

MINIATURIZED HIGH-PERFORMANCE MEMS ACCELEROMETER





Safran Group Overview



€15,3 **Billion in revenues** in 2021

3 fields

- Aircraft propulsion .
- Aircraft and defense . equipment
- Aircraft interiors .

No.3 Aerospace company worldwide

(excluding aircraft manufacturers)



12,000 +Employees involved in R&D



€1.43 **Billion** invested in R&D in 2021



- Engines Landing Systems Nacelles
- **Electrical Wiring**







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Safran Electronics and Defense

a world leader in our core markets

No.1 worldwide

FADEC (*) for civil aircraft

* Full Authority Digital Engine Control





No.1 in Europe

Inertial navigation systems

No.2 worldwide

Cockpit panels



No.2 worldwide

Flight data monitoring



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Safran Sensing Technologies

High-Performance MEMS

- Leader in high-perf. MEMS sensors
- 25+ Years experience in MEMS
- Proprietary 6" Wafer foundry
- Located in: Yverdon, Switzerland Horten, Norway





The strategic partner for MEMS in Safran, Safran Sensing Technologies is a worldwide leader in high-end inertial MEMS





Safran Sensing Technologies Switzerland Product Mapping

Potential Space application for MEMS accelerometers





BACKGROUND INFORMATION





MEMS Accelerometer Transducer

Capacitive Accelerometer

- 3-wafer bonded stack
- Suspended Proof-Mass between fixed electrodes → Pair of variable capacitors
 Spring Electrodes
- Accel. $\xrightarrow{\text{Spring}}$ Position $\xrightarrow{\text{Electrodes}} \Delta$ capacitance
- Key Specifications:
 - Bias (K0): Signal Output @ zero input
 - Scale Factor (K1): Output sensitivity



SEM cross-section





Open Package





MEMS Accelerometer Readout Electronics

Readout Electronics: Charge-Sense Amplifier

- Differential Charging of MEMS Capacitors $+q = \Delta C \cdot V_S$
- Amplifier equalizes charge on center node $-q = C_{CSA}V_{CSA} \rightarrow V_{CSA} = -\frac{\Delta C}{C_{CSA}}V_S$



Note: *V_S* modulated @ high frequency (outside MEMS bandwidth)





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Force Feedback Electronics (Closed-Loop):

- Controller input: Mass position
- Controller output: Acceleration
- Feedback Mechanism: Electrostatic Force
 - $\Delta V_+ > \Delta V_- \rightarrow$ push mass up
 - $\Delta V_+ < \Delta V_- \rightarrow$ push mass down







Aerospace MEMS Accelerometer Program History

SA0120 open-loop accelerometer

- Adapted from 1000-series sensor
 - 2-channel sensor (1 gee & 20 gee) → Better Spec. Coverage
- Rad-Hard electronics (digital, transistors in ASIC)
- Full performance characterized
- · Lessons learned:
 - Excellent spec. coverage → attractive SWaP-C (MEMS)
 - 2-Ch \rightarrow Enlarged package \rightarrow degraded thermal performance
 - Radiation → Scale Factor shift

SA0120







Aerospace MEMS Accelerometer

SA500 Closed-Loop Sensor

SA500 closed-loop accelerometer

- Adapted from 500-series sensor (20 gee)
- Improved MEMS die
 - Reduced bias error
- Closed-Loop Architecture
 - Improved Performance (1-Ch)
 - Reduced scale factor error
- Targeted Improvement:
 - Bias (K0) Error vs. Temperature: < 300 μgee [-40°C, +85°C]
 - Scale Factor (K1) Error post-rad: < 300 ppm [TID: 50 krad]



- Note: Breadboard model
 - Socket for MEMS
 - Discrete ICs for analog circuit





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PERFORMANCE TESTING







Performance Characterization *Activities*

Thermal Test

MEMS + Electronics in Oven



Radiation Test

MEMS only (ICs & MCU not Rad-Hard)

Thermal performance testing

Figure 7: Accelerometer breadboard with individual package MEMS sensor and discrete electronic components



Figure 6: Radiation testing of the MEMS accelerometer only





Objective & Test flow

Objectives: Demonstrate

- Improved Thermal Bias (K0) Error
- Improved Radiation Scale Factor (K1) Error
 - Embedded K1 Auto-Calibration







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Temperature Acceptance Test: Data Extraction

Process:

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- Measure Acceleration
- Separate into +1g & -1g
- Calculate $K_0 \& K_1 \rightarrow$ plot vs. Temperature
- 3rd order polynomial fit, extract: nominal values (K₀ & K₁ calibration), TC & residue



Improved thermal bias performance

Bias (K0) Residue: 100 µgee (rms)

Scale Factor (K1) Residue: 90 ppm (rms)







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Summary: Improved thermal bias performance

<u>x200</u> x3.3

Total improvement:

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- Smaller Package
- Improved MEMS die v2.0 x15
- Improved MEMS die v3.0 x4



Note: K0 error approx. proportional to FS accel.





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De-risking closed-loop breadboard

Post radiation Scale factor

Scale factor error (20°C): no change

post-Radiation [ppm] 500 500 error [ppm] 0 n ¥ error -500 -500 Кl -1000 -1000 2 3 5 6 -500 500 1 -1000 0 Cycle # K1 error pre-Radiation [ppm] Evolution: Res K1 Acc max XY: Res K1 Acc max 1000 1000 +#1 +#5 ŝ +#2 +#6 +#3 +#8 Ē, 800 +#4 800 c, ResK1Accmax [ppm] Radiatio 600 600 ē. max 400 400 K1 Acc 1 200 200 Res Ref. — most offset: 7 (ppm) Rad. = 1:1 slope 0 1 2 3 5 6 200 600 800 0 400 Cycle # Res K1 Acc max pre-Radiation [ppm]

+#1 +#5

+#2 +#6

+#3 +#8

+#4

1000

■ Ref. - 1:1 slope

SAFRAN

Rad.

XY: K1 error

1000

1000

Space

a Thales / Leonardo company

Repeatability: K1 error

1000

Scale factor Residue: no change

De-risking closed-loop breadboard

Summary: Improved post-radiation scale factor performance

Open-Loop Sensors (SA0120, TS1000 & MS1000)

4500 ppm shift @ 50 krad In-Spec @ ~3-4 krad

Closed-Loop Sensor (SA500)

- MEMS K1 Radiation ageing: <u>none</u>
- Embedded K1 auto-calibration: Validated
 - Correct analog circuit shifts (e.g., voltage reference).









PRODUCT **ARCHITECTURE & DEVELOPMENT LOGIC**







Product architecture & development logic

Product architecture SA500







Product architecture & development logic

Product architecture: SA500 vs. MS500



Next Step: Multichip Module

- MEMS + rad-hard ASIC (analog circuit)
- J-lead Ceramic LCC package

Client Interface

- SPI comm. (accel., temperature, status)
- Digital inputs (e.g. auto-calibration)
- +3.3V digital (MCU)
- +3.3V analog (DAC, ADC)
- ±15V analog (Feedback, CSA)

Development Logic

- Maintain BOM commonality (same MCM)
- Only MCU changes (rad-hard)





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