

MBSE 2022

A model based approach to budget management on the Earth Return Orbiter

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The Earth Return Orbiter

MBSE implementation within ERO

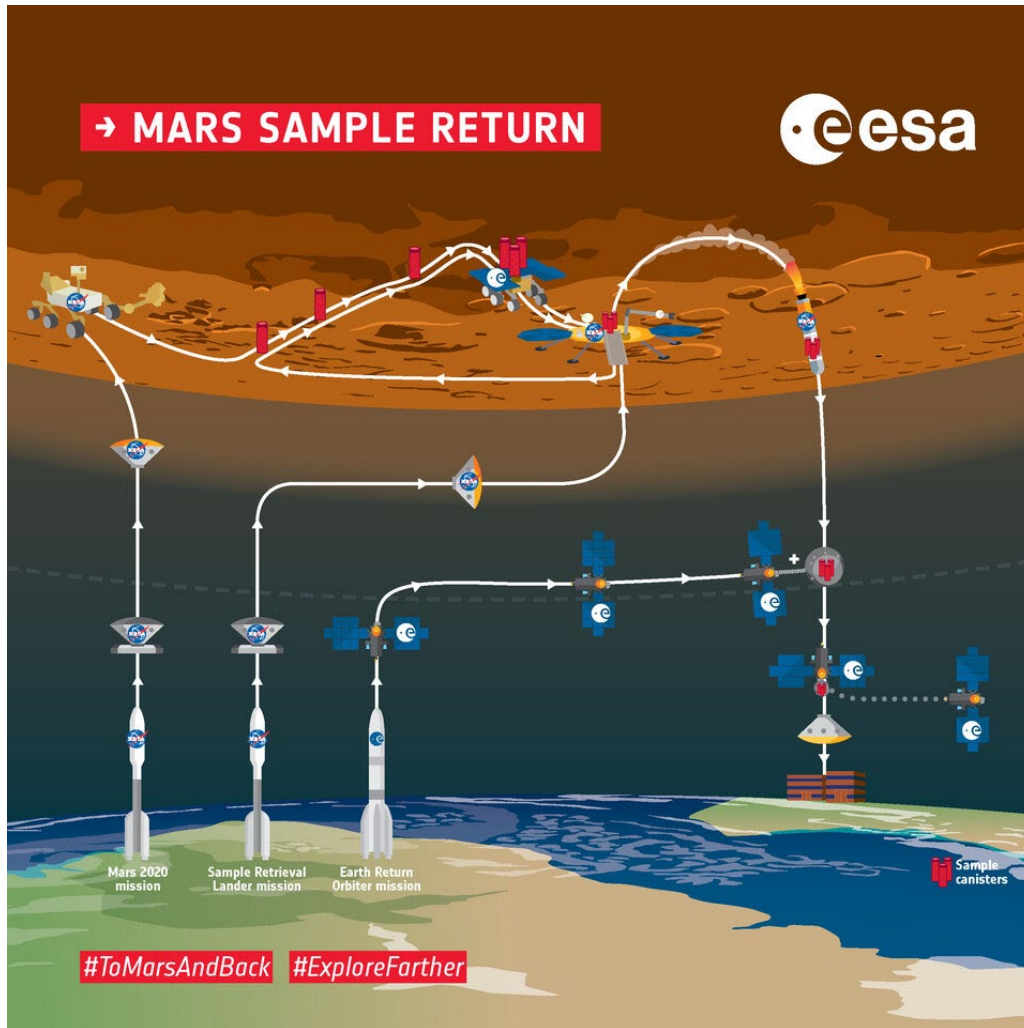
Budget management: MBSE

Budget management: Dashboard

Conclusion and Way forward

Q&A

The Earth Return Orbiter (ERO)



MSR Campaign goal

- Bring back Martian samples to be studied on Earth

ERO Mission goals

- Capture and contain samples in Martian Orbit
- Deliver samples to Earth

Organisation

- ESA's Human and Robotic Exploration Directorate (HRE)
- Airbus Defence and Space (ADS) as Prime contractor

Challenges

- Cross-agency program
- Complex technical challenges

Needs

- Robust systems engineering
- Effective communication between stakeholders and all disciplines involved

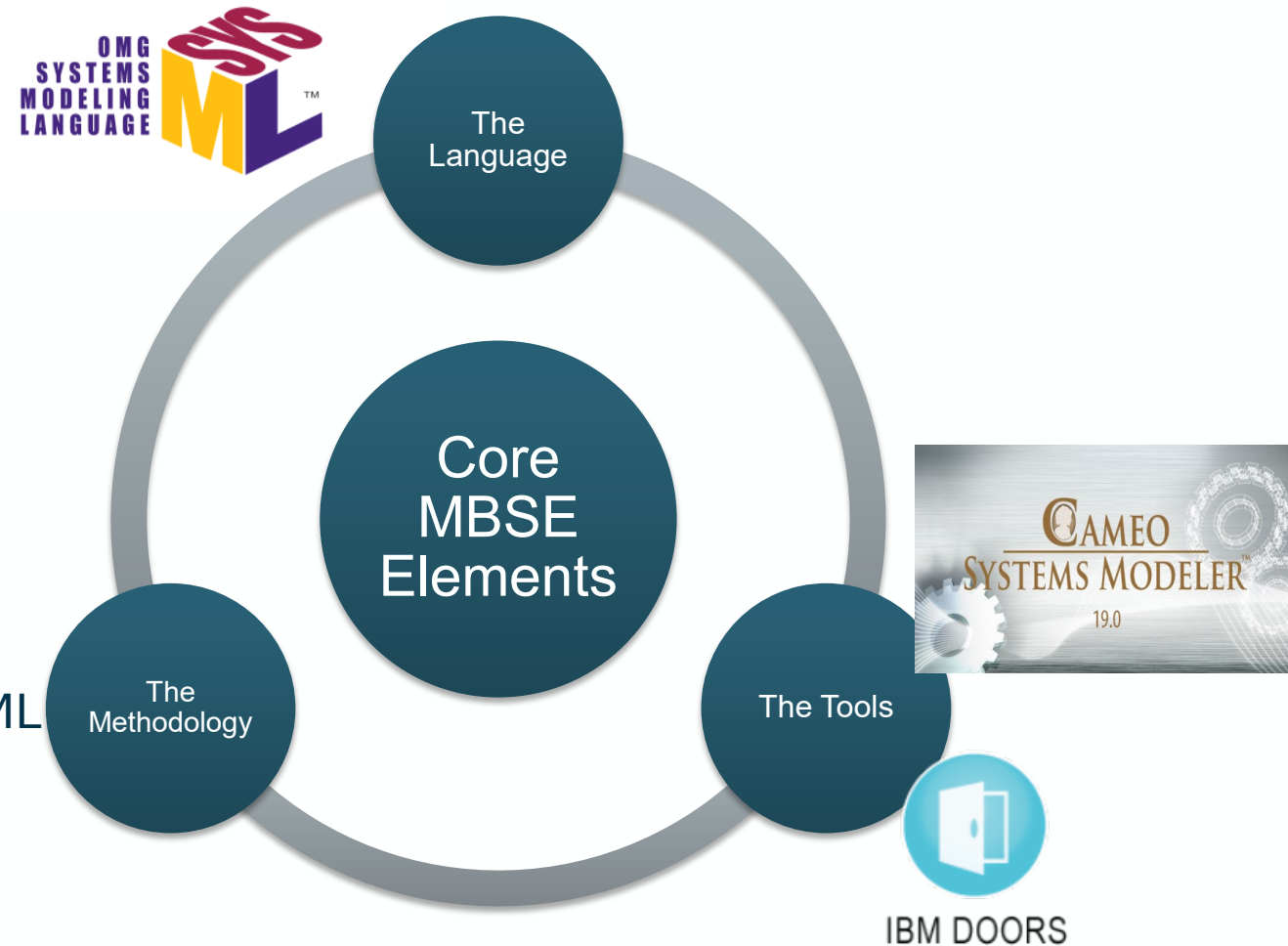
Model content:

- Pilar I: Requirements
- Pilar II: Operation Analysis
- Pilar III: Functional architecture
- Pilar IV: Logical Architecture
- Pilar V: Model traceability

Workflow:

- Modeling activities started in Phase B2
- First model delivered for S-PDR
- User need identification
- Model based mass budget management pilot

- 1) ESA SysML solution
- 2) Airbus MOFL(T)



CAMEO

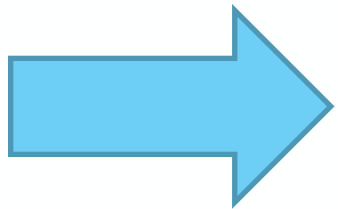
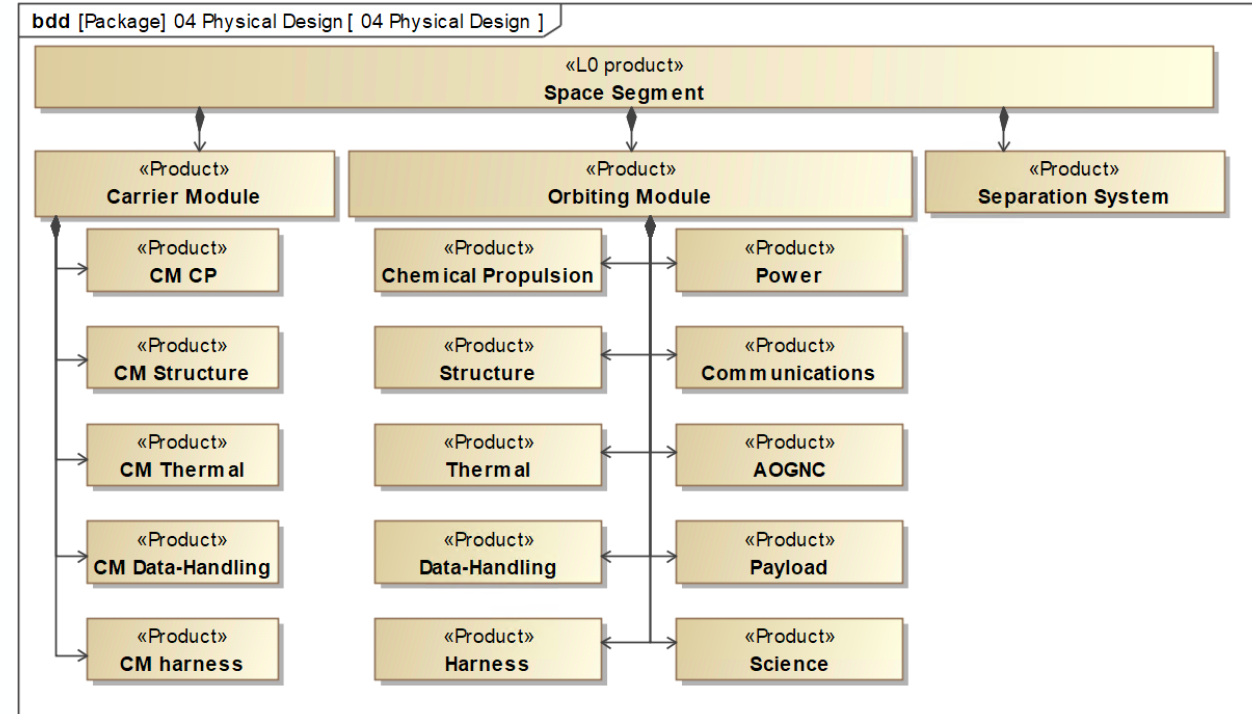
- Extension of methodology
- Formalized mass budget
- Version controlled and working views
- Connectivity to rest of the model

Dashboard

- Visual aid
- Communication tool
- Ease of access throughout stakeholder
- Output from CAMEO
- Consistency of information

Logical Decomposition

- SysML block
- Association links
- Hierarchical decomposition of system

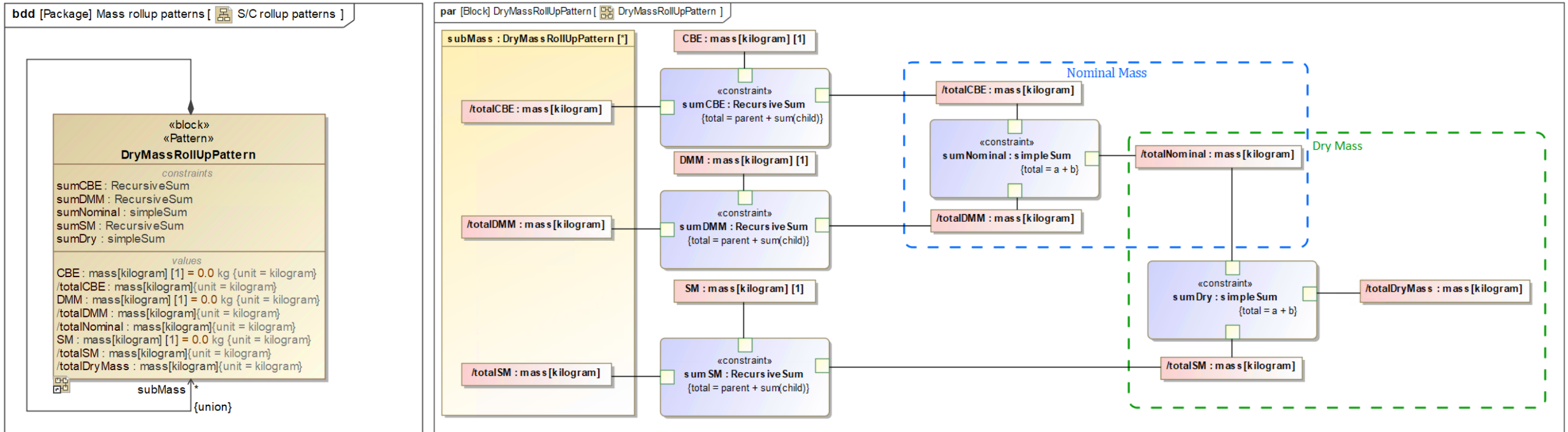


Both *ESA's SysML Solution* and *ADS's MOLF(T) methodology* use this approach

SysML Parametrics

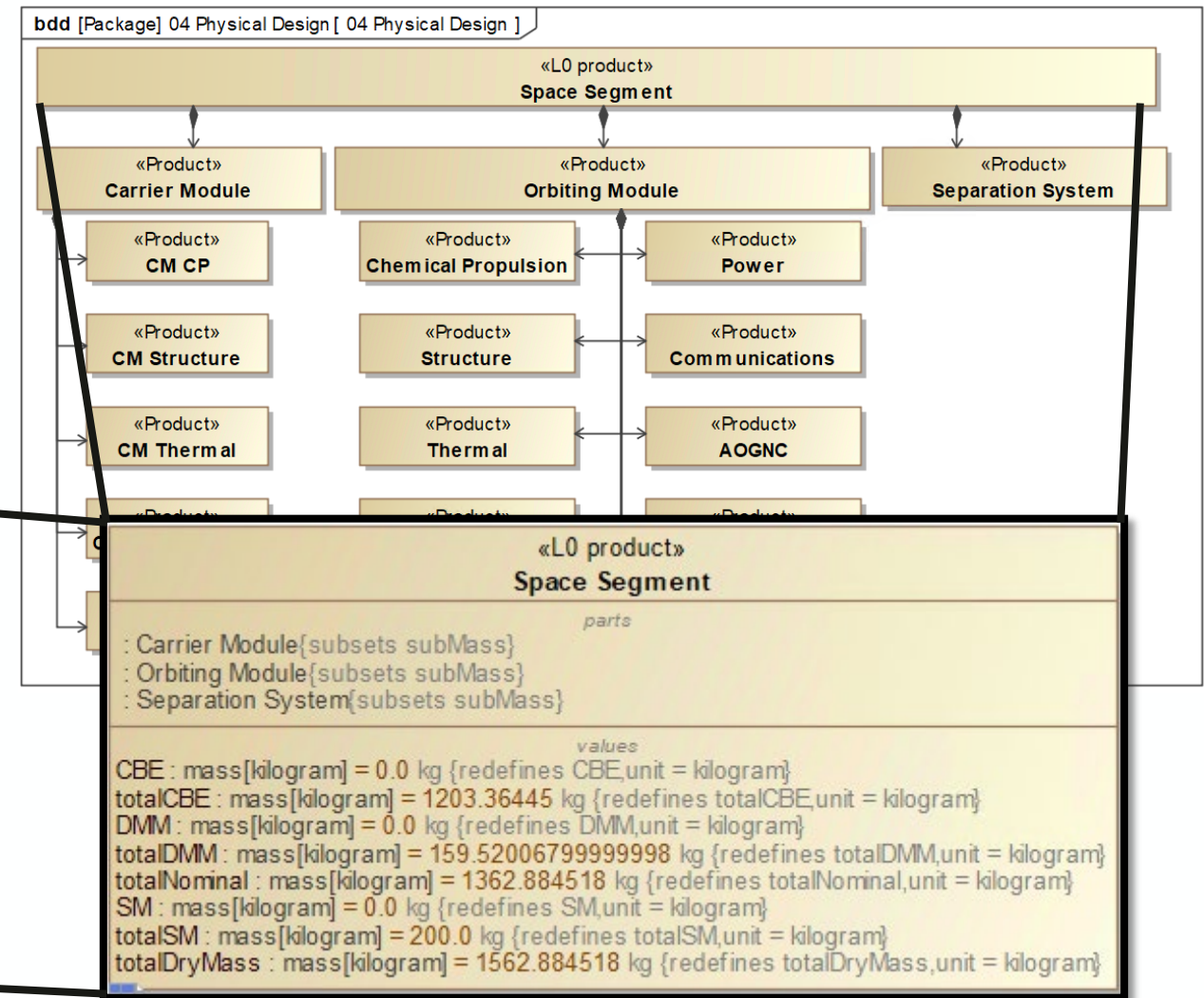
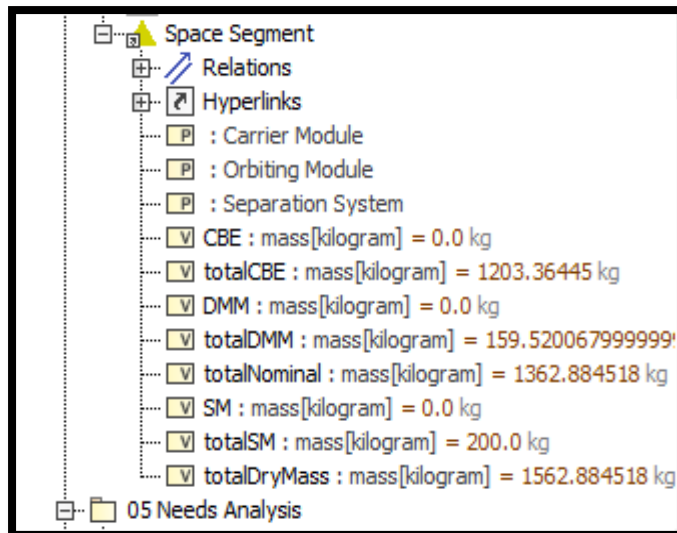
- Patterns
- Parametric diagrams

- Formalised budget management
- Visualisation of information flow
- Automated application to Architecture
- SysML inheritance hierarchy



SysML Parametrics

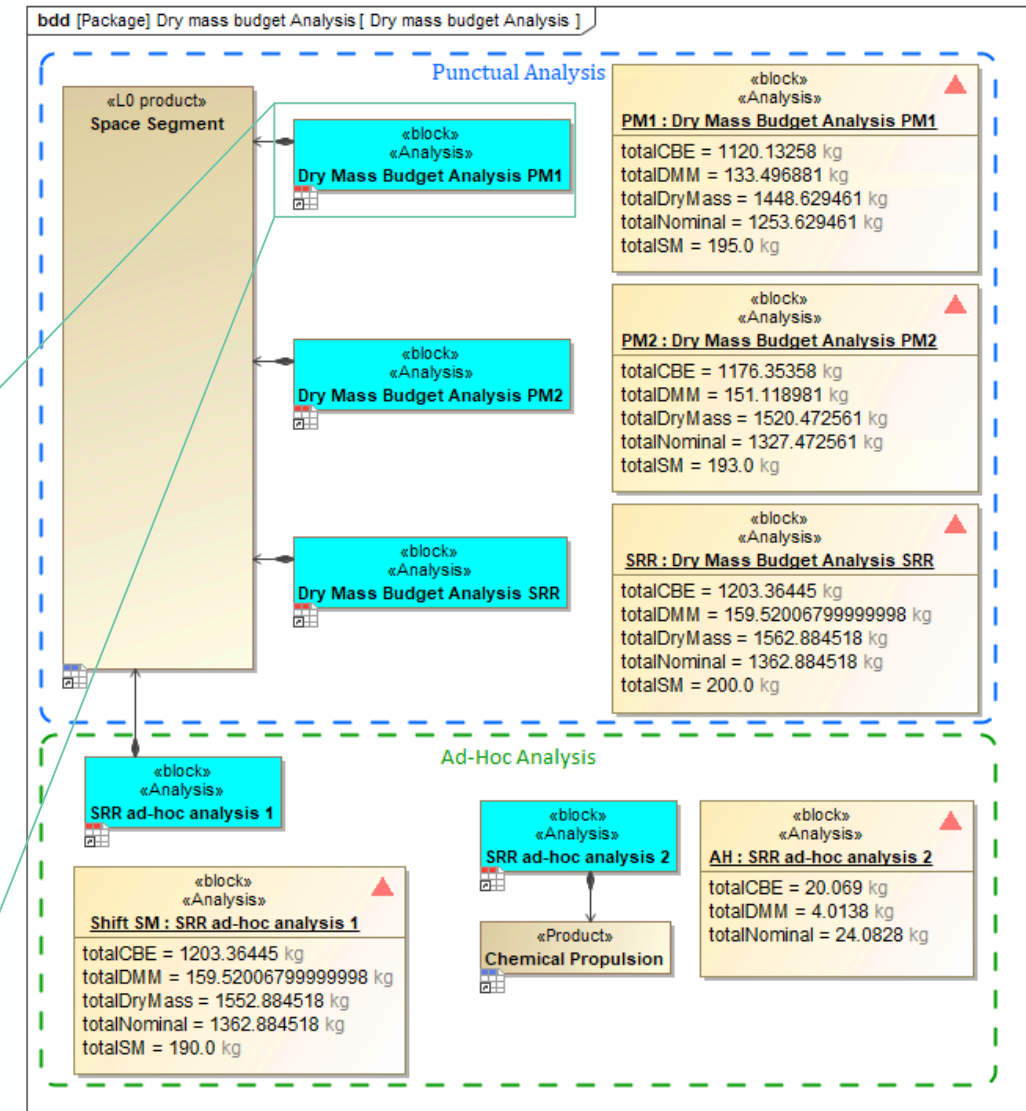
- Patterns
- Parametric diagrams



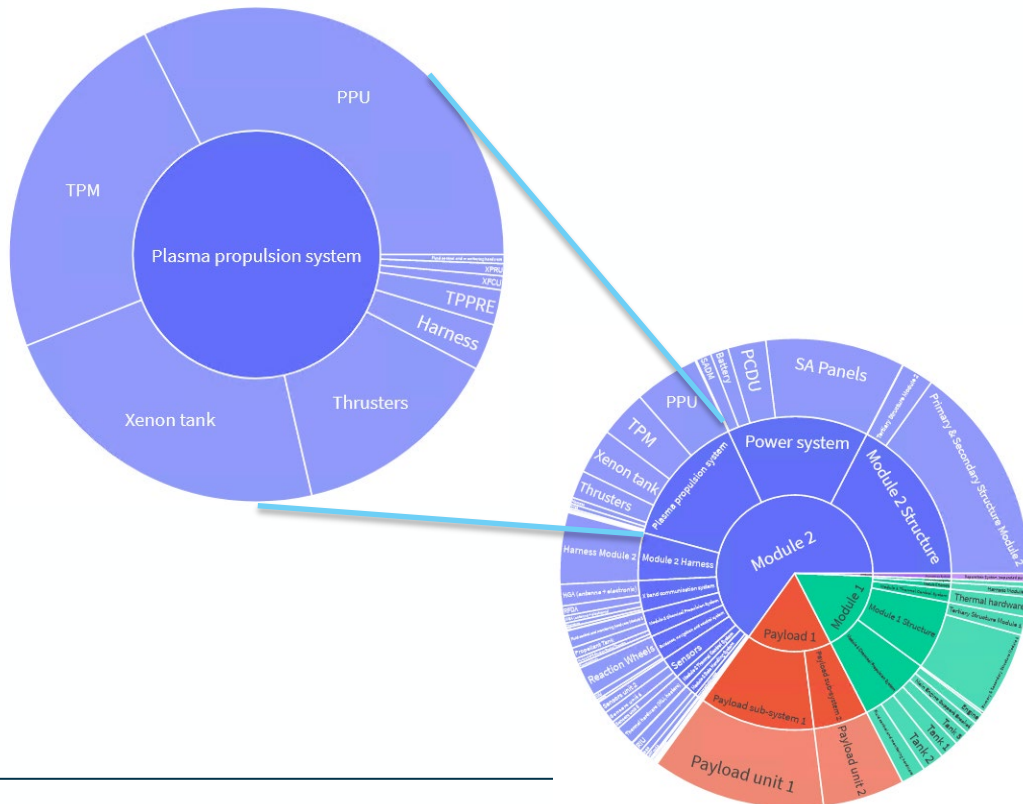
SysML instances

- Virtual copy of S/C
- Store information of past baselines
- Ad-Hoc analysis between milestones

#	Name	Classifier	totalCBE (kg)	totalDMM (kg)	totalNominal (kg)	totalSM (kg)	totalDryMass (kg)
1	PM1	Dry Mass Budget Analysis PM1	1120.1326	133.4969	1253.6295	195	1448.6295
2	dry Mass Budget Analysis PM1.space Segment	Space Segment	1120.1326	133.4969	1253.6295	195	1448.6295
3	dry Mass Budget Analysis PM1.space Segment.carrier Module	Carrier Module	448.9716	43.8684	492.84	50	542.84
12	dry Mass Budget Analysis PM1.space Segment.orbiting Module	Orbiting Module	651.161	87.6285	738.7894	141	879.7894
13	dry Mass Budget Analysis PM1.space Segment.orbiting Module.aognc	AOGNC	30.612	4.5918	35.2038	0	35.2038
18	dry Mass Budget Analysis PM1.space Segment.orbiting Module.chemical Propulsion	Chemical Propulsion	20.069	4.0138	24.0828	0	24.0828
22	dry Mass Budget Analysis PM1.space Segment.orbiting Module.communications	Communications	43.47	4.347	47.817	0	47.817
27	dry Mass Budget Analysis PM1.space Segment.orbiting Module.data-Handling	Data-Handling	42.72	4.272	46.992	0	46.992
29	dry Mass Budget Analysis PM1.space Segment.orbiting Module.harness	Harness	35.702	5.3553	41.0572	0	41.0572
30	dry Mass Budget Analysis PM1.space Segment.orbiting Module.payload	Payload	100	20	120	0	120
32	dry Mass Budget Analysis PM1.space Segment.orbiting Module.power	Power	95.4	9.54	104.94	0	104.94
37	dry Mass Budget Analysis PM1.space Segment.orbiting Module.science	Science	75.528	15.1056	90.6336	0	90.6336
41	dry Mass Budget Analysis PM1.space Segment.orbiting Module.structure	Structure	200.4	20.04	220.44	0	220.44
42	dry Mass Budget Analysis PM1.space Segment.orbiting Module.thermal	Thermal	7.26	0.363	7.623	0	7.623
46	dry Mass Budget Analysis PM1.space Segment.separation System	Separation System	20	2	22	4	26



- Outputs from CAMEO
- Python based (modular implementation)
- User defined views



Selection pan

Progress Meeting

Current

Spacecraft configuration

Full Spacecraft

Report type

Risk and opportunity

Export Report (Work in progress)

Excel

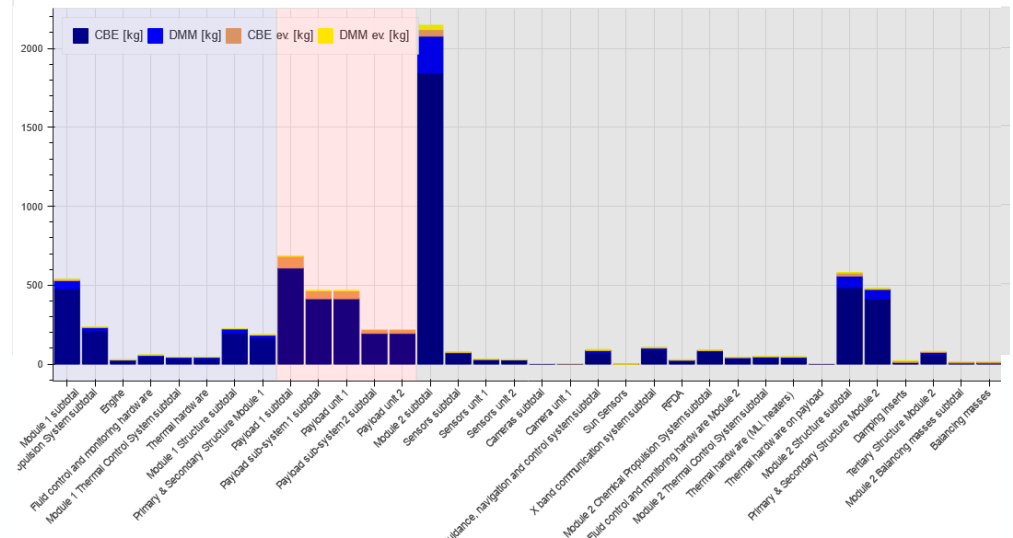
Export

Synch from (Work in progress)

CAMEO - DISM

Synch

Risk and Opportunities - October 2022											
Risk		Opportunity									
Overview		Detailed									
#	Module	System	Unit	CBE [kg]	CBE ev. [kg]	DMM [%]	DMM ev. [%]	DMM [kg]	DMM ev. [kg]	Nominal mass [g]	Nominal mass ev. [kg]
76	Module 2	Module 2 Chemical Pr	Module 2 CPS Reaction Control Thrusters	11.74	0	2	0	0.23	0	11.98	0
77	Module 2	Module 2 Chemical Pr	Fluid control and monitoring hardware Module 2	35.49	3	14.9	0	5.29	0.39	40.78	3.45
78	Module 2	Module 2 Thermal Con	Module 2 Thermal Control System subtotal	38.19	3	20	0	7.64	0.52	45.83	3.6
79	Module 2	Module 2 Thermal Con	Thermal hardware (MLI, heaters)	36.45	2	20	0	7.29	0.35	43.74	2.4
80	Module 2	Module 2 Thermal Con	Thermal hardware on payload	1.74	1	20	0	0.35	0.17	2.09	1.2
81	Module 2	Module 2 Structure	Module 2 Structure subtotal	484.52	19.3	15.21	0.11	73.68	3.06	558.2	22.82
82	Module 2	Module 2 Structure	Primary & Secondary Structure Module 2	412.67	5	14.6	0	60.25	0.64	472.91	5.73
83	Module 2	Module 2 Structure	Damping inserts	8.96	9.3	20	0	1.79	1.62	10.75	11.16
84	Module 2	Module 2 Structure	Tertiary Structure Module 2	62.89	5	18.5	0	11.63	0.8	74.52	5.92
85	Module 2	Module 2 Harness	Module 2 Harness subtotal	127.11	0	25	5	31.77	6.35	158.88	7.3
86	Module 2	Module 2 Harness	Harness Module 2	127.11	0	25	5	31.77	6.35	158.88	7.3
87	Module 2	Module 2 Balancing m	Module 2 Balancing masses subtotal	6.96	6	0	NaN	0	0	6.96	6
88	Module 2	Module 2 Balancing m	Balancing masses	6.96	6	0	0	0	0	6.96	6
89	Module 2	Module 2 units ground	Module 2 units grounding and fasteners subtotal	5.22	0	20	0	1.04	0	6.26	0
90	Module 2	Module 2 units ground	Units grounding and fasteners	5.22	0	20	0	1.04	0	6.26	0



First iteration implemented

- ✓ Tailoring of MBSE methodology to include parametrics
- ✓ Version controlled budget and work environment in CAMEO
- ✓ Dashboard for data viewing and sharing
- ✓ Initial versions well received within project

Deployment

- ☐ Pilot roll-out within the project pending
- ☐ Tracing of requirements to budget

Capabilities

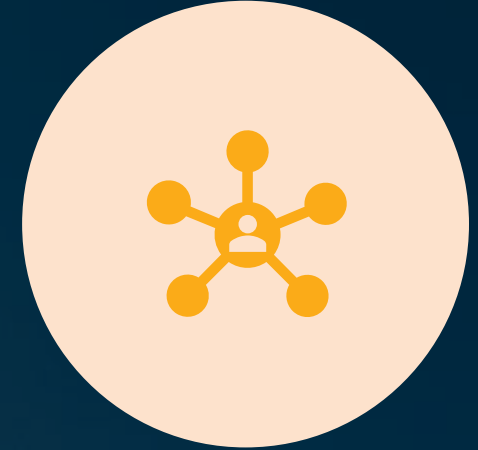
- Expand to other budgets (power, ...)
- Converter between budget philosophies
- User experience



**THANK YOU FOR
YOUR
ATTENTION**



**QUESTIONS ARE
WELCOMED**



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Additional materials

ERO at a glance

Earth Return Orbiter

Round trip to Mars Design

Stacked S/C (1/3 jettisoned mass)

-> Launch mass: 7.2T

-> Departure mass: 3.6T

-> Return mass: 3T

High Reliability

2-FT design for safety critical phases

CCRS

Earth Entry System

Deliver OS back to Earth surface

Capture, Contain & return System

Electric Propulsion

5 electrical thruster
High thrust & Isp:
250mN - 4000s
1203 kg of Xenon

Solar Array

Deployable 2 x 72m²
Power: 42 kW

Return Module

Main Engines

2 x 2 main insertion engines
400N

Orbit Insertion Module

CP Tanks

2025 kg of Bi-Prop

UHF Payload

Relay capability for Mars assets,
EDL & MAS ascent
2 Mbits/s

LIDAR & NAC

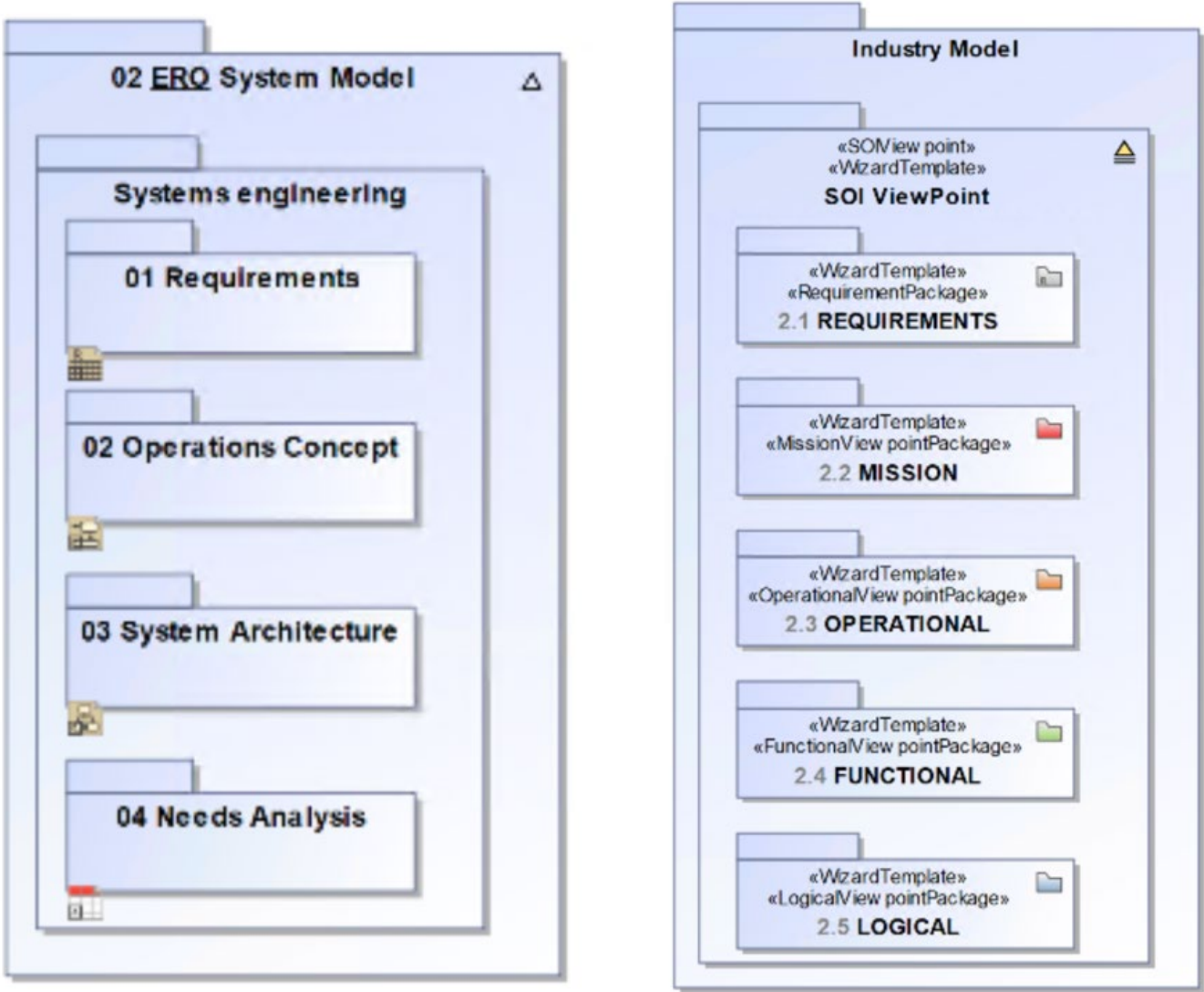
Locate samples in Martian Orbit
Jettisoned at Mars

Rendezvous Sensor Suite

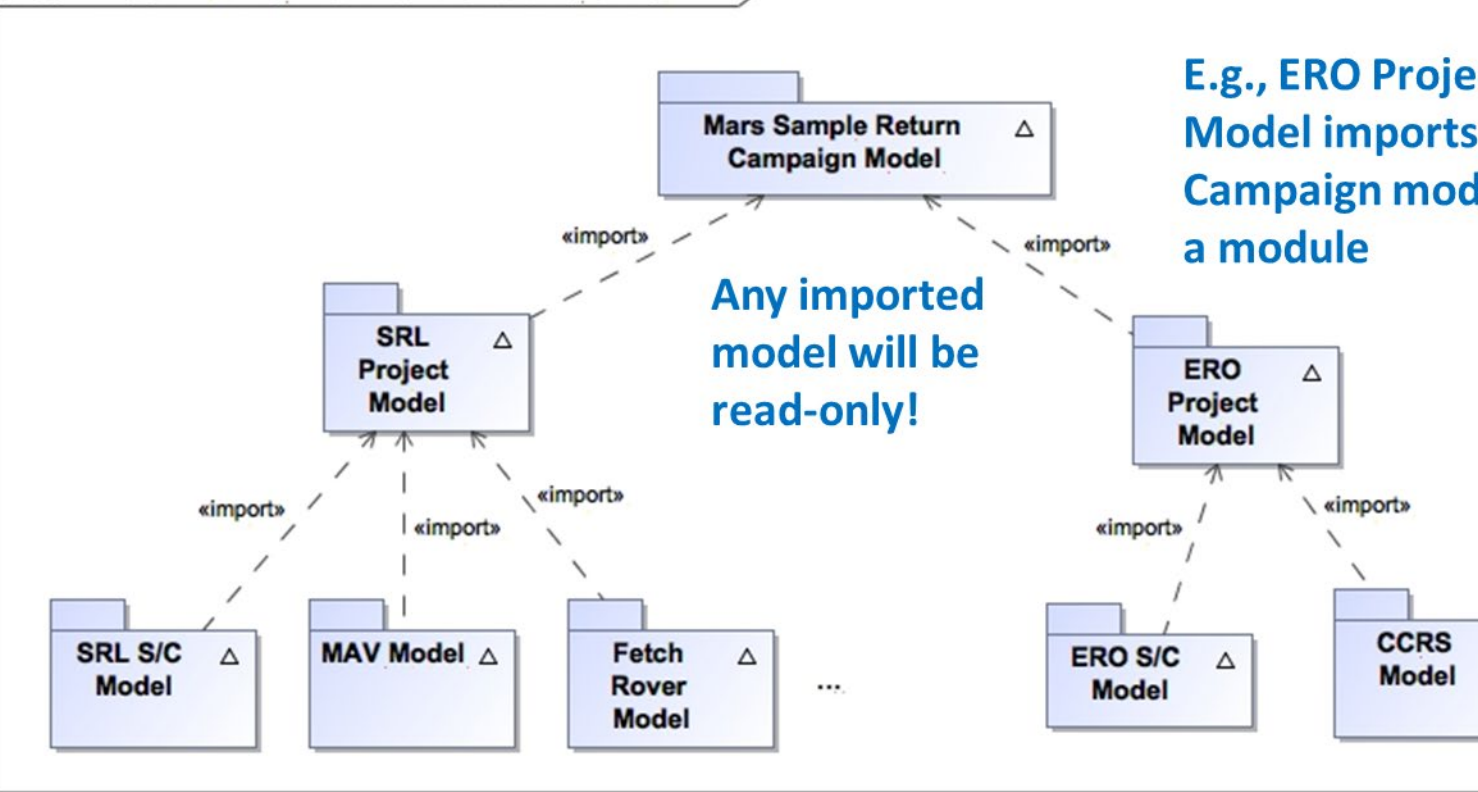
Return Module

X-Band Communication

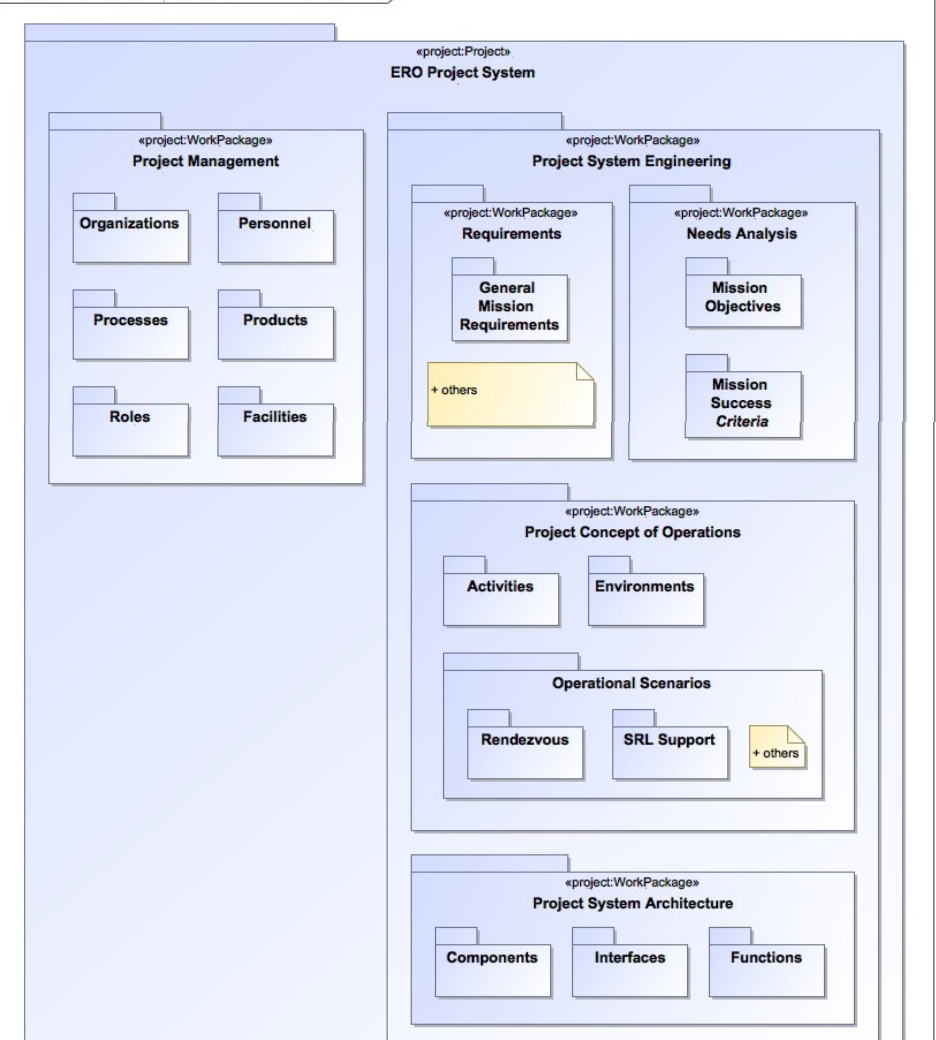
2-axis steerable HGA & MGA
2 x LGA
2.5Mbits/s @ 1AU -
150kbits/s @ 2.2 AU

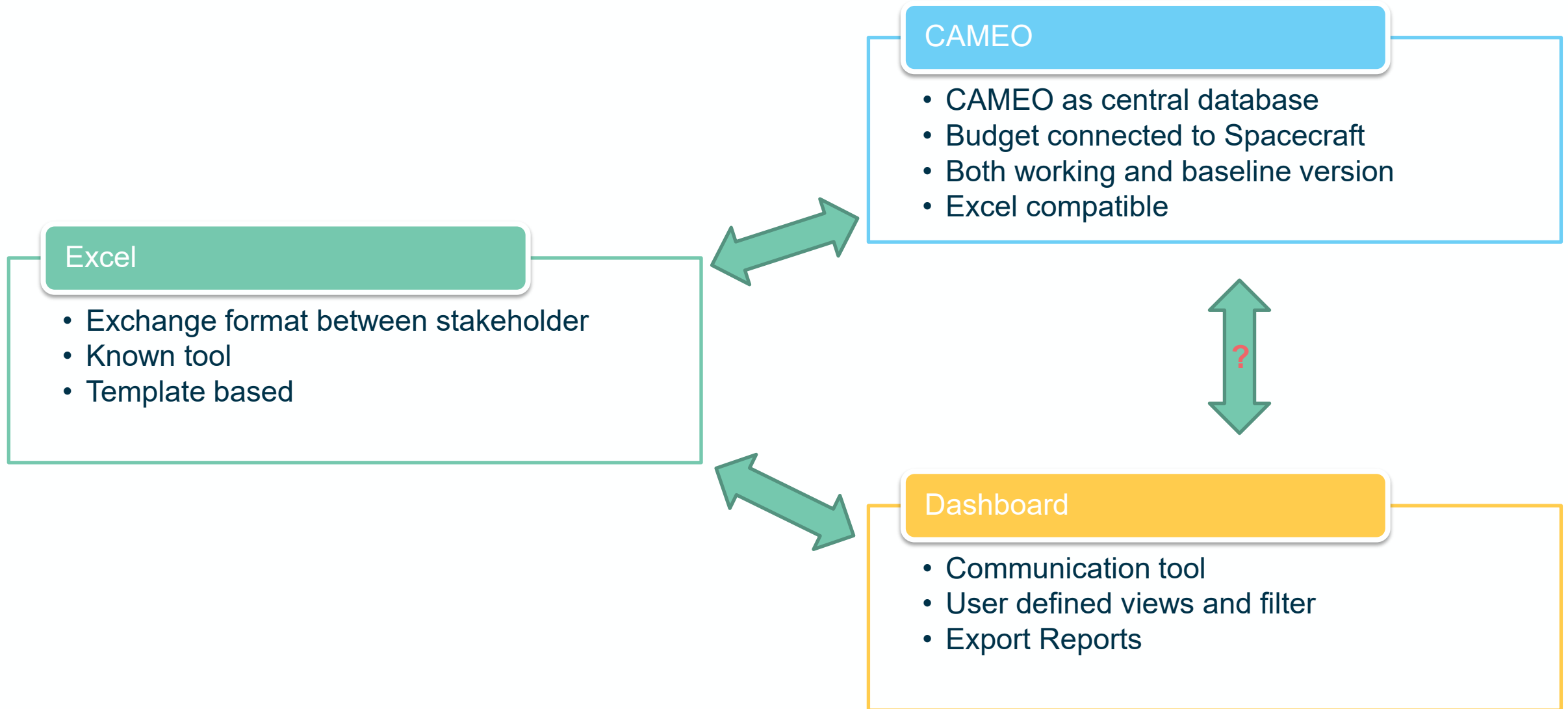


pkg [Package] Module Import Hierarchy Alternative[Import Graph Small]



bdd [Package] Module Import Hierarchy Minimal[Packages - ERO]





Online Aspects

General Introduction to MBSE

- Familiarize stakeholders to MBSE environment
 - SysML
 - CAMEO
 - MOFL

Online Review

- Consistency, Completeness and Correctness (3 c's)
- Specific model aspect mapped to relevant expert
- Issue raised directly into the model

Colocation

- Discussion of review comments generated offline
- Incorporate changes in the model

Offline Aspects

Definition of review scope and delivery of DISM

- Similar to classic review format
- Accessibility of delivery
 - CAMEO model
 - HTML – web access
 - Excel (capture raised issues)

Offline Review

- Offline reviewer have access to current model review status
- Raise issues:
 - Directly in the model
 - Excel (capture raised issues)