

Introduction

In light of the new ESA space debris mitigation requirements and further emphasis on pre-mission risk assessment, mission planners rely on models like MASTER or ORDEM to give accurate simulations of the debris environment. Unfortunately, these have significant discrepancies in the sub-millimetric realm.

While objects larger than 10cm can be reliably tracked from the ground, subcentimeter space debris can only be detected in-situ or by examining space-exposed surfaces returned from orbit. Although past validation of debris models relied on such surfaces, they only recorded impacts from 1984 to 2002 and covered a limited range of altitudes and inclinations [1]. In-situ detection methods have not been widely implemented yet, but missions like DEBIE and APID/ADLER have demonstrated the feasibility of such sensors [2,3].

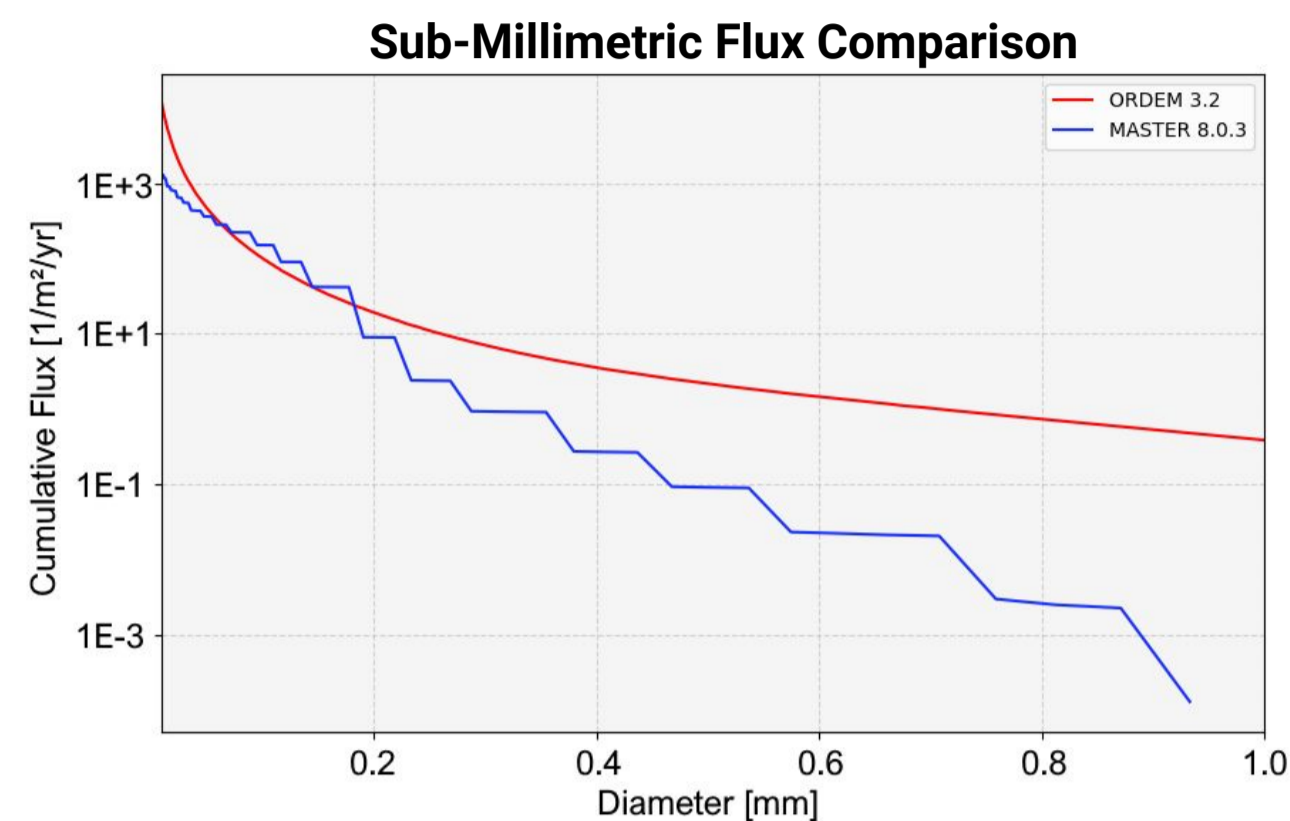


Figure 1: Difference in Sub-millimetric Flux in a 600 km SSO between MASTER 8.0.3 and ORDEM 3.2

Debris Density Retrieval and Analysis (DEDRA)

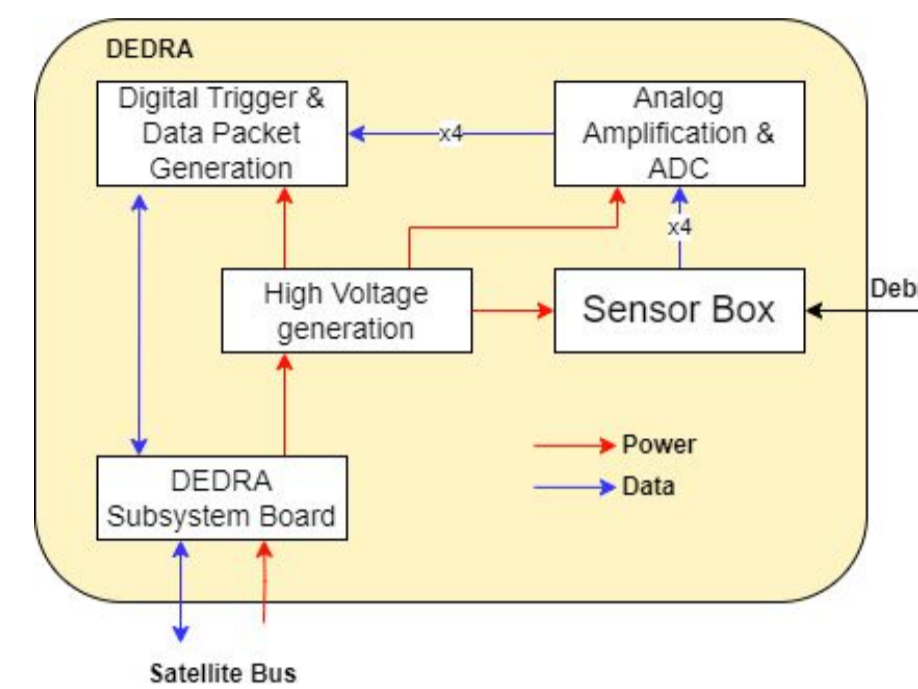


Figure 5: Schematic of the DEDRA sensor architecture

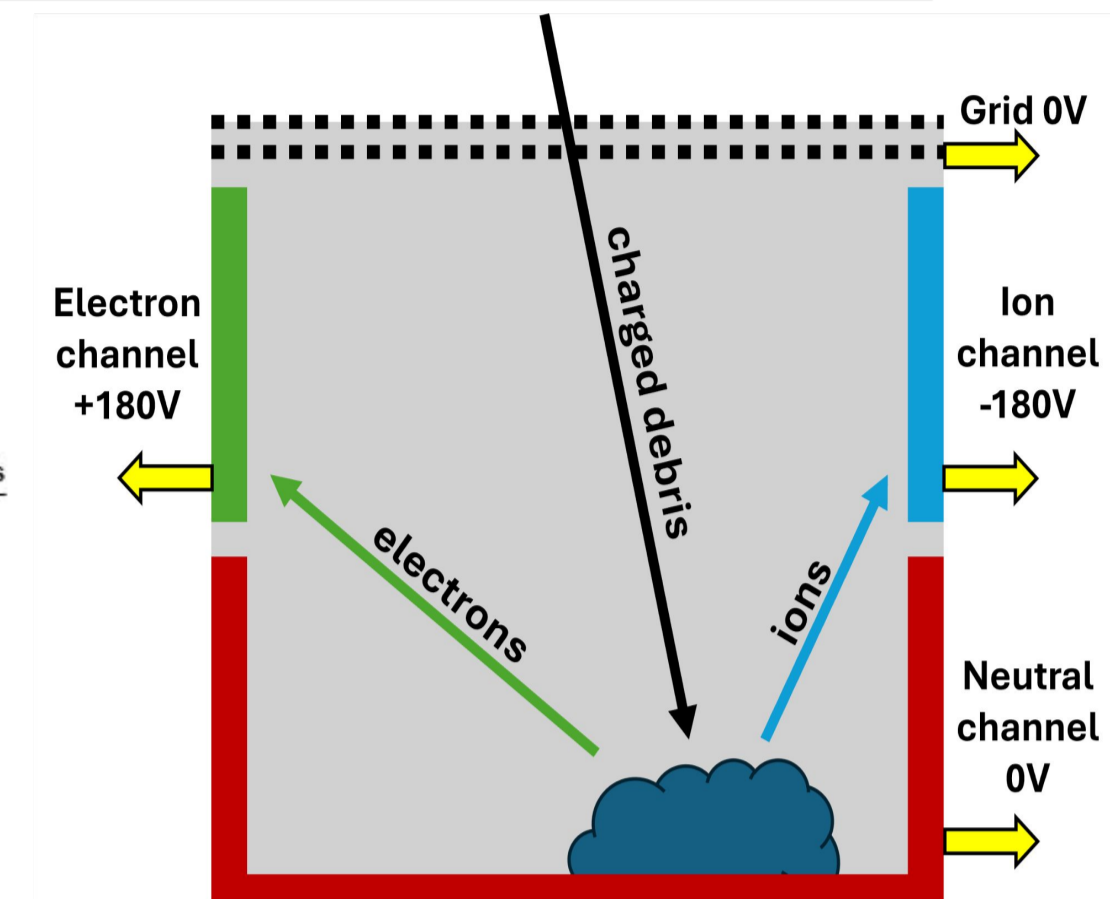


Figure 6: Simplified schematic of the DEDRA sensor box

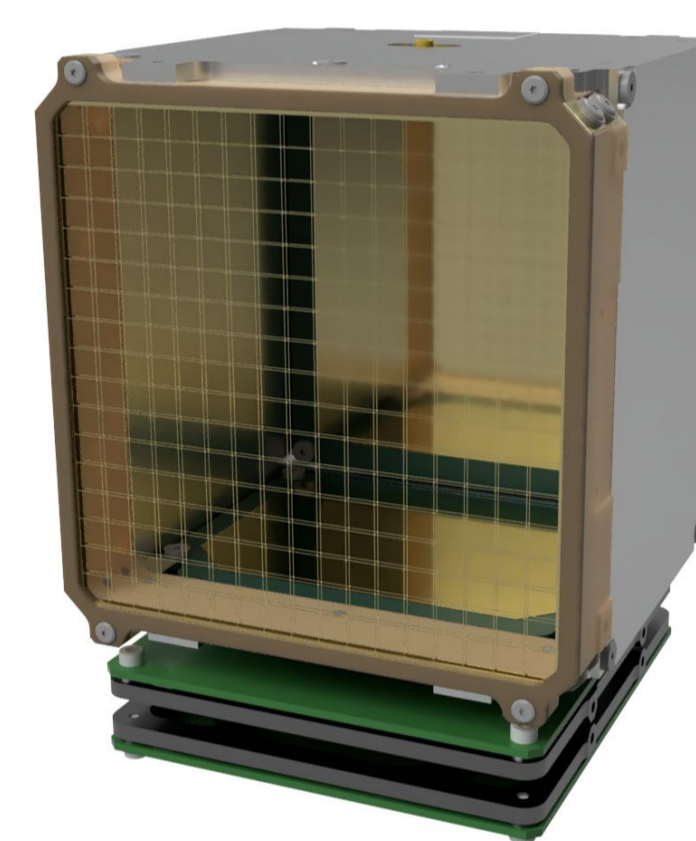


Figure 7: Rendering of the DEDRA CAD model

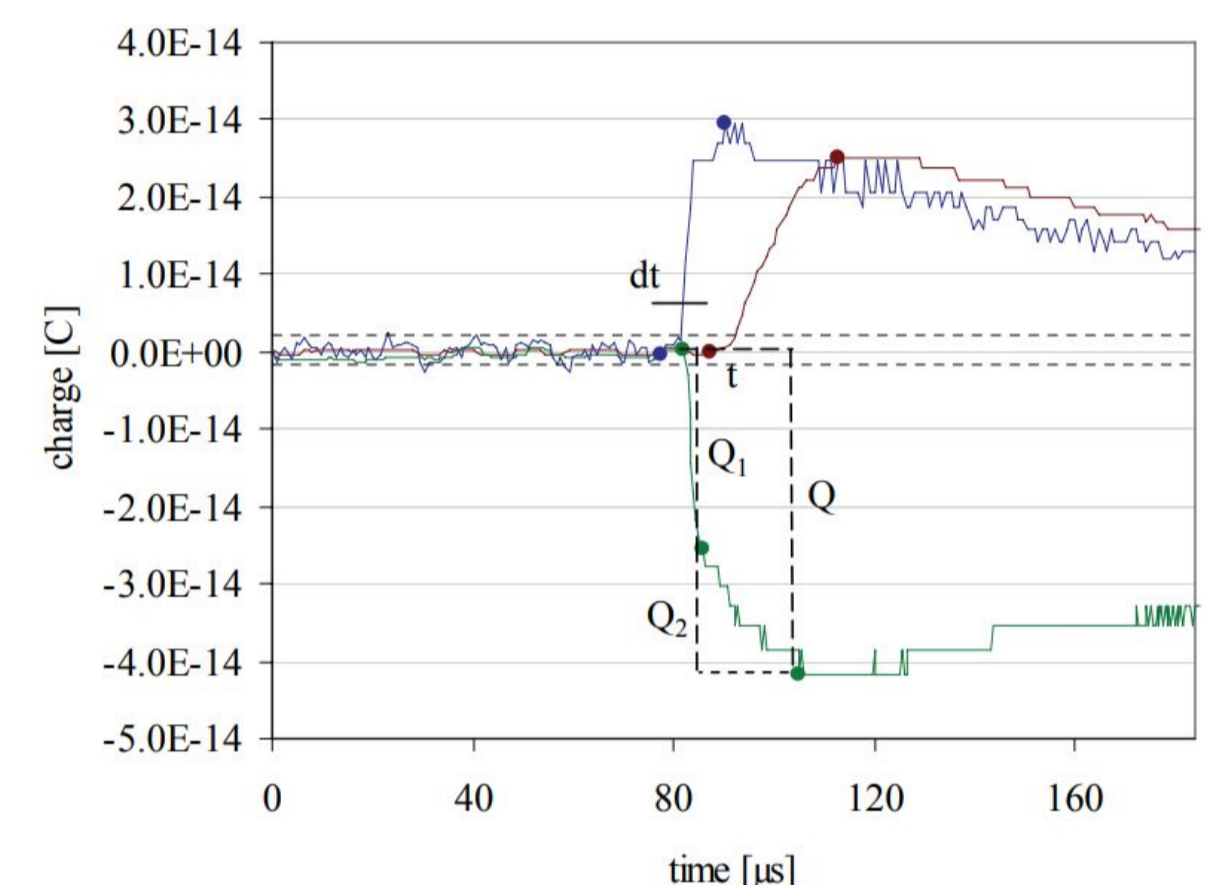


Figure 8: charge curve from exemplary impact during the Munich Dust Counter (MDC) calibration campaign [4]

Performance Simulations

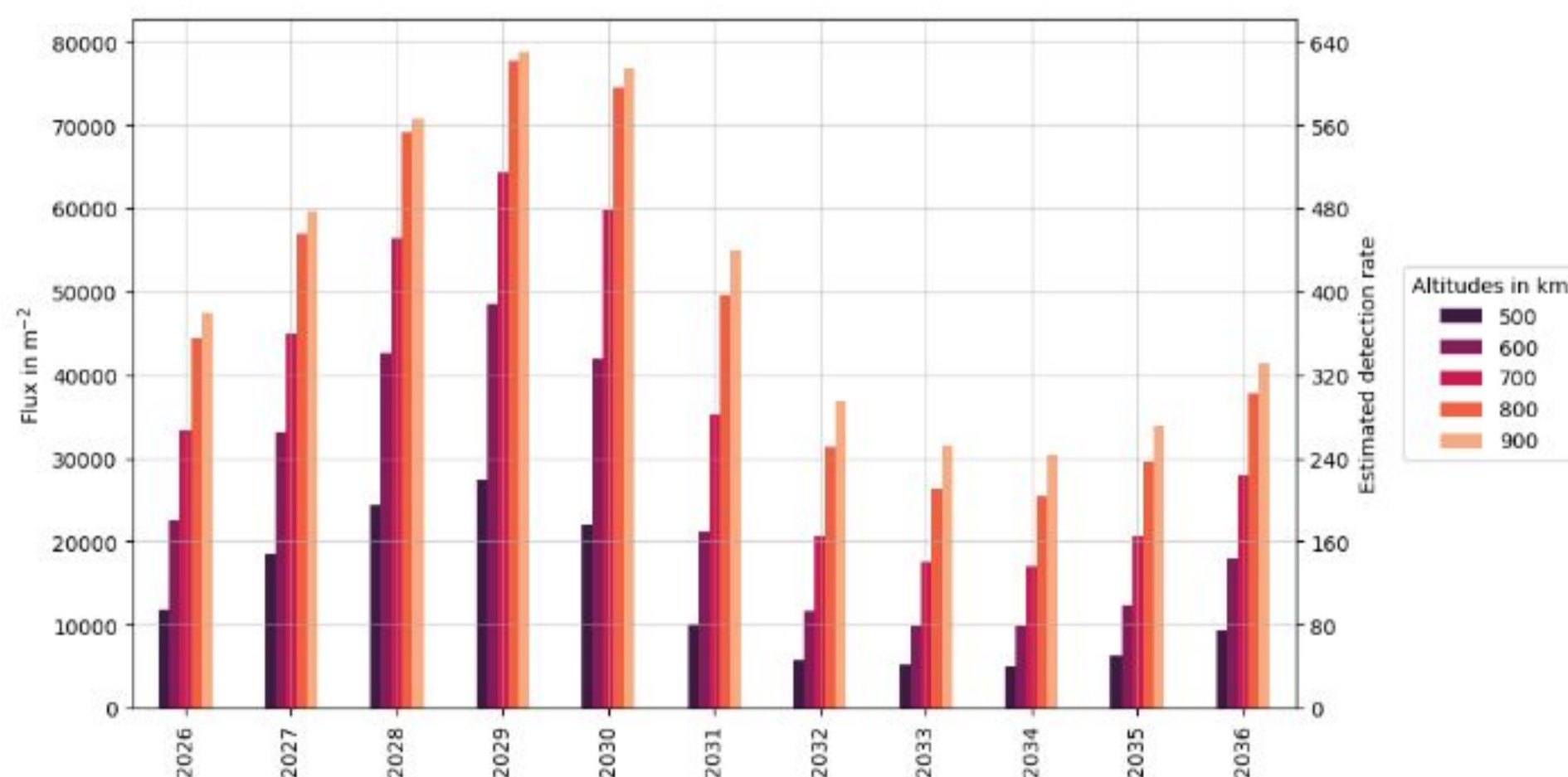


Figure 2: Variability of modelled sub-millimetric debris flux and estimated detection rates throughout a 11-year period for selected altitudes in SSO (ORDEM 3.2.0).

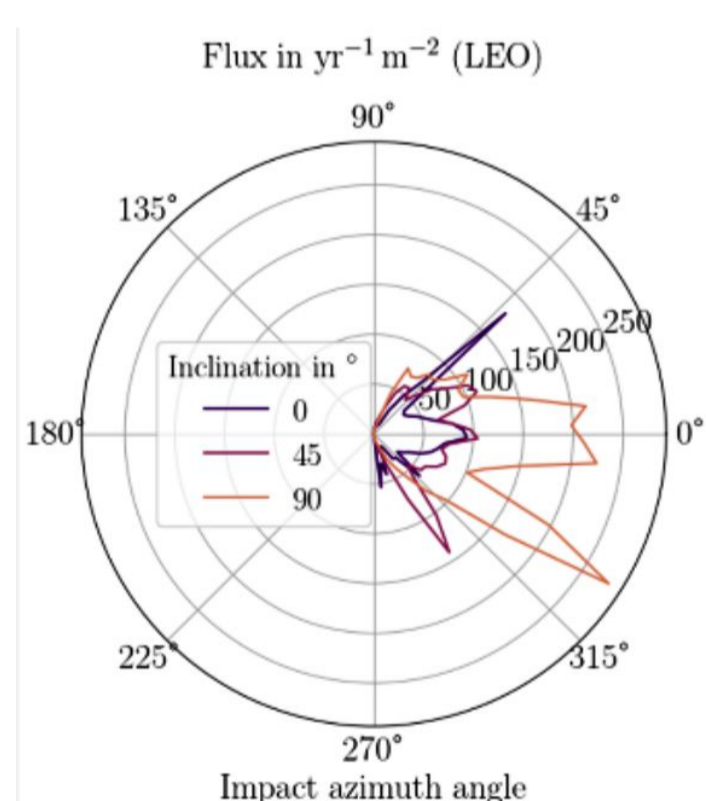


Figure 3: Variability of modelled sub-millimetric debris flux impact angle throughout a 11-year period for different orbital inclinations in LEO (MASTER 8.0.3)

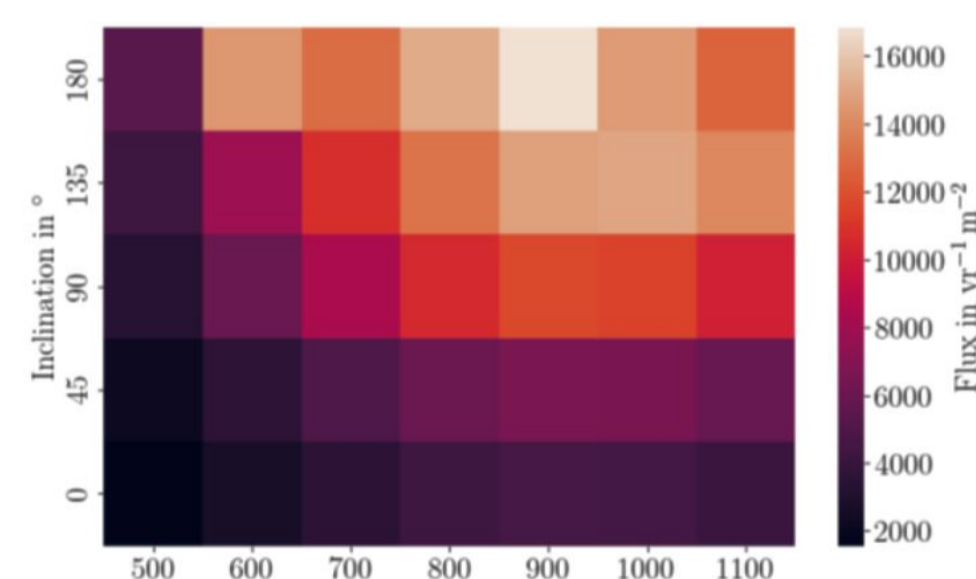


Figure 4: Heatmap of the total modelled sub-millimeter debris flux (11-year-average) in LEO for selected altitudes and inclinations (MASTER 8.0.3).

Sensor Characteristics:

- Particle Mass: 10^{-15} to 10^{-10} kg
- Mass Resolution: 2×10^{-15} kg
- Velocity: 7 to 30 km/s
- Velocity Resolution: 1 km/s
- Power Consumption: 3.7 W
- Size: 1U (Box only)

Challenges

- The upcoming IOD mission will show the effectiveness of the sensor's noise mitigation measures against both atmospheric interference, which largely affected DEDRA's predecessor MDC on BremSat, and periodic solar noise
- Although we expect best scientific results at 1000 km, the initial injection altitude is practically limited due to the 5 years LEO clearance requirement
- Calibration of this sensor remains a challenge as the impact ionization cannot be properly simulated. The use of machine learning algorithms helps us find correlations in the signals and particle characteristics. These are currently being trained on old MDC calibration datasets
- Once the concept of this sensor has been proven, many more must be flown to get optimal statistical significance in different orbits

Acknowledgements

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References

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- [3] Groemer, G. *ADLER-1 In-situ Space Debris Measurements*, Presentation at MASTER Modelling Workshop, ESA, 2021.
- [4] Senger, R. (2008) *Data handling and evaluation for autonomous experiments in interplanetary missions*. Planetary and space science Vol.56, p.1057-1059