



European Space Thermal Engineering Workshop
ESA/ESTEC, Noordwijk, Netherlands, 9 – 11 September 2025



Candidacy of Cryogenic Pulsating Heat Pipes for Space Applications



Tisha DIXIT¹

Lorenzo MARELLI², Gilles AUTHELET¹, Bertrand BAUDOUY¹

¹Commissariat à l'énergie atomique et aux énergies alternatives (CEA),
Paris-Saclay Centre, France

²Centre national d'études spatiales (CNES), Toulouse, France

Outline

- PHPs
 - Working mechanism
 - Non-cryogenic PHPs already in space
- Cryogenic PHPs
 - Motivations in the superconductivity world
 - Specific characteristics
 - State-of-the-art at CEA Paris-Saclay
 - Mono-sized 0.4 m long neon ($C_{th} \sim 6 \text{ W/K}$) and helium PHP ($C_{th} \sim 1 \text{ W/K}$)
 - First demonstrator of HTS-MI magnet (10 T, 30 K) cooled by cryogenic PHP
- Collaborative project with CNES
 - Objectives; first such project aimed for miniature cryogenic PHPs in space
 - Commissioning of on-ground test facility
- Take-away ideas



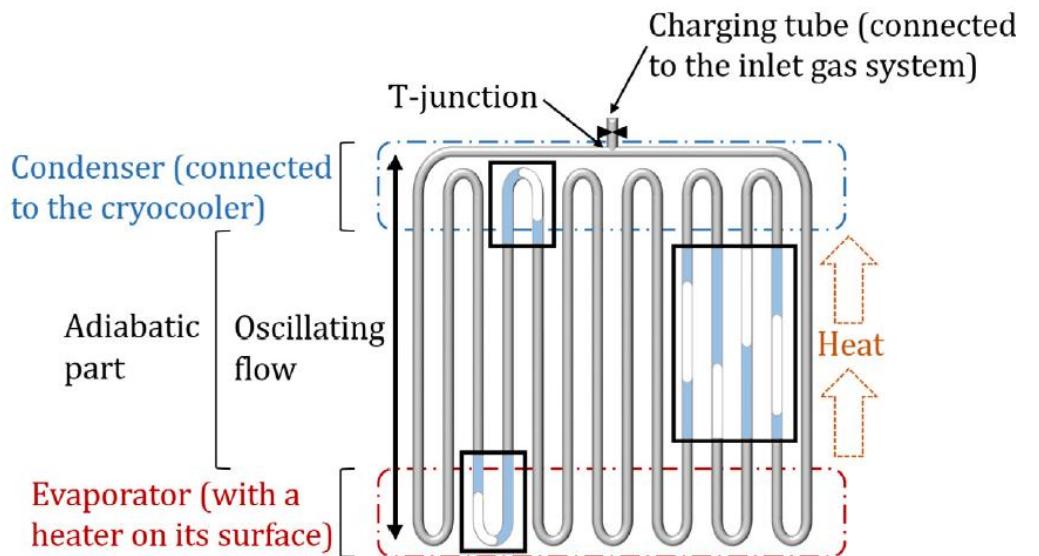
PHPs: Working Mechanism

What is a pulsating heat pipe (PHP) / oscillating heat pipe (OHP)?

- Two-phase passive heat transport device patented by Akachi in 1990
- Filled partially with working fluid operating around its saturation conditions
- Thermal transport carriers: oscillating train of vapor bubbles and liquid slugs



Visualization of water PHP in working
Credits: Prof. S Khandekar, IIT Kanpur, India



Attractive traits for space industry:

- ✓ High and variable heat transport capability: HT mechanism entails both in sensible and latent heat
- ✓ Ease of construction, lightweight, no moving parts
- ✓ High effective thermal conductivity compared to metallic counterparts
- ✓ « Independent » of gravity
- ✓ Possible heat switch



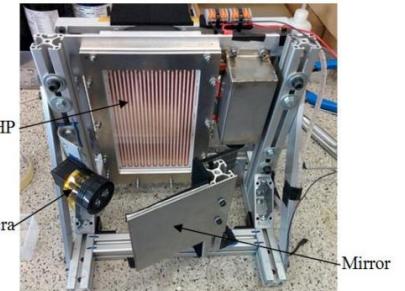
PHPs: Non-cryogenic PHPs already in Space

Two long-term reliability and operation on-orbit demonstrators of non-cryogenic PHPs by JAXA and U.S. AFRL

Several micro-/hyper gravity experiments aboard parabolic flights and sounding rockets by collaboration b/n institutes in Italy (*Università di Bergamo, Università di Parma, Università di Pisa, Politecnico di Milano*), UK (University of Brighton), France (*Institut Pprime CNRS – ENSMA – Université de Poitiers*), ESA Academy

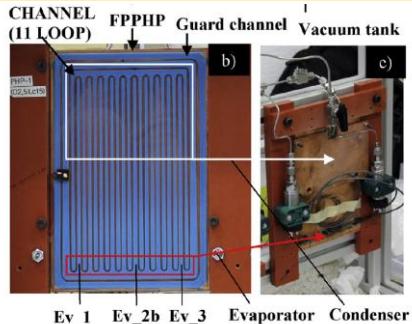
JAXA FOX on SDS-4	U.S. AFRL + ThermAvant Technologies ASETS-II
Type pf OHP	Tubular with check valves
Material	Al alloy
Working fluid	HFC-134a
Test duration	2012 – 2016
<div style="display: flex; justify-content: space-around;"> <div style="text-align: center;"> <p>FOX 140 mm</p> </div> <div style="text-align: center;"> <p>152.4 mm 46.25 mm</p> </div> <div style="text-align: center;"> </div> </div>	
<p><i>Ando et al. Applied Therm. Eng. 130 (2018) 552-560</i></p> <p><i>Drolen et al. J. Thermophys Heat Trans. 36 (2) (2022) 314-327</i></p>	

62nd, 64th ESA Parabolic Flight Campaign



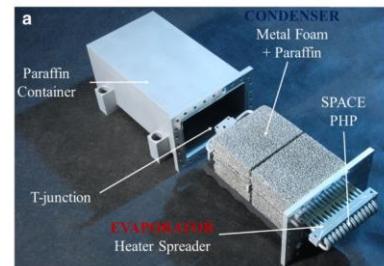
Tubular Cu PHP, FC-72
Ayel et al. Heat Tranf. Eng. 40 (2018) 227-237

65th ESA Parabolic Flight Campaign



Flat Cu PHP, water+SRWF
Cecere et al. Acta Astronautica 147 (2018) 454–461

ESA REXUS 22 Sounding Rocket



Tubular Al 6060 hybrid
LT /PHP, perfluorohexane

Mameli et al. Microgravity Sci. Tech. 31 (2019) 249-259

79th ESA Parabolic Flight Campaign



Deployable tubular Al 6063 PHP, HFE-7000

Perna et al. Therm. Sci. Engg. Prog. 61 (2025) 103479

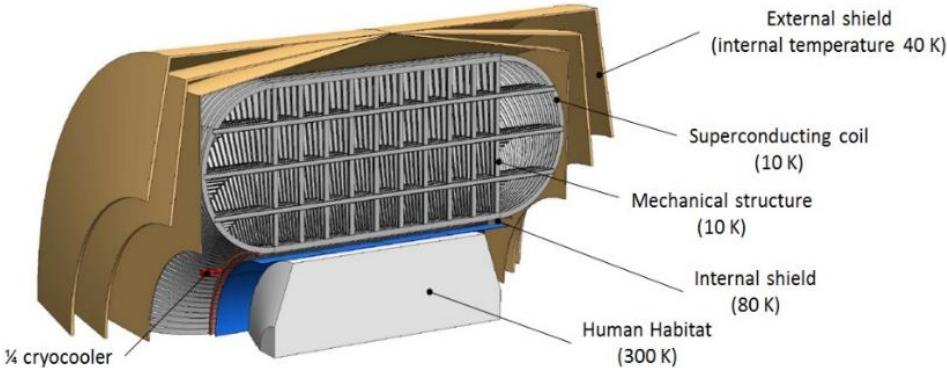


Cryogenic PHPs: Motivations in Superconductivity World

Migration of PHPs in cryogenics in 1998: investigations began at CEA in 2014

Efficient heat transfer device in “zero gravity” environment

- Cooling superconducting system for space application – **CEA conceptual idea**
- Cooling very high magnetic field superconducting magnet ($B > 20$ T) where magnetic force can oppose gravity – **CEA R&D**



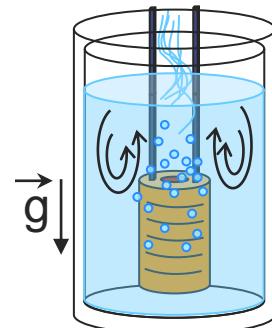
European Commission Grant ID:313224,
Project duration 2013 – 2015
Proposal to cool thermal shield b/n human habitat and
superconducting magnet with nitrogen PHP in space

Liquid helium bath
cooled 32.5 T
solenoid HTS magnet
NOUGAT

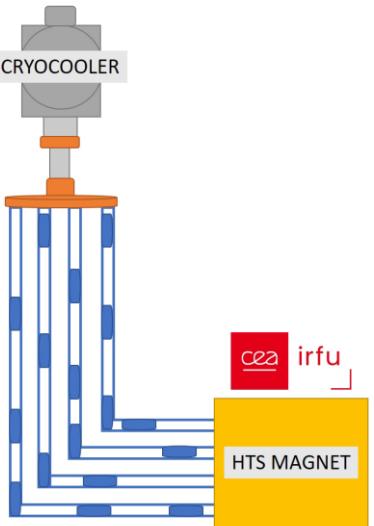


Fazilleau et al., Cryogenics
106 (2020) 103053

$$\vec{f}_{mag} \propto \frac{\chi}{2\mu_0} \vec{grad}(B^2)$$

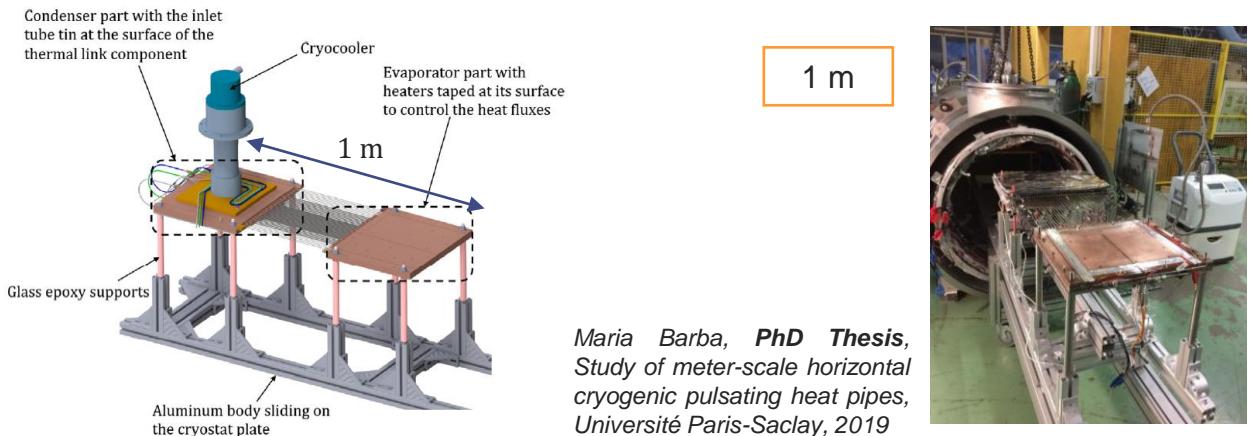
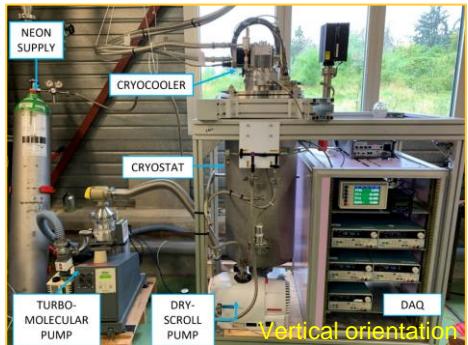
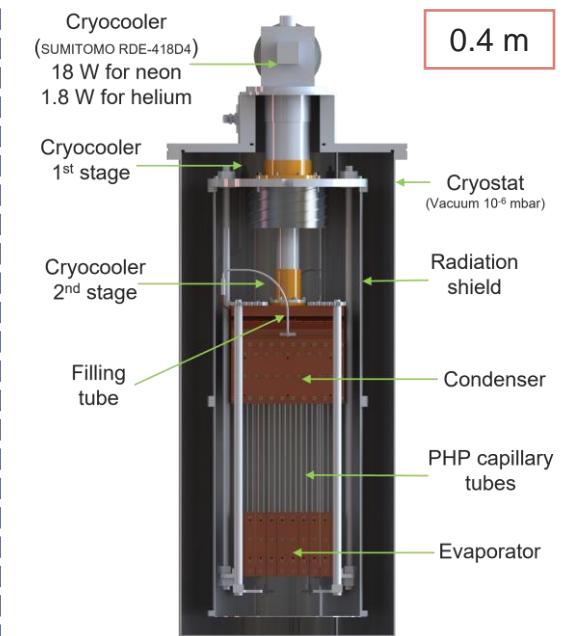
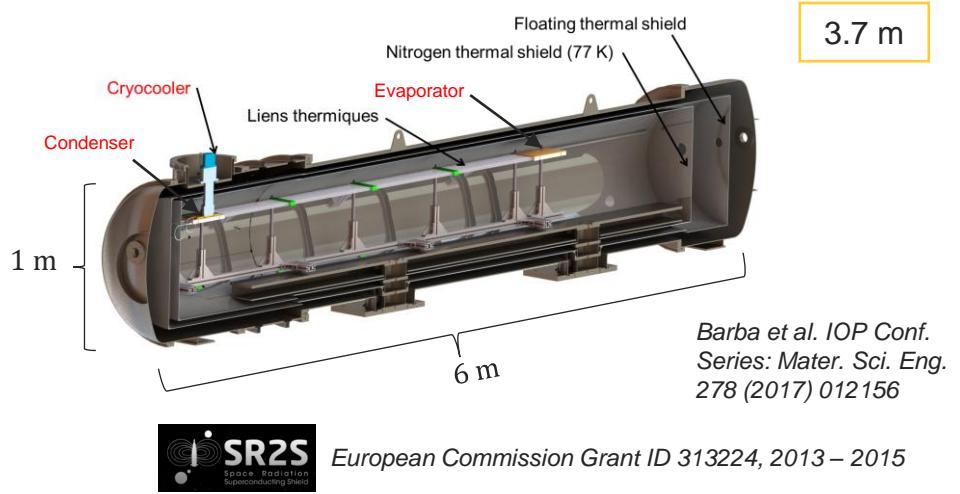


Simon Bagnis, **PhD Thesis**, Analysis of the effect of
magneto-gravitational forces on liquid helium cooling
of high magnetic field superconducting magnets,
CEA/IRFU/DACM Université Paris-Saclay 2023





Cryogenic PHPs: CEA State-of-the-Art 2014-till date



*Dixit et al. Cryogenics 132 (2023) 103670
Dixit et al. App. Therm. Eng. 251 (2024) 123613*

- Cryogenic fluids tested
- Argon (87.3 K or -185.9°C)
 - Nitrogen (77.3 K or -195.8°C)
 - Neon (27.1 K or -246.1°C)
 - Hydrogen (20.3 K or -252.9°C)
 - Helium⁴ (4.2 K or -268.9°C)

Cryogenic PHPs: CEA State-of-the-Art Mono-sized neon & helium PHP

Design parameters

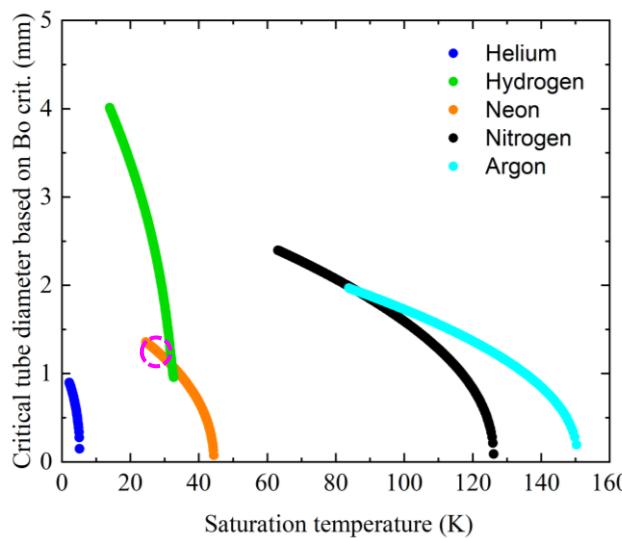
- Not straightforward – juggling of several parameters
- Guidelines not streamlined – still primarily dependent on experimental investigations

Physical characteristics

- ✓ Capillary tube inner diameter
- ✓ Number of tubes
- ✓ Length of evaporator/condenser/adiabatic part (1:1:1 or 1:2:1 or ...)

Operational characteristics

- ✓ Working fluid
- ✓ Condenser temperature
- ✓ Filling ratio
- ✓ Evaporator heat load



Choice of PHP capillary tube diameter based on widely accepted, fluid property dependent Bond number criterion

$$Bo = \frac{g(\rho_l - \rho_v)D^2}{\sigma} < 4$$

ρ_l = fluid saturation liquid density (kg/m^3)
 ρ_v = fluid saturation vapour density (kg/m^3)

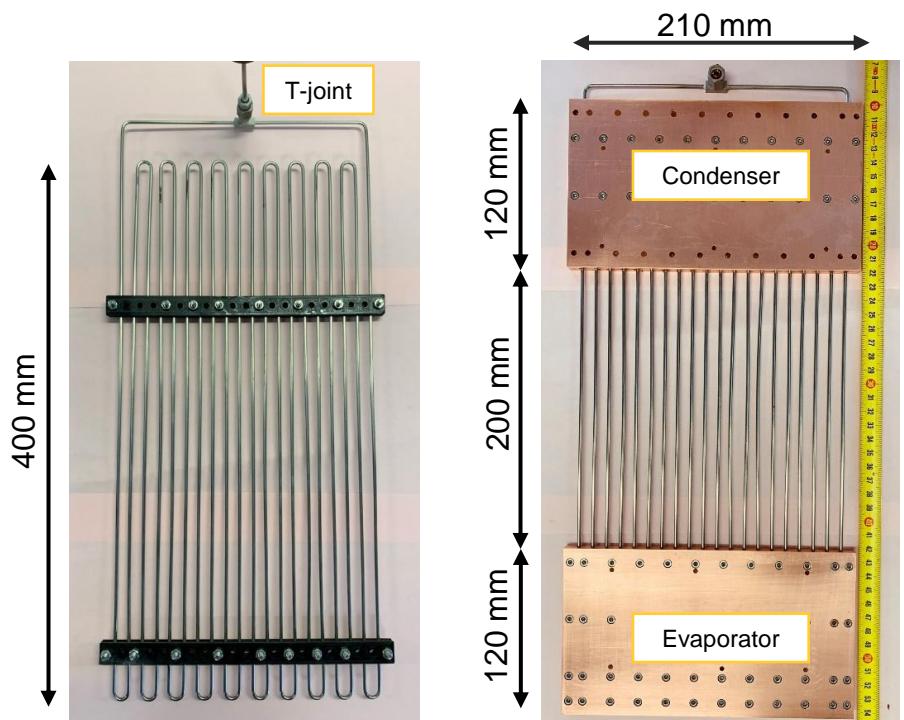
σ = fluid surface tension
 D = tube inner diameter (m)

Good starting point but validity of theory deliberated!

Dixit et al. "Oversized diameter helium pulsating heat pipe" Applied Thermal Engineering 251 (2024) 123613

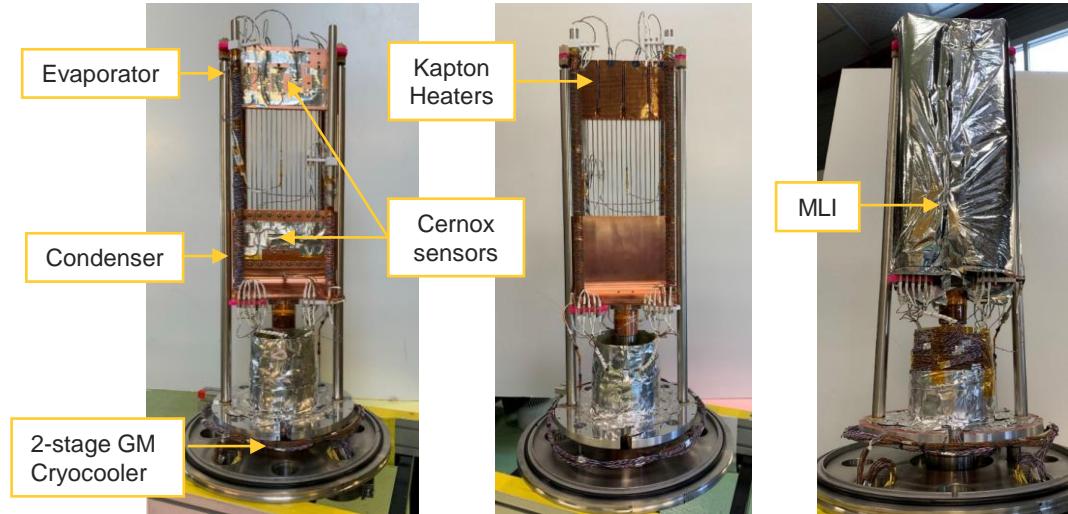
Chosen neon PHP parameters:

- Inner diameter = 1.0 mm (Bo ~ 2.5 at 27.1 K)
- Number of tubes = 20
- Projected length = 0.4 m
- Condenser/evaporator material = CuC2
- Tube material = SS316 L annealed
- Turns enclosed within the condenser/evaporator

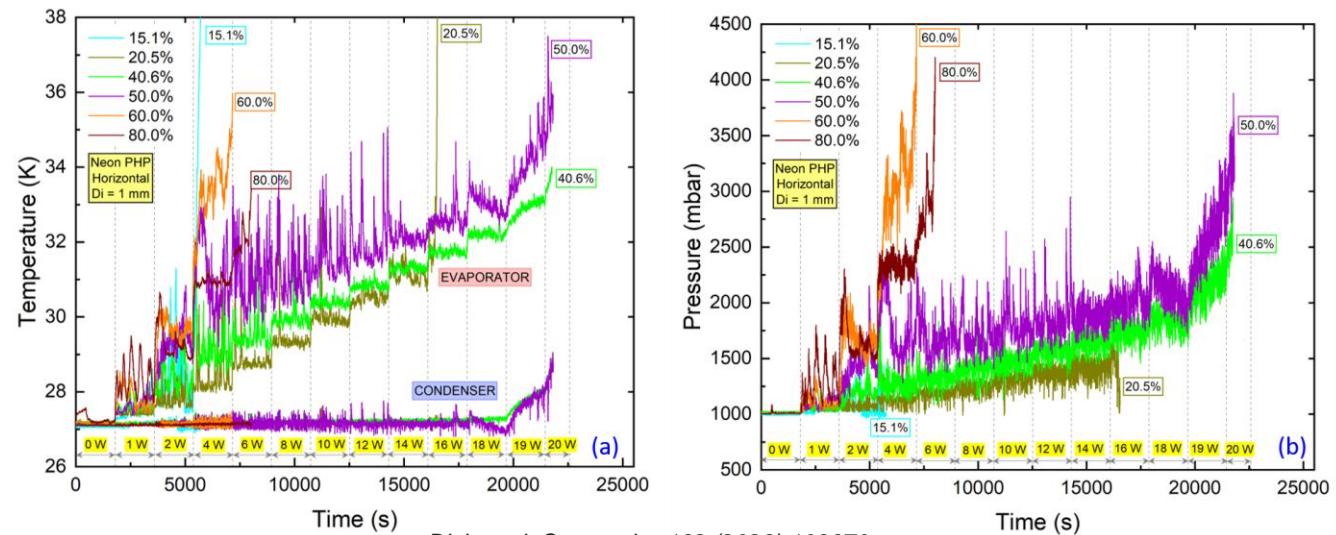
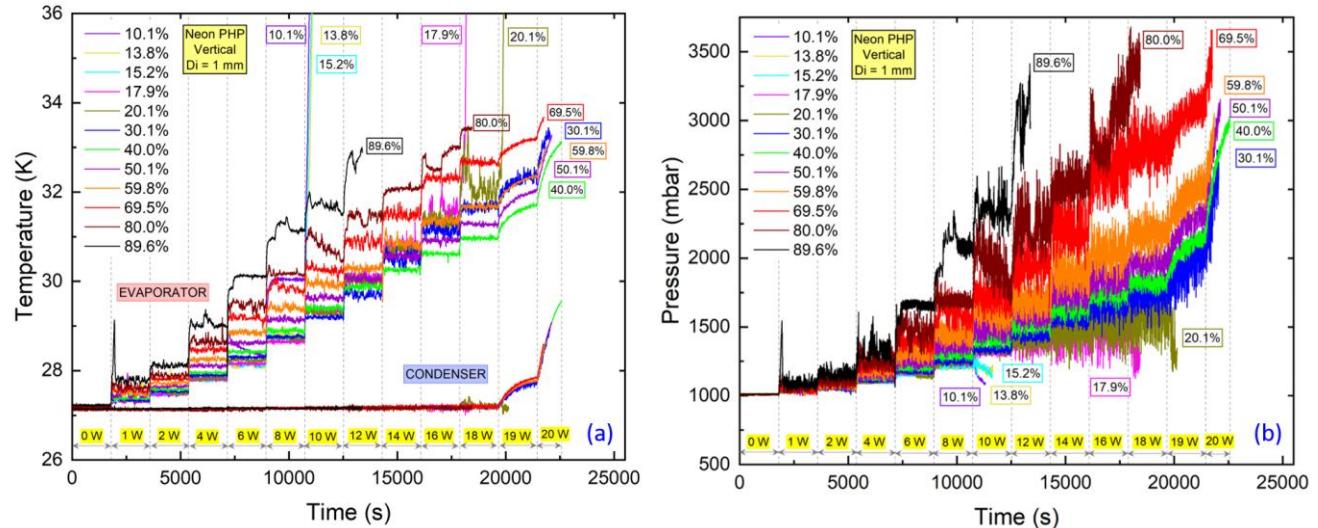


Cryogenic PHPs: CEA State-of-the-Art Mono-sized neon & helium PHP

Neon PHP



- Evaporator temperature, (regulated) condenser temperature, global pressure measurement
- Vertical and horizontal orientation
- Filling ratio (FR) of 10% - 90%
- Temperature range 27.1 K to 37 K
- Q_{evap} 0 - 18 W (cryocooler limit reached)
- Optimum FR of 40% for 18 W capacity

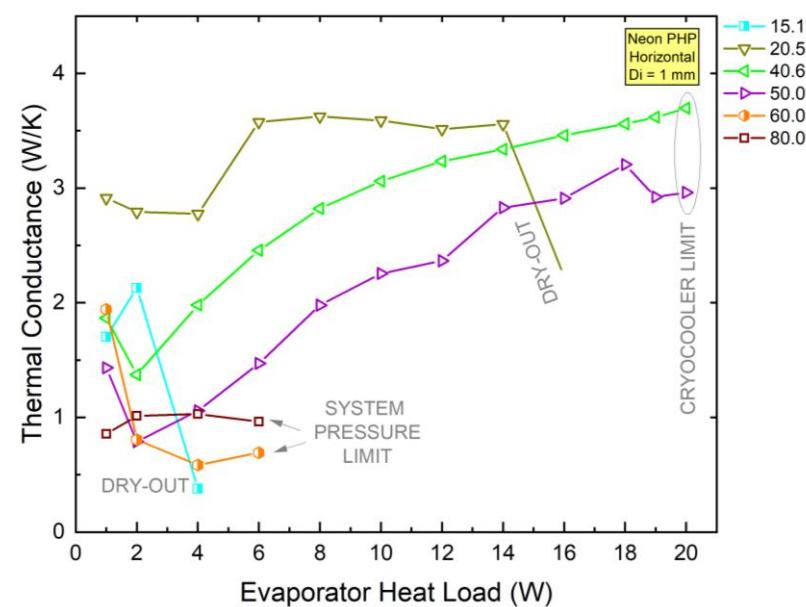
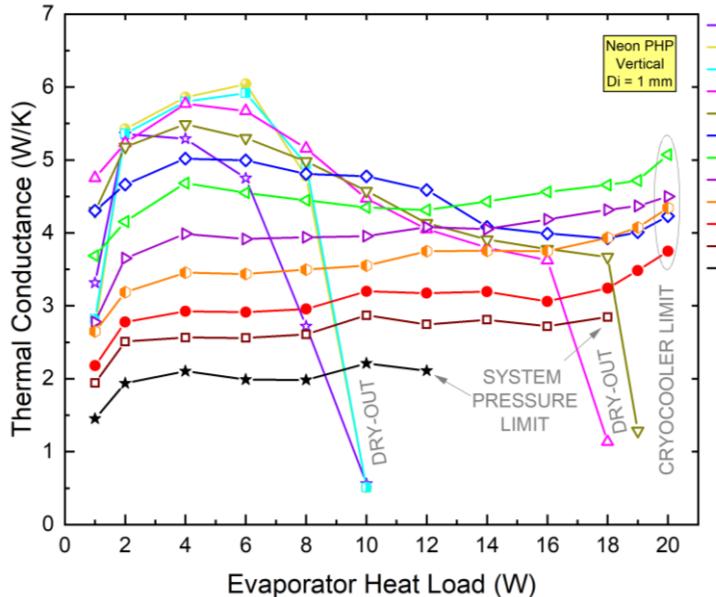


Dixit et al. Cryogenics 132 (2023) 103670

Cryogenic PHPs: CEA State-of-the-Art Mono-sized neon & helium PHP

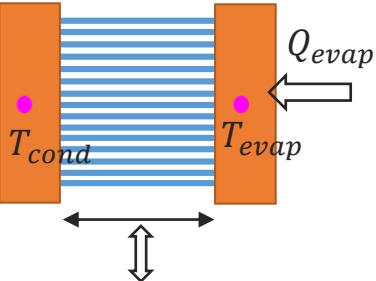
Neon PHP

- Thermal conductance in temperature range 27.1 K to 37 K
 - Highest C_{th} ~6 W/K in vertical and ~3.7 W/K in horizontal orientation
 - Best thermal performance ever measured for neon PHPs



- Function steadily under non-homogeneous evaporator heat load
- Long run stable functioning ~100 hours at constant FR and Q_{evap}

$$C_{th} = \frac{Q_{evap}}{T_{evap} - T_{cond}}$$



Comparison with classic metallic strap

Mass of PHP adiabatic part
 • $n = 20$, $L = 200$ mm b/n condenser & evaporator
 • SS316L tube

C_{th} at FR 13.8 % & 6 W

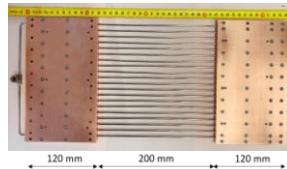
132 g

6 W/K

In 30 K temp range	Cu strap (k_{th} 2000 W/mK)	Al Strap (k_{th} 3000 W/mK)
Iso-conductance	1060 g	215 g
Iso-mass	0.75 W/K	3.8 W/K

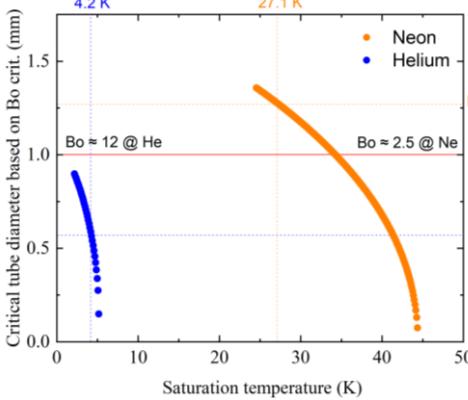
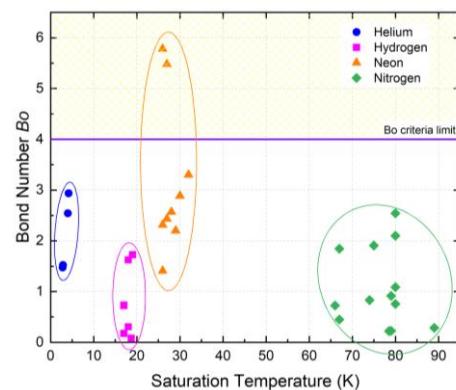
Cryogenic PHPs: CEA State-of-the-Art Mono-sized neon & helium PHP

Helium PHP



Breaching Bo criteria with helium

$$D_i = 1 \text{ mm} \rightarrow Bo \approx 12 @ 4.2 \text{ K}$$

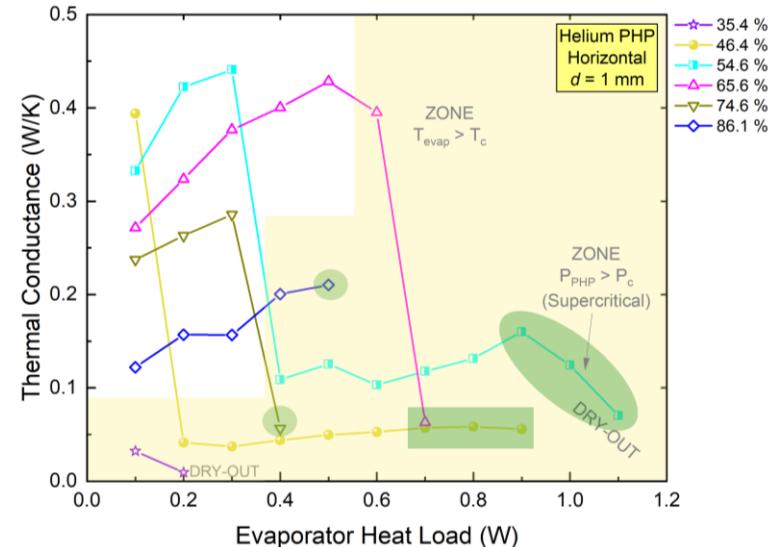
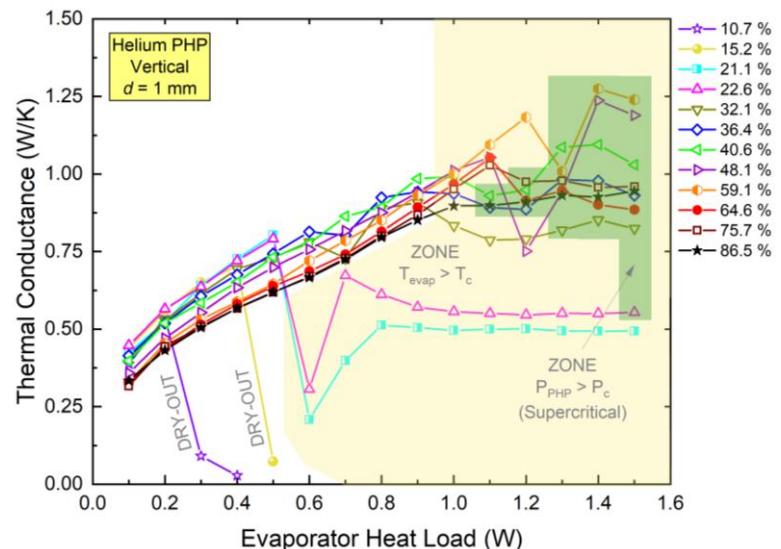


Dixit et al. App. Therm. Eng. 251 (2024) 123613

- Evaporator temperature, (unregulated) condenser temperature, global pressure measurement
- Vertical and horizontal orientation
- Temperature range 2.5 K to 10 K
- Q_{evap} 0 - 1.5 W (cryocooler limit reached)
- Successfully transport heat – stable working condition in **supercritical state** in evaporator ($T > T_c$ and $P > P_c$)
- ~65 hours stability test at constant FR and Q_{evap} in vertical/horizontal orientation

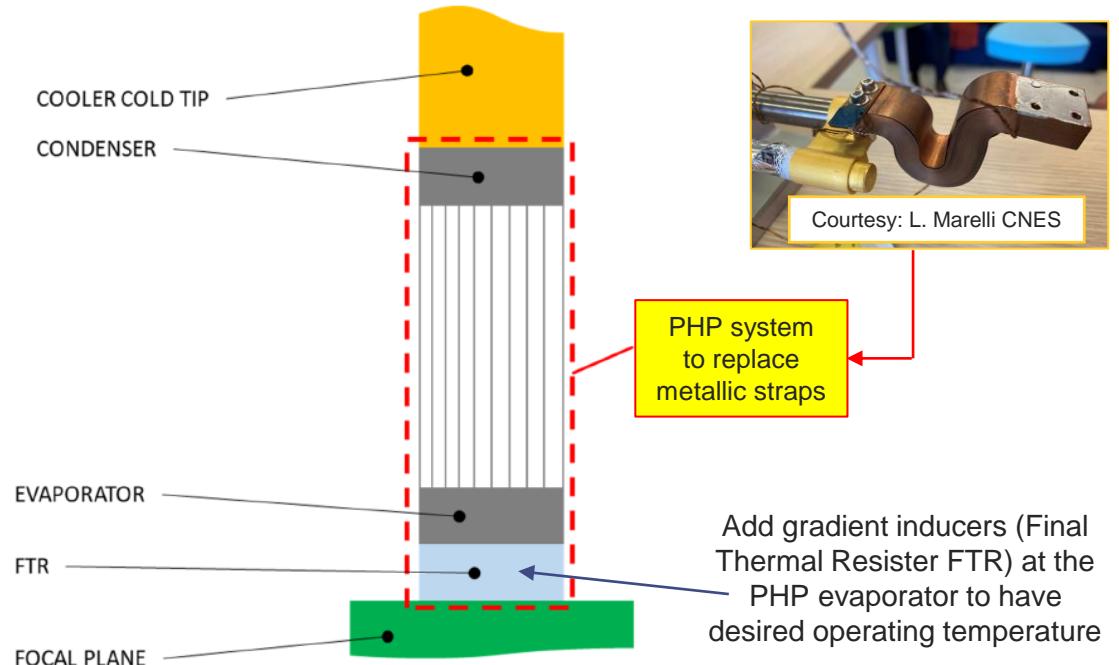
Comparison with classic metallic strap

In 4.2 K temp range FR 48% & 1 W	Present 0.4 m PHP	Cu strap (k_{th} 520 W/mK)	Al Strap (k_{th} 2500 W/mK)
	132 g	700 g	39 g
	1.0 W/K	0.19 W/K	3.15 W/K
Heat switch function	Yes	No	No



CNES-CEA Collaboration Project: Objectives

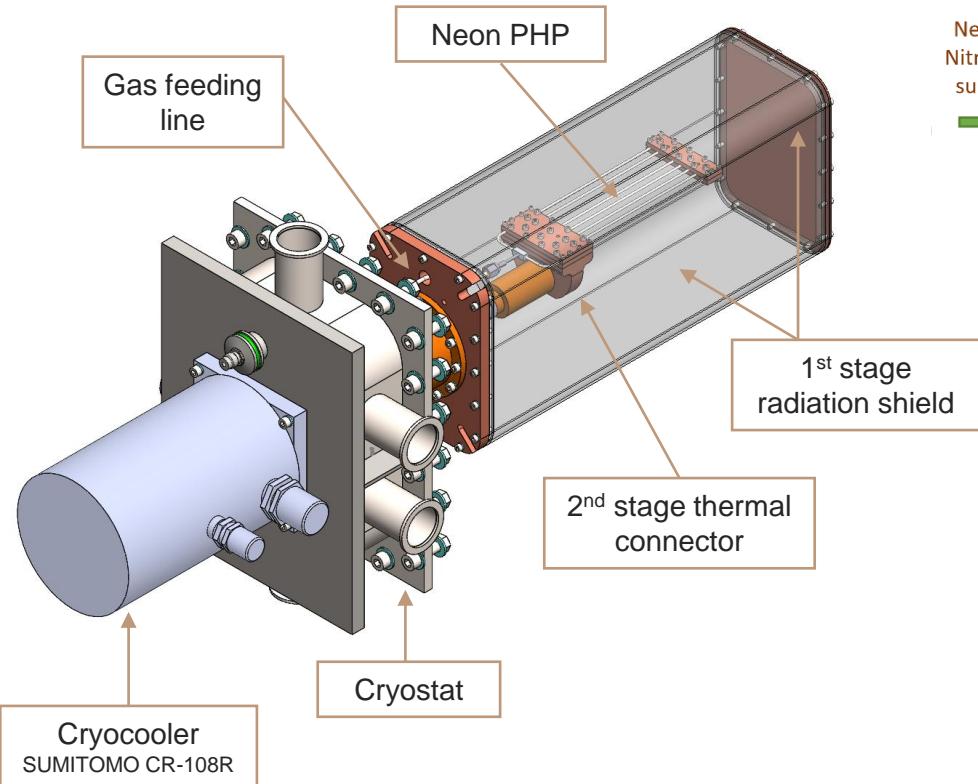
- Two-fold objectives:
 - ✓ Feasibility study of replacing metallic straps that thermally couple focal plane of an IR instrument to a cryocooler with miniature cryogenic PHPs
 - ✓ Viability check to see if heat-in-leak arriving from redundant cryocooler can be inhibited by having PHP in “OFF” mode
- Testing at two temperature range of focal planes: 50 K – 65 K and 77 K – 80 K with neon and nitrogen as working fluid



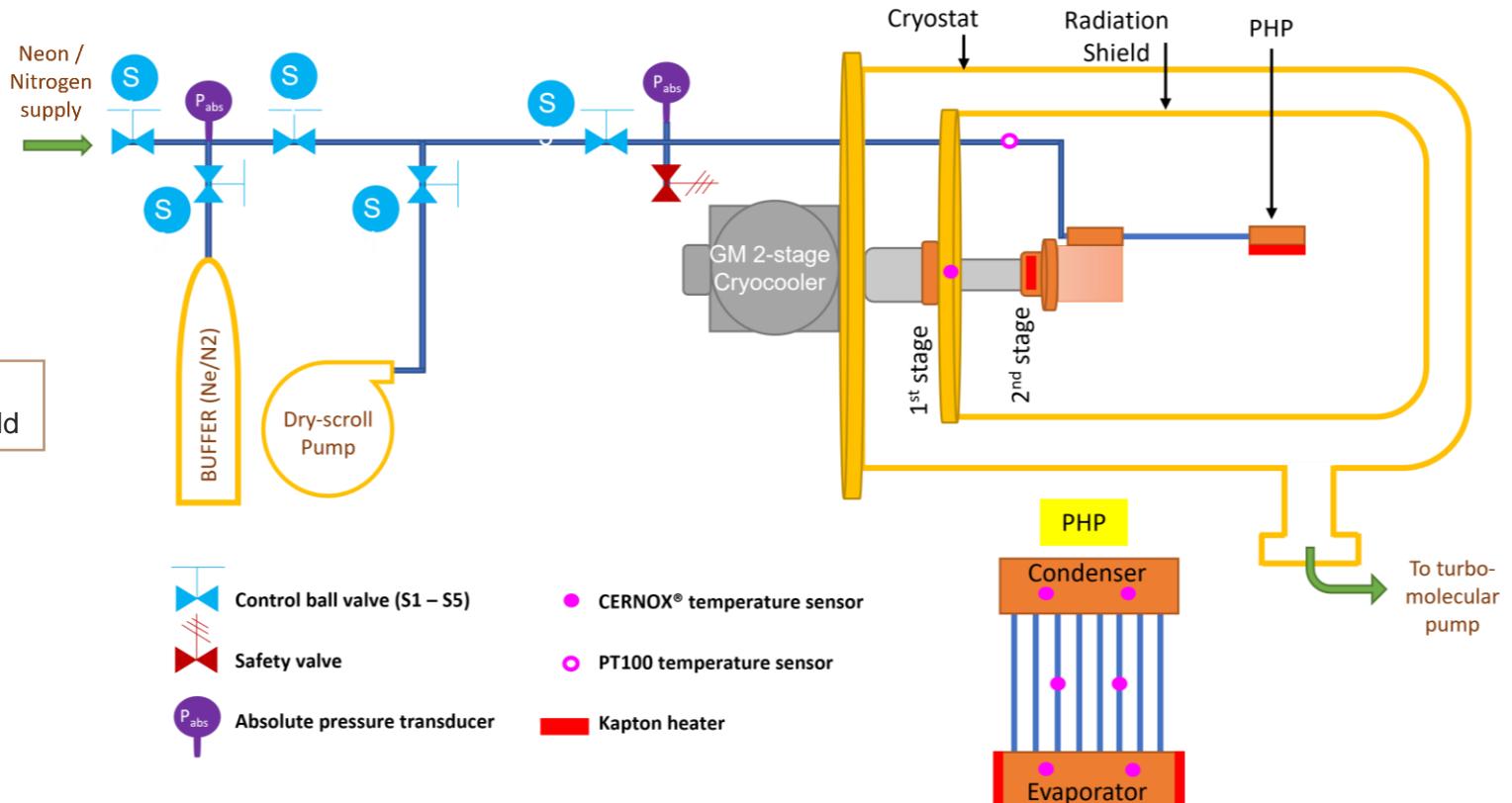


CNES-CEA Collaboration Project: On-ground Test Facility

Cryostat insert



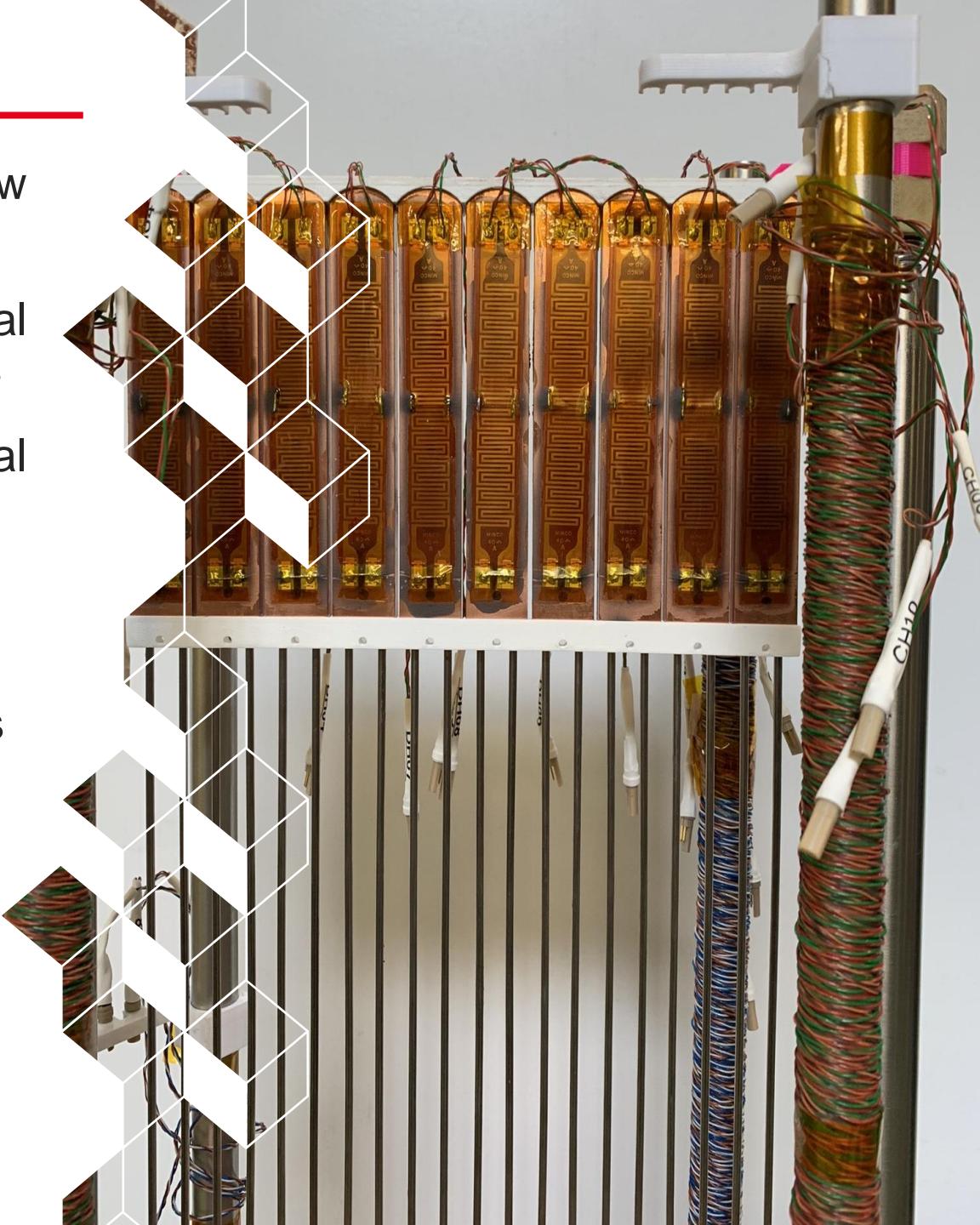
Set-up schematic

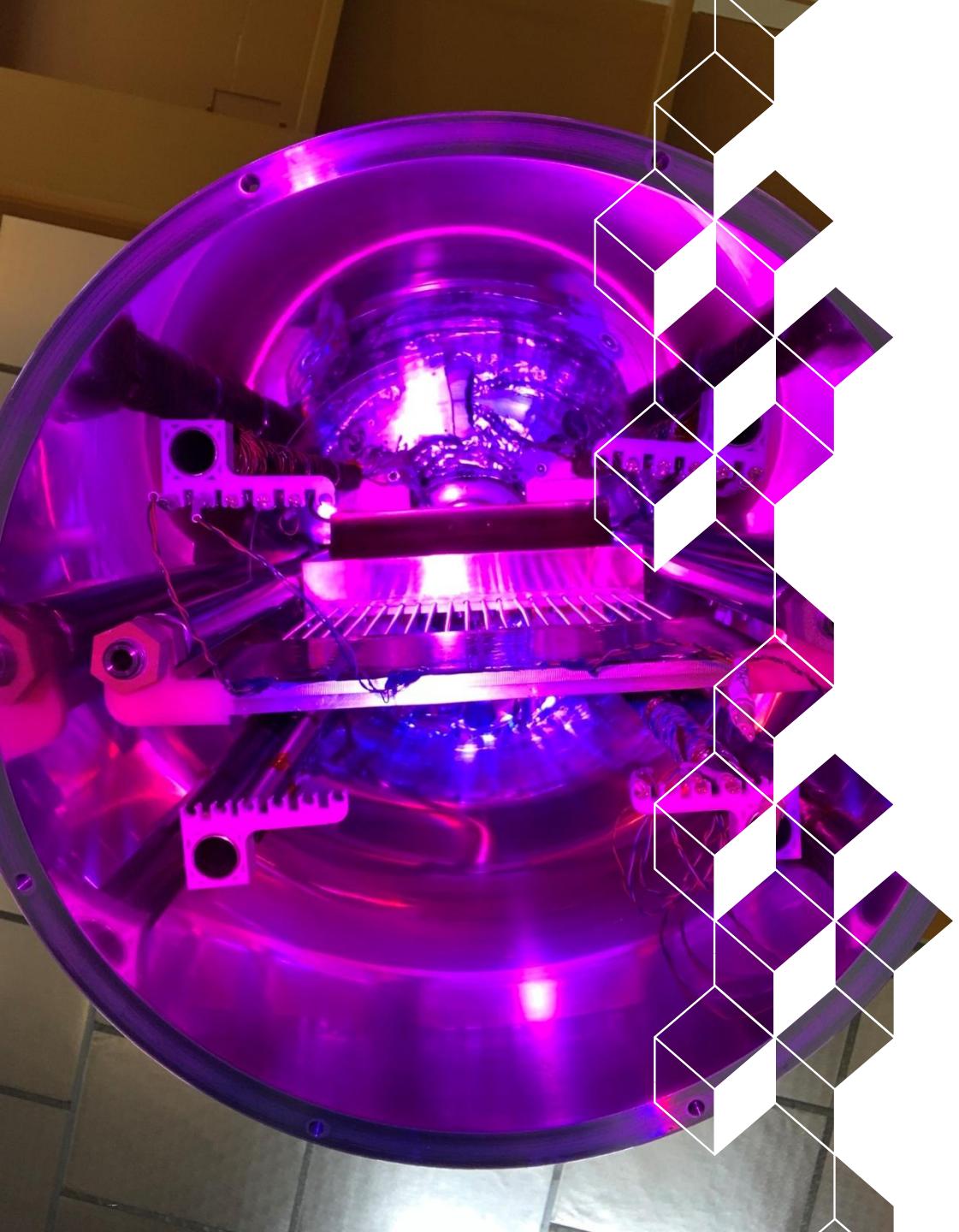


Next step: Manufacturing and Testing... Results next year!

Take-away Ideas: Cryogenic PHPs

- Passive heat transport solution that can serve low temperature range of 2.5 K to 100 K
- Gravity independent – Ground tests in horizontal orientation showcase behavior similar to that in space
- Mono-sized $\frac{1}{2}$ m long PHP is shown to have thermal conductance range
 - 3.7 – 6 W/K for neon 27 – 38 K
 - 0.45 – 1.25 W/K for helium 2.5 – 10 K
- Compared to classical metallic straps, multi-fold times
 - superior thermal performance at iso-mass
 - lighter at iso-conductance
- Heat switch: from insulator to heat transport device!





irfu



Thank you for your
attention!

tisha.dixit@cea.fr

