



maiaSpace

**Update on the implementation of LCA
and ecodesign in the development of a
semi-reusable minilauncher**

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2023 ESA Clean Space Industry Days
18 / 10 / 2023



Past participations and today's agenda

2021 Edition (ArianeWorks)

Preliminary LCA
Ecodesign vision

2022 Edition

Draft of sustainability strategy
Methodology (reusability, ecodesign)
Preliminary analysis of impact mitigation levers and their potentials
Analysis of the environmental benefits/drawbacks of reusability

2023 Edition - Agenda

1. Sustainability strategy and state of progress
2. Updated LCA
3. Methodological developments: focus on the derivation of GWP coefficients for launch
4. Ecodesign: methodology and tools, process, difficulties, first use case



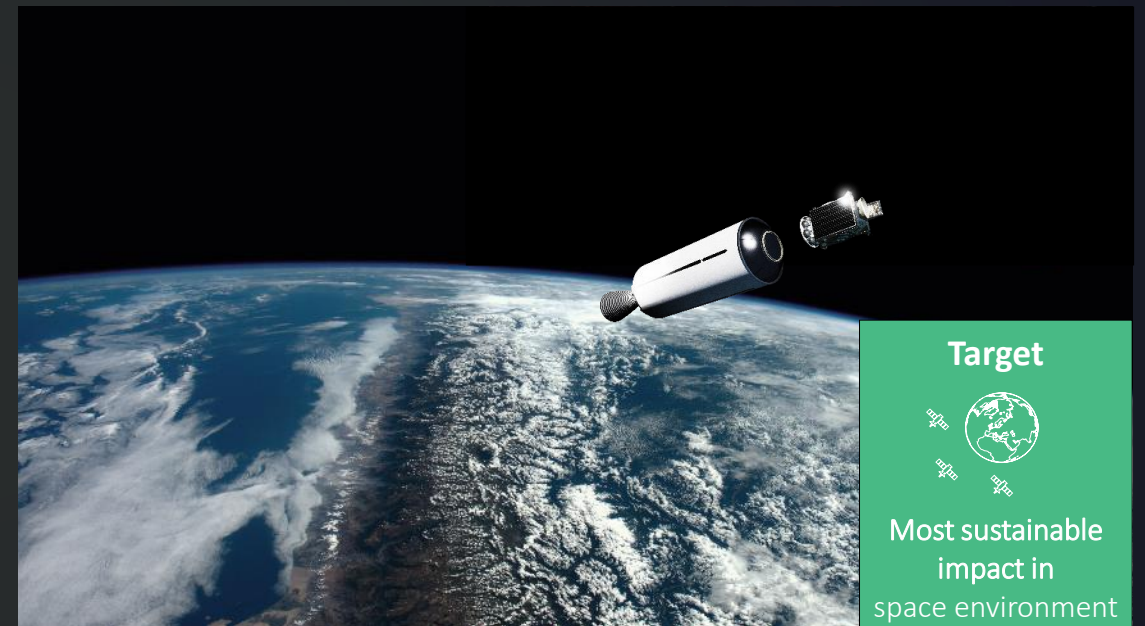
MaiaSpace's space transportation solutions

Reusable, eco-designed and dual-performance launcher

500kg SSO 500km (RLV) – 1500kg SSO 700km (ELV)

Regenerative in-orbit services

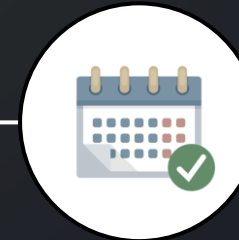
Last miles delivery, Debris Removal...



Prometheus engine



Colibri kick-stage



1st launch
End 2025



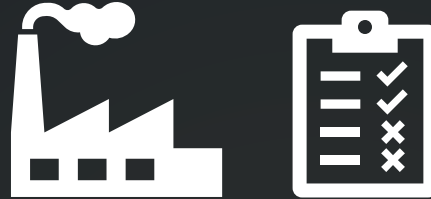
To fulfill its targets and vision, MaiaSpace has set a sustainability strategy based on 4 axes



Lifecycle phases of MaiaSpace's launch service



Research & Development



Manufacturing, Assembly,
Integration and Test (MAIT)



Transport
Europe – French Guiana

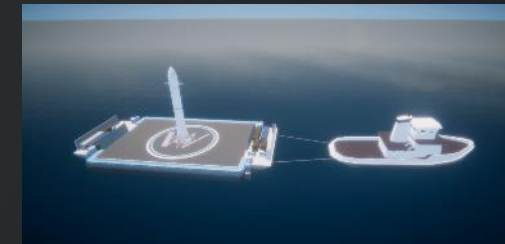
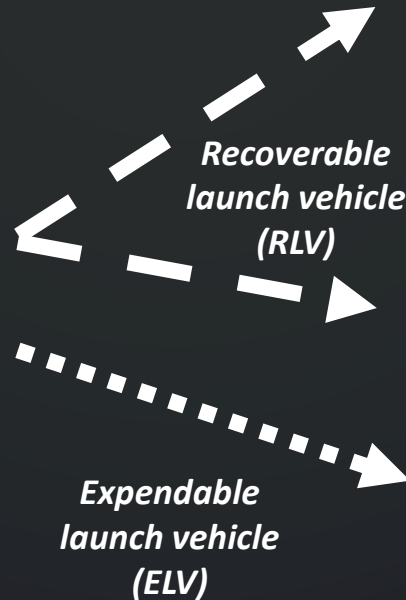


Launch Campaign

Propellant production



Launch Event



Recovery & Refurbishment (of the lower stage)

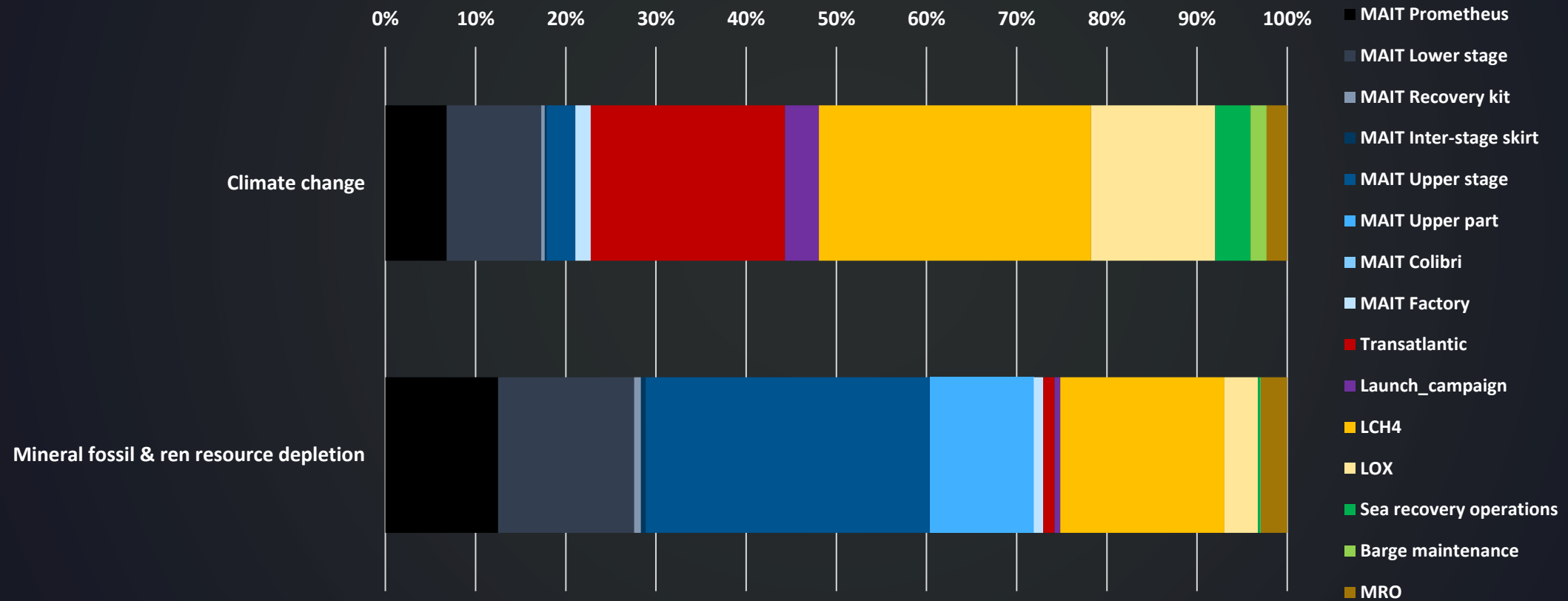
Disposal of the upper stage



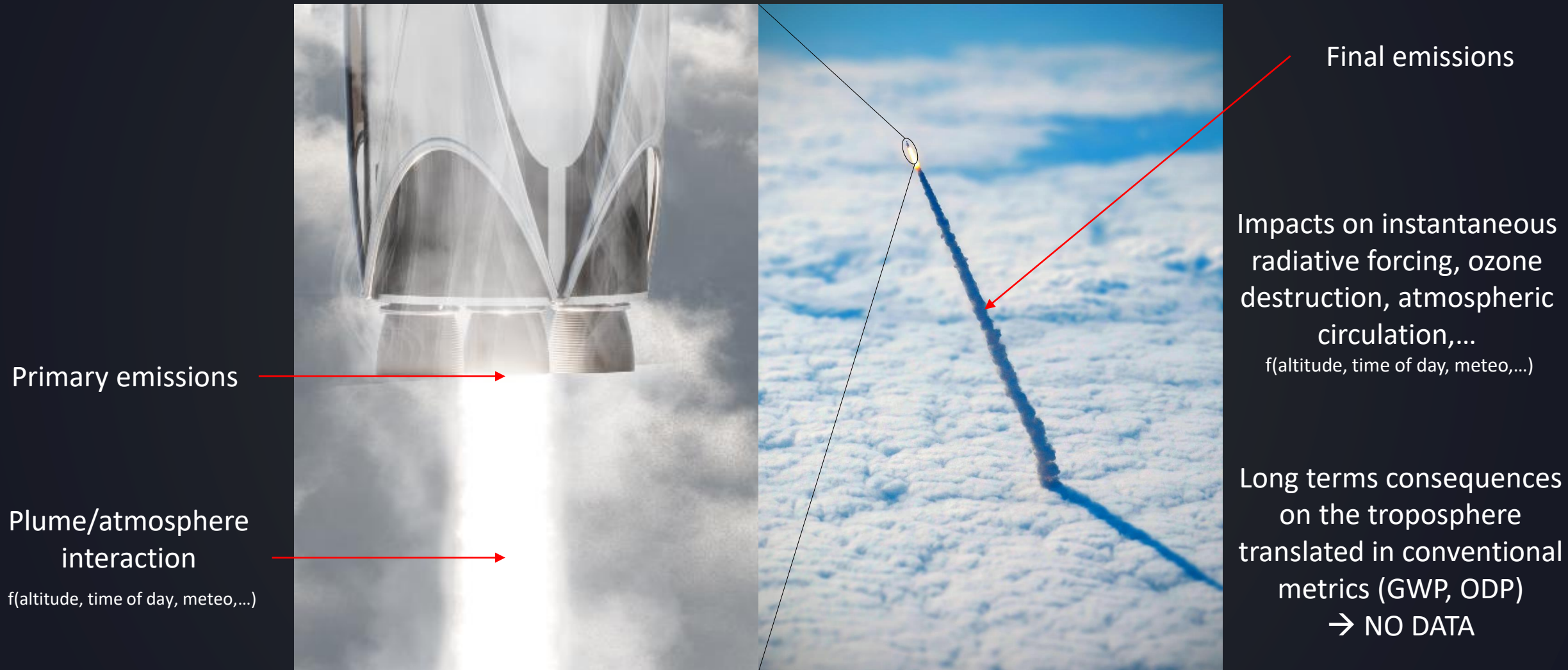
Disposal of all stages

Results over one year of operations

(at full operational capability)



Current knowledge gaps on the launch phase



Necessity of comparing the impacts of launch VS the rest

Which one is best?

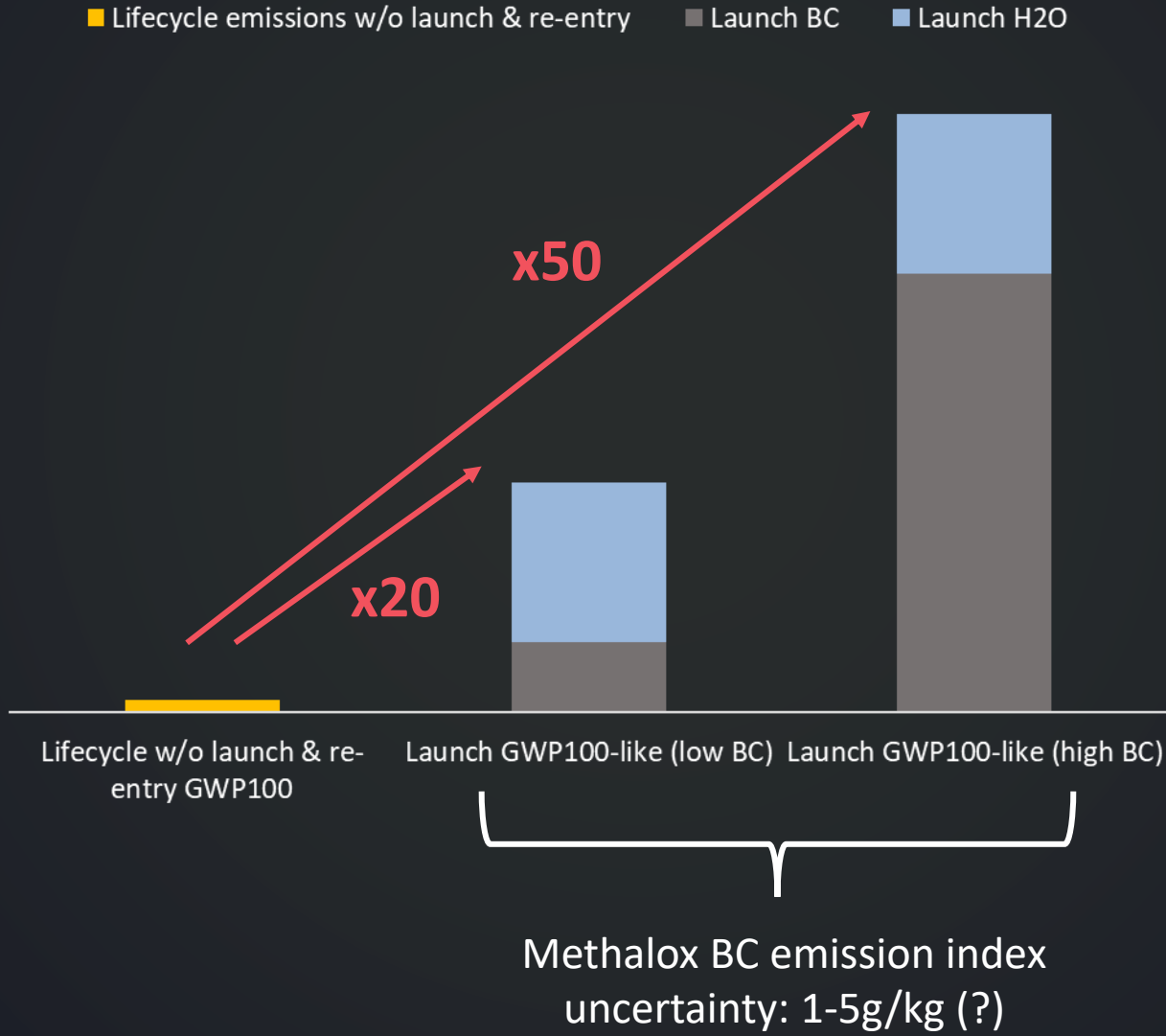


Available Global Warming Potential (GWP) coefficients

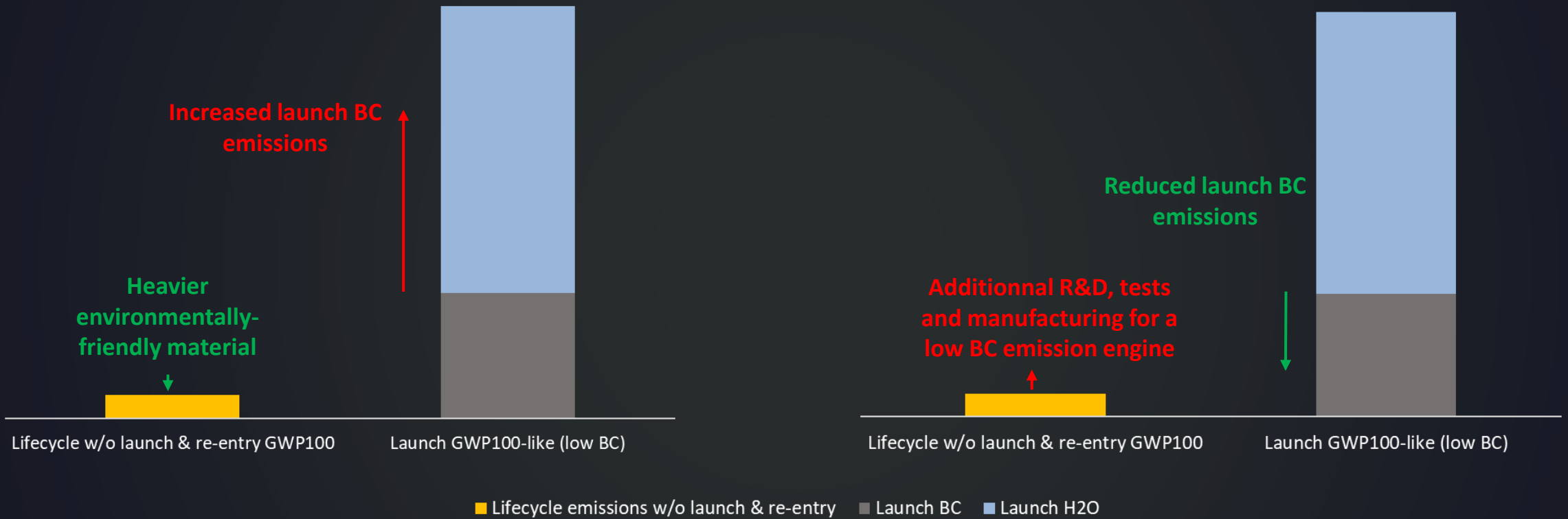
Ground-based and aviation-based climate change characterization factors (GWP100)
as a function of altitude + filled with in-house methodology

| | Altitude (km) | BC | Al2O3 | H2O | NOx |
|-------------------|---------------|--------|-----------|------|----------|
| Lower troposphere | 0-5 | 460 | 1.23 | ~0 | 8.5 |
| Upper troposphere | 5-15 | 1166 | ? -> 1.23 | 0.06 | 114 |
| Stratosphere | 15-50 | 310906 | 60156 | 854 | ? -> 114 |
| Mesosphere | 50-85 | 310906 | 60156 | 854 | ? -> 114 |
| Space | >85 | 0 | 0 | 0 | 0 |

Lifecycle GWP100 is significantly smaller than launch GWP100-like



Issues to manage a much larger effect and uncertainty



Precautionary principle: prevent the increase of atmospheric emissions?

Minimizing atmospheric emissions = N°1 mitigation strategy

→ **The case for performance optimization?**





Example: Colibri's structure trade-off

Order of magnitudes

$$\Delta CC_{41} = CC(4) - CC(1)$$

Measures the "stake" of the tradeoff for CC

Methodology

Rankings

Direct impact only over 1 year

| | CC | RD | Mass |
|----------------|----|----|------|
| A | 4 | 4 | 3 |
| B | 3 | 3 | 1 |
| C | 2 | 2 | 2 |
| D (fictitious) | 1 | 1 | 4 |

$\Delta CC_{41} \approx 6\%$ of total MAIT/yr

Impact including variation of launcher's performance

1kg gained in Colibri → 1kg gained on payload
No effect on filling rate

| | CC | RD |
|---|----|----|
| A | 3 | 3 |
| B | 2 | 2 |
| C | 1 | 1 |
| D | 4 | 4 |

Although D is much better initially, it performs worse due to higher mass

2kg of additional mass of D is enough to erase its CC benefits!

$\Delta CC_{41} \approx 100\%$ of total MAIT/yr

Impact including variation of launcher's performance AND launch phase with high-altitude effects

| | CC | RD |
|---|----|----|
| A | 3 | 3 |
| B | 1 | 1 |
| C | 2 | 2 |
| D | 4 | 4 |

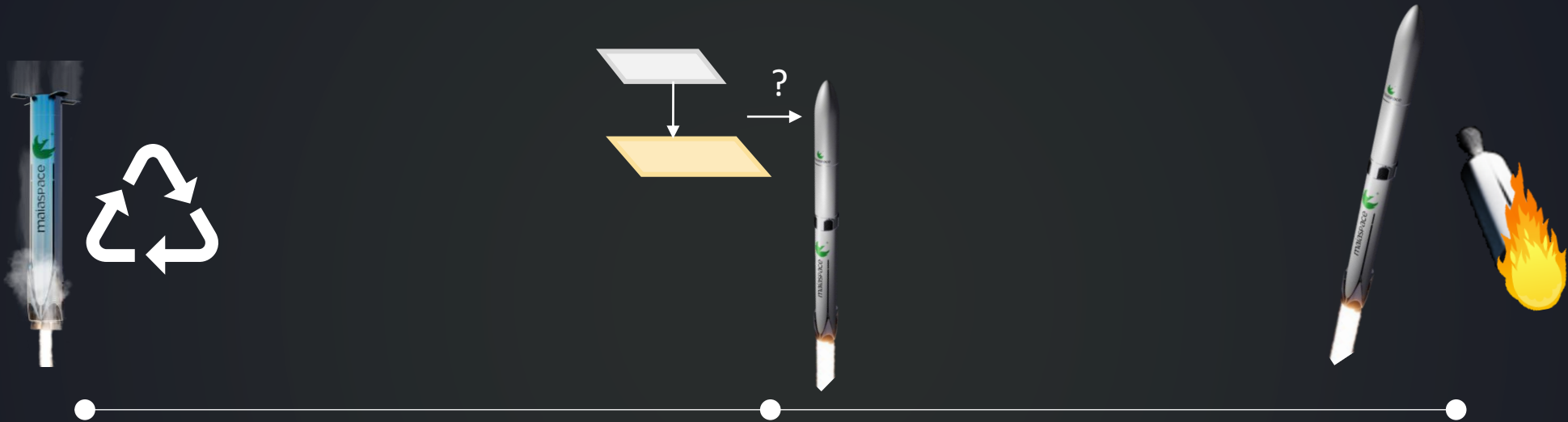
0.1kg of additional mass of D is enough to erase its CC benefits!

$\Delta CC_{41} \approx 4500\%$ of total MAIT/yr



MaiaSpace's methodology objective

Tackle 3 issues not currently addressed by existing methodologies



Management of reusability

Functional Unit
Parametrization
Impact allocation conventions



Management of vehicle performance implications

Highlight multiplicity of possible consequences
Mass \rightarrow performance laws
Conventions



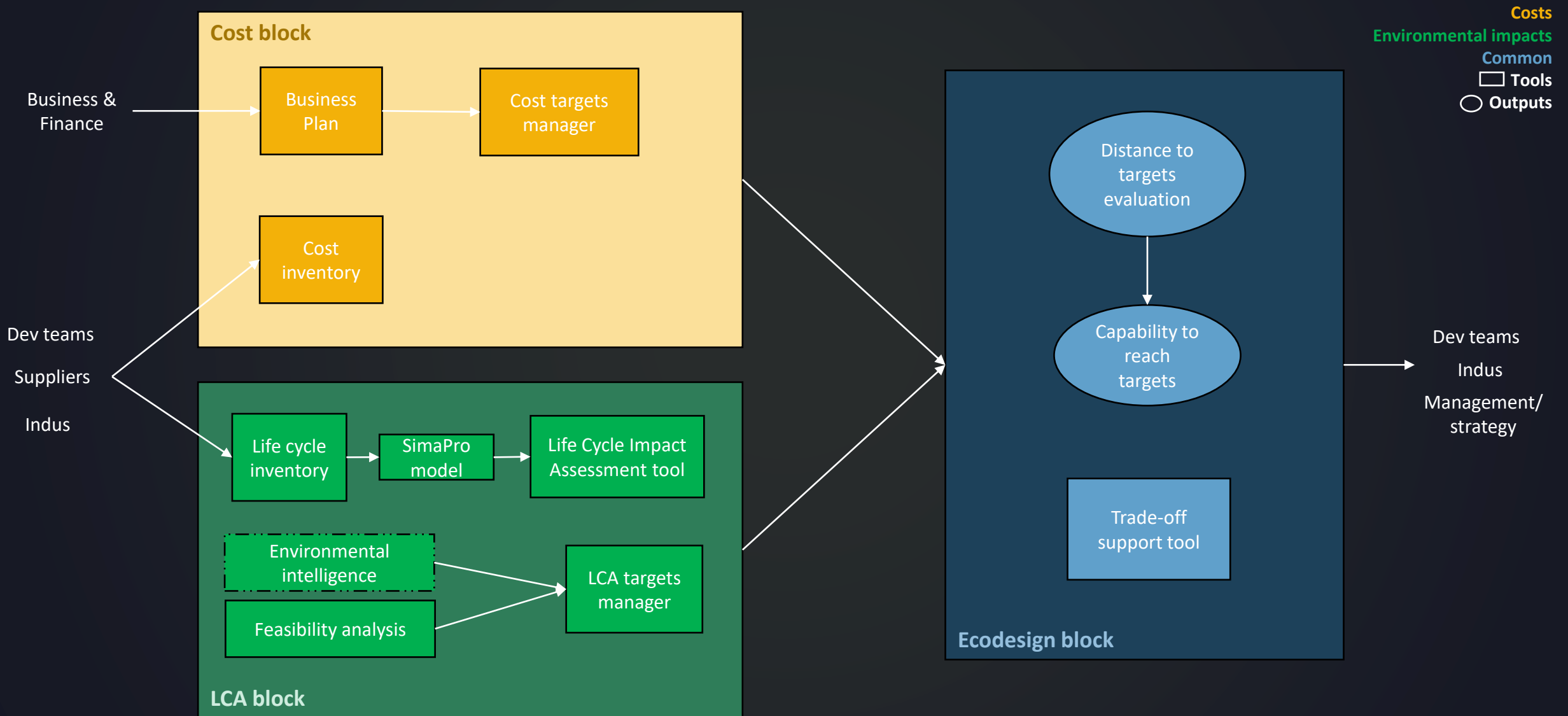
Effects of launch and re-entry

Approximative methodology proposal ✓
Critical reviews from atmospheric scientists ✓

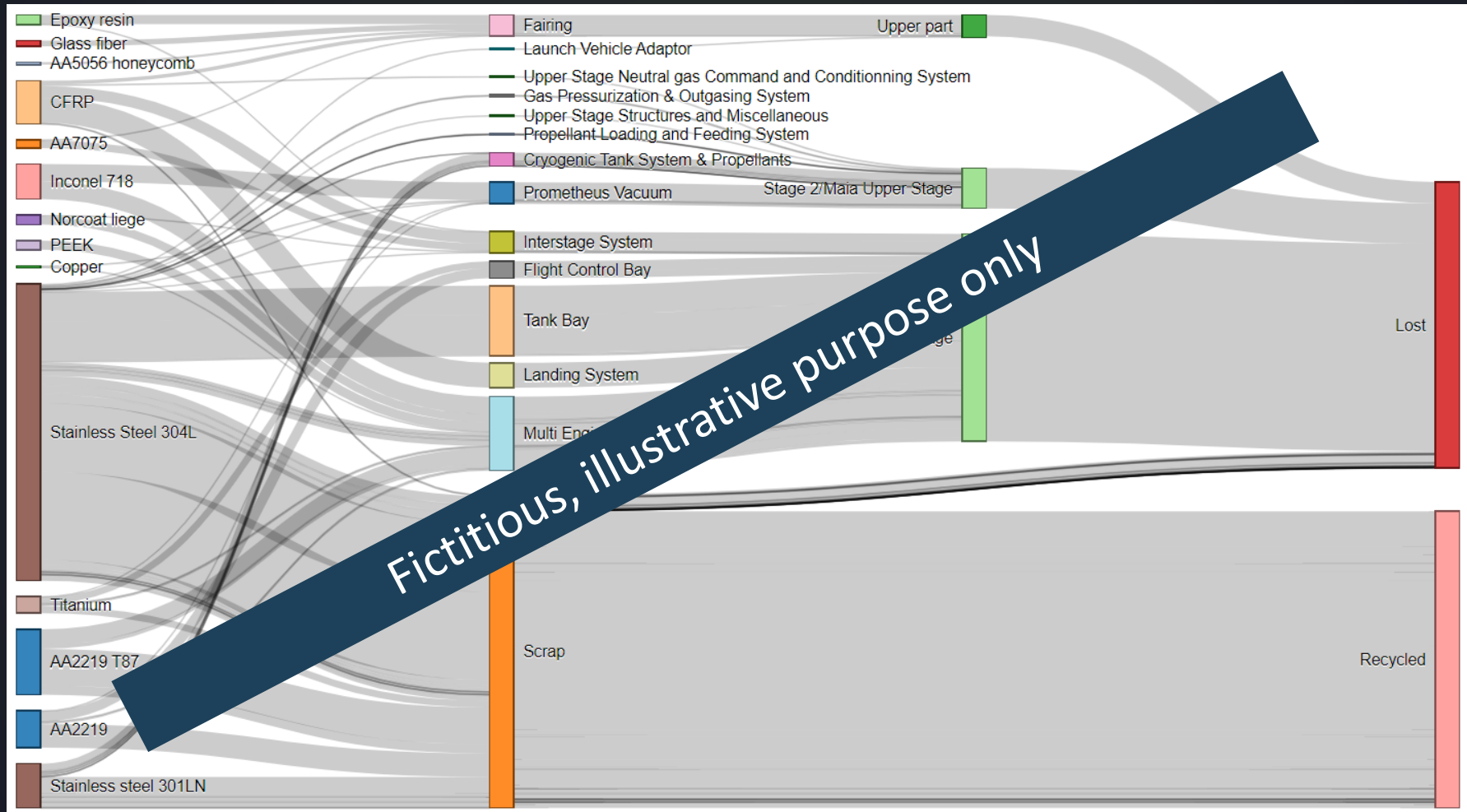
Co-funding of PhD ■■■

➔ Provide a methodological brick to the standardization effort (ESA Handbook, EC PEF)

Ecodesign tools



Mapping material and energy fluxes: a 1st step for understanding vulnerability to systemic risks



Key takeaways

MaiaSpace has made progress towards its sustainability objectives

- Methodological efforts conducted and disseminated
- LCA model updated
- First ecodesign cases (in addition to good early overarching design choices)
- Good feedback from potential customers and partners
- Many challenges must still be overcome

However, ecodesign on the launcher itself is currently not robust due to knowledge gaps on the launch phase

- Methodology to derive GWP100-like proposed
- Suggests that performance optimization reduces atmospheric impacts /kg payload
- PhD project initiated and co-funded on LOX/LCH4 emissions

“Sustainability” is not limited to LCA/ecodesign: first discussions on climate resilience and supply chain vulnerability initiated.

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