



BEYOND THE CHALLENGE

Development of a 140 Am^2 Magnetorquer designed for Improved Demisability

Agenda Presentation

What we do, is who we are.



- About Lusospace
- Magnetorquers
- Demisability
- MRQ140
- Conclusions

A dark, atmospheric image of space with the Earth's horizon visible at the bottom. Two large, complex satellite or space station components are shown in orbit. The component on the left is a large, circular structure with a central opening. The component on the right is a more rectangular, box-like structure with various panels and antennas. The text 'About LusoSpace' is centered over the image in a large, white, serif font.

About LusoSpace

Skills

Main activity areas

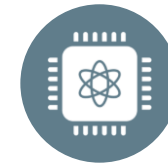
Lusospace engineering teams support companies and organizations in the activities of research and development of advanced technology.

With specialists in Photonics, Electronics, Mechanics and Software development, our multidisciplinary team covers a wide range of knowledge areas.



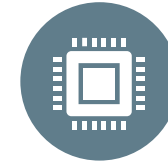
AOCS

Magnetometers, Magnetorquers



Photonics

OGSE, Laser Communications, Lidars, Telescopes, ...



Electronics

EGSE, Precision amplification, Front Ends, Ancillary Electronics



Mechanical

Large Systems, Composites



Software

Embedded (FPGA), Augmented Reality

Lusospace numbers

In 16+ years of space engineering



Heritage

Reliable track in space

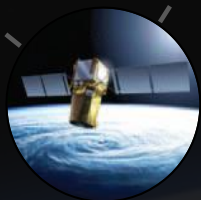
SENTINEL 1
A, B, C, D



PROBA-2



AEOLUS



SENTINEL 3
A, B, C, D



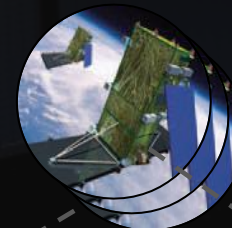
SEOSAT



SENTINEL 5



RADARSAT
CONSTELLATION



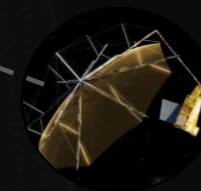
Space Rider



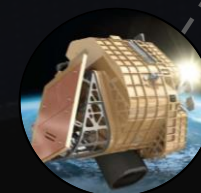
JUICE



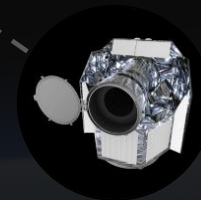
BIOMASS



PRISMA



CHEOPS



Facilities

+500m² office space, +300m² lab space



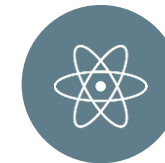
300 m² of lab space

Split between 2 locations, Lusospace offices occupy more than 500m², including 300m² of labs.



Labs

- 15m² ISO 5 Clean Room (class 100)
- 100m² ISO 7 Clean Room (class 10,000)
- 15m² Electronics Lab
- 20m² Optics Lab
- 20m² Testing Room
- 130m² Potential expansion



Main equipment

- Thermal-vacuum chamber
- Thermal cycling chamber,
- Helmholtz Calibration system
- Laser Interferometer
- Fiber Splicing Machine
- Others

Partners

Building strong and long relationships



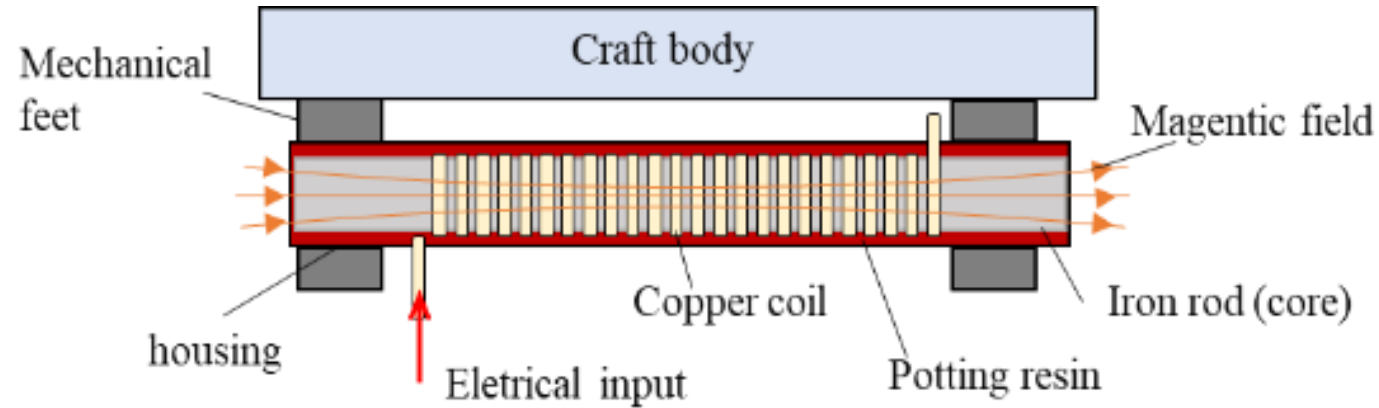
Best service our company

Since 2002, Lusospace has established long and prosperous relationships with our partners.



The background of the slide is a dark, deep blue space scene. In the lower foreground, the curved horizon of the Earth is visible, showing a thin layer of white clouds and a blue sky. Two spacecraft components are shown in orbit. On the left is a large, cylindrical module with a complex internal structure visible through a circular opening. To its right and slightly higher is a larger, more rectangular satellite structure with various panels, antennas, and a prominent yellow circular dish. The text 'Magnetorquers (MTQ)' is centered in the middle of the image in a large, white, sans-serif font.

Magnetorquers (MTQ)



MTQs are S/C elements that use a ferrous core to amplify the magnetic field of a coil of wire.

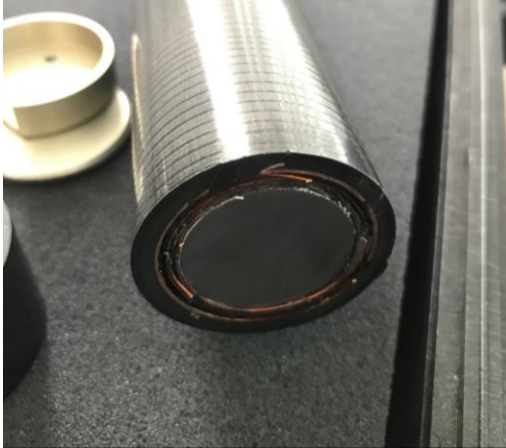
MTQs generate a torque on the S/C by interacting with Earth's magnetic field.

MTQs are S/C elements that, during uncontrolled re-entries, have high probability of surviving re-entry and hitting the ground.

A dark, atmospheric image of space with the Earth's horizon at the bottom. Two large, complex satellite or space station components are visible. One is a large, rectangular structure with various panels and antennas, and the other is a more compact, cylindrical module. The word 'Demisability' is centered in white text over the image.

Demisability

Magnetorquer Demisability



*Non demisable Magnetorquer (LusoSpace)
Source: ESA Cleanspace webpage*



*Non demisable Magnetorquer in wind-tunnel
Source: Belstead UK with Lusospace MTQ*

Due to the size, layered structure, high melting temperatures, and often late separation from the S/C, the core of the MTQ will often survive re-entry.

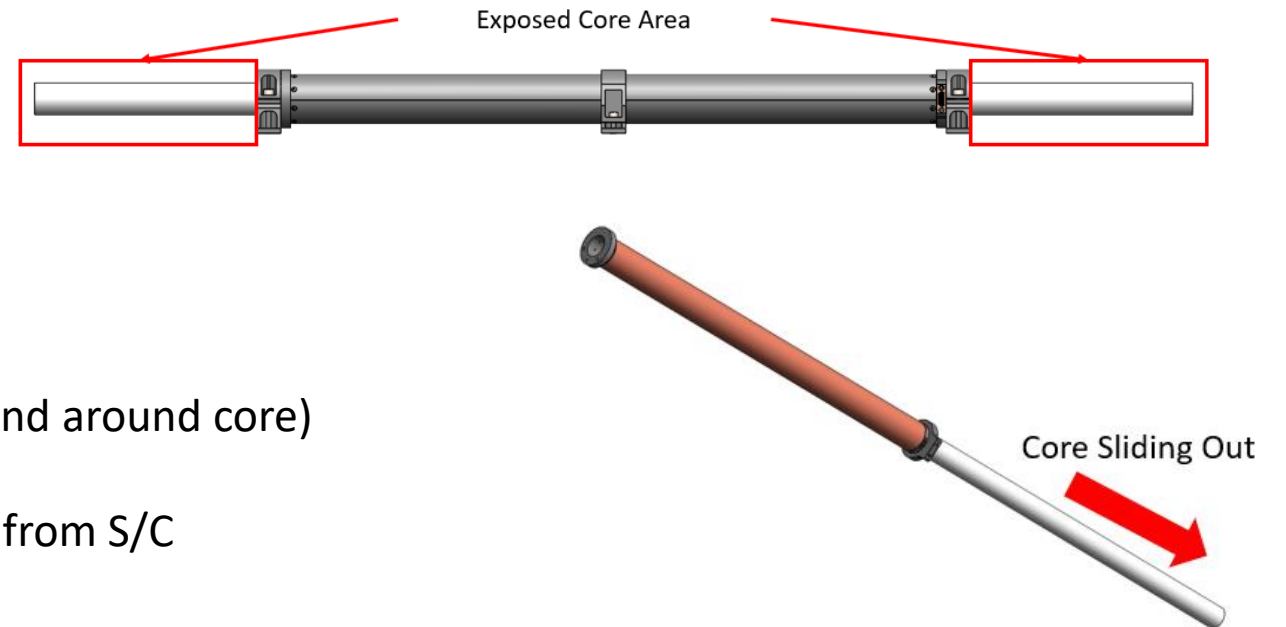
A baseline MTQ without design for demisability was tested in the plasma wind tunnel at DLR with the help of Belstead UK.

A design study was performed to analyse possible design changes to increase the demisability of the MTQ.

Demisable Magnetorquer

BB design: Design for demisability highlights

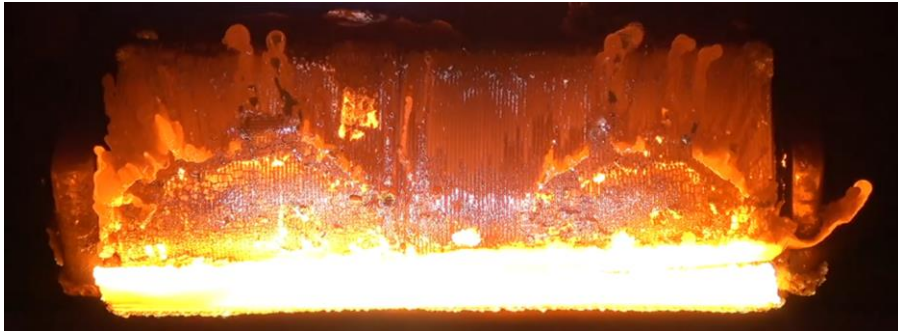
1. Core exposed, for earlier thermal exposure
2. Long and thin diameter
3. No potting material to allow winding separation
4. External housing made in aluminium
5. Internal housing allowing core slide (vs. coil wound around core)
6. S/C interface screws in PEEK to promote release from S/C



Demisable Magnetorquer

BB test: demisability test comparison to baseline

Baseline: Potted Coil after CFRP demise



Demisable:
Un-potted Coil with Aluminium housing



Demisable Magnetorquer

BB test: demisability test highlights

1. The exposure of the core showed that heating of the core would begin earlier and conduct to the inner core allowing earlier heating
2. Long and thin diameter generally gives more surface area and demises faster in simulations
3. The absence of potting material allowed winding separation
4. External housing made in aluminium demises faster than CFRP
5. Internal housing did not permit 'slide out' for this implementation, and this feature was dropped to reduce development time.
6. S/C interface was not tested, due to test setup limitations (plasma chamber size)

Demisability simulation comparison of DMTQ to Standard with Aluminium and CFRP housing.

Release Altitude (km)	Demise Probability (out of 1)		
	DMTQ	Standard Aluminium	Standard CFRP
78	0.98	0.59	0.56
75	0.80	0.35	0.29
70	0.26	0	0
65	0	0	0

MTQ-140



MTQ140



LS used the D4D guidelines it helped develop to design a MTQ140 Am²

A contract is in place to supply the Space Rider mission (3 units, MTQ140 Am²)

ESA is involved to support the material and process qualifications to mitigate risks

Material coating qualification tests passed.

TVAC cycles completed for winding process qualification.

MTQ140

Performance and specifications



Property Under Test	Criterion	Result
Length	$620 \pm 2 \text{ mm}$	622mm
Mass	$< 3.8 \text{ Kg}$	3.14 kg
Impedance	$54 \pm 5 \text{ } \Omega \text{ @ } 22\text{C}$	54.3 Ohm @20C
Inductance	$6 \text{ H} < L < 20 \text{ H}$	8.7 H
Time Constant	$< 200 \text{ ms}$	163 ms
Power Consumption (140)	$< 5 \text{ W}$	4.25 W @20C
V @ 140 Am ² @ 20 °C	$< 28 \text{ V}$	15.2 V
Linearity Error (140 Am ²)	$\leq 0.5\%$	0.25±0.04%
Residual MDM (140 Am ²)	$\leq 0.3\%$	0.21±0.05%
Alignment of field to MIRF	$< 1^\circ$	0.67°±0.3°

A dark, atmospheric image of a space station in orbit above Earth. The station consists of several interconnected modules. On the left is a large, circular module with a central opening. To its right is a larger, more complex structure with multiple windows and external equipment. The Earth's horizon is visible at the bottom of the frame, showing a thin blue line against the blackness of space.

Conclusions

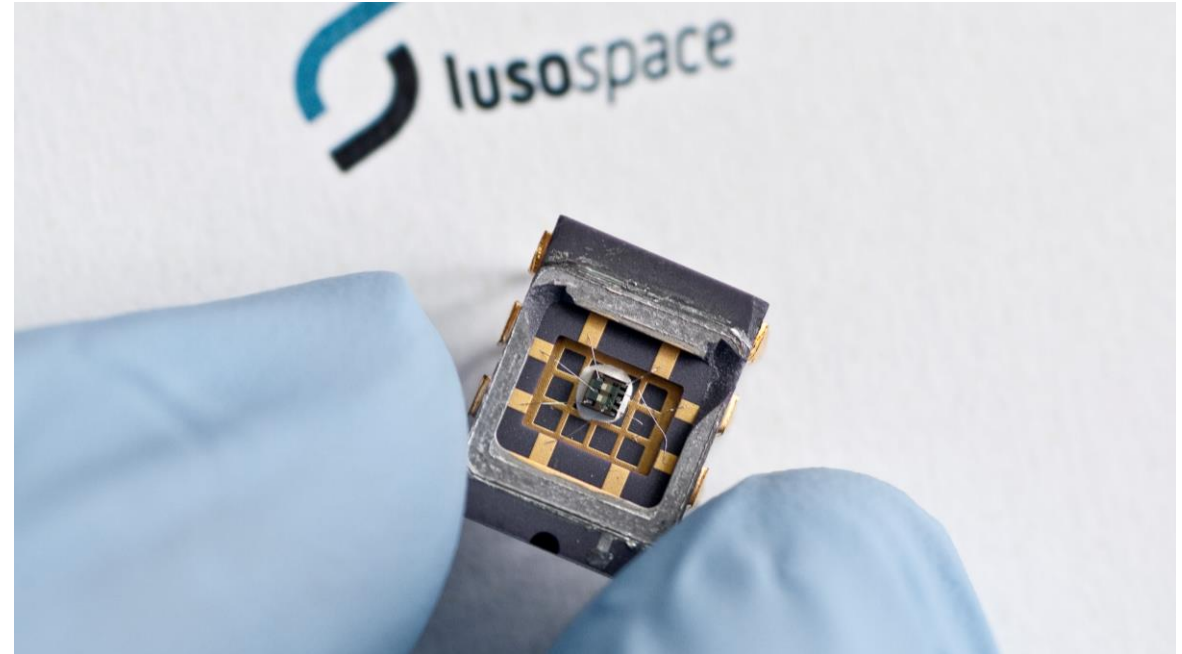
Conclusions



1. LS developed, with ESA and Belstead, design features for magnetorquers that increase their demisability, and can scale with MTQ size.
2. The improved demisability was confirmed in tests and simulation.
3. Using the D4D design features, LS designed and manufactured a 140Am² MTQ Qualification Model (QM).
4. Qualification of materials and processes are ongoing but promising.
5. Performance of the QM meets general requirements of Industry for small MTQ.
6. Environmental tests of QM await final qualifications of materials and processes.

Get In Touch

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