

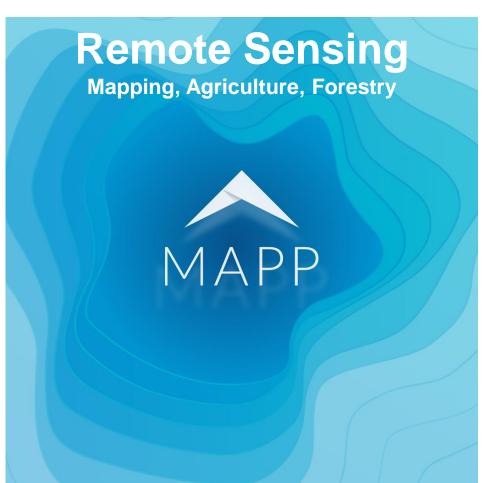
# CAMERA-LIDAR HAZARD DETECTION AND AVOIDANCE SYSTEM: FLIGHT TEST RESULTS

13<sup>th</sup> ESA Workshop on Avionics, Data, Control & Software Systems (ADCSS 2019)

November 14, 2019

SPIN.WORKS, S.A.





### **REMOTE SENSING**

# **MAPP.IT**



# **Satellite**

Multispectral Image 10 m/pixel update every 5 days global

# **Online App**

Algorithm generated insights for farmers to take action without expert knowledge

# **Drone**

Multispectral Image
5 cm/pixel
on-demand
3rd party local operations

S20: Fixed wing 2kg, 2h



**S250**: Fixed wing 25kg, 10h



**Platforms** 

**AVERT**: Rotating, 25kg



VTOL: Fixed wing 2kg



**UP:** nano satellite





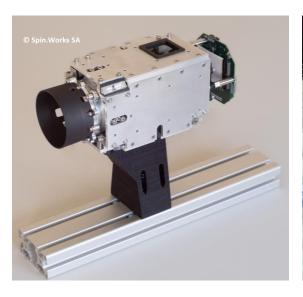


### **PAYLOAD**



**INFANTE:** CubeSat camera

**LITMUS:** micro LIDAR





### **EARTH OBSERVATION - PROCESSING - DATA**



AINGEO: hi-res geo (ESA)



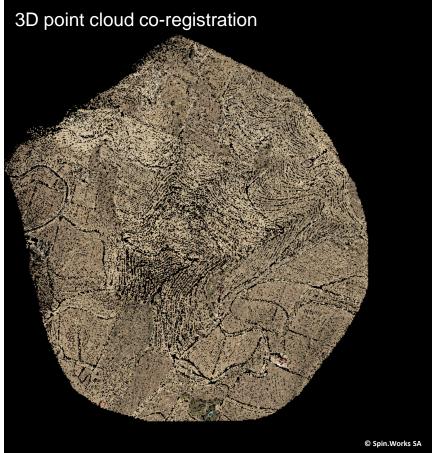
Data

**CORDS:** co-register (ESA)



ML: Machine Learning





# HAZARD DETECTION AND AVOIDANCE

# **HAZARD DETECTION AND AVOIDANCE (HDA)**

# **Hazard Detection and Avoidance (HDA)**

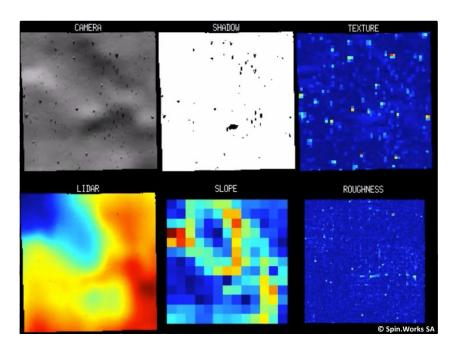
-Key technology for autonomous Safe Precision Landing

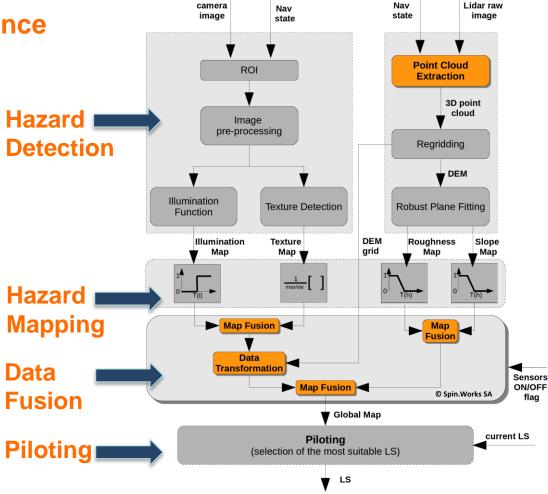
# **HDA System Design Challenge:**

- -Minimum sensor suite to enable Hazard Detection (mass, power, Hazard Detection)
- -Heterogenous Data Fusion (sensor data, information, etc.)
- -HDA strategy (trajectory, #HDA opportunities)
- -Onboard memory and processing power
- -HDA real time software safe site selection <10sec
- -Enable Safe Landing anytime anywhere (Moon, Mars, Minor bodies).

# HYBRID CAMERA-LIDAR HAZARD DETECTION AND AVOIDANCE SYSTEM (H2DAS)

Hybrid Hazard Detection & Avoidance (H<sup>2</sup>DAS)

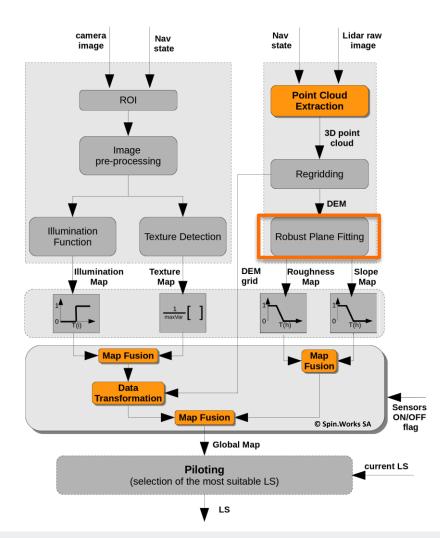




# HYBRID CAMERA-LIDAR HAZARD DETECTION AND AVOIDANCE SYSTEM (H2DAS)

# H<sup>2</sup>DAS real-time sw (FPGA accelerated)

- -RTEMS real-time operating system
- -Firm deadline of 10 seconds
- -Flight hardware with CPU clocked at 36MHz takes 16 seconds to execute all H<sup>2</sup>DAS tasks
- -H<sup>2</sup>DAS CPU+FPGA accelerated version returns a Safe Landing Site in 8 seconds
- -The hardware accelerated version uses a modified Sobel filter and a mean filter to extract the slopes and terrain roughness

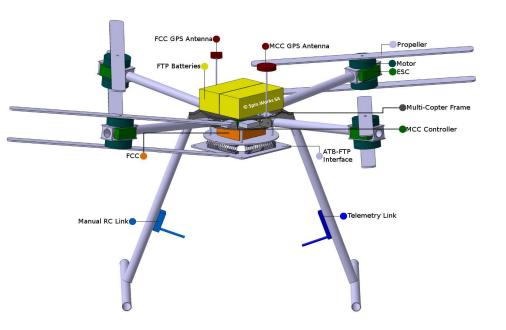


# **FLIGHT TEST PLATFORM**

### **FLIGHT TEST PLATFORM**

# **Platform Summary**

Dimensions [m]	Platform Weight [Kg]	Maximum Take-off Weight [Kg]		ATB Dimensions [cm]	ATB Weight [Kg]
1.9 x 1.9 x 0.78	13	24	9 / 13	33 x 33 x 29	6.8

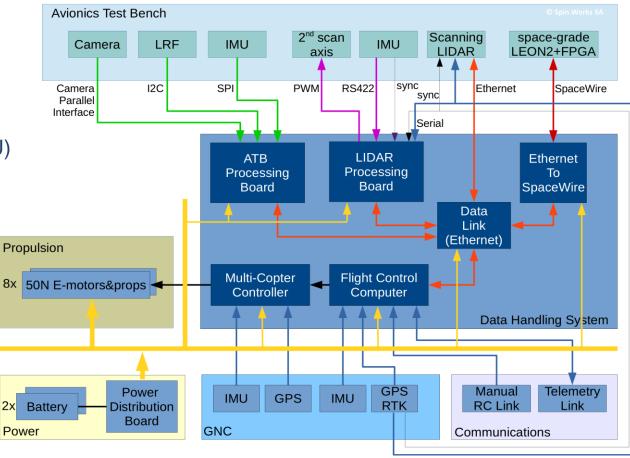




# **Avionics Architecture**

- -Flight Test Platform (FTP)
- -Avionics Test Bench (ATB)
  - -Camera
  - -Laser Range Finder (LRF)
  - -Inertial Measurement Unit (IMU)
  - -LiDAR System
  - -Space-grade LEON2+FPGA
  - -ATB Processing Board





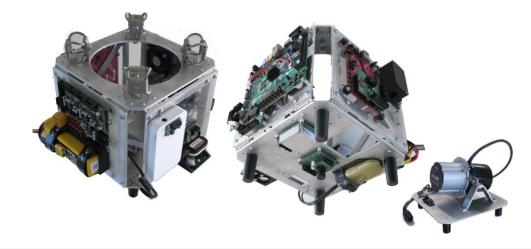
# **Avionics Test Bench (ATB)**

- -VN&HDA sensor suite (Camera, LIDAR, IMU, LRF)
- -Visual Based Navigation (VBN) Computer:
  - Zedboard
  - ARM Cortex-A9 hard-core CPU @ 665MHz
  - Zynq-7020 FPGA (from Artix family) @ 66MHz
- -HDA Computer:
  - Planetary Landing Descent Processing Unit (PLDPU)
  - LEON2 soft-core CPU @ 36MHhz
  - ProAsic3E FPGA @ 18MHz



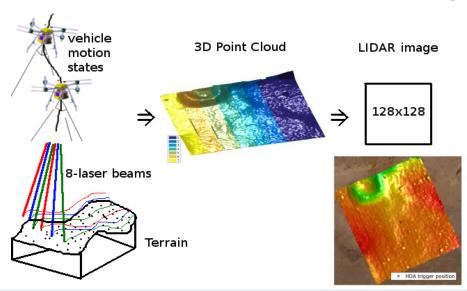






# **LIDAR Imaging System**

- -Quanergy M8 scanning LIDAR Laser Unit,
- -KVH 1725 high accuracy FOG IMU,
- -PYNQ (CPU+FGPA) LIDAR Processing Board,
- -2<sup>nd</sup> scanning axis mechanism (servo actuated)
- -GPS time&position solutions and PPS signal.





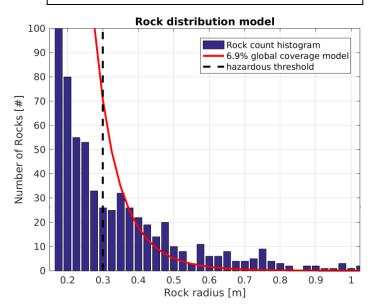
Specifications	Flight Tests		
Image size [pixels]	128 x 128		
Field of View [deg]	25		
Pixel angular res. [mrad]	3.4		
Range accuracy [cm]	10		
(3-sigma @ altitude)	@ 83m		
O	Scanning		
Sensor Technology	LIDAR		
Frame rate [Hz]	1/10s < x < 1/5s		

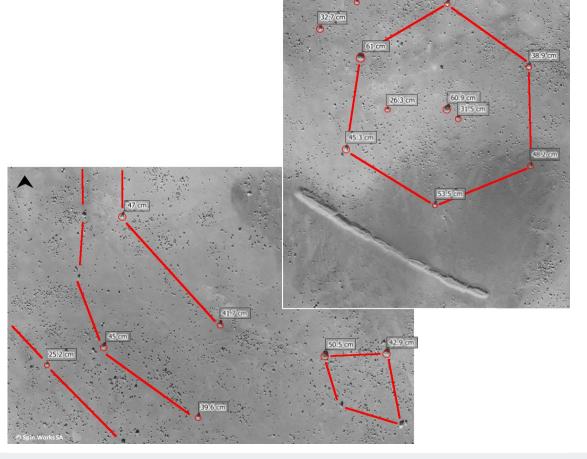
# TEST SITE (MARS ANALOGUE TERRAIN)

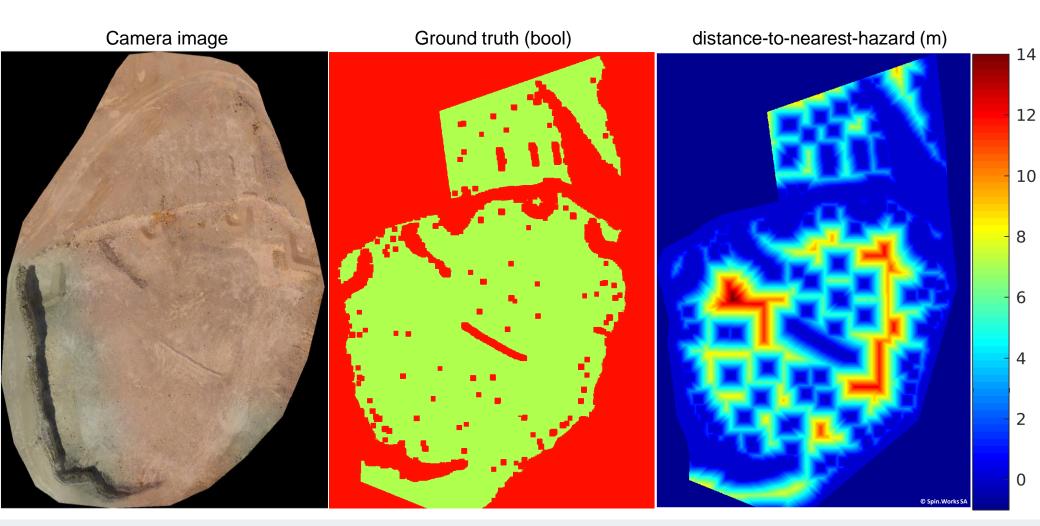




Safety rate	AVERT Test Site		
Slopes	88.80%		
Roughness	90.54%		
Illumination	99.33%		
Global Safety rate	79.09%		

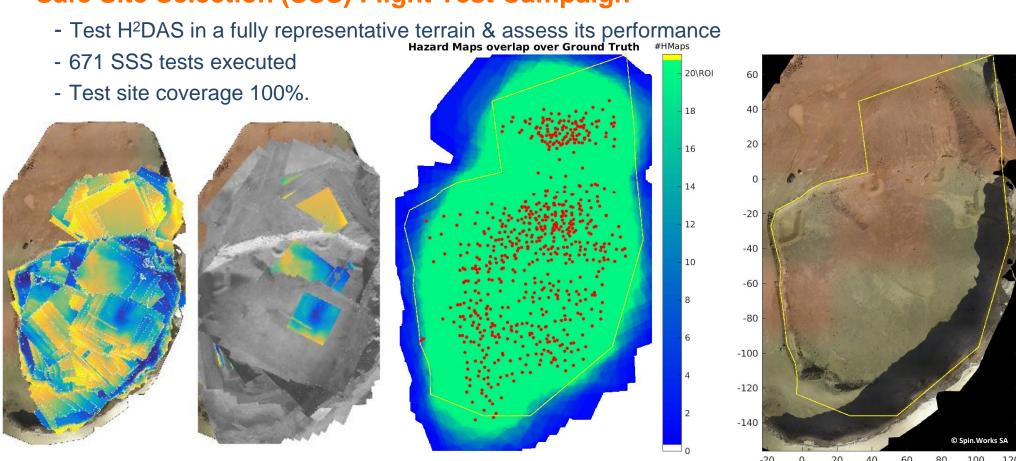






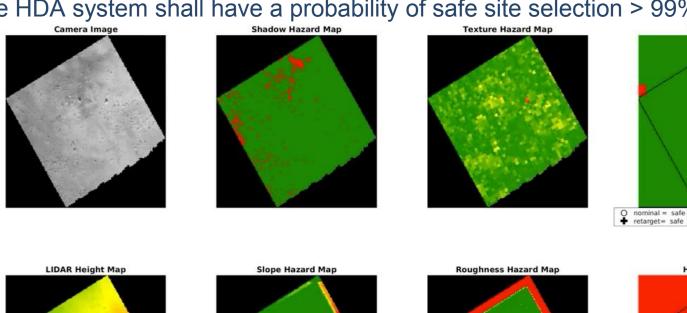
# AVERT SYSTEM SAFE SITE SELECTION FLIGHT TEST CAMPAIGN

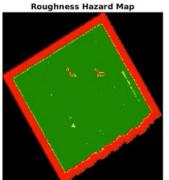
# Safe Site Selection (SSS) Flight Test Campaign

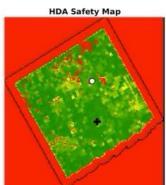


# Safe Site Selection (SSS) HDA Results

-"The HDA system shall have a probability of safe site selection > 99%"







**Ground Truth** 

# Safe Site Selection (SSS) HDA Results

- -"The HDA system shall have a probability of safe site selection > 99%"
- -671 SSS tests/data points flying at about 80m height AGL @ test site
- -SSS batch processed 2 times by the PLDPU (Camera+LIDAR & Camera only)

		H <sup>2</sup> DAS		
	LIDAR	ON	OFF	
Number of SSS points (# SSS)  Nominal safe blind landings (#safe nominal LS)  Required diverts (#unsafe nominal LS)		671		
		597 (89.0%)		
		74 (11.0%)		
HDA safe retargeting ( <b>PSSS</b> (#safe retargeting LS)	)	669 ( <b>99.7%</b> )	659 ( <b>98.2%</b> )	
HDA Miss-detections (#unsafe retargeting LS)		2 (0.3%)	12 (1.8%)	

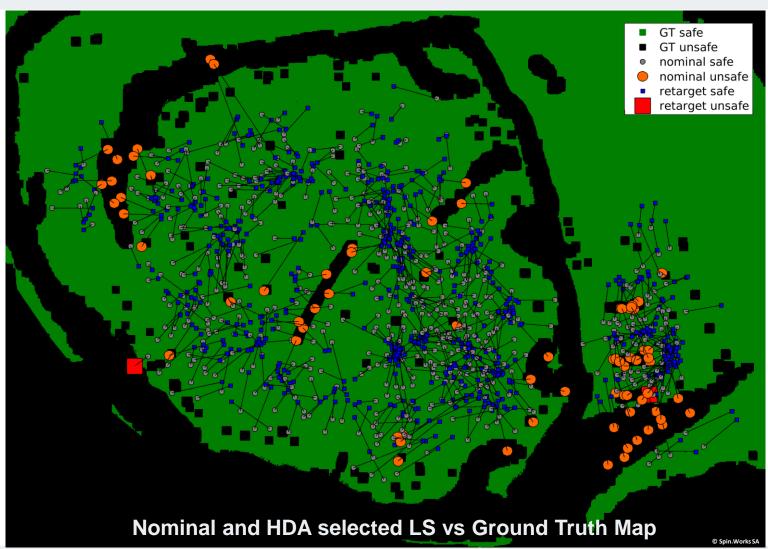
# Safe Site Selection (SSS) HDA Results

# **Confidence levels**

-H<sup>2</sup>DAS meets the target PSSS with confidence levels of at least 95%

HDA algorithm	H <sup>2</sup> DAS		
LIDAR		ON	OFF
Selected Landin			
Max. allowed prob	1%	10%	
Elight Tooto	HDA measured prob. of SSS	99.7%	98.2%
Flight Tests	Confidence level (CL) for Pfmax	96.4%	>99.9%
CII TDM (Forth)	HDA measured prob. of SSS	99.7%	90.5%
SIL TDM (Earth)	Confidence level (CL) for Pfmax	>99.9%	63.2%
SIL MPL (Mars)	HDA measured prob. of SSS	99.8%	89.8%
SIL WIFL (Wats)	Confidence level (CL) for Pfmax	>99.9%	41.8%

# SAFE SITE SELECTION FLIGHT TEST CAMPAIGN



### SAFE SITE SELECTION FLIGHT TEST CAMPAIGN

