Uncertainty in Life Cycle Assessments of Reusable Launchers:

Preliminary findings

Guillermo J. Dominguez Calabuig Early Stage Researcher in System Aspects of Reusable Launch Vehicles at German Aerospace Center (DLR) / Visiting researcher University of Strathclyde Co-authors:

- Andrew Ross Wilsonb, Massimiliano Vasileb, Sifeng Bib. University of Strathclyde Martin Sippel. DLR Cem Berk Senel, Ozgur Karatekin, Orkun Temel, Royal Observatory of Belgium, Brussels, Belgium
- •

2023 Clean Space Industry Days, 16–19 Oct 2023, ESTEC







Content

Introduction

- Launch vehicle fleets
- LCA Methodology
- Climate Change impacts of RLVs
- GCM study
- Conclusions and Way Forward
- Appendix (LCIA, CFs, GCM ejecta/residence time)

Strathclvde



SART Systemanalyse Raumtransport

Introduction



Launch vehicle fleets



-VTHL RLVC4, VTVL LCH4, VTVL, LH2 families. From [9,10]

SART Systemanalyse Raumtransport



ARI



Figure 7: Building-Block launcher family LOX-LH2 combination



Figure 8: Building-Block launcher family LOX-LCH4 combination

www.DLR.de\SART



LCA Methodology: Space Based SSSD



LCA Methodology: Launcher foreground case

Infrastructure not considered (only

recovery specific infrastructure)

Activities based on proxies:

Phase C+D

Propellants life cycle

Production

Clean room fuelling

Decontamination

General handling

Storage

Production of launcher components

Production of Stage 1 / Common Core+Boosters

Production of Stage 2

- Production of Stage 3
- Production of Fairing

What about allocation of Phase A+D?

Dominated by protoypes/test firings/travel

Phase E1

Assembly, Integration and Testing

Launch Campaign

Launch Event

Phase F

Recovery Operations

Refurbishment of Stage 1/Boosters



SART Systemanalyse Raumtransport

University of Strathclvde

Climate Change Impacts of RLVs: Fleet impacts

 Number of launches from accompany paper
RLVC4 VTHL lowest impact

Significant differences
between climate metrics
Assuming ground based
emissions result in
significantly lower impacts
(especially for LCH4 fleet
with GWP20)

Are these CF's adequately modelling the launcher impact?

SART Syst

Strath

Launcher	n	GWI	P100	GW	P ₂₀	GTF	100
		$[CO_{2,eq}]$	[% G.]	$[CO_{2,eq}]$	[% G.]	$[CO_{2,eq}]$	[% G.
L-RLV: $H240 + H61$	55.4	4.73	6.53	4.02	4.68	6.31	4.0
XXL-RLV: 2 H240 + H240 + H61	84.2	7.98	1.4%	11.84	4.9%	6.49	0.2%
XXL-RLV, exp. core.: 2 H240 + H240 + H61	45.9	9.48	1.2%	13.39	4.3%	7.96	0.2%
L-RLV: M520 + M110	61.7	12.08	6.9%	24.87	15.2%	7.58	1.29
XXL-RLV: 2 M520 + M520 + M110	58.7	23.10	9.9%	55.12	18.7%	12.01	2.19
XXL-RLV, exp. core.: 2 M520 + M520 + M110	69.0	24.95	9.1%	57.06	18.0%	13.82	1.89
RLVC4 VTHL Mini-TSTO	39.9	5.93	1.0%	7.97	3.7%	5.13	0.1%
RLVC4 VTHL TSTO	78.0	7.07	1.1%	9.42	4.2%	6.15	0.2%
RLVC4 VTHL 3STO	70.7	7.49	1.1%	9.95	4.1%	6.53	0.29



Climate Change Impacts of RLVs: Contributional Analysis (GWP100)





SART Systemanalyse Raumtransport

University of Strathclyde

Glasgow

AR

t www.DLR.de\SART

Climate Change Impacts of RLVs: Contributional Analysis (GWP20)





SART Systemanalyse Raumtransport

University of Strathclyde

Glasgow

AR

www.DLR.de\SART

Climate Change Impacts of RLVs: Use of Analogue CFs derived from RF [11]



SART Systemanalyse Raumtransport

University of Strathclyde Glasgow



www.DLR.de\SART

GCM Modelling



University of Strathclyde Glasgow

OYAL OBSERVATO

GCM Input: Emission profiles



GCM Results: Radiative Forcing



Conclusions

RLVC4 had the lowest impact (except in GTP100)

In-air capturing approach showed lower impact than the sea fleet required within DRL operations

Launch event impacts appear as a dominating process, specially for the LCH4 fleet and when employing GWP20, because of the non-co2 emissions modelled.
Uncertainty in results may be completely dominated by high altitude impacts from launch and re-entry emissions when using analogue CFs

Significant challenge to incorporate these within LCA's

Climate simulations ongoing to verify analogue. Currently no major signature identified (different from past studies). Uncertainty in:

- Reflectivity properties vs wavelength might not be adequate for study case (based on large forest fires)
- Pulse vs Sustained Emissions
- Particle size distribution for methalox required
- Plume post-combustion model

Future studies shall address these high-altitude impacts, fugitive emissions, and include additional LCI processes as:

- Launch infrastructure (launch sites/landing sites),
- Transportation processes
- Development phases and test firings





Any Questions?

Email: guillermo.dominguezcalabuig@dlr.de

Acknowledgments: Lois Miraux, Andrew R. Wilson, Alberto Sartizu, Cem Berk Senel, ASCenSlon network..







SART Systemanalyse Raumtransport

University of Strathclyde Glasgow

European Commission

AILO

Horizon 2020 European Union funding for Research & Innovation

Credits: Sciency Words: Ideal Rocket Equation

https://planetpailly.com/2015/04/17/sciency-words-ideal-rocket-equation/



The project leading to this application has received funding from the European Union's Horizon 2020

research and innovation programme under the Marie Skłodowska-Curie grant agreement No 860956.



www.DLR.de\SART

HANG ON-I DIDN'T SAY YOU COULD LEAVE-

Bibliography

- (1) Miraux, Loïs; Wilson, Andrew Ross; Dominguez Calabuig, Guillermo J., "Environmental sustainability of future proposed space activities", Acta Astronautica, 2022
- (2) Guillermo J. Dominguez Calabuig, Lois Miraux, Andrew Ross Wilson, Alberto Sarritzu, Alberto Passini, "Eco-design of future reusable launchers: insight into their life cycle and atmospheric impact". In 9th European Conference for Aeronautics and Space Sciences (EUCASS), 2022
- (3) Ross, M.N. and Sheaffer, P.M. (2014), Radiative forcing caused by rocket engine emissions. Earth's Future, 2: 177-196. DOI: 10.1002/2013EF000160
- Martin N. Ross, Karen L. Jones, Implications of a growing spaceflight industry: Climate change, Journal of Space Safety Engineering, Volume 9, Issue 3, 2022, Pages 469-477, ISSN 2468-8967, https://doi.org/10.1016/j.jsse.2022.04.004.
- (5) Ryan, R. G., Marais, E. A., Balhatchet, C. J., & Eastham, S. D. (2022). Impact of rocket launch and space debris air pollutant emissions on stratospheric ozone and global climate. Earth's Future, 10, e2021EF002612. <u>https://doi.org/10.1029/2021EF002612J</u>.
- (6) Maloney, C. M., Portmann, R. W., Ross, M. N., & Rosenlof, K. H. (2022). The climate and ozone impacts of black carbon emissions from global rocket launches. Journal of Geophysical Research: Atmospheres, 127, e2021JD036373. https://doi.org/10.1029/2021JD036373
- Leonard Schulz, Karl-Heinz Glassmeier, "On the anthropogenic and natural injection of matter into Earth's atmosphere", Advances in Space Research, Vol. 67, Issue 3, 2021, Pages 1002-1025, ISSN 0273-1177, DOI: 10.1016/j.asr.2020.10.036.
- Pletzer, J. and Hauglustaine, D. and Cohen, Y. and J\"ockel, P. and Grewe, V., The climate impact of hydrogen-powered hypersonic transport, Atmospheric Chemistry and Physics, Vol 22, 2022, Number 21, DOI: 10.5194/acp-22-14323-2022
- Jascha Wilken. Cost estimation for launch vehicle familities considering uncertain market scenarios. In ASCenSIon Conference, Dresden, 12-14th September, 2023., 2023
- (10) Martin Sippel, Steffen Callsen, Jascha Wilken, Kevin Bergmann, Ingrid Dietlein, Loenid Bussler, Guillermo Joaquin Dominguez Calabuig, and Sven Stapper. Outlook on the new generation of european reusable launchers. In ASCenSIon Conference, Dresden, 12-14th September, 2023, 2023.
- (11) Lois Miraux. Derivation of analog gwp coefficients for the high altitude effects of launches. In Workship on Life Cycle Assessment of Space Transportation Systems, 4-6 July, 2023, Stuttgart,, 2023.



SART Systemanalyse Raumtransport

Strathclvde

Appendix: LCIA methods

. , .							
Species	GW	P_{100}	GWP_{20}		GTI	Reference	
	Aviation	Ground	Aviation	Ground	Aviation	Ground	
H_2O	6×10^{-2}	5×10^{-4}	0.22	-1×10^{-3}	0.008	0.	[26]
NO_x	114	8.5	619	31.5	13	-0.65	[27]
H_2	12.8	12.8	40.1	40.1	2.3	2.3	[28]
CH_4	29.8	29.8	82.5	82.5	7.5	7.5	
CO	4.0	4.0	9.2	9.2	1.95	1.95	
BC	1166	900	4288	3200	161	130	

Table 2: Climate change life cycle impact for a kg of emission. Values from IPCC [24] for ground based emissions and Lee et al [25] unless otherwise stated.

- Global Warming Potential over 100 years (GWP100)
- Global Warming Potential over 20 years (GWP20)
- Global Temperature Change Potential over 100 years (GTP100)

www.DLR.de\SART

17

- Aviation based impact factors assumed as default for the assessment
 - Sensitivity was performed with ground based emissions

Strathclvde

SART Systemanalyse Raumtransport



Appendix: Past studies and climate metric derivation for LCA's

- Based on [11]
- From instantaneous stratospheric radiative forcing (A):

 $RF_i(t) = A_i R_i(t)$

- Not TOA or Tropopause, nor RF or ERF
- Assumes exponential decay

$$R_i(t) = e^{-\frac{t}{\tau_i}}$$

- Lifetime assumed from averaged stratospheric circulation
- Where does it sink?

Table 1: Instantaneous stratospheric Radiative Forcing A for emissions of different species in the stratosphere obtained from past studies

Species		$A_i (W/m^2/Tg)$				
BC H2O Al2O3 CO2 NOx Cloudiness	Ross et al.[3] 34 300 31.8 6000 0.017	Ryan et al,[5] 8720 -24.0	Pletzer et al,[8] 1.58, 1.90	This study 34 300 31.8 6000 0.017		

Absolute Global Warming Potential:

$$AGWP_{i}(H) = \int_{0}^{H} RF_{i}(t)dt = A_{i}\tau_{i}\left(1 - e^{-\frac{t}{\tau_{i}}}\right)$$

Global Warming Potential

$$GWP_i(H) = \frac{AGWP_i(H)}{AGWP_{CO_2}(H)}$$

www.DLR.de\SART

SART Systemanalyse Raumtransport

Appendix: Past studies and climate metric derivation for LCA's



Appendix GCM Results Residence Time

