

Avionics-Power interfaces standardisation

An exportable approach?

Power and Avionic Work Group*



Presented by

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16/10/2020

Outline

- Preamble
- Motivation
- Subjects and objectives
- WG
- Methods
- Work in progress
- Conclusions.....slide 22

Avionics-Power interfaces / PREAMBLE

Acronyms

- **ADHA** Advanced Data Handling Architecture parallel studies (RUAG/ADS, TAS-I)
- **DOD** Depth of Discharge
- **OBDH** On-board Data Handling (SubSystem)
- **EPS** Electrical Power SubSystem
- **MBSE** Model Based System Engineering
- **MPPT** Maximum Power Point Tracker
- **PCDU** Power Conditioning and Distribution Unit
- **S3R** Sequential Shunt Switching Regulator
- **SA** Solar Array
- **SADE** Solar Array Drive Electronics
- **SADM** Solar Array Drive Mechanism
- **SAR** Solar Array Regulator
- **SOC** State of Charge
- **SOH** State of Health

(Key) definitions

- **Observability of a system**

A system is said to be **observable** if its current state can be estimated using only the information from its outputs.

In other words, one can determine the behaviour of the entire system from the system's outputs.

- **Functional/logical view**

The perspective of a system that identifies its functionality *without* specifying the physical allocation of the relevant functions

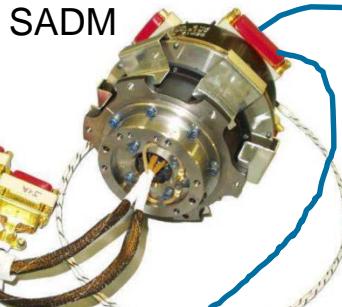
- **Physical view**

The perspective of a system that identifies its functionality *including* the physical allocation of the relevant functions

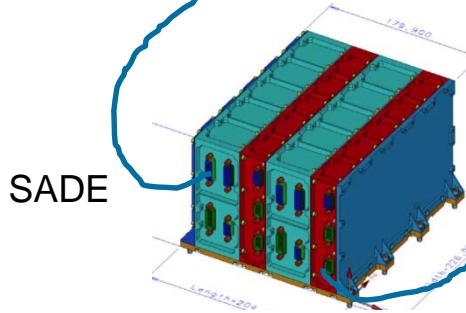
Electrical Power subsystem – a real view



Solar Array

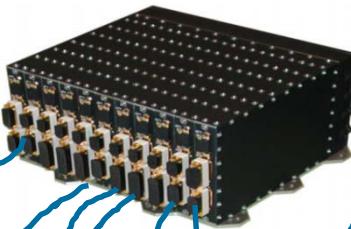


SADM

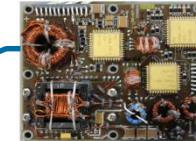


SADE

PCDU



Secondary DC/DC converters



Battery

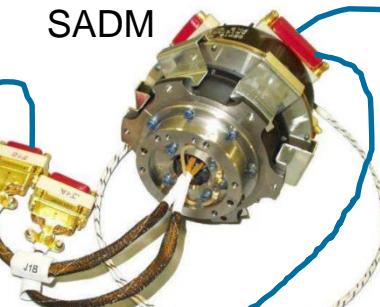


Electrical Power subsystem

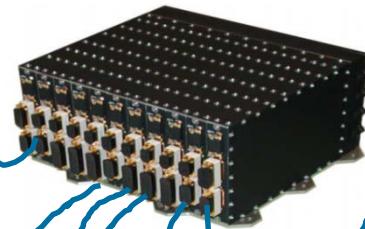


Solar Array

SADM



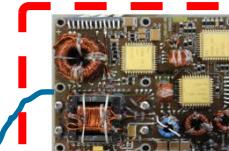
PCDU



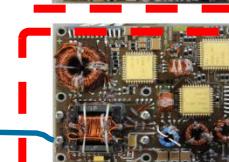
Secondary DC/DC converters



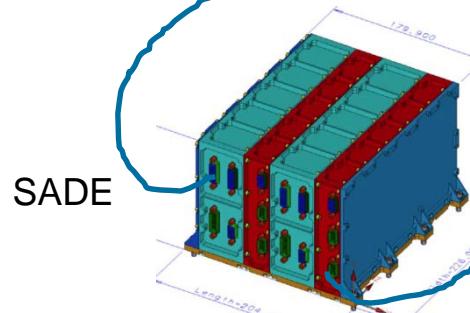
Equipment 1



Equipment 2



Equipment n



Battery



Electrical Power Subsystem (EPS)

Electrical Power Subsystem variability

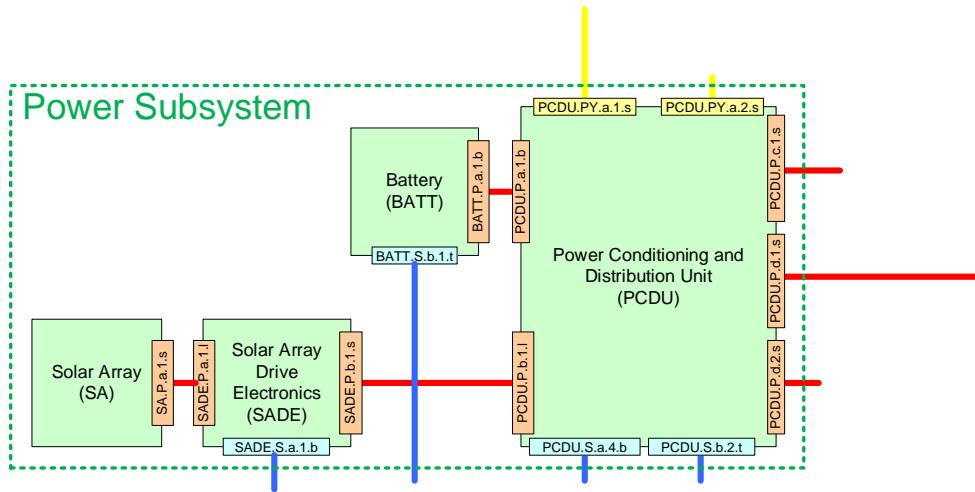
- Many variants, according to mission (environment, orbits)
 - Different solar arrays
 - Different batteries
 - Different PCDUs with different internal architectures
 - Regulated or unregulated bus
 - 28V, 50V, 100V regulated; around 28V or 50V unregulated (but not real standard)
 - Different or no SADM/SADE...
 - Different distribution lines (for essential, non essential equipment, heater lines, etc)

Power Subsystem Reference Architecture ... (to be continued)

- It makes sense to be as general as possible...

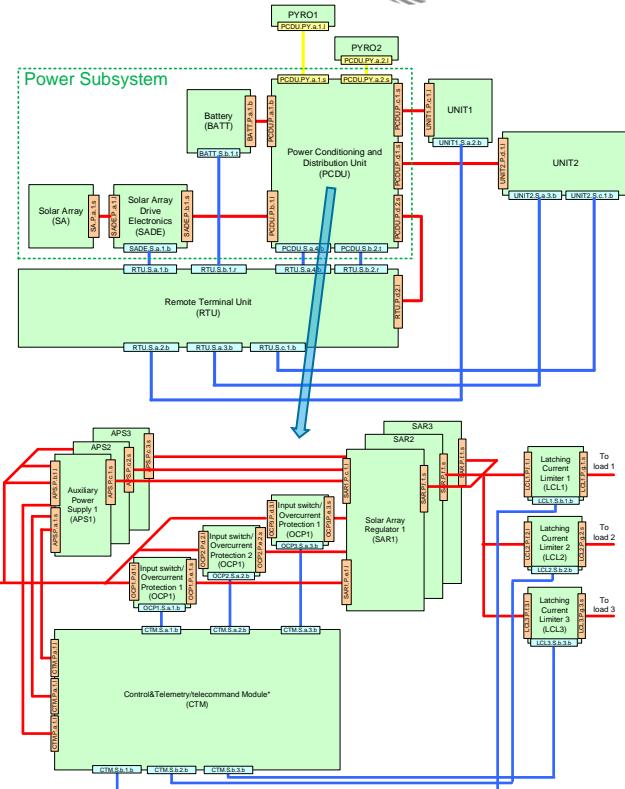
Legenda

	Element1 (E1)
	Power line
	Signal line
	Pyro line
	APS element, Interface P (power), interface type a, interface instance 1, source side s
	TMTC element, Interface S (signal), interface type b, interface instance 2, receiver side r
	PCDU element, Interface PY (pyro), interface type g, interface instance 3, load side l



=> Standardisation effort

- At **interface level**
- As a **first step** towards more recurrent products
- With a **realistic** attitude based on consensus
- Making treasure of previous successful experiences
- With the aim to be **MBSE ready!**



Avionics-Power interface / SUBJECTS AND OBJECTIVES

- Gives account of Power Subsystem **large variability**
- **Realistic**
 - Covering those interfaces that are actually **already** largely used in applications
 - No projection of too idealistic / too theoretical ideas
- **Opportunistic**
 - Covering those (mature) interfaces for which standardization seems possible
 - Considering that effective standardization is consensus based

Power Interfaces Standardisation – key topics

Failure Detection,
Isolation and
Recovery

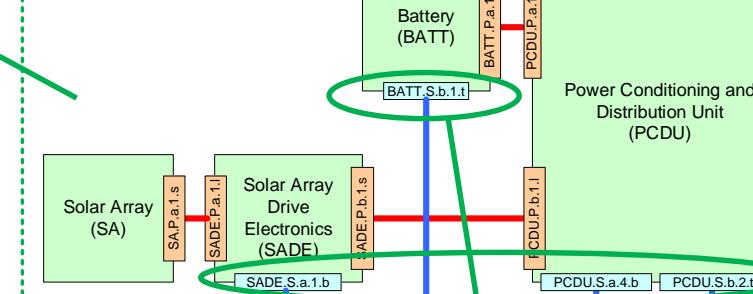
Actuators power
interfaces
(source and load)
ECSS-E-ST&HB-20-
21, 2019

Power
distribution
interfaces
(source &
load) based
on Latching
Current
Limiters
ECSS-E-
ST&HB-20-
20, 2016

Legenda

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 APS element, Interface P (power), interface type a, interface instance 1, source side s
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Power Subsystem



TM/TC interfaces

+
Secondary
power
distribution

Power interfaces and SAVOIR, Prerequisites

- Identification and agreement on electrical **power subsystem reference architectures**
- Cooperation with any standardisation activity concerning **back plane definition for avionic equipment** (ADHA, other)

Avionics-Power interfaces / (SAVOIR)WORKING GROUP



ToR agreed mid March 2020

Participation

ADS, N. Neugnot, J.. Seronie-Vivien

CNES, C. Elisabelair

DLR, N. Aksteiner

ESA, S. Landstroem, O. Mourra, F. Tonicello

OHB, J. Caudepon

RUAG, H. Myllymaki

SENER, C. Tato

TAS, G. Bouhours, J.L. Bolsee



7 Web-meetings performed so far

Reference architecture definition!

- Two step approach, **functional/logical** layer first, **physical** layer after that
- Most importance given to the precise correspondence of **verbal** definitions and assumptions with the relevant **reference block diagrams**

What do we cover?

which are the reference functions?

what are the reference functions doing?

How do we allocate (fully, partially) the ref. functions within the physical view?

Reference architecture definition!

- Two step approach, **functional/logical** layer first, **physical** layer after that
- Most importance given to the precise correspondence of **verbal** definitions and assumptions with the relevant **reference block diagrams**

What do we cover? What do we cover instantiation? How do we cover instantiation of the different physical views? What functions are doing? How do we locate (fully, partially) the ref. functions within the physical view?

the ref. functions which are doing?

1. Broad system/subsystem
functional/logical definition
2. Relevant (verbal) functions
functional/logical description
(as exhaustive as possible)

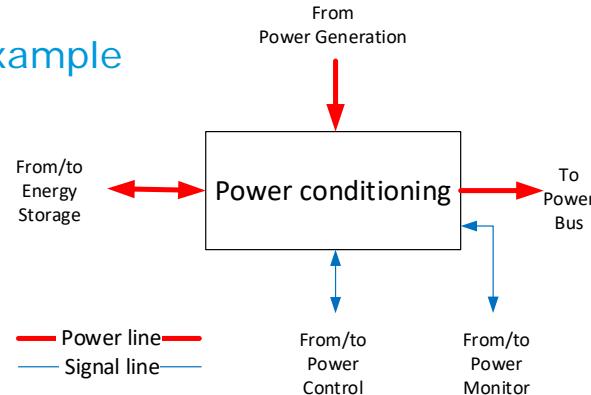
Example

Power conditioning

is the function of the power subsystem dedicated to the conversion of power coming from power generation, and from or to energy storage to the power bus. The power conditioning function is regulated by the power control function.

1. Broad system/subsystem functional/logical definition
2. Relevant (verbal) functions functional/logical description (as exhaustive as possible)
3. Block diagram ("twin") functional/logical function description

Example



The arrows indicate
the flow of power or
of the signal

ESA UNCLAS

Electrical Department | ESA-TEC-HO-020560 | ESTEC | 16/10/2020 | Slide 17



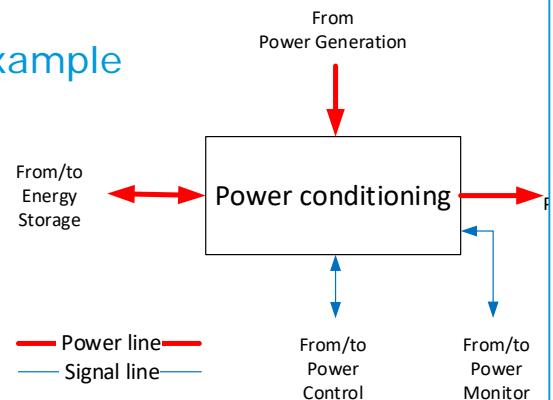
European Space Agency

1. Broad system/subsystem functional/logical definition

2. Relevant (verbal) functions functional/logical description (as exhaustive as possible)

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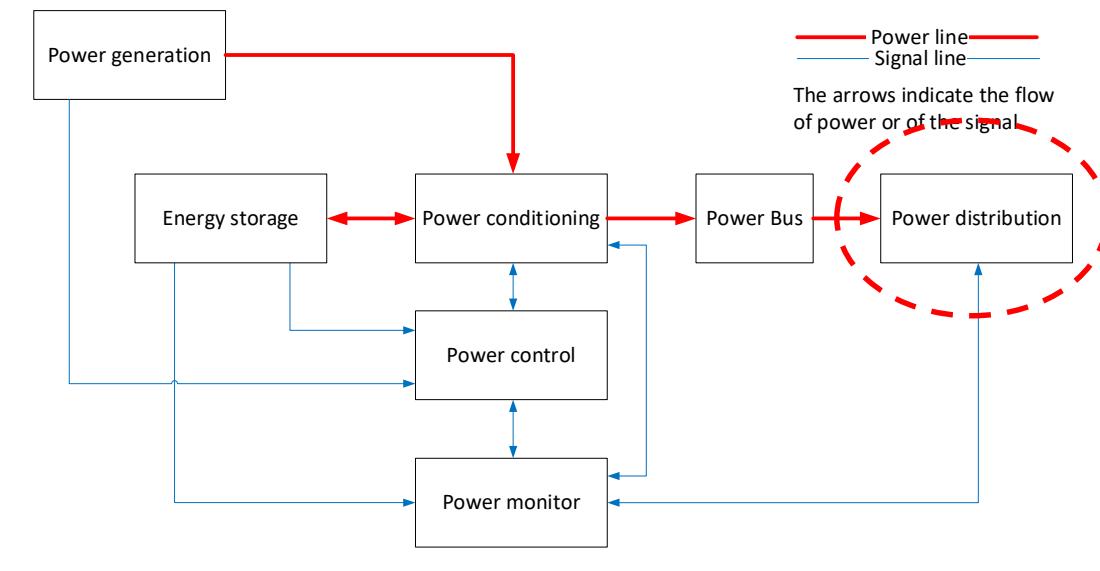
Example



The arrows indicate the flow of power or of the signal

4. Overall Block diagram ("twin") functional/logical function description

Example



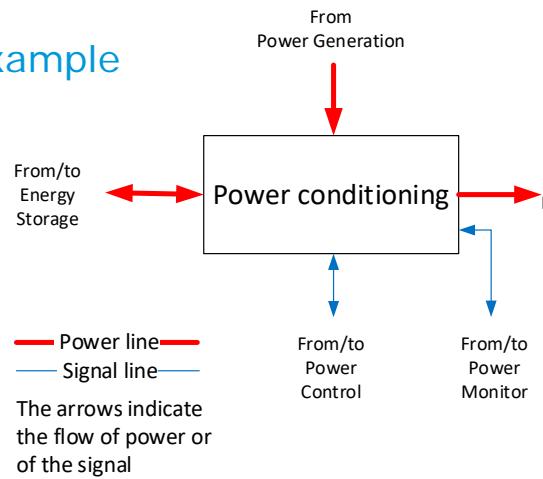
Power line
Signal line
The arrows indicate the flow of power or of the signal

1. Broad system/subsystem functional/logical definition

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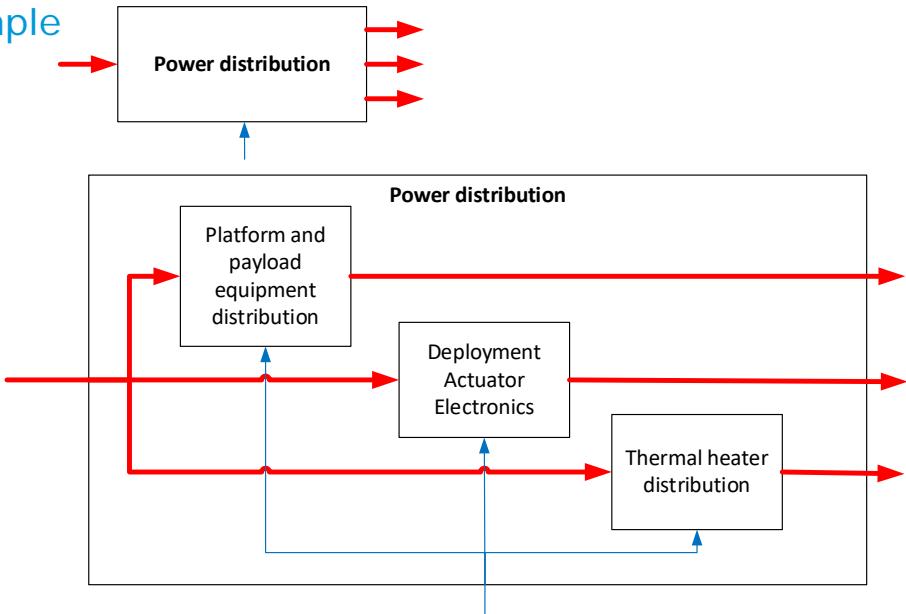
ESA UNCLAS



4. Overall Block diagram ("twin") functional/logical function description

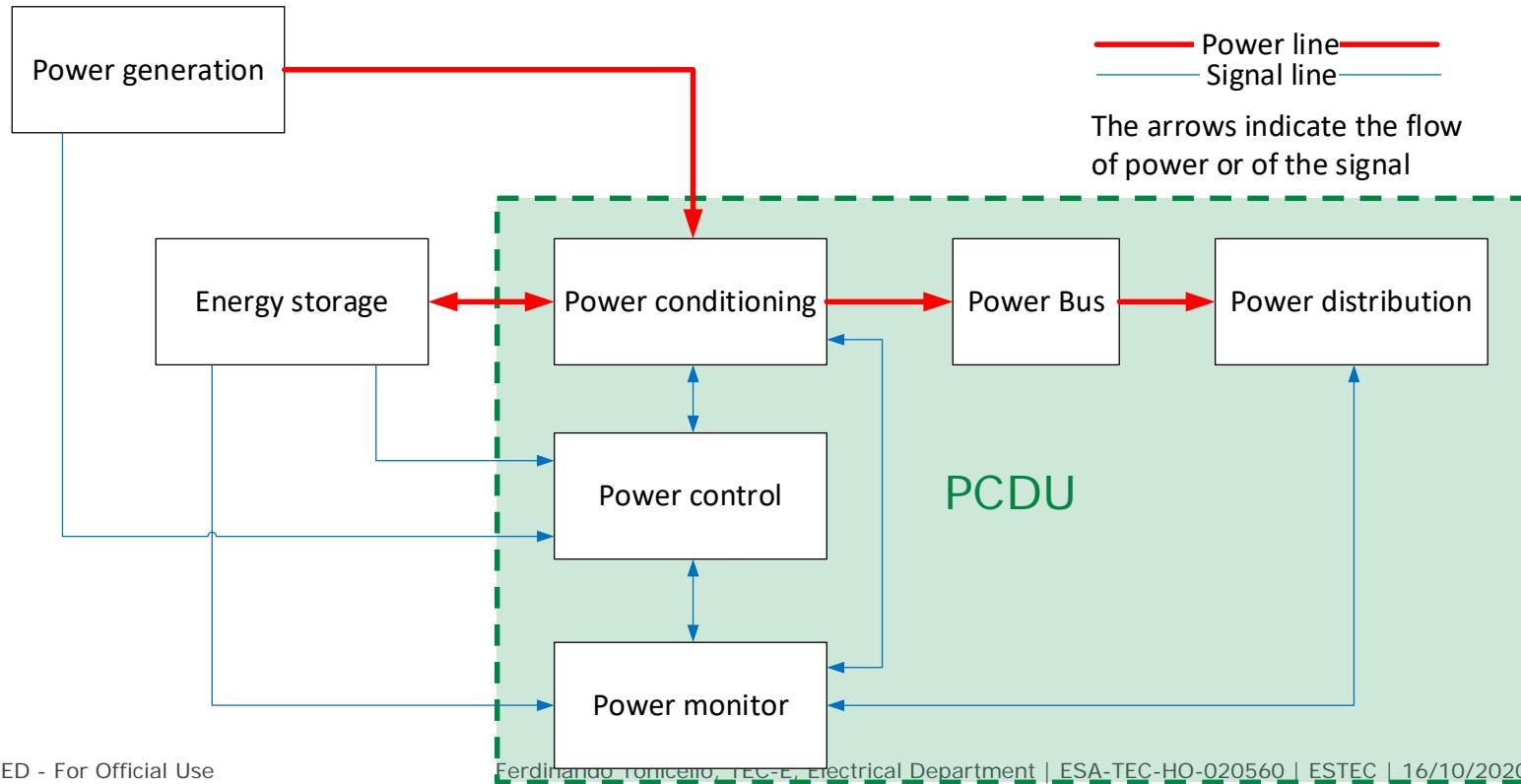
5. "Get inside the box", functional/logical function description

Example



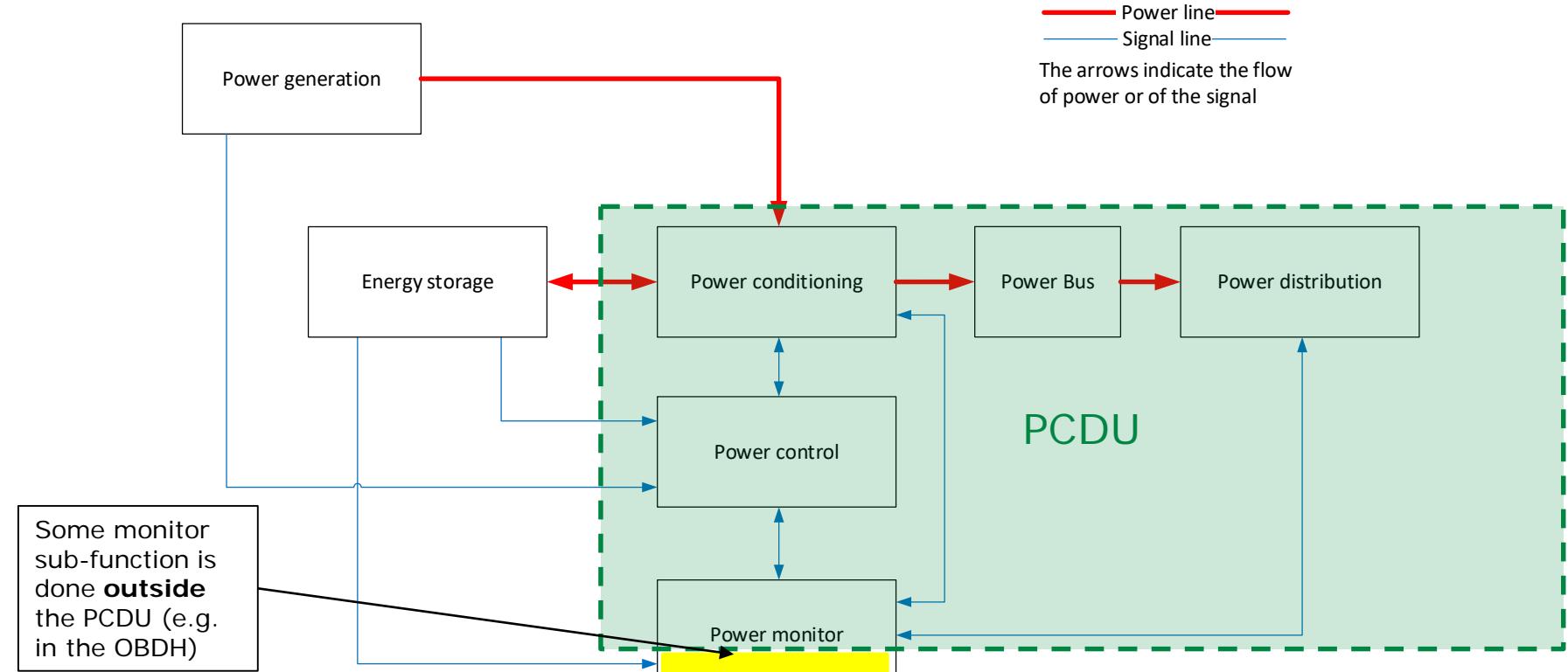
Avionics-Power interface / METHODS / in practice...

From functional/logical to physical block diagram / "allocation"



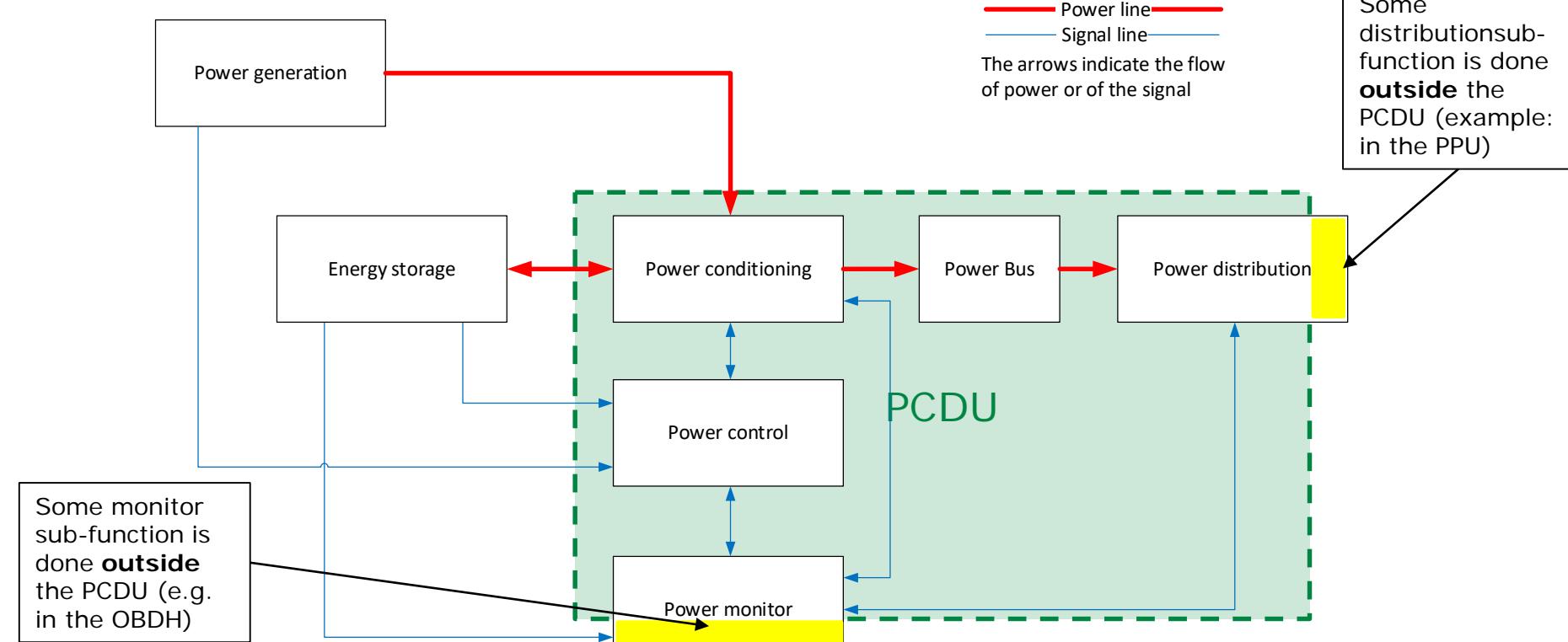
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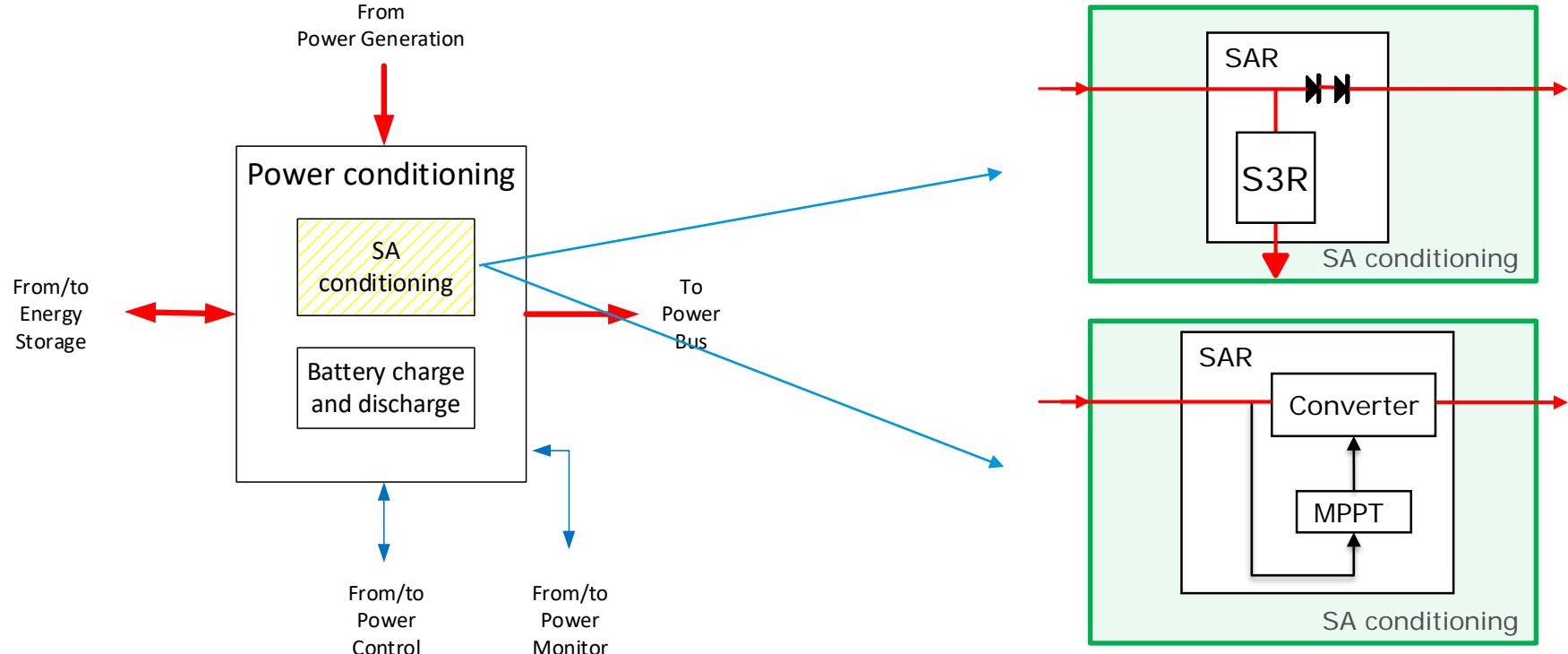


Avionics-Power interface / METHODS / in practice...

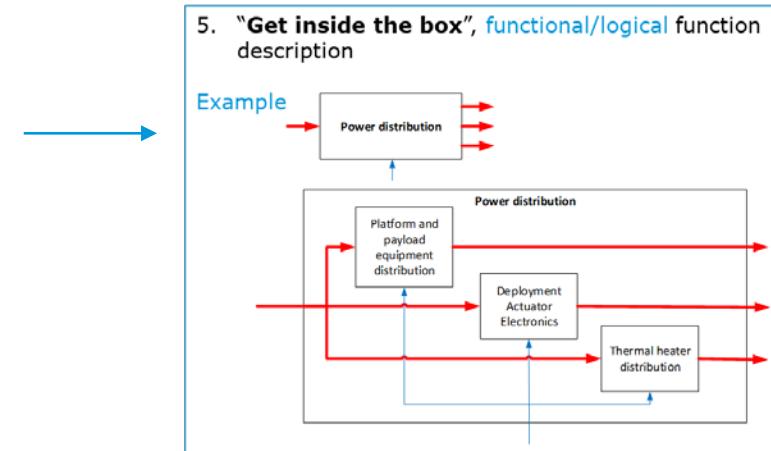
From functional/logical to physical block diagram / "allocation"



Instantiation of **physical** functions, where interfaces are impacted



- Reaching agreement on
 - Functional/logical EPS description, including both nominal and contingency aspects
 - The level of depth to go to (see "Get inside the box" , slide 15)
 - Possible instantiation options on EPS physical view
- Starting to identify the observability and commandability needs of all Primes/LSI and translate this need into standard interfaces where possible and convenient



Discussion on

EPS **observable** items, as an **input** for the **agreed telemetry set** (and the relevant interfaces, according to possible instantiation of the physical layer)

Observable items	Information available for Flight/Ground/AIT purposes	Processing	Accuracy
Battery DOD	Flight	second	±3% of DOD
Battery SOC	Flight	second	±5% of SOC
Battery SOH	Ground	month	±3% of SOH
Battery internal impedance (DC, AC)	Ground	hour	±10% of full scale
Battery temperature	Ground	minute	±2°C
SA instantaneous output power	Ground	second	±3% of full scale
SA available power (of possible)	Ground	second	±3% of full scale

**Primary focus on the functional requirements and
not in the design specification**

Avionics-Power interface / CONCLUSIONS



Too early to have some definitive conclusions, but...

1. We identified a **workable** way to have a **MBSE ready approach** to reference architecture! (see “METHOD” slides)
2. Whatever the **level of interface definition** we will achieve, it will be **agreed** by European Large System Integrators, European Agencies and by some of the most important power equipment manufacturers and users
3. This (interesting) work continues...