

# Life cycle of propellants

## Environmental benchmark of current "green" propellants

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# Methods



## The study includes

- *RP-1/LOx*
- *LH<sub>2</sub>/LOx*
- *CH<sub>4</sub>/LOx*
- *UDMH/NTO*
- *Ammonium perchlorate composite propellant (APCP)*

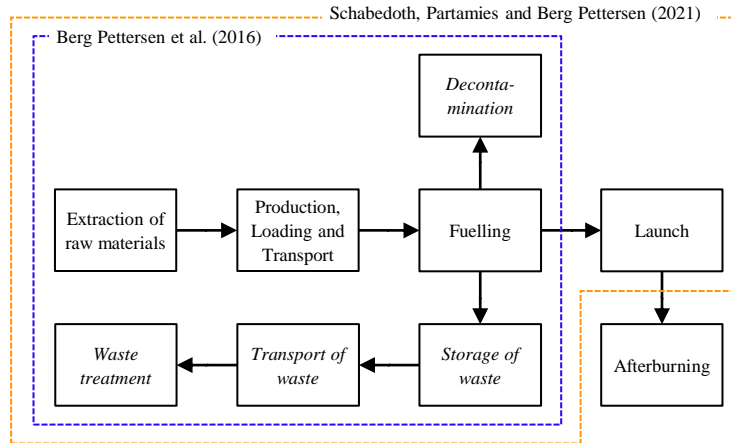
Climate change  
GWP<sub>100</sub> (CO<sub>2</sub> eq.)

Ozone depletion  
ODP (CFC-11 eq.)

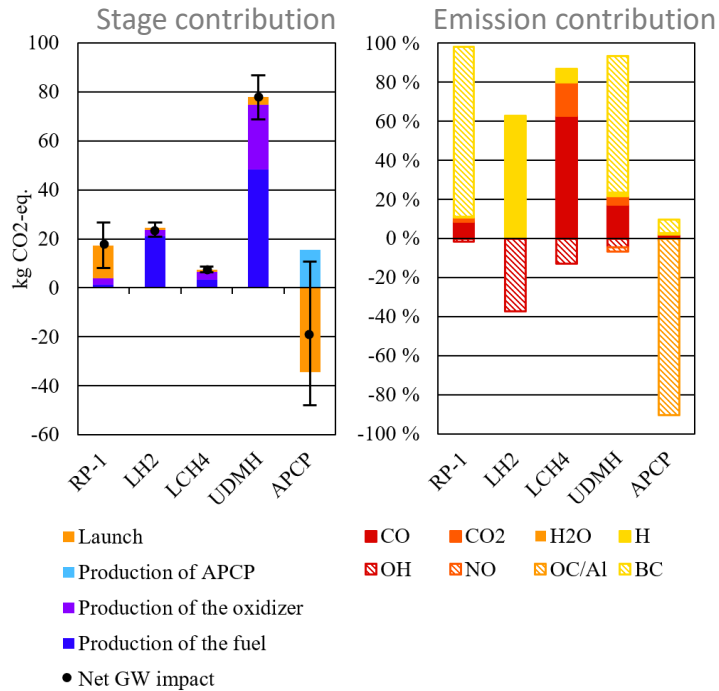
GWP<sub>100</sub> beyond standard factors:  
CO, H<sub>2</sub>O, H<sub>2</sub>, OH, NO, BC, OC, alumina (≈OC)  
ODP: NO, Cl and HCl

CEARUN, i.e., no afterburning  
Black carbon (BC) from literature

ESA Space LCA Database  
for stages up to launch



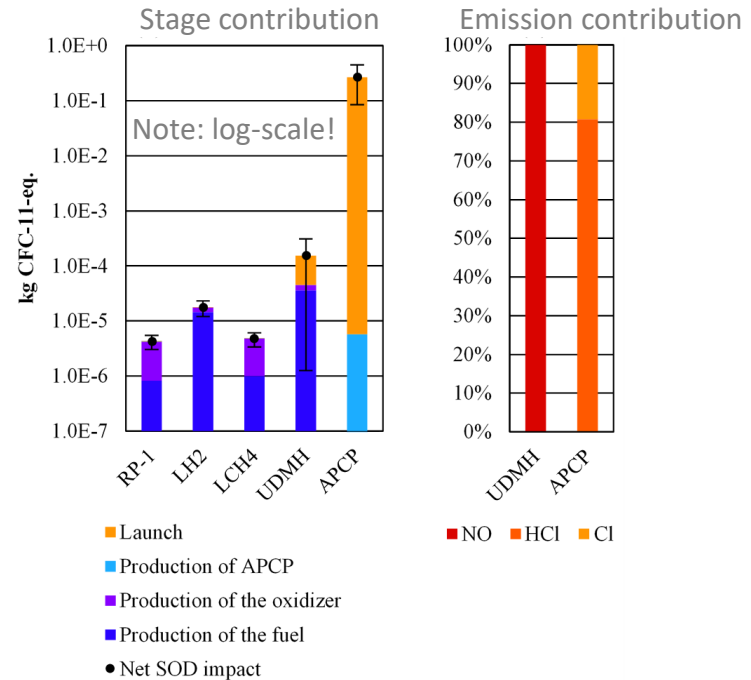
Life cycle impacts weighted by  $I_{SP}$



- UDMH/NTO: production
- LH<sub>2</sub> production
- APCP: cooling effect from alumina particulates (assumed as organic carbon, OC)
- RP-1: black carbon (BC) from launch

Climate change  
(CO<sub>2</sub> eq.)

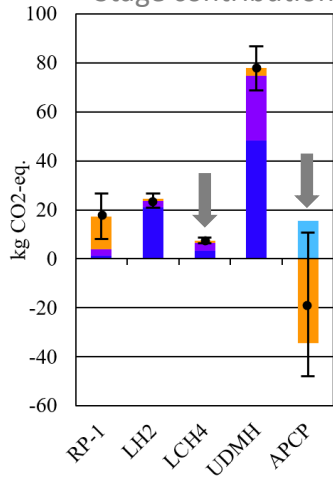
- Fuel (& oxidizer) production
- APCP: HCl & Cl-emissions from launch



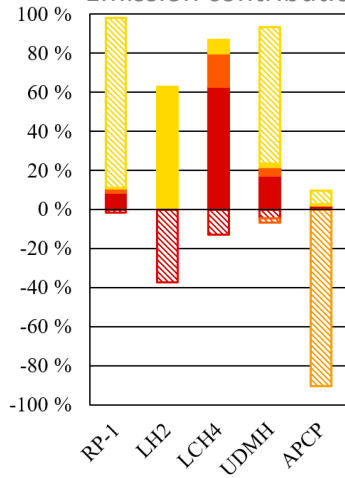
Ozone depletion  
(CFC-11 eq.)



Stage contribution



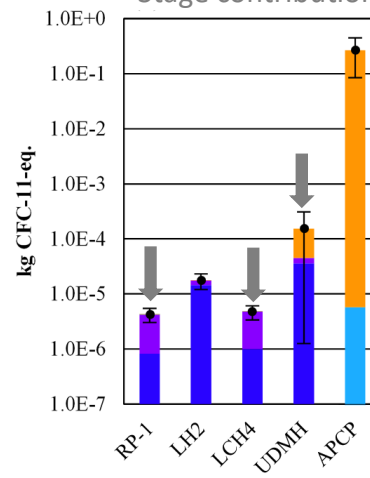
Emission contribution



- Launch
- Production of APCP
- Production of the oxidizer
- Production of the fuel
- Net GW impact

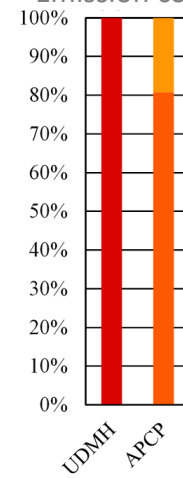
- CO
- CO2
- H2O
- H
- OH
- NO
- OC/Al
- BC

Stage contribution



- Launch
- Production of APCP
- Production of the oxidizer
- Production of the fuel
- Net SOD impact

Emission contribution



- NO
- HCl
- Cl

Climate change  
(CO2 eq.)

**LCH<sub>4</sub> and APCP preferred options**

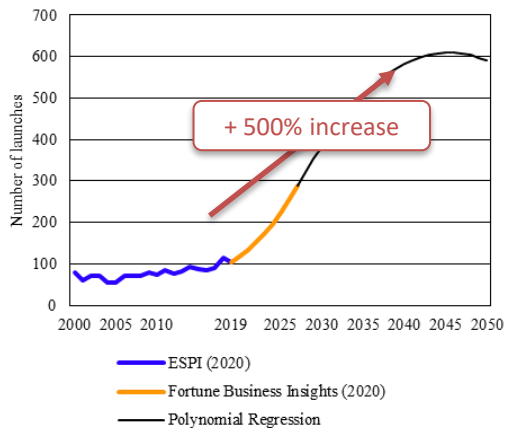
Ozone depletion  
(CFC-11 eq.)

**RP-1 and LCH<sub>4</sub> preferred options**  
**Uncertainty for UDMH**

# Forecast



## Launch rates



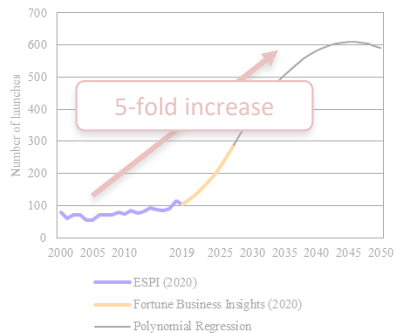
## Trends

Propellant	Influence	Source	Positive / Negative	Starting from	Strength of influence
RP-1	Planned upscaling of the launch rate of the Falcon 9 and Falcon Heavy.	FAA (2020)	Positive	2020	2
	Introduction of Irtysh and Angara	ITAR-TASS News Agency (2019a, 2019b); Kyle (n.d. -b)	Positive	2020	2
	Introduction of the Long March 9	Andrew Jones (2018); Xinhua News Agency (2018)	Positive	2028	
LH <sub>2</sub>	Cheap and easy to produce	Dallas et al. (2020)	Positive	2020	1
	Introduction of the SLS	FAA (2018; 2020)	Positive	2021	2
LCH <sub>4</sub>	Introduction of the Long March 9	Andrew Jones (2018); Xinhua News Agency (2018)	Positive	2028	2
	High I <sub>sp</sub> makes it attractive to use in the future	Dallas et al. (2020)	Positive	2028	1
UDMH	Introduction of the Starship	Henry (2019)	Positive	2021	3
	Introduction of New Glenn	Blue Origin (n.d.)	Positive	2021	2
	High I <sub>sp</sub> , cheap	Pettersen et al. (2016)	Positive	2020	1
APCP	Replacement of the Proton, Rockot until 2030	Harvey (2019)	Negative	2020	-9
	Phasing out of UDMH fuelled Chinese until 2030	Harvey (2019)	Negative	2020	-9
APCP	Industry trend towards green propellants	Gohardani et al. (2014)	Negative	2020	-2
	Introduction of the SLS and Ariane 6	ESA (n.d.); FAA (2018); Foust (2020)	Positive	2020	2
	APCP is suitable for small launchers	Dallas et al. (2020)	Negative	2020	-1

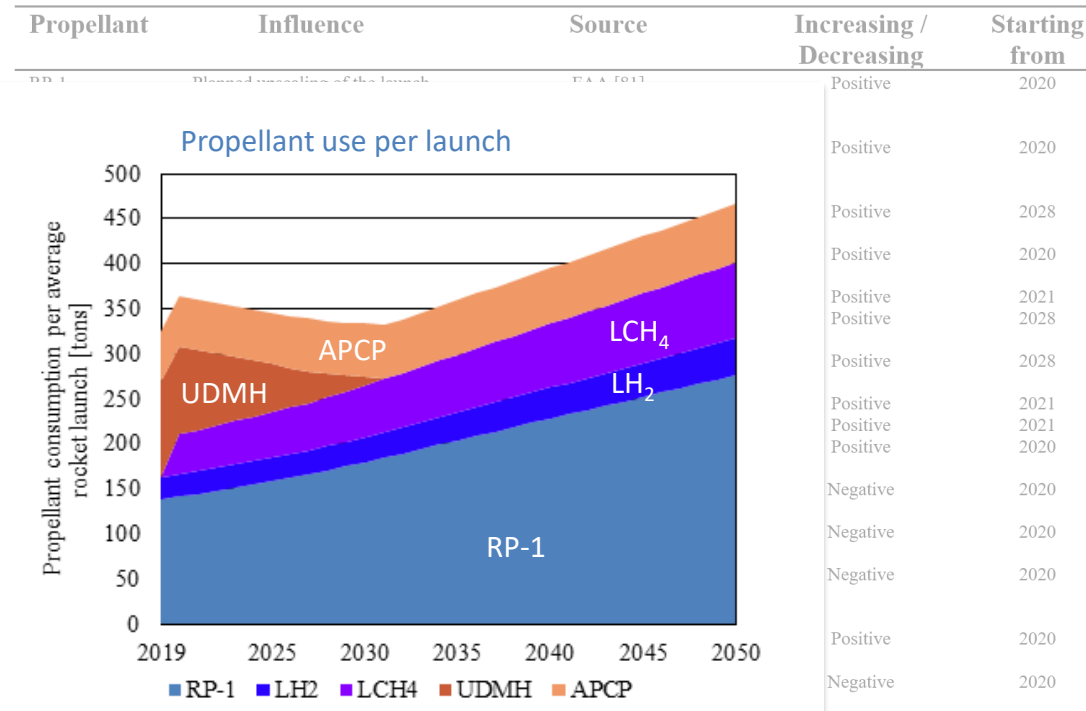
# Forecast



### Launch rates

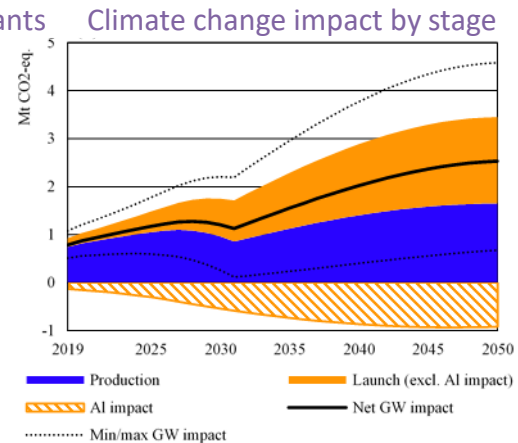
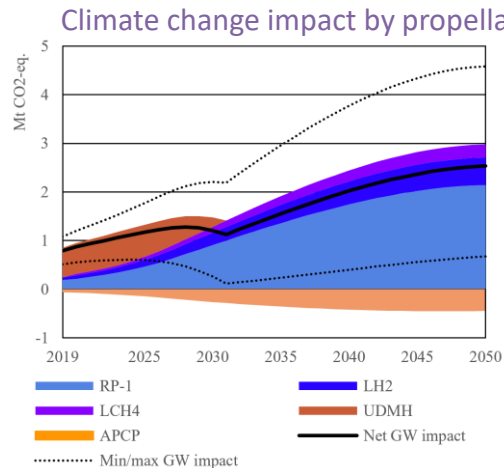
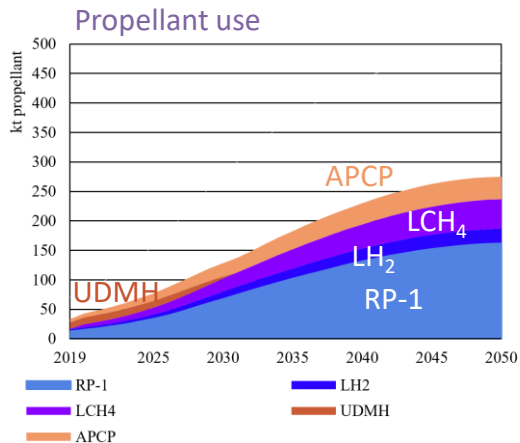


### Trends



Climate change  
(CO<sub>2</sub> eq.)

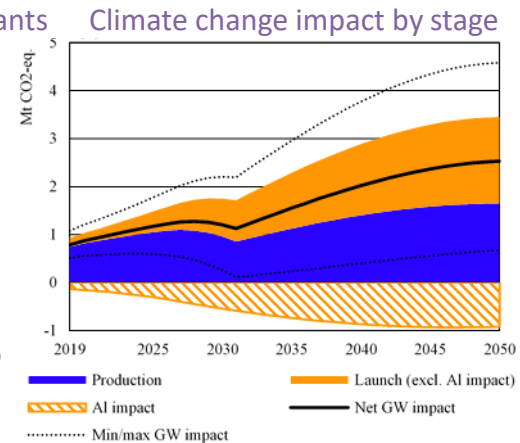
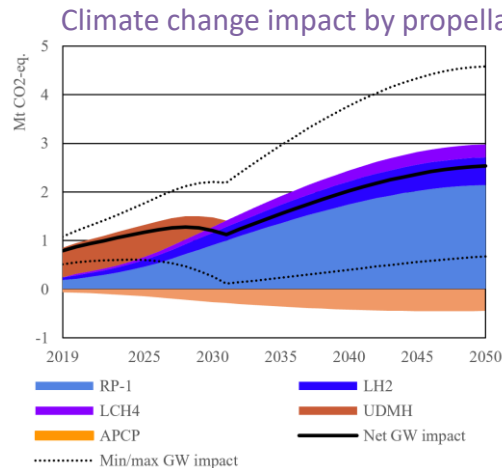
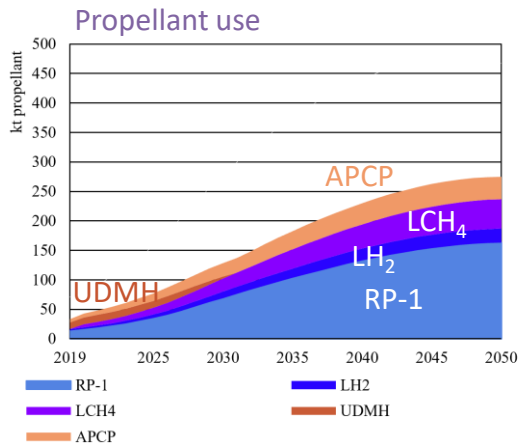
Slightly more than  
doubled towards  
2050





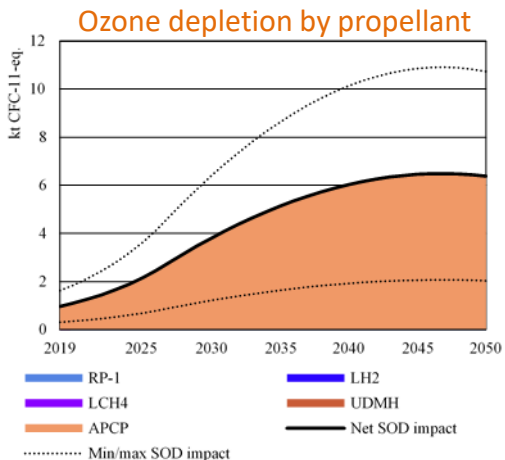
Climate change  
(CO<sub>2</sub> eq.)

Slightly more than  
doubled towards  
2050



Ozone depletion  
(CFC-11 eq.)

More than 5-fold  
increase towards  
2050



# Conclusions



- Large variation in climate change and ozone depletion performance
- Some propellants stand out as better candidates
- Life cycle perspective is necessary to evaluate the alternatives
- Stabilization of global impacts from rocket launches requires transitioning towards propellants with lower life cycle impacts

# Limitations



This represents a **first** attempt at consistent life cycle assessment of propellants

- Launch stage emissions are estimated from simulation: could be improved or validated
- We have estimated ODP factors for launch rate emissions, especially chlorines
- We have adopted  $GWP_{100}$  factors for the range of emissions
- We have assumed that alumina particulates have a cooling effect (uncertain)
- Emissions were assumed to be emitted at ground level due to methodological constraints

# Abstract



## Life cycle of propellants: environmental benchmark of current "green" propellants

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*European Space Agency (ESA) has previously established life cycle data for life stages up to launch. In this work we extend the previous ESA LCA data and present complete life cycle assessment of several current propellants, including propellant chemical production, loading and launch stage emissions with impacts to climate change and ozone depletion. CEARUN was used to estimate launch stage emissions. The life cycle performance of RP-1/LOx, LH2/LOx, CH4/LOx, UDMH/NTO and solid ammonium perchlorate composite propellant (APCP) is benchmarked per specific impulse.*

*Results clearly show the importance of including emissions both before and during launch, e.g., production stage emissions dominate for climate change emissions from hydrogen and UDMH, and launch emissions from APCP overrule any other contribution to ozone depletion. Some of the propellants carry climate cooling effects through emissions of reflective particulates, while others contribute to increased radiative forcing by emission of black carbon.*

*We make emission forecasts from global launch rates towards 2050, to project climate change emissions (GWP100) and ozone depletion, and findings from these underline the importance of the ongoing shift towards certain propellants. We conclude that, under some conditions, hydrogen and methane appear good candidates for the future. Results have been submitted to a relevant journal (in prep.)*