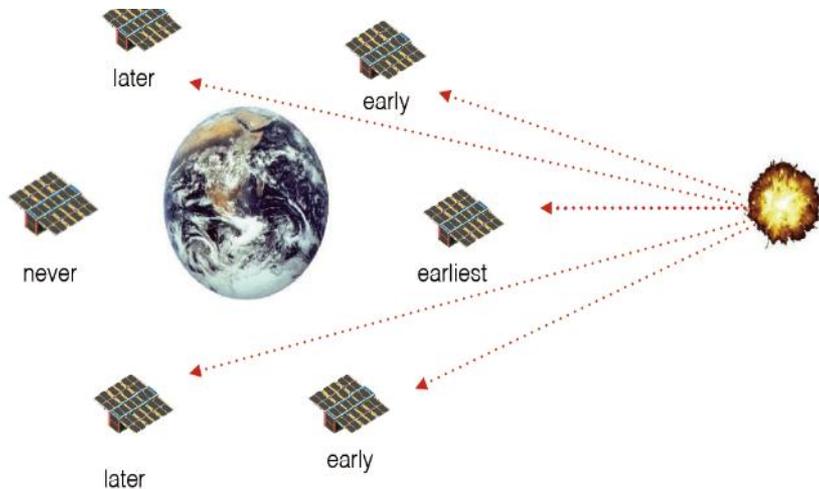
- Simulation of photoelectron tracks.
- 100% polarized incident X-rays are irradiated.
- Diffusion of the charge inside the detector is NOT taken into account.

Detector Development

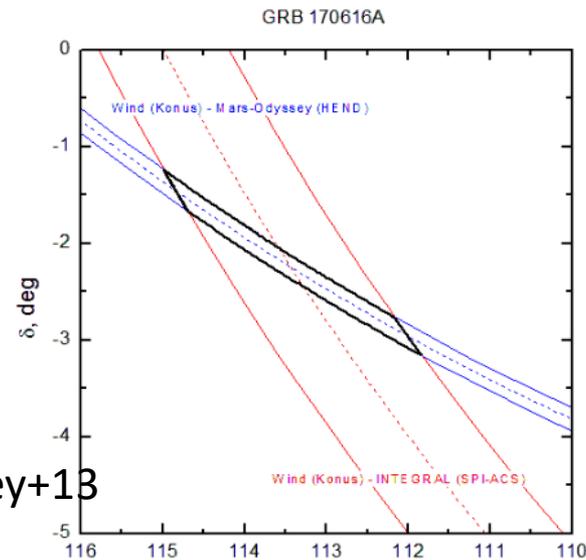
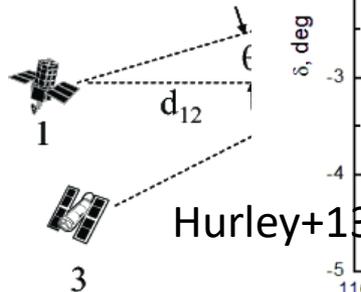


- Geant4 is used to simulate the detector behavior
 - X-ray polarization detection with precise CMOS image sensor
 - **Optimal scintillator design using G4 optical photon simulation for GRB detection fleet**

fleet of cube-satellite w/ triangulation



$$\cos \theta_{12} = c\Delta t_{12} / d_{12}$$



□ dozen cube satellite constellation

- ➔ all-sky coverage
- (no earth shadow)
- huge effective area

□ localization by photon arrival time

- High timing synchronization by GPS
- ➔ <100 μ-sec timing accuracy results
- several arcmin localization accuracy
- (Pal+18 SPIE)

CAMELOT (Cubesat Applied for MEasuring and Localising Transients) (Werner+18 SPIE)



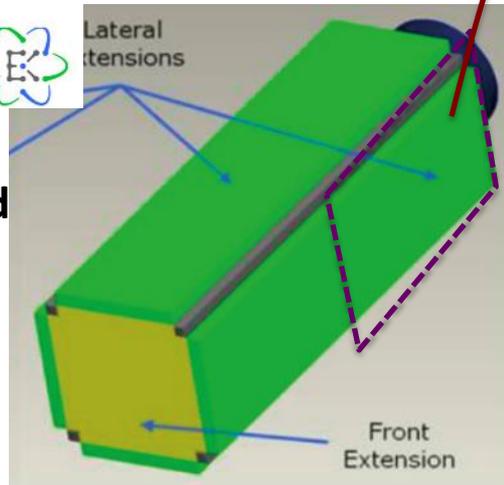
Imperial College London

Astronika

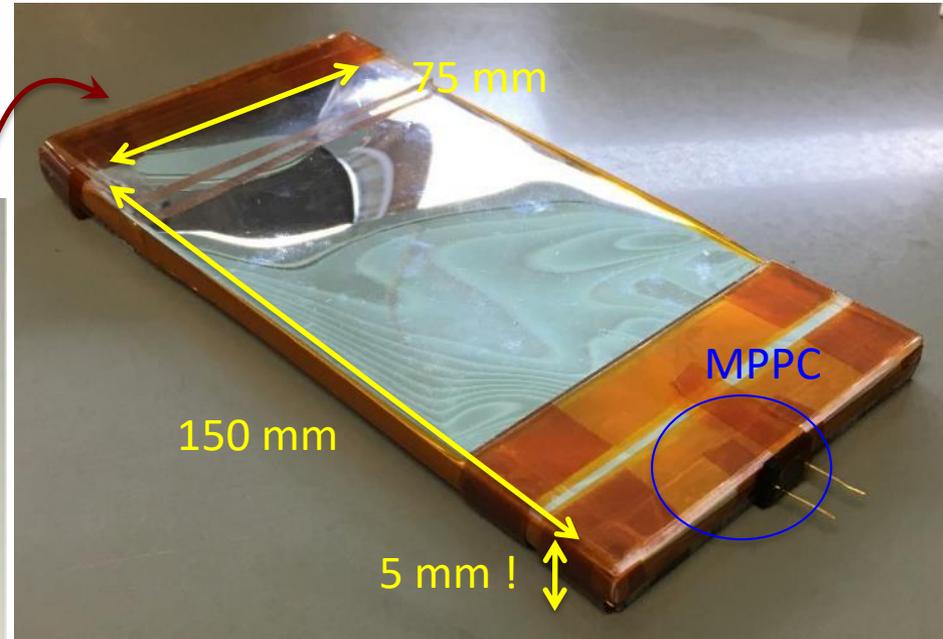
COMPLEX SYSTEMS & SMALL SATELLITES



Lateral tensions



Satellite platform :
3U Cubusat designed for “RADCUBE” mission. Hungarian 3CS company + ESA consortium



- two sides of lateral extension :
~268 cm² x 9mm is a basic idea to maximize geometrical area
- challenging for such “large and thin” detector with small readout area

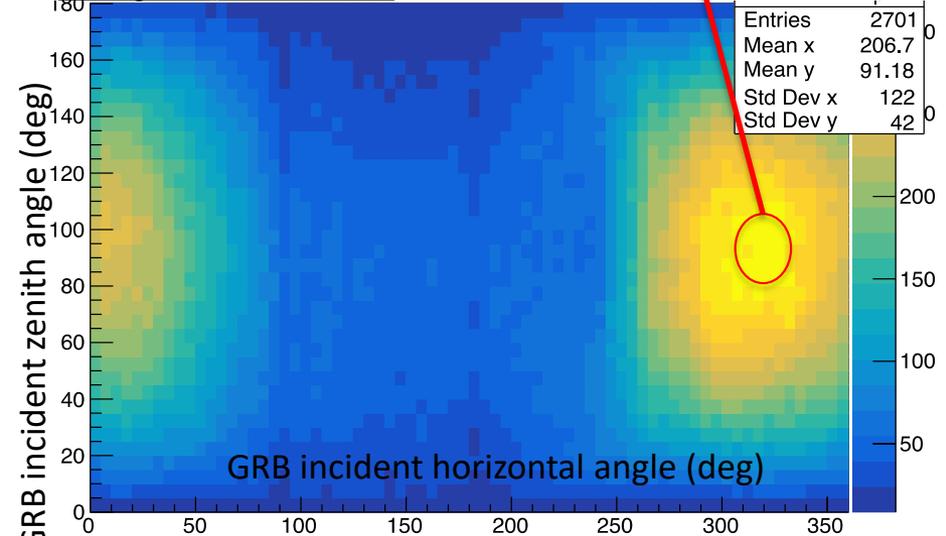
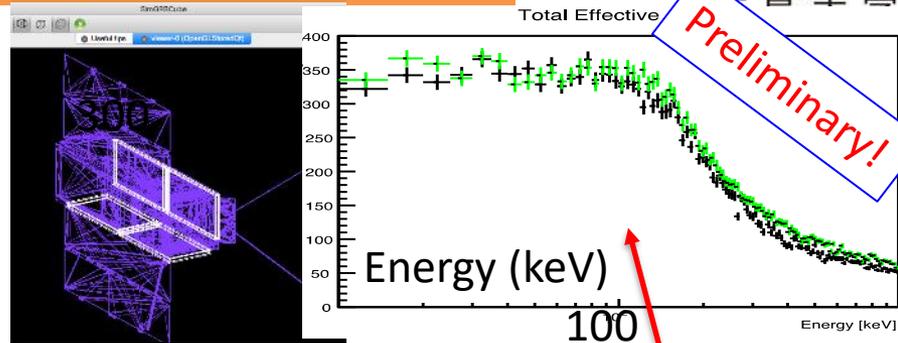
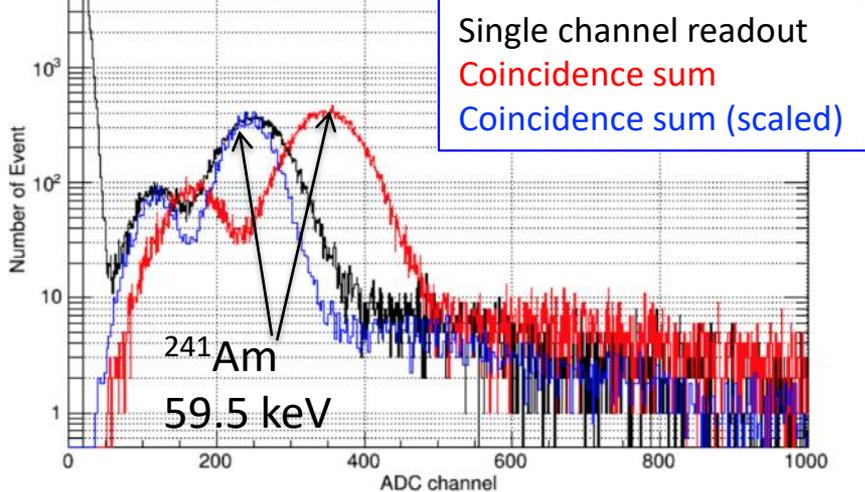
CsI scintillator readout by multi pixel photon counter (MPPC) is now being evaluated as the first feasibility study for their large light yield, compact readout area and low power consumption.

Detector feasibility

By courtesy of M.Ohno (Hiroshima-U)



Torigoe+ NIMA in press (Poster#**)



✓ Developed a multi-readout system for high light yield and redundancy

✓ Effective area simulation including

Comparable observational capability to the one detector of Fermi-GBM with our single unit of Cubesat

from GRBs

absorption

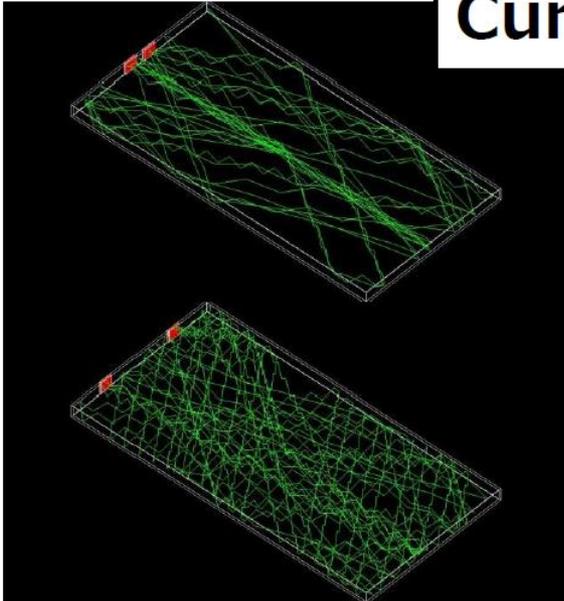
In order to improve the light yield, we developed a readout system with MPPCs

- Light yield and uniformity was improved
- Energy threshold is ~ 10 keV at 25 °C

By courtesy of K.Torigoe (Hiroshima-U)



Current study



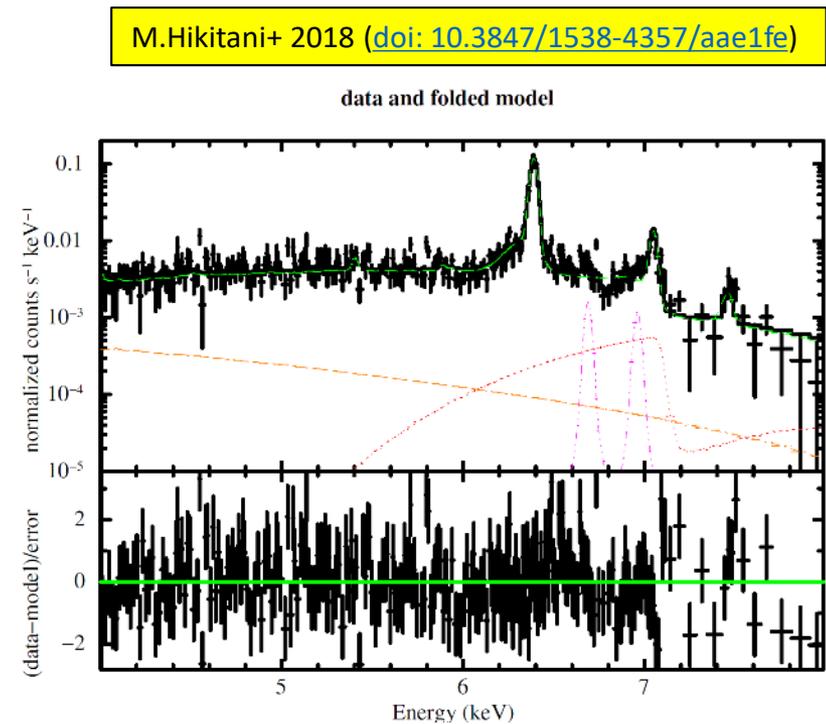
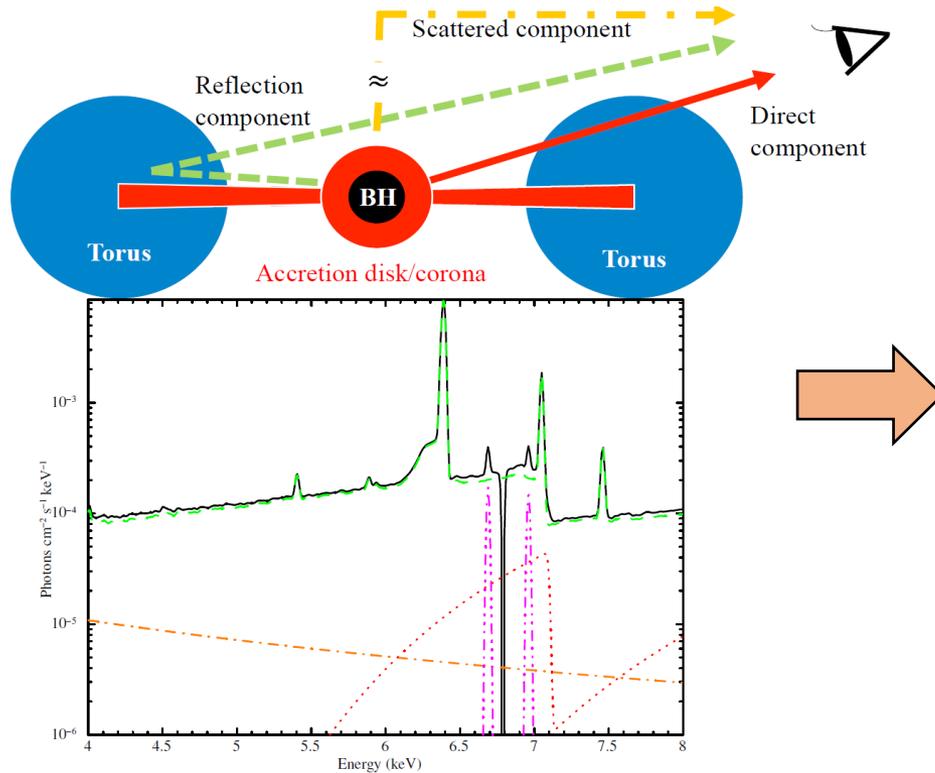
Optimum position of two MPPCs based on Geant4 simulation



Effect of radiation damage to MPPCs in orbit by the proton beam test

Astrophysical Simulations

- Geant4 simulation in astrophysical scale (e.g., pc to Mpc) to represent observations
 - Hitomi results ← will be presented 3rd day
 - Chandra spectrum of active galactic nuclei



Summary

- Geant4 not used frequently for spacecraft or manned-structure designing by engineers
 - Maybe some use PHITS
- Used for detector design/calibration and astrophysical simulations by scientists
 - Payload design/calibration: Hitomi/XRISM, Suzaku, Arase, Cubesat
 - Detector development: Scintillator design, CMOS X-ray polarimeter
 - Astrophysical simulations: photon simulations observed by Chandra, Hitomi