



**ariane**GROUP

# **WATER PROPULSION**

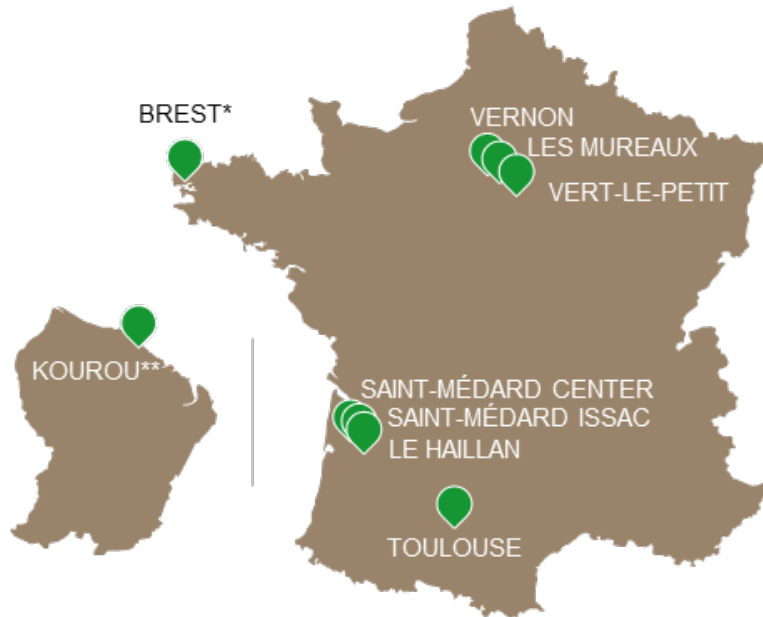
**The ultimate Green Propellant**

Malte Wurdak, Nicolas Harmansa, Ulrich Getzig

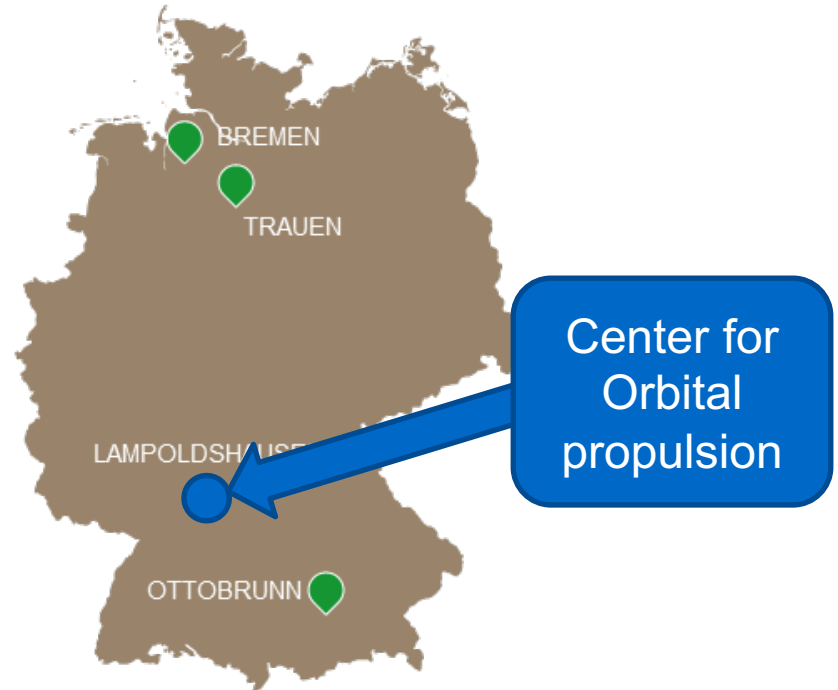
Clean Space Industry Days, Noordwijk

# ARIANEGROUP - LOCATIONS

## FRANCE



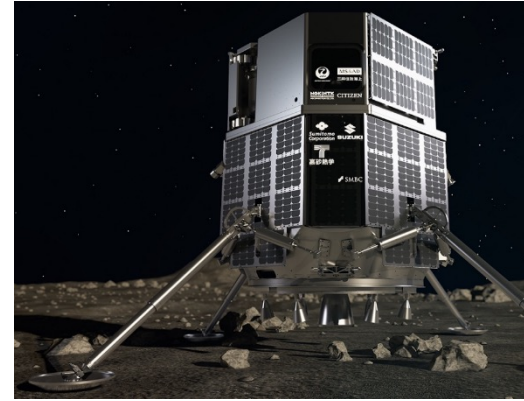
## GERMANY



\* Secondment to the French Navy site on Île Longue

\*\* Locations on the Guiana Space Center (CSG)

# ArianeGroup's Applications in Propulsion



Electric Propulsion is a game changer



# A BRIEF EXCURSION IN ROCKET HISTORY



Hydrazine Hydrate „C-Stoff“ ME 163

While H<sub>2</sub>O<sub>2</sub> was initially used for rocket propulsion [H. Walther] it was replaced by hydrazine due to its better stability and higher performance.

At that time toxicity was not a real issue ...





# Motivation for Green Propellants

## Economic and Legislative

### Advantages

- Lower handling cost
- Fast turnaround times
- Potential higher performance (ISP, density ISP)
- More players in the field

### Challenges

- Heritage, Reliability, Performance, Cost
- Availability of system components

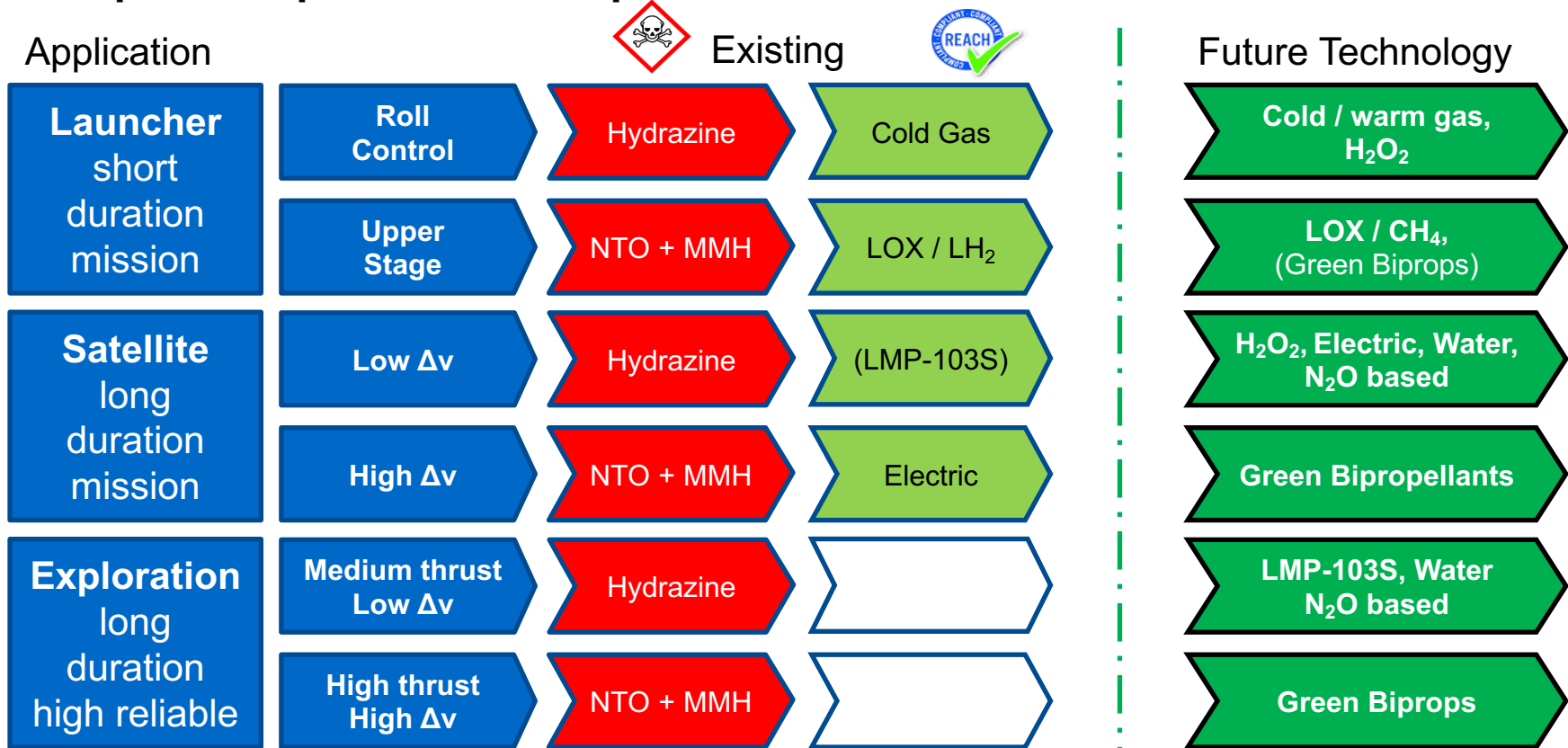
### Legislative

- In 2011 Hydrazine was identified as Substance of Very High Concern and included in Europe's REACH candidate list with a potential ban of its use – since then no further evolution
- Clean Space



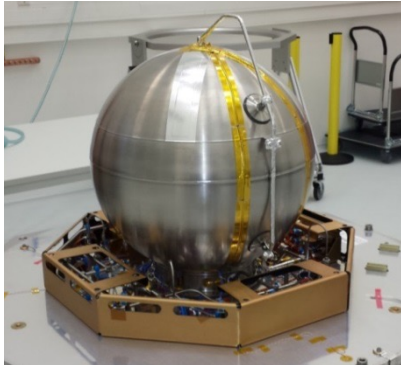
# APPLICATIONS AND TECHNOLOGIES

## for Space Propulsion in Europe

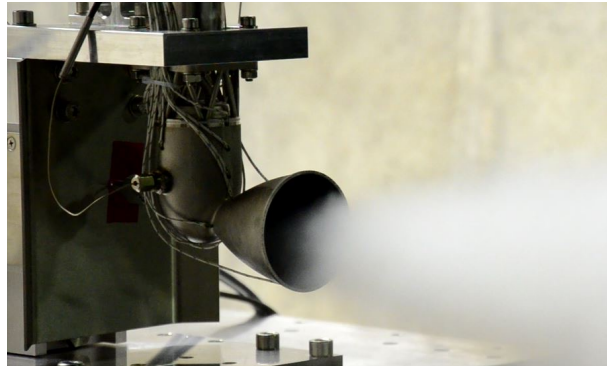




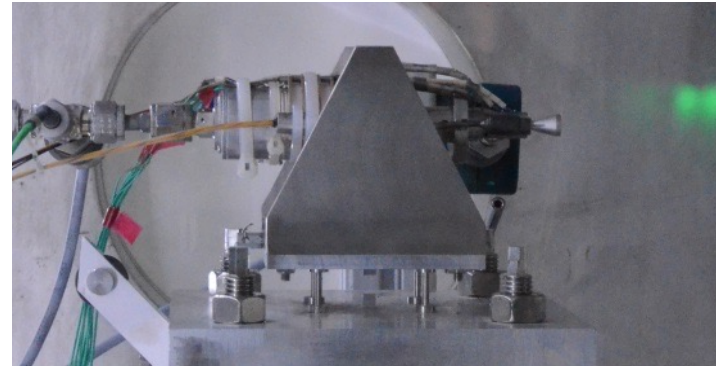
# Investigated Technologies at a glance



LMP-103S System



ALM printed 240N H<sub>2</sub>O<sub>2</sub> Thruster (87,5%)



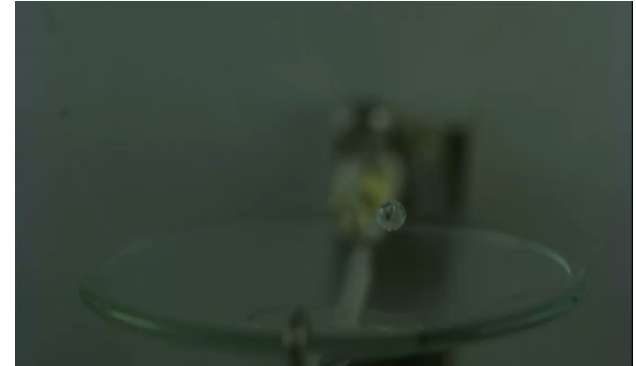
1N H<sub>2</sub>O<sub>2</sub> Thruster (98%)



Stoichiometric GO<sub>2</sub> / GH<sub>2</sub> Thruster



High Pressure PEM Electrolyzer



Hypergolic Bipropellants (H<sub>2</sub>O<sub>2</sub> 97%)



*“In the end, SES does not specify the propulsion technology, but specifies the capability, need date and price target.*

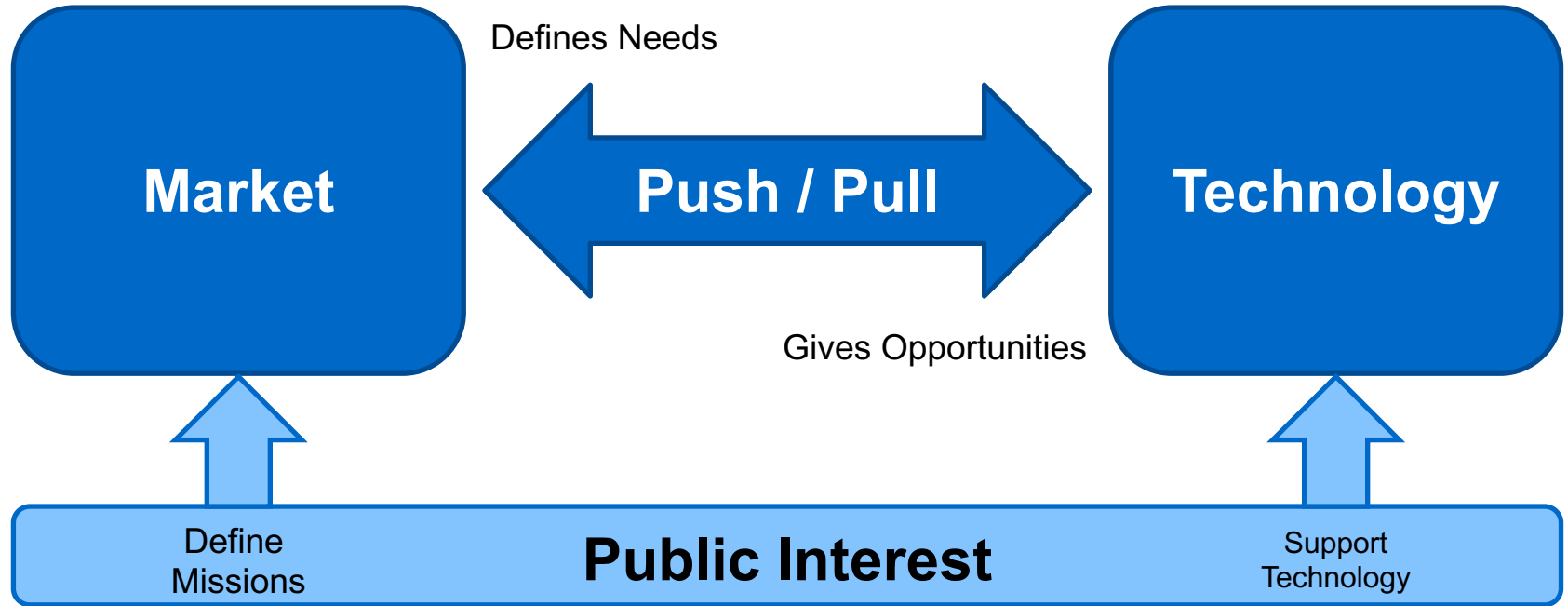
*The satellite vendor presents the most optimal propulsion subsystem(s) to SES for each specific mission”*

*EPIC workshop 2017*












# Challenges to bring Green Propulsion into Orbit

## Propulsion Needs and Trends



# Green Propellant Properties / Technologies

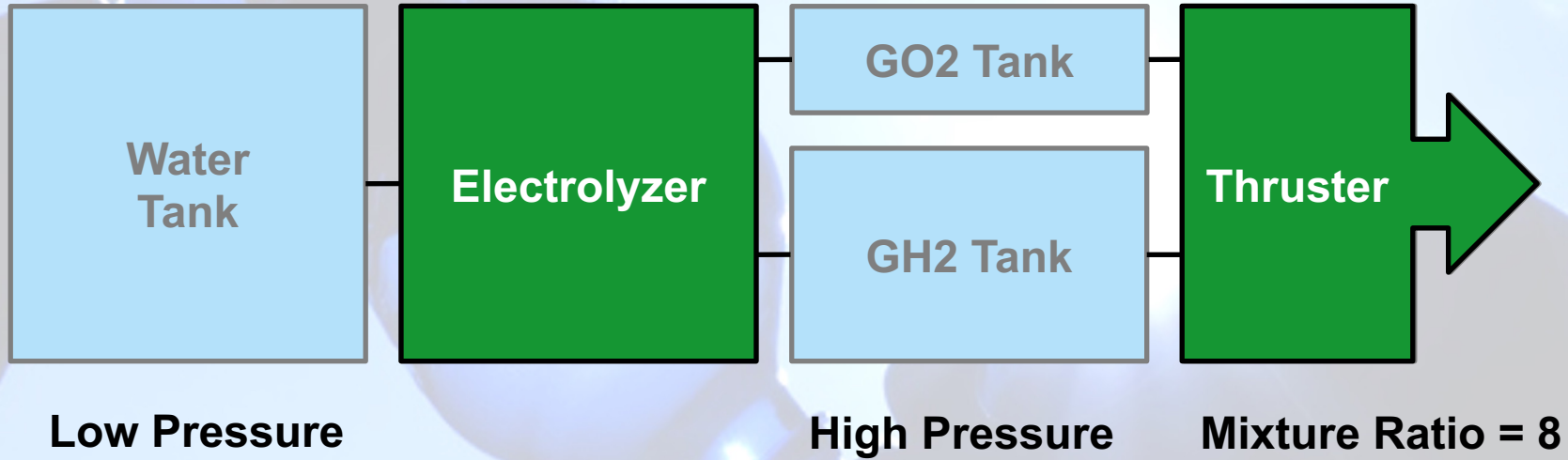
Energetic Propellants cannot be harmless

Hazards	 Flamable	 Corrosive	 Toxic	 Health	 Environm.	 Explosive	 Harmful	 Oxidizing	 Compress.
Hydrazine	X	X	X	X	X	(X)	X		
Ammonium Dinitramide (ADN)	X			X		X	X		
Hydroxylammonium Nitrate (HAN)			X	X	X	X	X		
Hydrogen Peroxide		X				(X)	X	X	
Nitrous Oxide						(X)	X	X	X
Water									

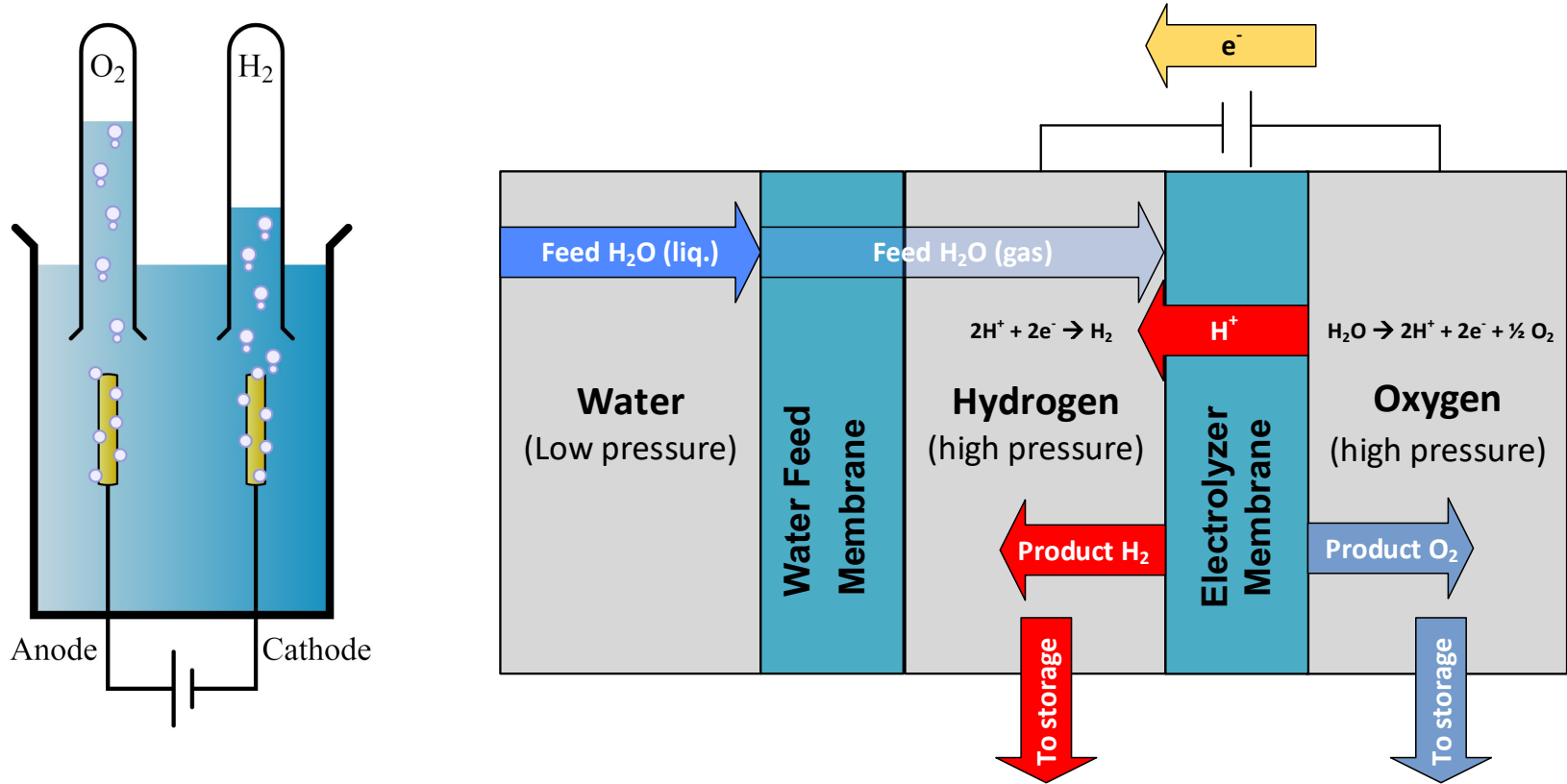
**Still Green when the entire lifecycle is considered?**



# WATER PROPULSION – THE ULTIMATE GREEN



# Challenges and Solutions for the Electrolyzer

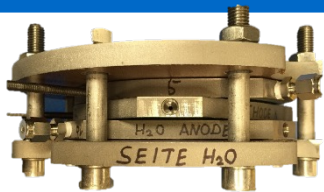


# ELECTROLYZER DEVELOPMENT @ AGG

## Heritage

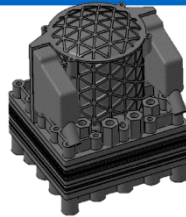
- In house developments
- Key component
- Built and tested to gain confidence
- 100 bar operation

### Lab Model



- Single Cell
- Proof of Concept
- Basic Characterisation

### CubeSat EM



- 2 Cells
- Higher power
- Compact
- Integration of System Infrastructure
- 25 bar
- 10 W per cell

### Upscale DM



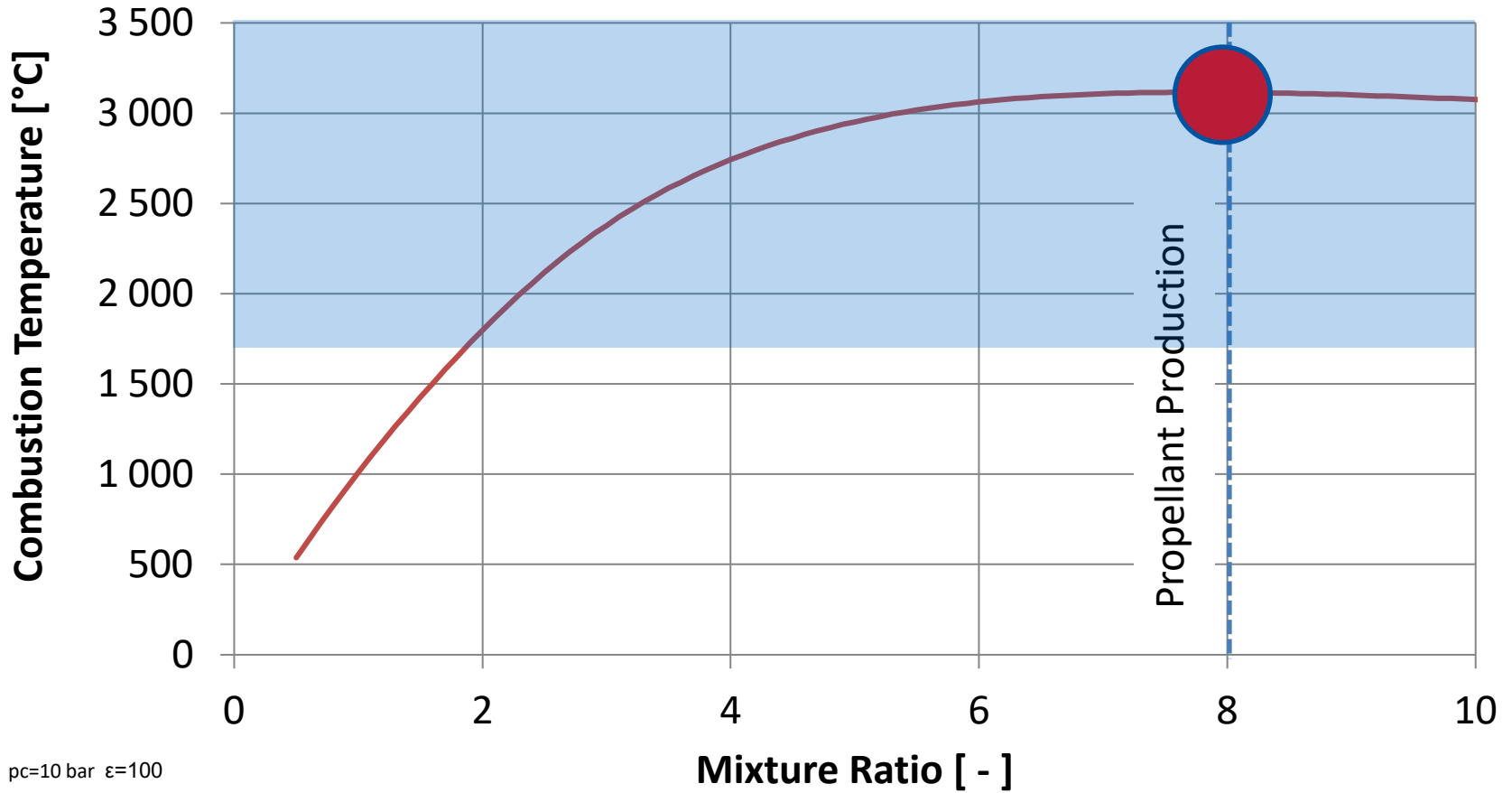
- 4 Cell Design
- Higher power
- High Pressure
- 70 bar
- 35 W per cell

### Flight Design

**Protected**

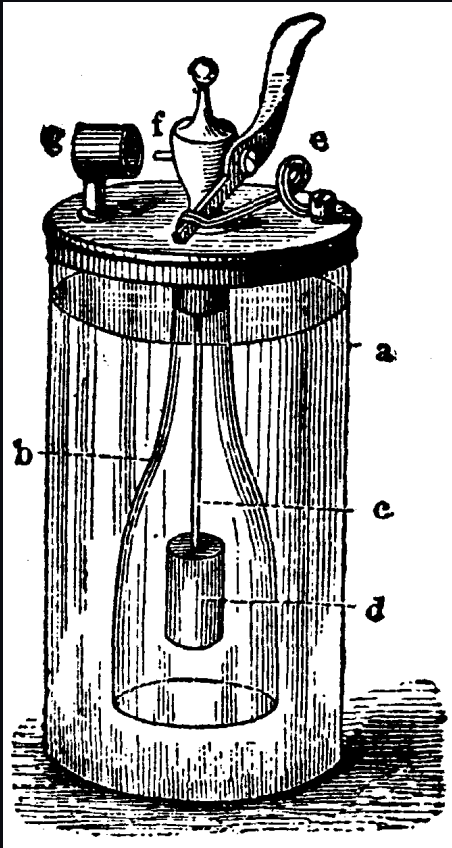
- 4 Cell Design
- Upscale for higher power
- 100 bar
- 50 W per cell

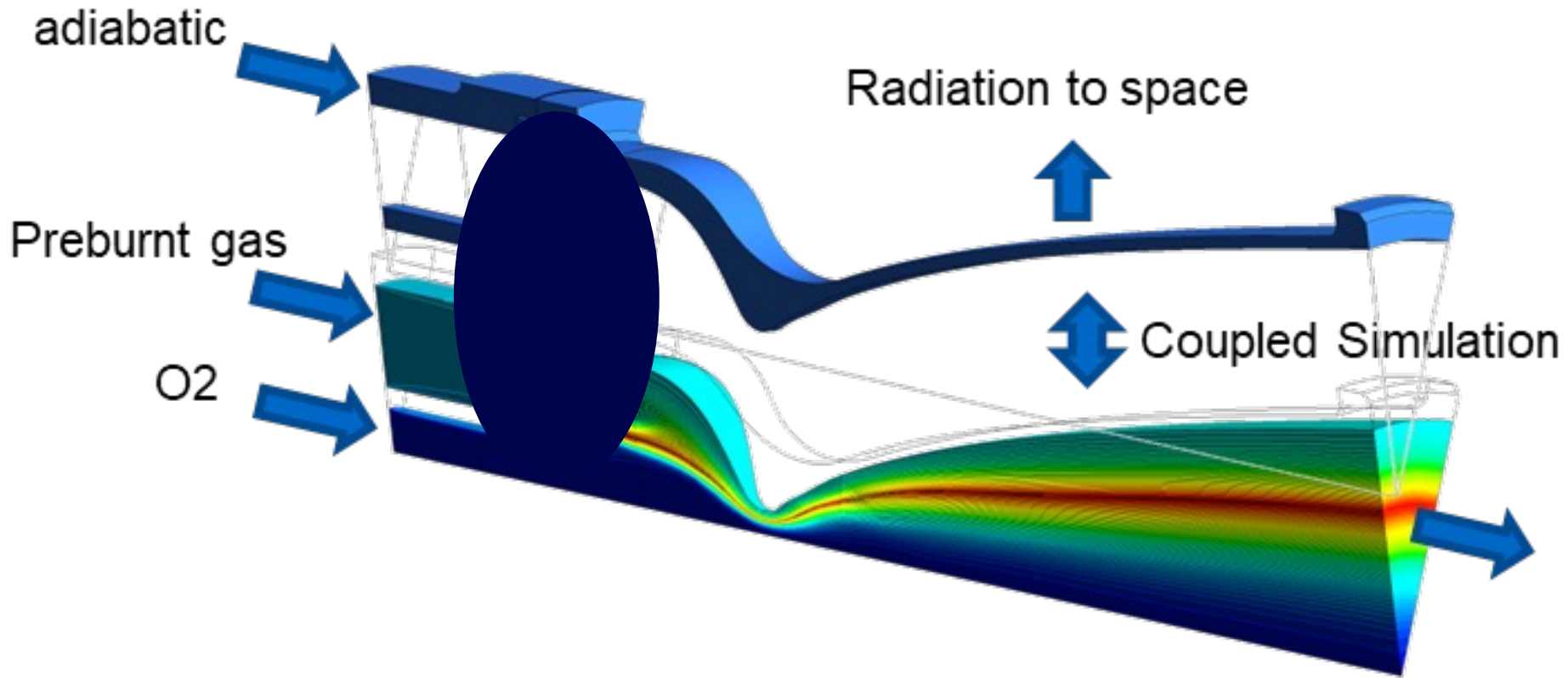




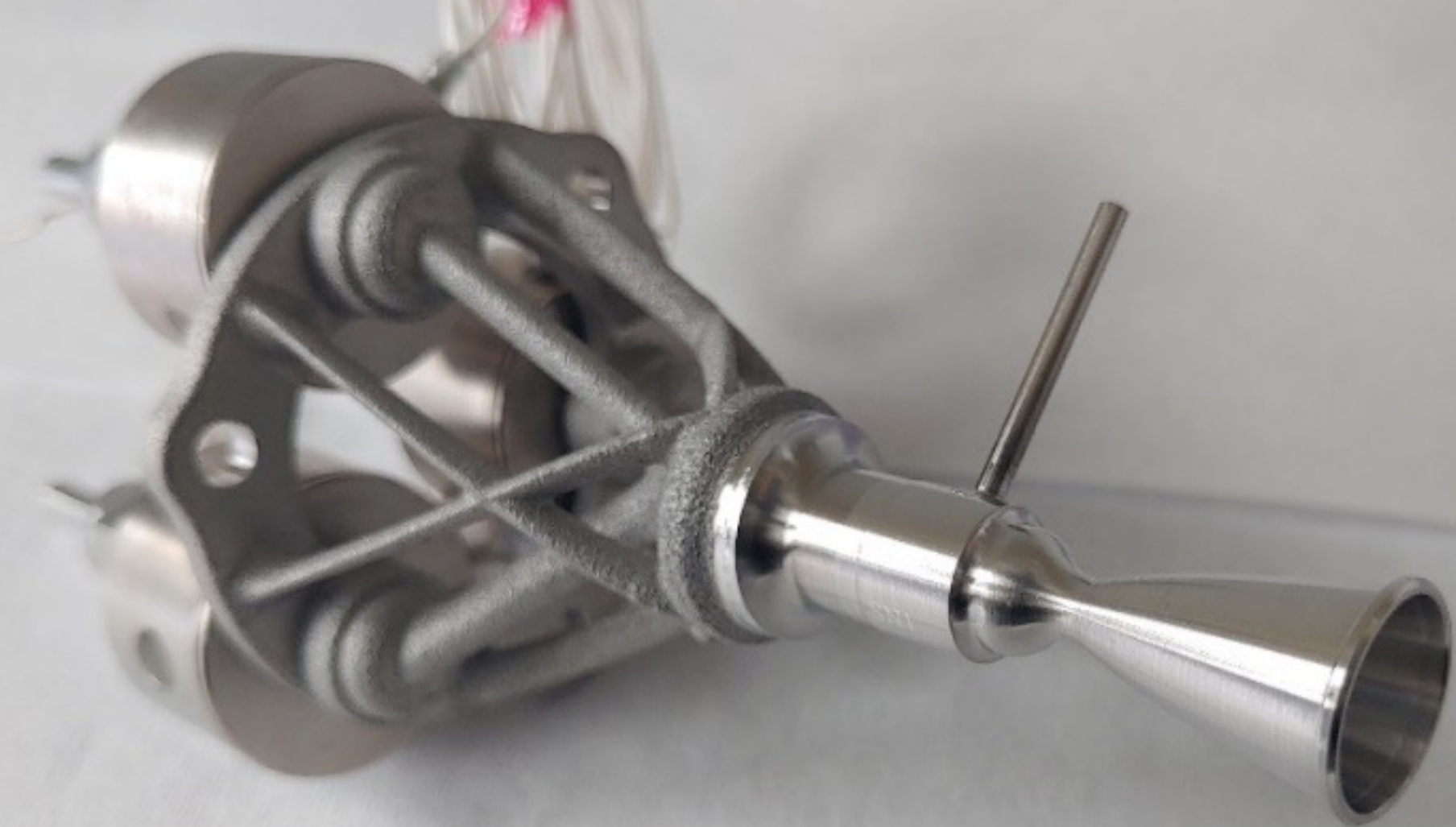
$p_c=10 \text{ bar}$   $\epsilon=100$

# How do we ignite?







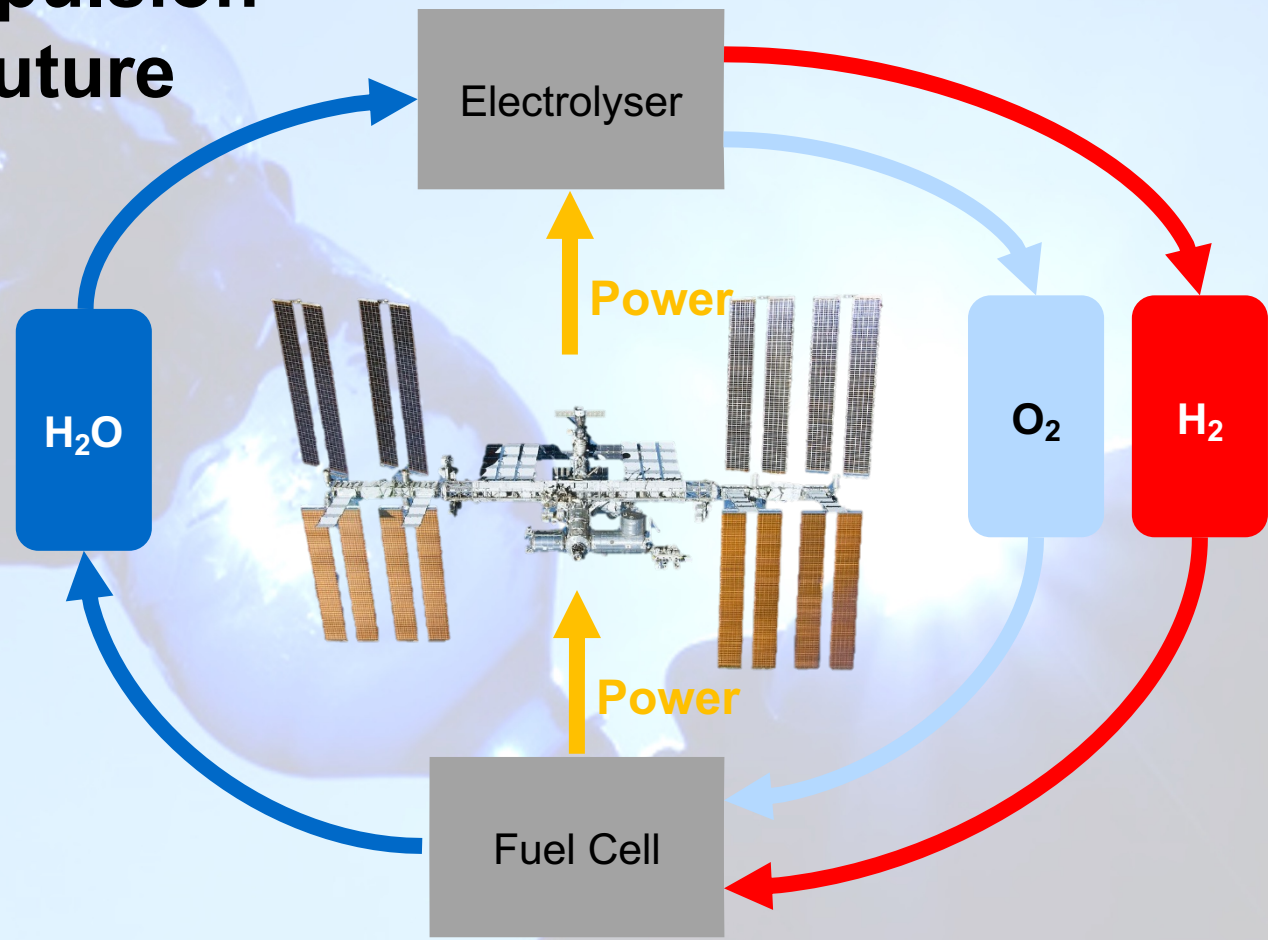


# Thruster Test Program and Results

- Variation of Cat Bed mixture ratio
- Steady state Operation from 7,5 bar to 82,5 bar in continuous and blow down mode
- Equal pressures and 5 bar pressure offset
- Pulse mode firing (MIB, ON- and OFF modulated)
- Coupled Test with Gases produced by the PEM Electrolyser
- Cold gas operation



# Water Propulsion – The Future

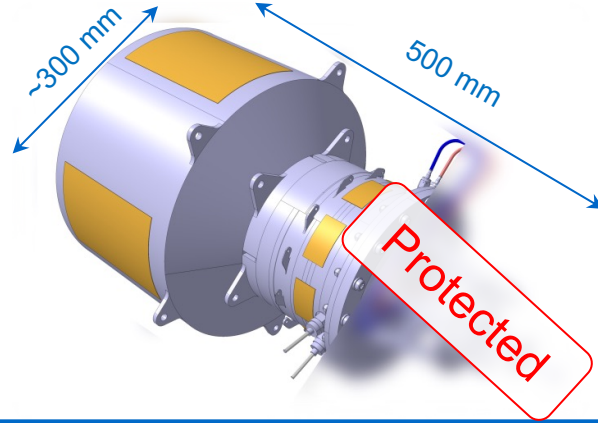




# WATER PROPULSION FLIGHT DEMONSTRATION

## Propulsion Pod

- Highly integrated
- Scalable and modular
- 1 or 4 thrusters
- 60 kNs of water capacity
- 250 W electrolyser
- 40 Ns manoeuvres
- 2N Thruster (ISP > 300s)
- Power and Avionics Unit



Flight  
Demonstrator  
Development  
Kick-Off

Components  
Test

System  
Test

System  
PDR

System  
CDR

System  
DRB  
Integration

Launch

2023

2024

2025

2026

# Water Propulsion



**Has lower overall lifecycle cost**



**Allows quick turnaround times**



**Has a high Performance**



**Is environmentally friendly**



**ariane**GROUP

**THANK YOU FOR YOUR  
ATTENTION!**

**Contact: [malte.wurdak@ariane.group](mailto:malte.wurdak@ariane.group)**