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Impact of IRENE models on Total Ionizing Dose, Single Event Effects and solar cell degradation, using OMERE and FASTRAD softwares

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Overview

- OMERE software
- FASTRAD software
- Impact on Total Ionizing Dose (TID)
- Impact on Single Event Effect (SEE) rate calculation
- Impact on Solar cells degradation
- Conclusion





OMERE software



The OMERE software

• The project

- Since 1999
- TRAD development with CNES support
- Freeware for space radiation environment and effects on electronic components
- Stand alone software (no internet connection needed)
- Conceived to meet industrial requirements
- Integrates ONERA models
- Integrates outcomes of Research and Technology projects financed by CNES
- Coupling with FASTRAD.





The OMERE software







- Many other functionnalities
- User Manual





FASTRAD software



Created in 2001

3D interface

User Manual

Mappings,

The FASTRAD[®] software

Modeling step Materials • Input: Radiation model Components ٠ +150 users worldwide Space environment OMERE / Other files **TID/TNID** calculations Internal charging calculations Multithreading calculations, ...





Impact on Total Ionizing Dose (TID)



Missions and space environment models

- Four missions:
 - LEO: 800 km, 98°, 10 years
 - MEO GALILEO: 23222 km, 56°, 12 years
 - GEO: 15 years,
 - EOR (LEO 250 km to GEO): 130 days

| Mission | Trapped electrons | Trapped protons |
|------------------|-------------------|------------------|
| | AE8 max | AP8 min |
| LEO, MEO and EOR | AE9 mean | AP9 mean |
| | AE9 PM-mean | AP9 PM-mean |
| | AE9 PM-80/90/95% | AP9 PM-80/90/95% |
| GEO | All + IGE 2006 | All |





Methodology



Models: Platforms, units and components



Unit A

Tests & radiation

cnes

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Unit B



Unit C



ICARE







| Models | Equipments | Detectors |
|-------------|------------|-----------|
| G1_unitA | 10 | 530 |
| G1_unitB | 10 | 730 |
| G1_unitC | 10 | 530 |
| G2_unitA | 10 | 530 |
| G2_unitB | 10 | 730 |
| G2_unitC | 10 | 530 |
| JASON_ICARE | 6 | 2262 |
| Total | | 5842 |

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Dose ratio for trapped electrons



- Ratios between 0.3 and 6.
- EOR and MEO: AE9 underestimates the dose compared to AE8, low range between min and max ratio → Same behavior for all the detectors,
- **GEO**: AE9 overestimates the dose compared to IGE2006 (from 2 to 5),
- LEO: Huge impact of percentile parameter. High range between min and max ratio → Hard to say if using AE9 will
 under or overestimate the dose.





Dose ratio for trapped protons



- Ratios between 1.2 and 4.
- LEO and EOR: AP9 overestimates the dose compared to AP8,
- The dose slightly increases with the percentile parameter,
- LEO: Low range between min and max ratios \rightarrow Could be used for predictive behavior,
- EOR: Max ratios near the average value, meaning that only few detectors experiment a low dose level.







Impact on SEE rates



Missions and components

- Two missions:
 - LEO: 800 km 98°
 - EOR (LEO 250 km to GEO)
 - Neither GEO nor MEO (no sufficient energetic trapped protons to cross the shielding)

| Mission | Trapped protons | |
|-------------|------------------|--|
| | AP8 min | |
| LEO and EOR | AP9 MC-mean | |
| | AP9 MC-80/90/95% | |



Components

- Realistic component database

- Realistic component database Different cross section sensitivity profiles 9 components: 4 with test data 5 using Heavy lons test data and PROFIT 90° model





SEE rate ratios VS Missions and models



- Ratios between 1.8 and 5.5,
- AP9 overestimates the SEE rate compared to AP8,
- High impact of the percentile parameter on the SEE rate,
- This overestimation has to be analyzed considering the margin applied by industry companies (ex: margin of 10 for ECSS standard).





Impact on solar cells degradation



SADC tool

• SADC (**Solar Array Degradation Calculator**) developed in the frame of ESA ARTES AT-4F.126 project (TRAD prime with ONERA, OHB, ADS & TAS)

- Integration of OMEP model developed by ONERA (p+ 30keV 20 MeV for EOR orbit).
- Global Remaining Factors of different electrical parameters in output,
- Based on NRL method,
- Detailed model of the solar cell:
 - o No limit on materials or shielding thicknesses
 - $\circ~$ User can import NIEL data
 - $\circ~$ No limitation on materials of the active layer
 - ightarrow Study of newer technologies
- Darkening of coverglass is now considered,
- Propagation of measurements uncertainties.

| SADC | ~ | | | |
|-------------|------------------------------|----------------|--------------|-------------------------------|
| SADC S | pecific | | | |
| ✓Layer | list | | | \odot |
| YOpt | ical Layer - coverglass | | | Θ |
| … Ту | pe | 0 | ptical Layer | |
| ~ 0p | otical Layer Parameters | | | |
| ··· I | layer name | coverglass | | (|
| F | Family | | Glass | |
| | Thickness | 100 | | |
| | Unit | um 🔻 | Density: | 2.61 g/cm ^³ |
| | Formula | SiO2 | | |
| | Transmission path | oss_table - CV | G from ON | ERA.dat 📑 |
| | Transmission error | 0 % | | |
| ⇒Opt | ical Layer - cg_adhesive | | | Θ |
| -> Opt | ical Layer - InP | | | Θ |
| ⇒ Acti | ve Layer - GaAs | | | Θ |
| ⇒ Shie | lding Layer - Ge | | | Θ |
| ⇒ Shie | lding Layer - BackContact | | | Θ |
| Shie | lding Layer - back_adhesive | | | Θ |
| ·-> Shie | lding Layer - Aluminium back | plate | | Θ |
| | | | | |
| | | | | |
| | | | | |





Solar cell model

- **3G30** cell, widely used in space industry
 - Active part: GaAs

| Layer | Type/Function | Material (Density in g.cm ⁻³) | Thickness (μm) |
|-------|----------------|--|----------------|
| 1 | Coverglass | SiO ₂ (2.61) | 100 |
| 2 | Adhesive | SiOC ₂ H ₆ (1.03) | 20 |
| 3 | Optical | InP (4.81) | 1 |
| 4 | Active | GaAs (5.12) | 2 |
| 5 | Semi-conductor | Ge (5.35) | 140 |
| 6 | Back contact | Ag (10.5) | 5 |
| 7 | Shielding | Al (2.7) | 1200 |

| Solar Cell Degradation | ? | × |
|---|------|-----------|
| Flux Type: From mission data | v 📦 | \$ |
| Degradation Model SADC SADC Specific SADC Coverglass Optical Layer - coverglass Optical Layer - cg_adhesive Optical Layer - lnP Active Layer - GaAs Shielding Layer - Ge Shielding Layer - BackContact Shielding Layer - back_adhesive Shielding Layer - Aluminium back plate | | |
| C:\Users\damien.herrera\Documents\OMERE 5.9\solarcells.dat | | ۲ |
| Calculations | (⊗ca | ancel |

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Missions and models

| Mission | Trapped protons | Trapped electrons | Solar average protons | |
|---------------------------|-----------------|-------------------|-----------------------|--|
| LEO (600km SSO) | AP8-min | AE8-max | ESP 90% | |
| 10 years | AP9-PM-mean | AE9-PM-mean | | |
| | AP8-min | MEOv2-average | ESP 85% | |
| MEO (GALILEO) 12 years | AP8-min | AE8-max | | |
| | AP9-PM-mean | AE9-PM-mean | | |
| EOR (129 days) | OMEP | AE8-max | ESP 95% | |
| | AP8-min | AE8-max | | |
| | AP9-PM-mean | AE9-PM-mean | | |
| GEO 15 years | AP8-min | IGE2006-average | ESP 80% | |
| | AP8-min | AE8-max | | |
| | AP9-PM-mean | AE9-PM-mean | | |

diations HECO CONVERSION AND COLOR SPATIALES Solar cell degradation VS Mission and models



- LEO : Ax9 PM \approx Ax8 \rightarrow Low DDD so RF still in the first part of the curve
- MEO : MEOv2 worst-case compared to Ax9 (DDD mainly due to e-)
- EOR : Ax8 worst-case compared to Ax9 (both p+ and e-)
- GEO : Ax9 worst-case compared to IGE2006 (due to e-)







- Using IRENE models instead of historical ones has a noticeable impact,
- It strongly depends of the effect and the mission considered,
- Lower solar cell degradation compared to Ax8,
- Higher SEE rate compared to AP8 min,
- Dose level, compared to Ax8, depends of the mission.





Thank you for your attention

For further information on:

<u>www.trad.fr</u> – <u>www.fastrad.net</u> <u>www.rayxpert.com</u> – <u>www.r2cots.com</u>



TRAD Tests & Radiations @TRAD_Officiel







Back-up





Ray-Tracing calculation

- Ray-Tracing
 - Division of the sphere surrounding the sensitive area in identical solid angles,
 - Calculation of the dose in each solid angle based on the dose-depth curve,
 - Average over all the solid angles.







Dose contribution



TRAD Tests & Radiations



AE9 and missions



• LEO:

- Ratio strongly dependent of the received dose level,
- For dose level > 1krad, ratios between 1.5 and 3.

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Dose ratio in Reverse Monte-Carlo Trapped electrons



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Representativeness of 3D models

• Shielding profile of detectors for G1_unitA and JASON_ICARE models





TID: LEO (800km SSO)





SEE: LEO (800km SSO)



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TID: MEO





























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Solar cells degradation: MEO





Solar cells degradation: EOR



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Solar cells degradation: GEO



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