

ESA'S CLEAN SPACE INDUSTRY DAYS (2021 CSID)

ELECTRICAL PASSIVATION - PCDU UPGRADE FOR POWER PASSIVATION

TAS-B TECHNOLOGICAL TESTS RESULTS ON COMPONENTS MOUNTING

Date: 20/09/2021

Ref: PCDU-CLEAN-TASB-PPT-0044

Template: 83230347-DOC-TAS-EN-010

PROPRIETARY INFORMATION

This document is not to be reproduced, modified, adapted, published, translated in any material form in whole or in part nor disclosed to any third party without the prior written permission of Thales Alenia Space.
© 2021 Thales Alenia Space All right reserved

THALES ALENIA SPACE LIMITED DISTRIBUTION

TABLE OF CONTENTS

1 Scope & Objectives

2 Main requirements

3 Trade-off solution

4 Technological test evaluation results

5 Conclusion

SCOPE AND OBJECTIVES

/// Objective : Isolation of the solar arrays embedded in the PCDU

- Identification of the components involved in the isolation of the solar array during the disposal phase.
- Manufacturing of technological breadboards and technological test evaluations

MAIN REQUIREMENTS

A2. Specific requirements

A2-010	The isolation of the solar array shall be designed to be implemented within the Power Conditioning (and Distribution) Unit.
A2-020	The isolation of solar array shall remain active even in case of a main power bus powered down to 0V as by a depleted battery.
A2-030	The isolation of solar array shall provide a SA passivation capability by short-circuiting or open-circuiting all SA sections, so that SA power is no more transferred to the main bus and battery charge becomes impossible.
A2-040	Two fully independent commands shall be used for passivation (for example arming command and firing command). At least one of the commands shall be a direct command from ground (HV-HPC as per AD10).

MAIN REQUIREMENTS

A2-240	<p>During the satellite operational phase the following temperature range at TRP shall be considered:</p> <table border="1" data-bbox="454 285 1580 445"> <thead> <tr> <th data-bbox="454 285 807 347">°C</th> <th data-bbox="807 285 962 347">Min Op</th> <th data-bbox="962 285 1116 347">Max Op</th> <th data-bbox="1116 285 1271 347">Min Non-Op</th> <th data-bbox="1271 285 1425 347">Max Non-Op</th> <th data-bbox="1425 285 1580 347">Min Start up</th> </tr> </thead> <tbody> <tr> <td data-bbox="454 347 807 380">Design Temperature limits</td> <td data-bbox="807 347 962 380">-20</td> <td data-bbox="962 347 1116 380">+50</td> <td data-bbox="1116 347 1271 380">-30</td> <td data-bbox="1271 347 1425 380">+60</td> <td data-bbox="1425 347 1580 380">-30</td> </tr> <tr> <td data-bbox="454 380 807 412">Acceptance Temperature</td> <td data-bbox="807 380 962 412">-25</td> <td data-bbox="962 380 1116 412">+55</td> <td data-bbox="1116 380 1271 412">-35</td> <td data-bbox="1271 380 1425 412">+65</td> <td data-bbox="1425 380 1580 412">-30</td> </tr> <tr> <td data-bbox="454 412 807 445">Qualification Temperature</td> <td data-bbox="807 412 962 445">-30</td> <td data-bbox="962 412 1116 445">+60</td> <td data-bbox="1116 412 1271 445">-40</td> <td data-bbox="1271 412 1425 445">+70</td> <td data-bbox="1425 412 1580 445">-30</td> </tr> </tbody> </table>	°C	Min Op	Max Op	Min Non-Op	Max Non-Op	Min Start up	Design Temperature limits	-20	+50	-30	+60	-30	Acceptance Temperature	-25	+55	-35	+65	-30	Qualification Temperature	-30	+60	-40	+70	-30
°C	Min Op	Max Op	Min Non-Op	Max Non-Op	Min Start up																				
Design Temperature limits	-20	+50	-30	+60	-30																				
Acceptance Temperature	-25	+55	-35	+65	-30																				
Qualification Temperature	-30	+60	-40	+70	-30																				
A2-250	During the satellite operational phase the solution shall be able to withstand 15 thermal cycles per day between 20°C and 40°C.																								
A2-260	During the disposal phase the solution shall be able of keeping the SA isolation considering a temperature range of -50 °C to 80°C.																								
A2-270	During the disposal phase the solution shall be able to withstand 15 thermal cycles per day between 60°C and 80°C.																								
A2-280	The passivation function shall be testable during the AIT phase at satellite level without the use of any specific test command.																								

TRADE-OFF

Trade-off results for Isolation of the solar array in the PCDU:

- S3R : shunting solution
- S2R : opening solution
- MPPT : opening/shunting solution

50V (un)regulated bus platform		Trade-off parameters						Results
		Passivation reliability	Mission reliability	Market coverage	Mass	Price	Efficiency	
Weight		10	40	10	15	15	10	
S3R	Commun Relay Shunting	9	10	8	0	3	10	716
	Shunting Relay (1 contact per section)	10	10	6	2	4	10	743
	Series Relay	10	7	10	0	2	6	563
	Relays passivation main bus regulation	5	7	2	9	3	10	637
	MOSFET D-G S/C	4	7	3	9	6	10	680
	Serial Electronic Switch	0	4	10	8	0	0	374
S2R	Serial Relay	10	7	10	4	6	4	658
	Serial Electronic switch	3	10	10	9	5	10	828
MPPT	Serial Relay	10	7	10	8	10	0	740
	Common SA shunt	10	10	8	8	10	10	940
	Electronic input switch	9	10	10	10	9	10	970
	1 buck per SA section	9	10	3	10	9	10	900
	Galvanic Insulation	6	5	3	0	0	0	288
Passivation module		10	10	10	0	0	10	700

Table 65 - Tradeoff summary for 50V (un)regulated bus platform

TECHNOLOGICAL TESTS – TESTED COMPONENTS

/// List of packages submitted to technology evaluation

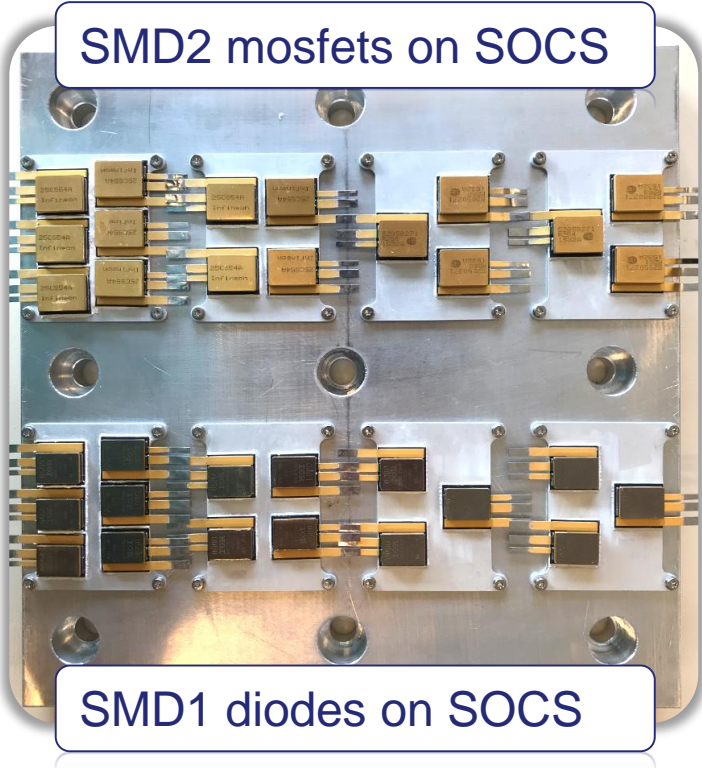
- / SMD2 MOSFET
- / SMD05 MOSFET
- / SMD1 DIODE
- / TO258 DIODE
- / POWER RELAYS
- / HIGH POWER TRANSFORMER
- / LOW LEVEL RELAY
- / FP14 COMPARATOR

PROPRIETARY INFORMATION

This document is not to be reproduced, modified, adapted, published, translated in any material form in whole or in part nor disclosed to any third party without the prior written permission of Thales Alenia Space.
© 2021 Thales Alenia Space All right reserved

TECHNOLOGICAL TESTS - TESTS VEHICLES

SMD2 mosfets on SOCS



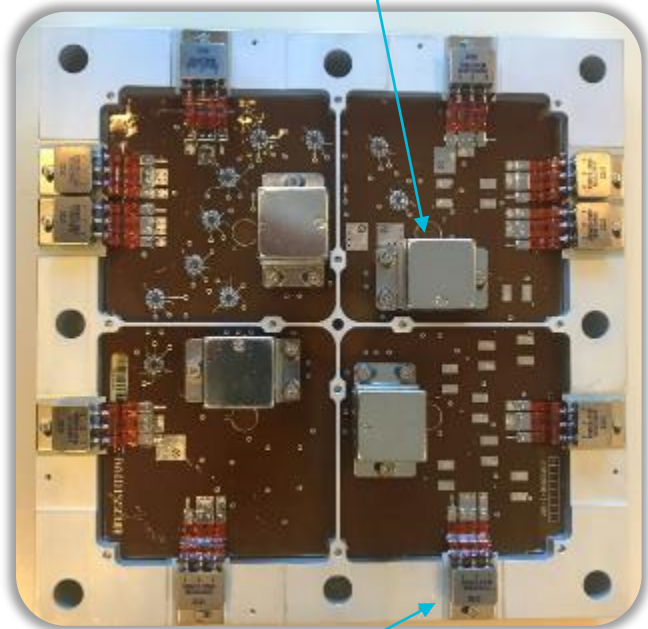
SMD05 mosfets on SOCS



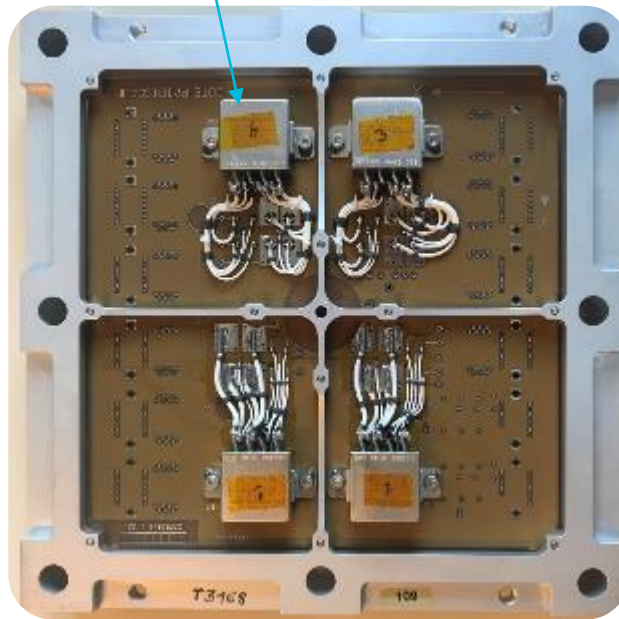
SMD1 diodes on SOCS

TECHNOLOGICAL TESTS - TESTS VEHICLES

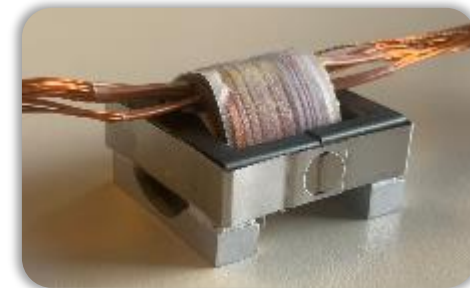
EL415 relays



EL215 relays



Power transformer



TO258 diodes

TECHNOLOGICAL TESTS - TESTS DESCRIPTIONS

/// Mechanical tests

/ VIBRATIONS

- High level sine vibrations
 - Random vibrations

/ SHOCKS

/// Thermal cycling tests file

/ MISSION PHASE

/ DISPOSAL PHASE

Electrical monitoring performed on all components during all the thermal cycling with low current (100 mA)

Mission phase :

	Components on PCB	Components on SOCS
Temperature range	-55 °C / +110 °C	-55 °C / +110 °C
Number of cycles	1337	2241

Disposal phase :

	Components on PCB	Components on SOCS
Temperature range	-55 °C / +110 °C	-55 °C / +110 °C
Number of cycles	1217	2050

/// Power cycling tests file

/ MISSION PHASE

/ DISPOSAL PHASE

Electrical monitoring performed with specific test set-up depending of mission or disposal phase

Parameters :

- SA current
- VOC
- VBus

All the tests were performed in order to cover mission and disposal phase needs in term of ageing, electrical functionality, mechanical and thermal stress

TECHNOLOGICAL TESTS - TESTS RESULTS

/// Power components

	Component to evaluate	Mounting	Quantity & quality definition	result
Power relay	EL415	on PCB	4 : thermal cycling, EM grade 4 power cycling, grade 3+	Succeed
	EL215	on PCB	4 : thermal cycling 4 : power cycling	Succeed
Power diode	SMD1	on SOCS	10 : thermal cycling, grade 1 6 : power cycling, grade 1	Succeed Succeed
	TO-258	on PCB (mechanical structure + flex)	10 : thermal cycling, grade 1 6 : power cycling, grade 1	Succeed Succeed
Electronic switch	SMD2	on SOCS	10 : thermal cycling, EM grade 6 : power cycling, grade 1	Succeed Succeed
	SMD05	on SOCS	10 : thermal cycling, grade 1 6 : power cycling, grade 1	Failed Results under analysis
Galvanic isolation	ETD transformer	on mechanical structure	2	Succeed
			2	

CONCLUSION

/// All the technological test evaluation are done successfully except SMD05 mosfet.

PROPRIETARY INFORMATION

This document is not to be reproduced, modified, adapted, published, translated in any material form in whole or in part nor disclosed to any third party without the prior written permission of Thales Alenia Space.
© 2021 Thales Alenia Space All right reserved