

# Simulating the detector response of MAVEN's Solar Energetic Particle (SEP) instrument with Geant4

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- \* Many slides previously shown in presentations by:
  - Bruce Jakosky, Principal Investigator, LASP, University of Colorado, Boulder
  - Davin Larson, SEP Instrument Lead, SSL, University of California, Berkeley
  - Robert Lillis, SEP Scientist, SSL, University of California, Berkeley

1. Mars Atmosphere and Volatile Evolution (MAVEN) Mission
2. Solar Energetic Particle (SEP) Instrument
3. Geant4 Simulations for SEP



## What Will MAVEN Do? (1 of 2)

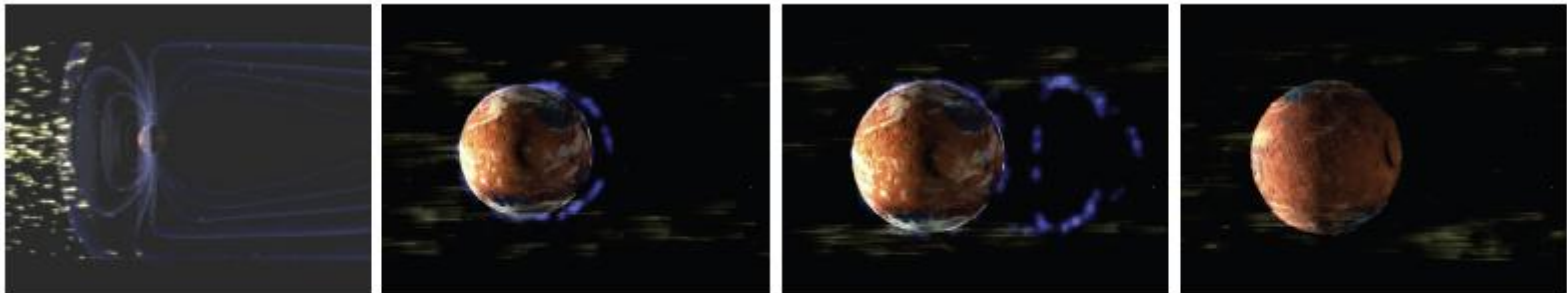


*Ancient Valleys*

Mars' atmosphere is cold and dry today, but there was once liquid water flowing over the surface.

*Where did the water and early atmosphere go?*

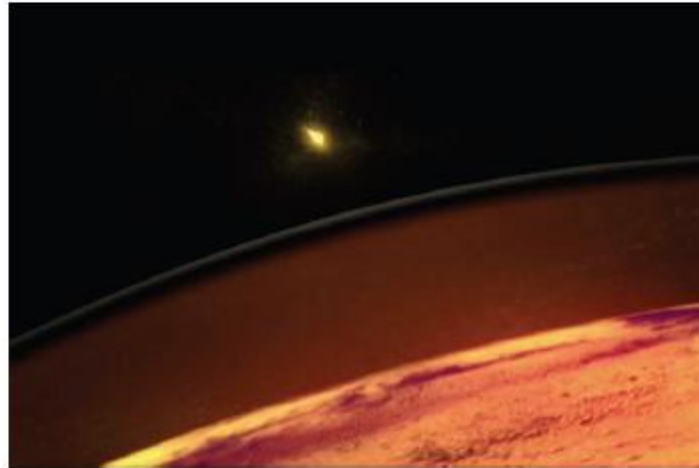
- *H<sub>2</sub>O and CO<sub>2</sub> can go into the crust or be lost to space.*
- *MAVEN will focus on volatile loss to space.*



*Turn-off of the Martian magnetic field allowed turn-on of solar-wind stripping of the atmosphere ~ 3.7 billion years ago; combined with solar-EUV-driven loss, resulted in the present thin, cold atmosphere.*



## What Will MAVEN Do? (2 of 2)



- Determine the structure and composition of the Martian upper atmosphere today
- Determine rates of loss of gas to space today
- Measure properties and processes that will allow us to determine the integrated loss to space through time

*MAVEN will answer questions about the history of Martian volatiles and atmosphere and help us to understand the nature of planetary habitability.*



## The MAVEN Spacecraft

- 3-axis attitude control (wheel based)
- Mono-propellant propulsion system
- Single-fault tolerant during all critical events
- Launch (Wet) Mass: 2550 kg max
- Observatory Dry Mass: 903 kg max
- Power: 1135 W at Mars Aphelion





# The MAVEN Science Instruments

## Mass Spectrometry Instrument



*Neutral Gas and Ion Mass Spectrometer; Paul Mahaffy, GSFC*

## Particles and Fields Package



*Solar Energetic Particles; Davin Larson, SSL*



*SupraThermal and Thermal Ion Composition; Jim McFadden, SSL*

## Remote-Sensing Package



*Imaging Ultraviolet Spectrometer; Nick Schneider, LASP*



*Solar Wind Electron Analyzer; David Mitchell, SSL*



*Solar Wind Ion Analyzer; Jasper Halekas, SSL*



*Langmuir Probe and Waves; Bob Ergun, LASP*

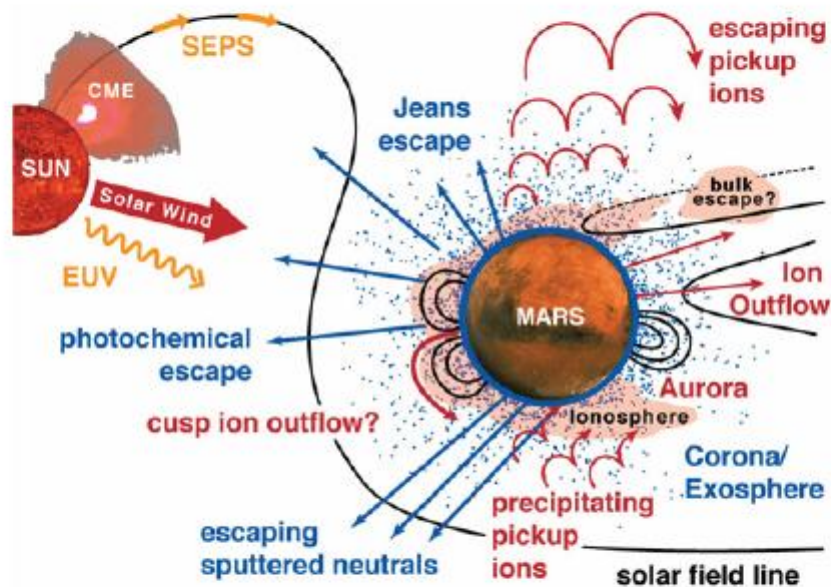


*Magnetometer; Jack Connerney, GSFC*



# MAVEN Will Measure the Drivers, Reservoirs, and Escape Rates

## Solar Inputs



## Plasma Processes



## Neutral Processes



- MAVEN will determine the present state of the upper atmosphere and today's rates of loss to space.

- Measurements will allow determination of the net integrated loss to space through time.

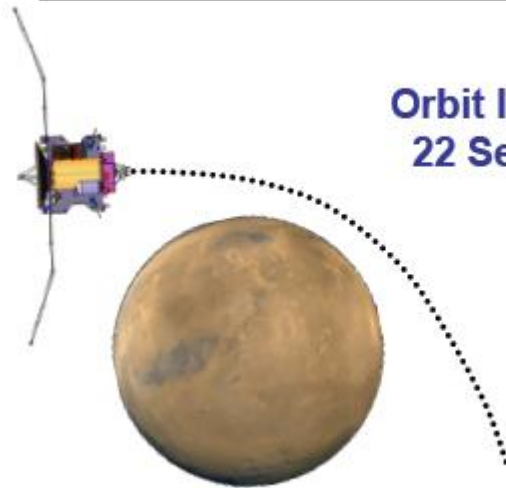
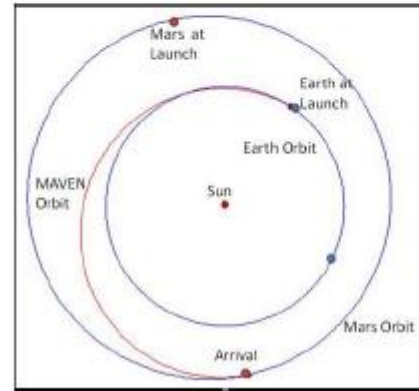


# MAVEN Mission Architecture



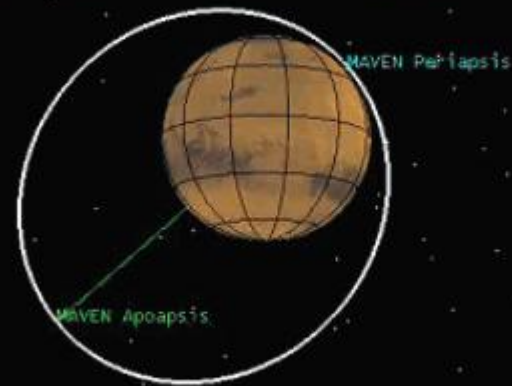
**20-Day Launch Period:  
November 18 –  
December 7, 2013**

## Ten Month Ballistic Cruise to Mars



**Orbit Insertion:  
22 Sept 2014**

## One Year of Science Operations





# Solar Energetic Particle (SEP) Instrument



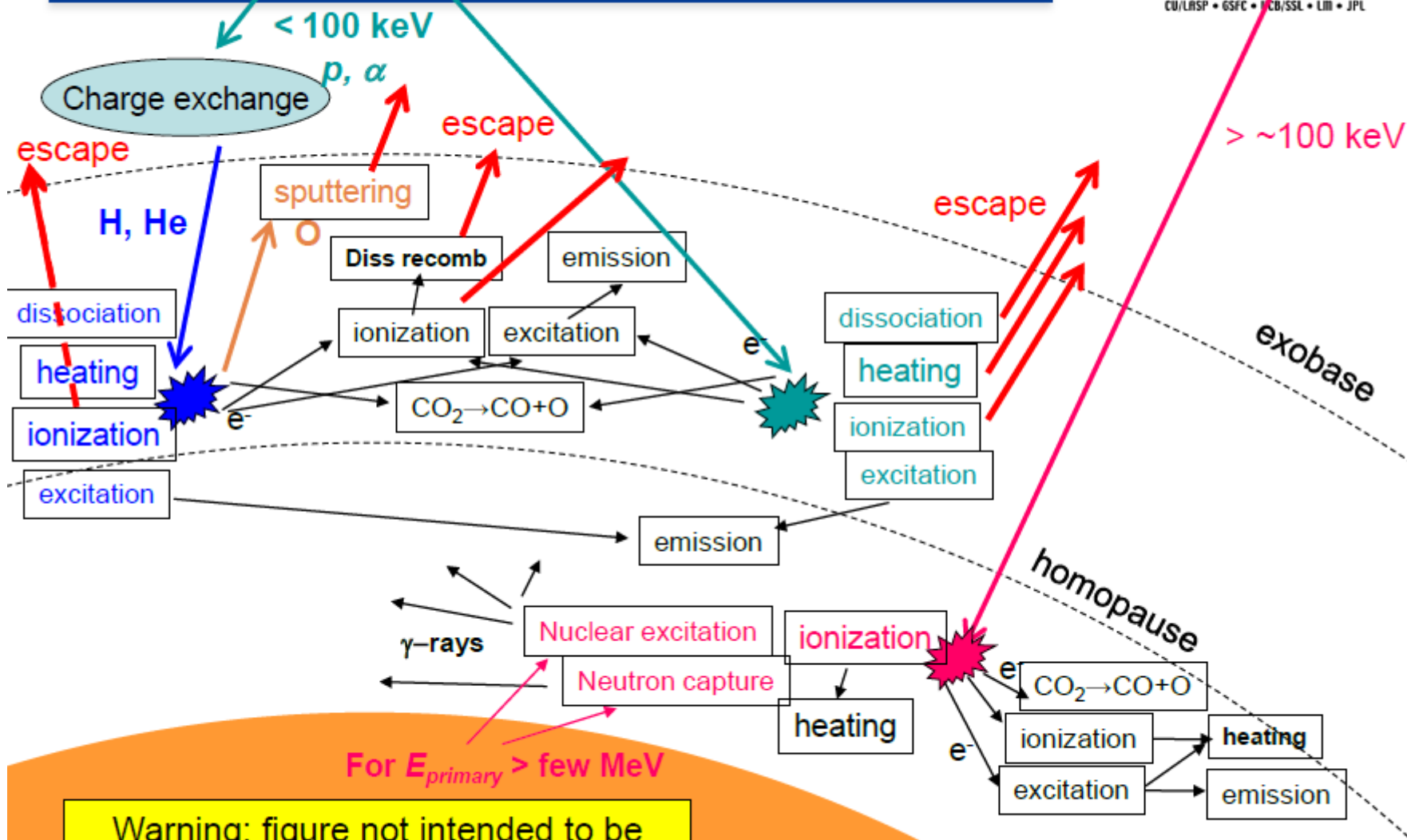
## Measurements will help address important SEP-related questions

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- 1) How & to what degree is the Mars atmosphere shielded
  - by the planetary-scale magnetosphere?
  - by the crustal magnetic fields?
  
- 2) What are the ways in which incident SEPs of various energies affect the atmosphere?
  - SEP spectrum degradation with altitude?
  - Sputtering by energetic neutrals created via charge exchange?
  - Bulk heating from SEP collisions with neutrals ?
  - SEP impact ionization → dissociative recombination → escape
  - Electronic excitation → auroral emission?
  - Molecular dissociation → effect on atmospheric chemistry?
  - Ionospheric currents?

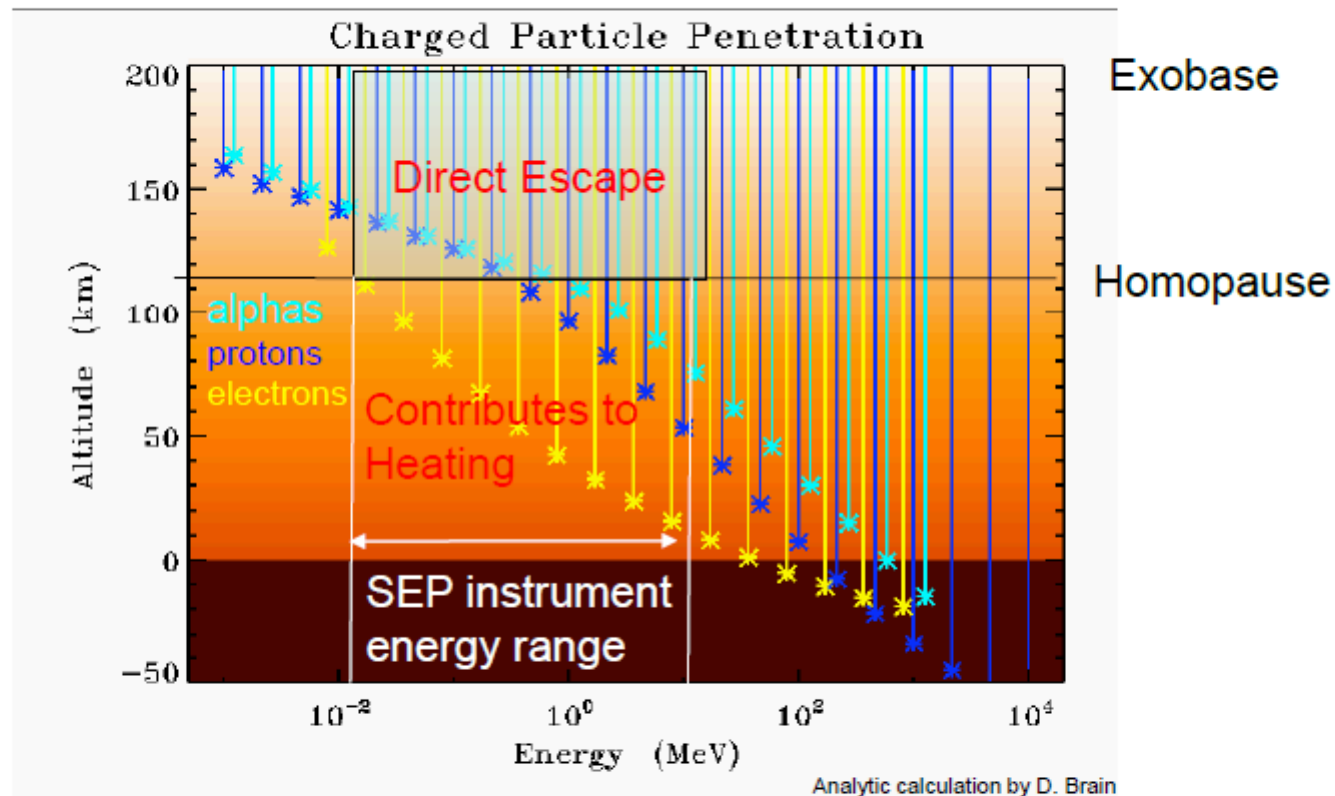
# SEPs in the atmosphere: energy partitioning is complicated and energy dependent



For  $E_{primary} > \text{few MeV}$

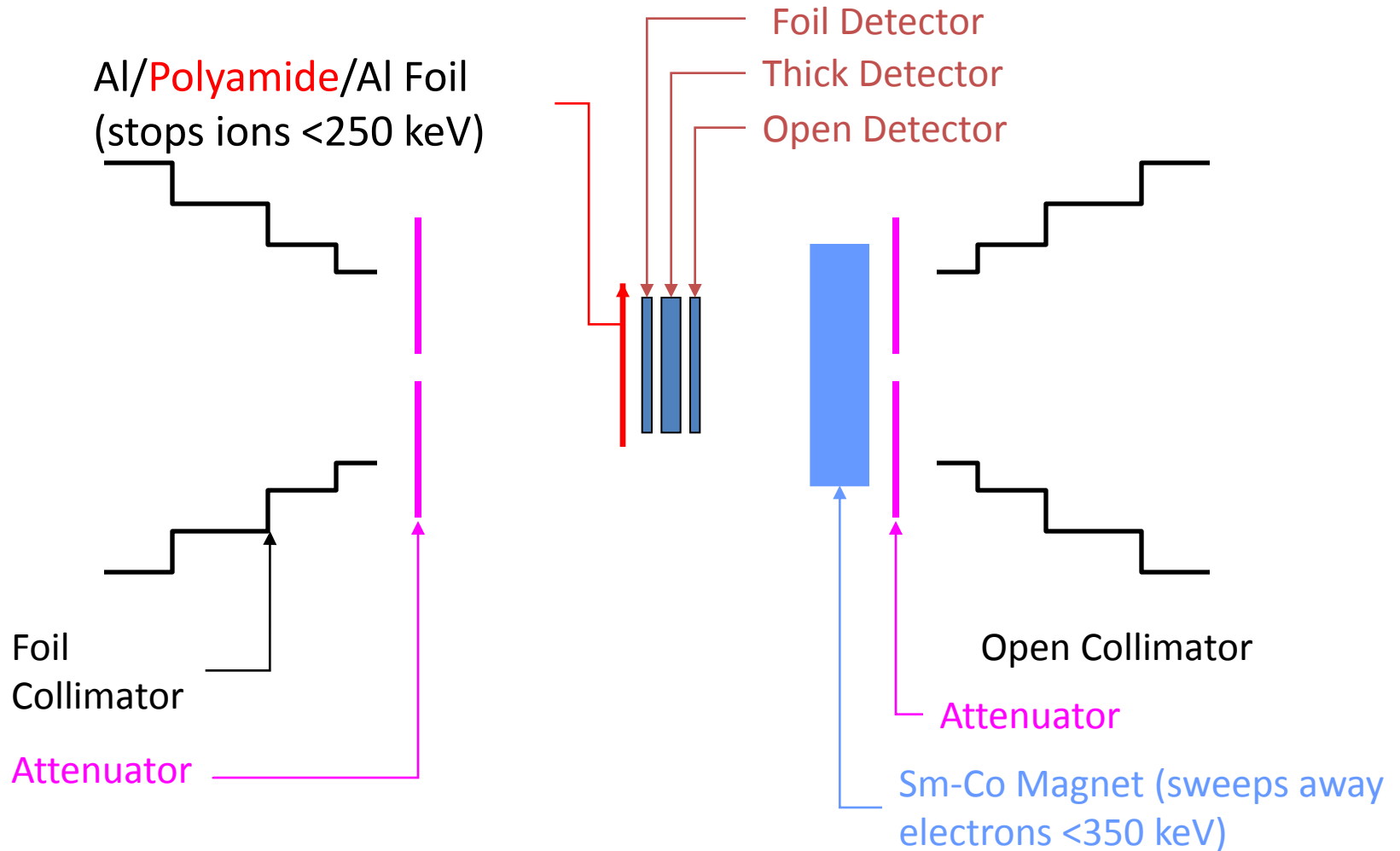
Warning: figure not intended to be fully comprehensive!

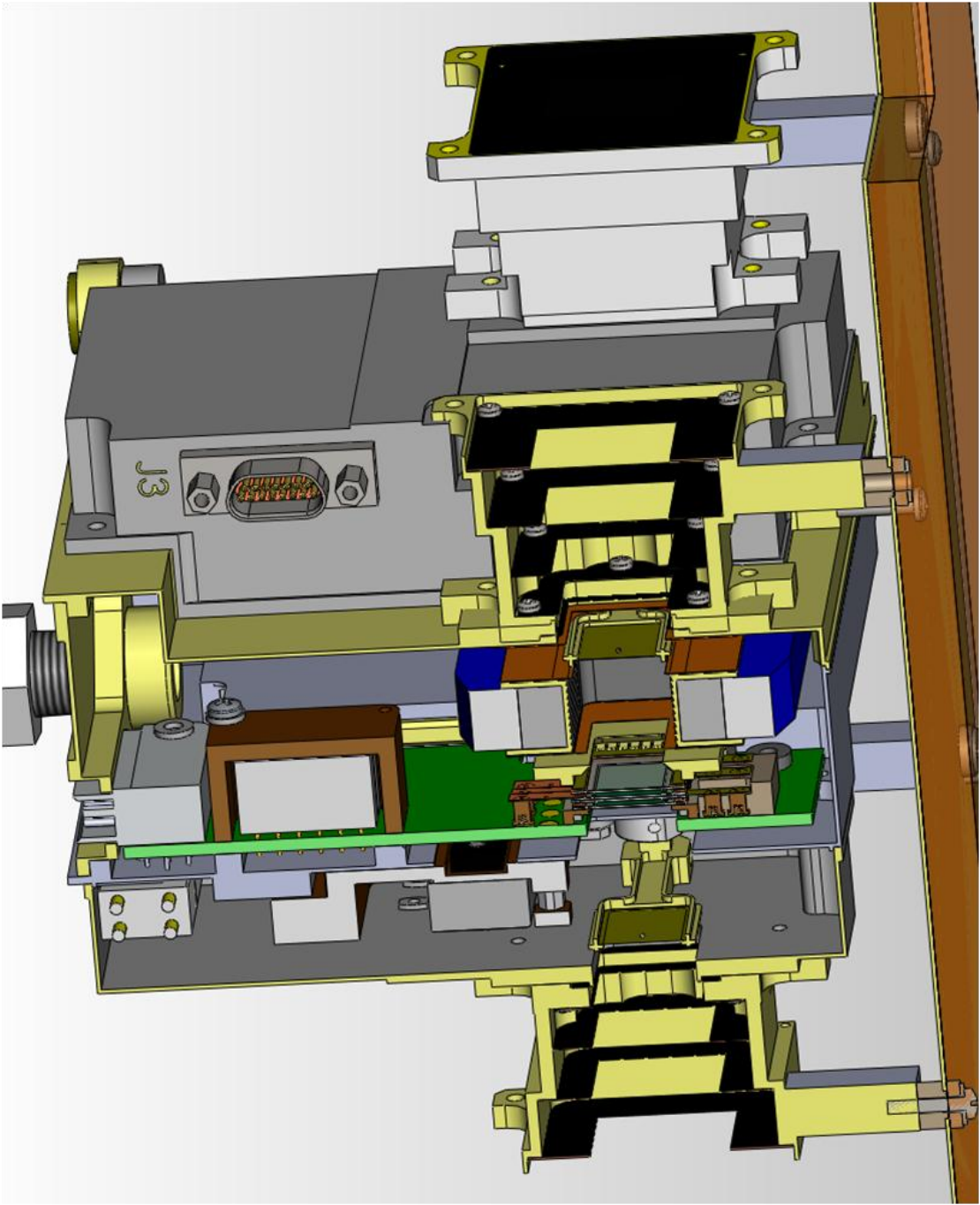
SEP instrument will measure particles that penetrate to altitudes important for escape processes.

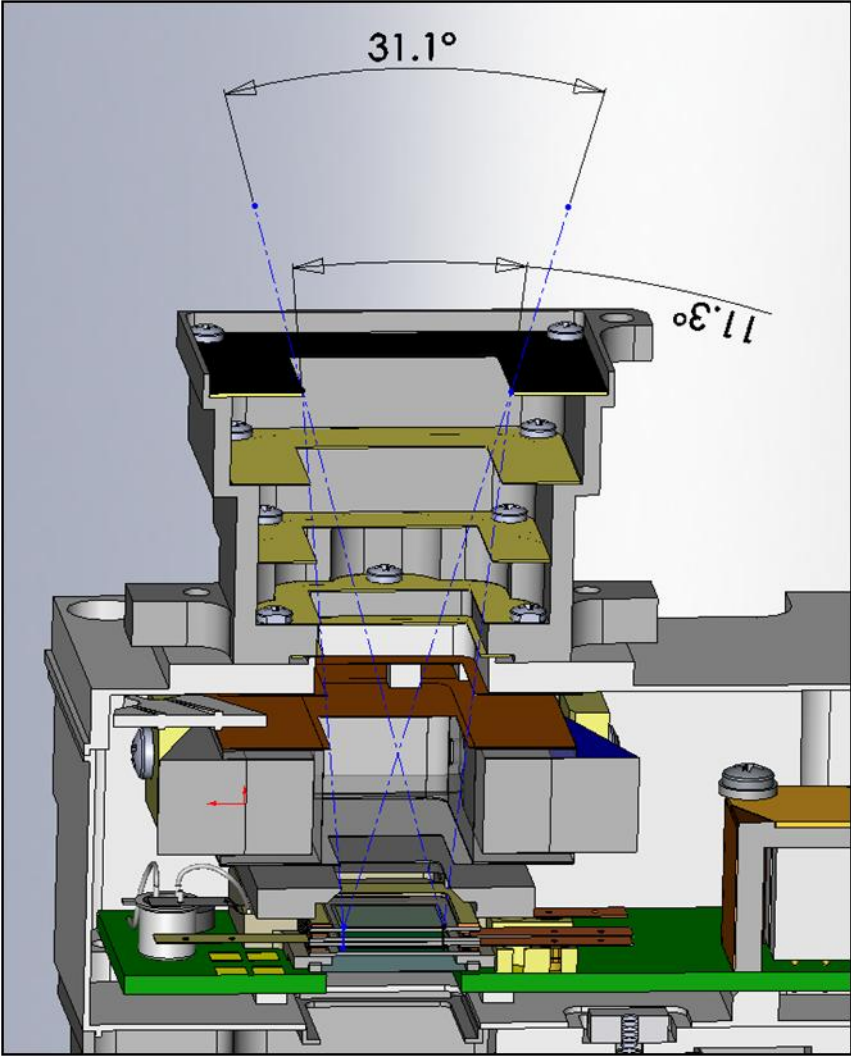


- The bulk of SEP event total energy is generally below 50 keV, deposited mostly between 100 km and 130 km [LeBlanc et al., 2002], though events widely vary.
- We will measure particles that penetrate to 50 km-150 km, providing important constraints on modeling of atmosphere/ionosphere dynamics.

# Basic design of SEP instrument



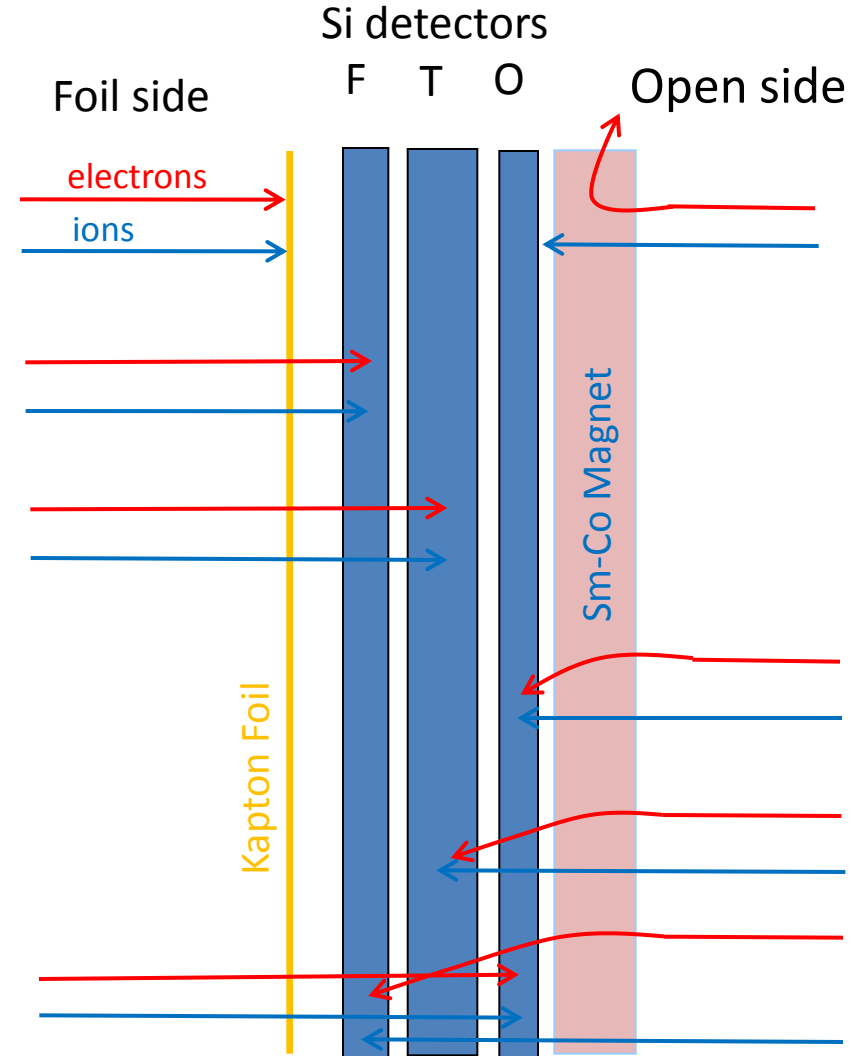




# Basic separation strategy: 3 detectors, 2 filters

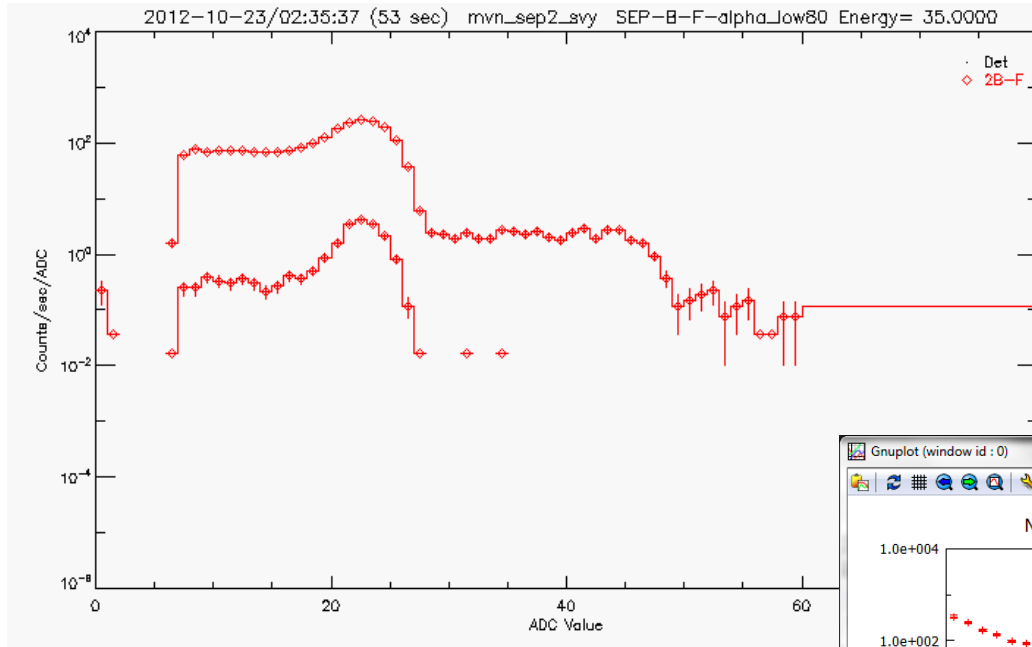
Energy ranges for counted events

keV	Electrons		Ions	
	Foil side	Open side	Foil side	Open side
No count	<20	<350	<250	<15
F	20-700	X	250-6000	X
FT	350-1300	X	6000-11,000	X
O	X	350-700	X	15-6000
OT	X	350-1300	X	6000-11,000
FTO	300-2000		>11,000	



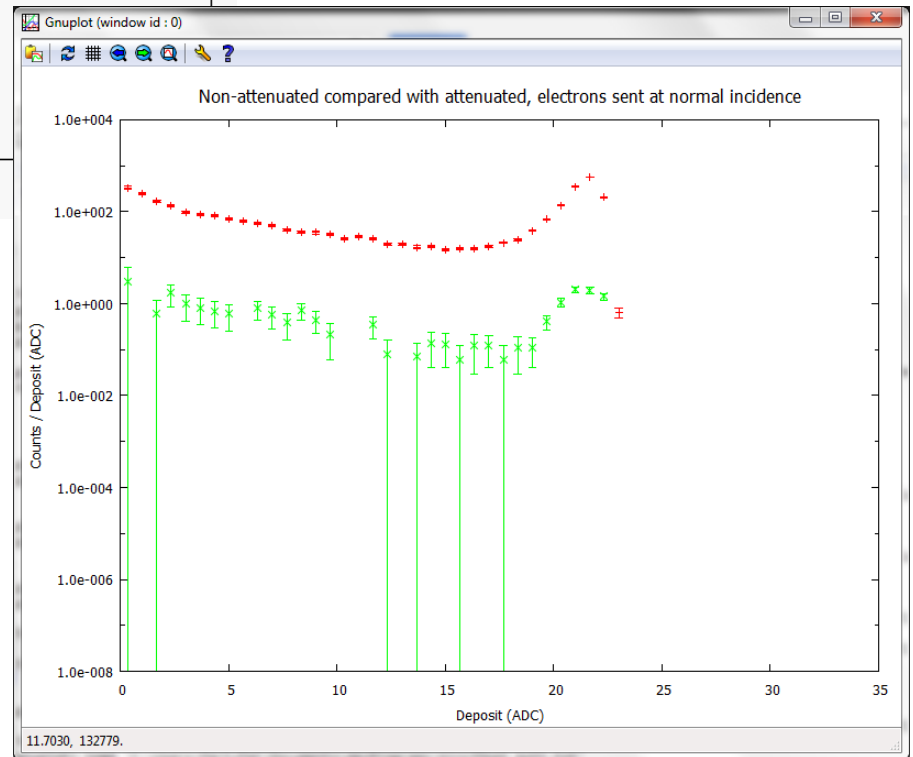


# GAINING THE TRUST OF AN INSTRUMENT LEAD

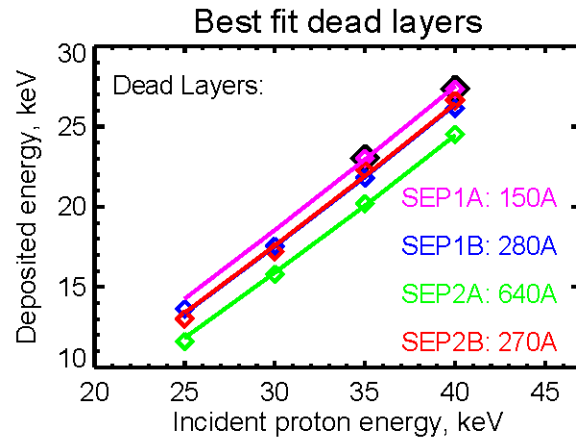
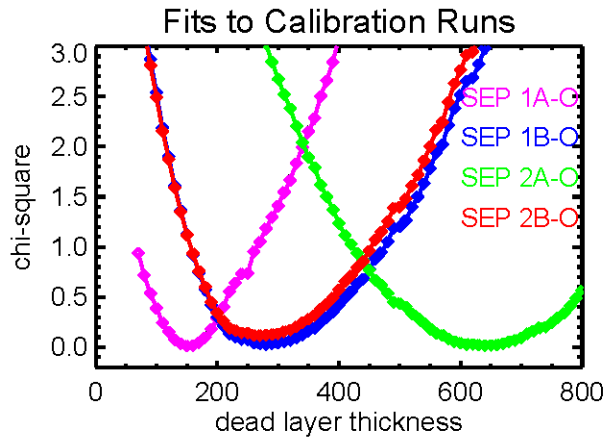


*Experimental results,  
electron gun*

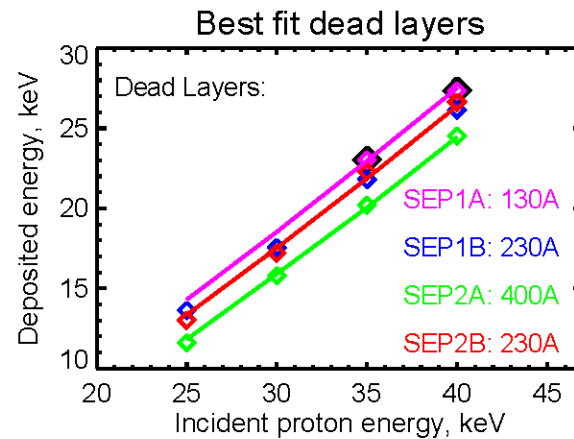
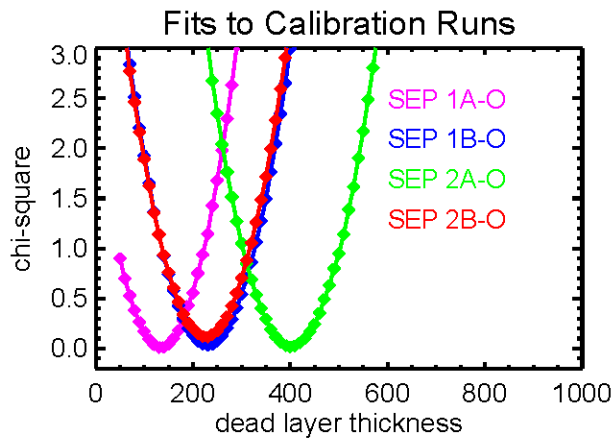
*Simulation results, Geant4*



# DETERMINING THICKNESS OF DETECTOR "DEAD" LAYERS

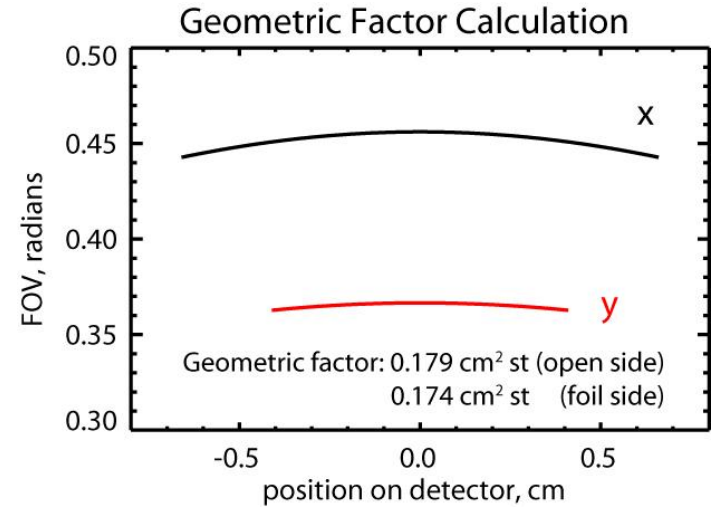
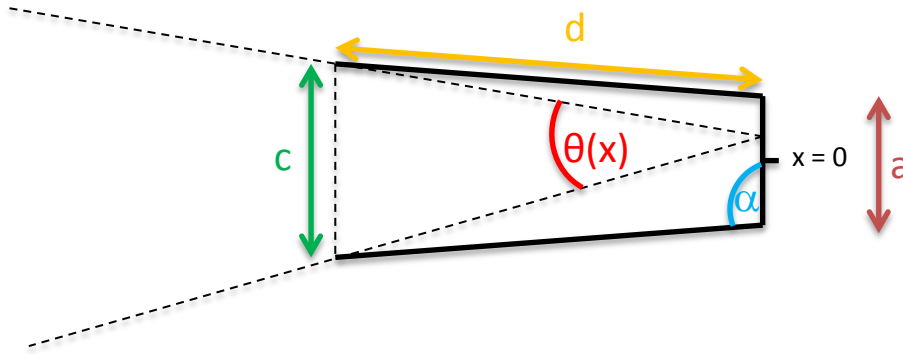


*Oops!  
Misplaced  
volume.*



*OK, much  
better! 😊*

# GEOMETRIC FACTOR CALCULATED GEOMETRICALLY



- Rectangular field of view: consider  $x, y$  separately
- Determine opening angles  $\theta(x), \phi(y)$  as a function of position on the detector.
- Integrate across the detector in  $x$  and  $y$ .
- Geometric factors:
  - 0.179 cm<sup>2</sup> st (open side)
  - 0.174 cm<sup>2</sup> st (foil side)
- GEANT4 simulations on following slide.

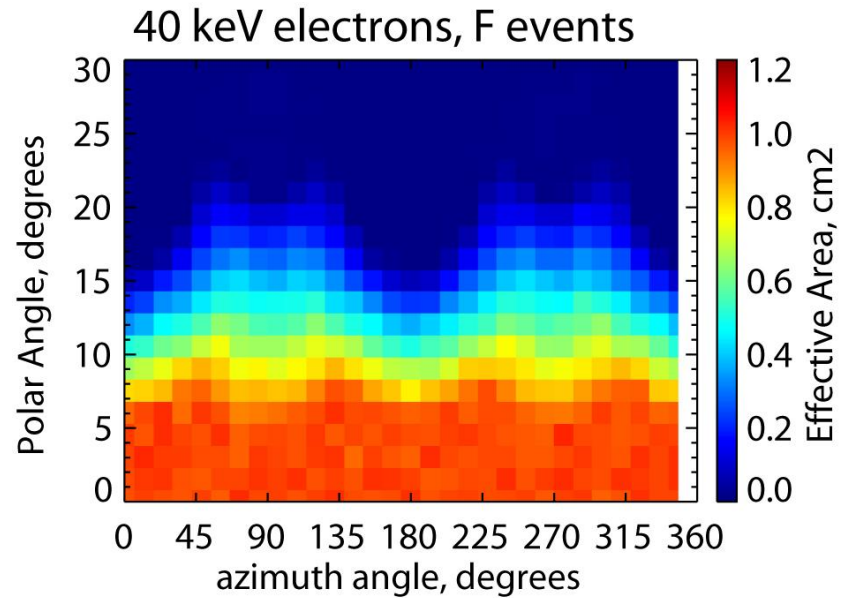
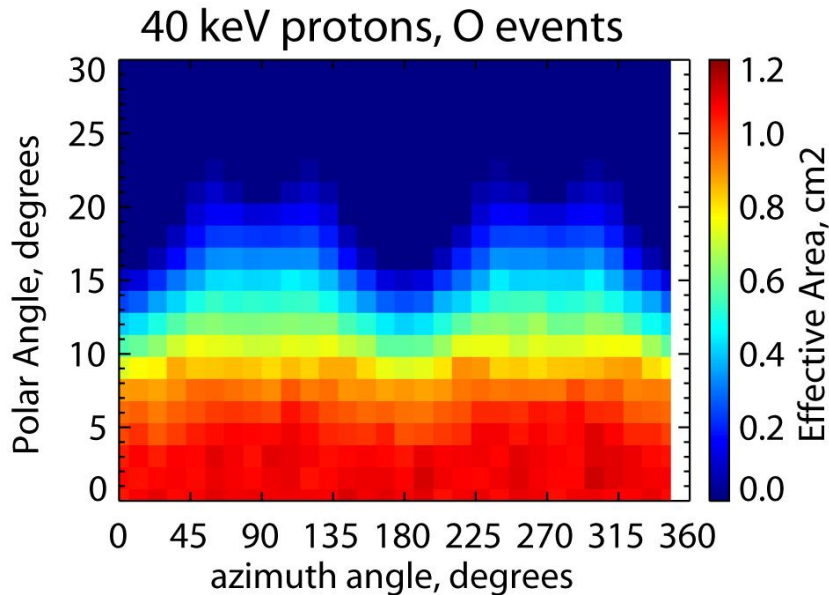
$$p_1^2 = d^2 + \left(\frac{a}{2} + x\right)^2 - 2d\left(\frac{a}{2} + x\right)\cos\alpha$$

$$p_2^2 = d^2 + \left(\frac{a}{2} - x\right)^2 - 2d\left(\frac{a}{2} - x\right)\cos\alpha$$

$$\theta(x) = \cos^{-1}\left(\frac{p_1^2 + p_2^2 - c^2}{2p_1p_2}\right)$$

$$GF = \int_{-b/2}^{b/2} \int_{-a/2}^{a/2} \theta(x)\phi(y) dx dy$$

# GEOMETRIC FACTOR CALCULATED WITH GEANT4 SIMULATIONS



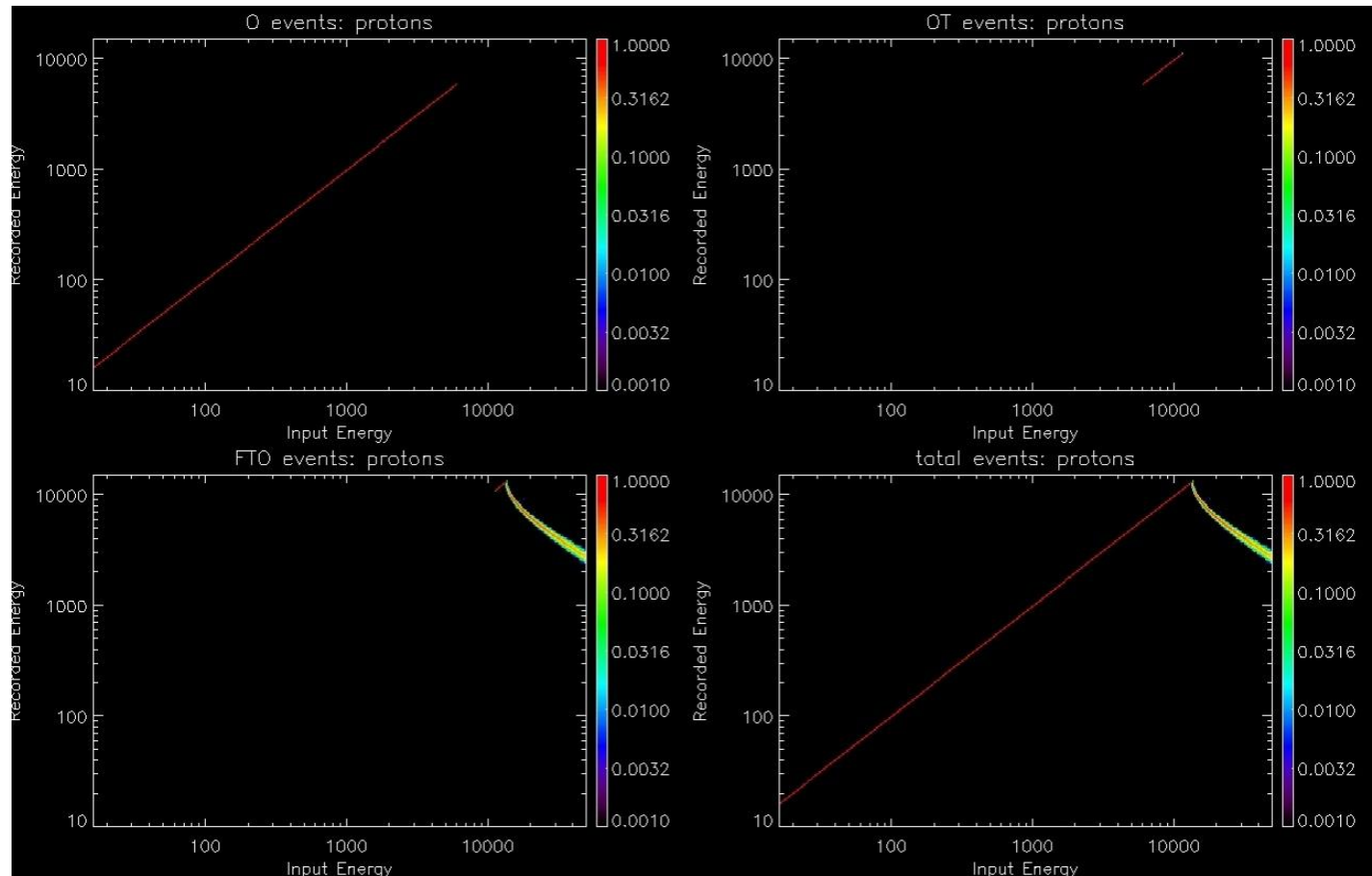
- 40 keV electrons. Foil side.
- Beam Area : 7.07 cm<sup>2</sup>
- # particles per angle: 20,000
- $GF = 0.1689 \pm .0025$  cm<sup>2</sup> sr
- 1.47% error from counting statistics
- 2.87% lower than simple geometric value of 0.174 cm<sup>2</sup> sr (see previous slide)

- 40 keV protons. Open side.
- Beam Area : 7.07 cm<sup>2</sup>
- # particles per angle: 20,000
- $GF = 0.1769 \pm .0027$  cm<sup>2</sup> sr
- 1.54% error from counting statistics
- 1.17 % lower than simple geometric value of 0.179 cm<sup>2</sup> sr (see previous slide).

$$GF = \frac{A_{beam}}{N_{incident}} \int_0^{2\pi} d\phi \int_0^{\pi/2} N_{detected}(\phi, \theta) \sin \theta d\theta$$

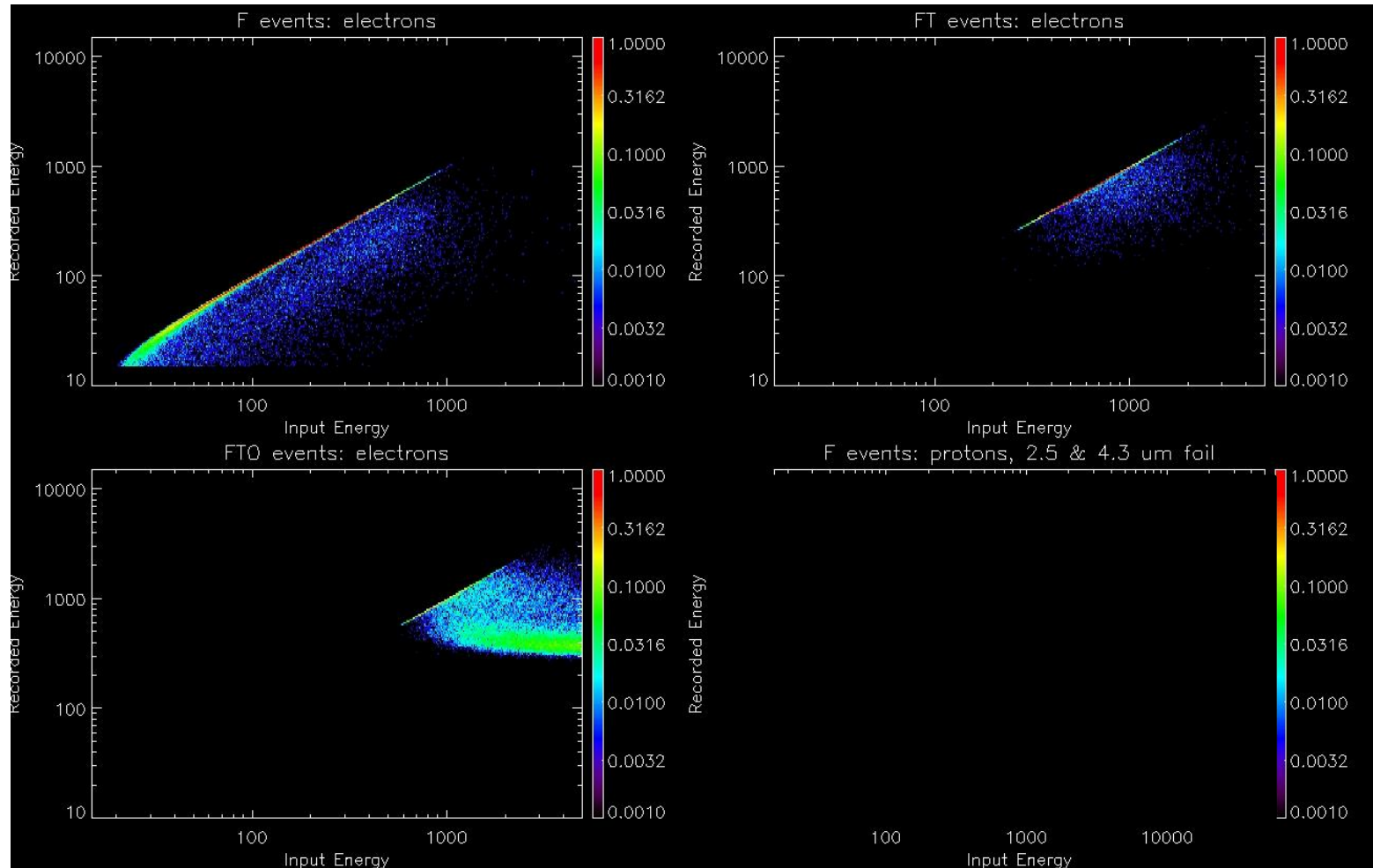
# Ion response matrices

- Since ions deposit most of their energy at the very end of their trajectory, proton response matrix for O, OT and FTO events is quite clean.



# Electron response matrices

- Electrons deposit their energy over a much longer distance than ions. Response matrices are much less diagonal.



# SUMMARY

## SEP tasks already completed with Geant4:

- Calculation of width of “dead” layer coating on detectors
- Calculation of low-energy (25 keV) geometric factor
- Minimum energy for Oxygen ions ( $O^+$ ) to reach detector

## SEP tasks yet to be completed (ongoing) with Geant4:

- Response matrices of detectors
- ???