



Ingeniería y Sistemas S.A.

## Modular and Incremental Framework for GNC Subsystem Validation and Verification

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## Modular and Incremental Framework

SCOPE: Overall GNC/AOCS Validation and Verification process

- early prototyping with generic pre-validated models
- an effective validation of the GNC/AOCS application software
- incremental development of the hardware-in-the-loop (HITL) facility

1. **Algorithm Design Environment:** design of the GNC/AOCS algorithms using generic pre-validated models
2. **Functional Engineering Simulation:** design and validation of the GNC/AOCS application software. Final tuning process is performed in this facility.
3. **Avionics Test Bench:** design qualification of incremental approach and provides an hybrid (fully simulate vs HW stimuli) hardware. (High Fidelity Models replace long leads items or units not allowing HW in the loop)
4. **System Verification:** final verification of the system for those aspects that cannot be verified in previous stages.



## Major Simulator models types, with respect to the item function

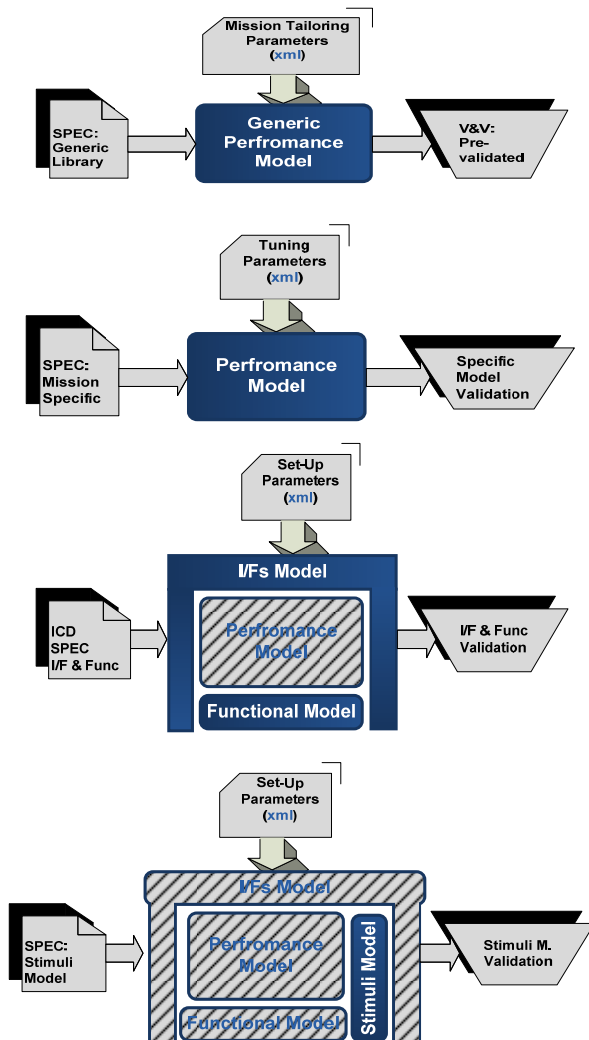
- Sensors and Actuators
- On-board computer
- Subsystem Models (such as Thermal SS, Power SS, etc..)
- Dynamics, Kinematics and Environment Models (including Flexible dynamics)
- Mass, Centring and Inertia Models
- Ground segment model



## Applicability Domain of the Validation and Verification Models

- **DESIGN PHASE:**
  - Generic set of simulator models and rapid prototyping tools like Matlab Simulink.
  - In-house development.
  - Project to project reuse and improvement (lessons learnt).
- **DESIGN VALIDATION PHASE:**
  - Representative of real units' performance with validation against
    - real unit data or
    - mission specific high-fidelity simulated data.
  - Hardware chain latency and jitter.
  - On-board software scheduling.
- **MODEL-BASE VERIFICATION:**
  - High Fidelity models interfacing:
    - Application Software
    - OBC Emulator or electrically to OBC
- **HW-IN-THE-LOOP VERIFICATION:**
  - Enhance the model with the ability:
    - to monitor actuator
    - Stimulate sensors' electronics
  - Hybrid configuration of with both Simulation Models and HW models (modularity & real-time)

## Different Models need, re-usability and impact on system validation



### GENERIC PERFORMANCE MODELS:

- These models are oriented to preliminary design and analysis.
- Mission tailoring parameters are in xml.*

### PERFORMANCE MODELS:

- Mission specific for detailed design validation at CDR level,
- Its re-usability as a core of the H-F Models.
- Tuning parameters are in xml.*

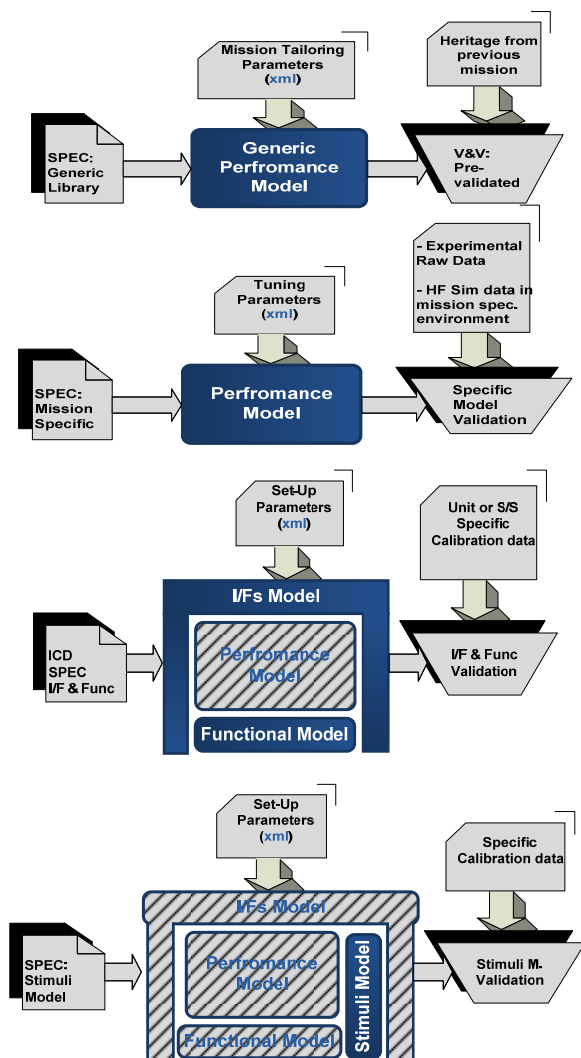
### HIGH FIDELITY MODELS:

- Fulfill part of qualification when no further stage is possible or
- Advance/Prepare avionics qualification with HITL
- Enhance with interfaces and functional model to allow a **Mode-Based Verification**. (*Set-Up Param. xml*)

### HITL MODELS:

- Confirm that electrical interfaces and real units interaction is not affecting functional and performance behavior of the system.
- Enhanced to allow hardware stimulation and monitoring of both sensors and actuators

## Models validation needs throughout the system and subsystem lifecycle



### GENERIC PERFORMANCE MODELS:

- Generic library derived on in-house heritage from previous missions.

### PERFORMANCE MODELS VALIDATION BASED ON:

- Experimental raw data provided by units' supplier
- High fidelity simulation data applicable to the specific mission environment

### H-F MODELS VALIDATION PERFORMED AGAINST:

- Specific calibration data obtained on dedicated test campaign or
- Measurement performed at system, subsystem and units level on mission dedicated hardware.

### HITL MODELS VALIDATION ARE ACHIEVED ON:

- Specific calibration data and
- reference cases

## Most relevant systems properties required in each stage of the V&V process and their function in the applicable domain

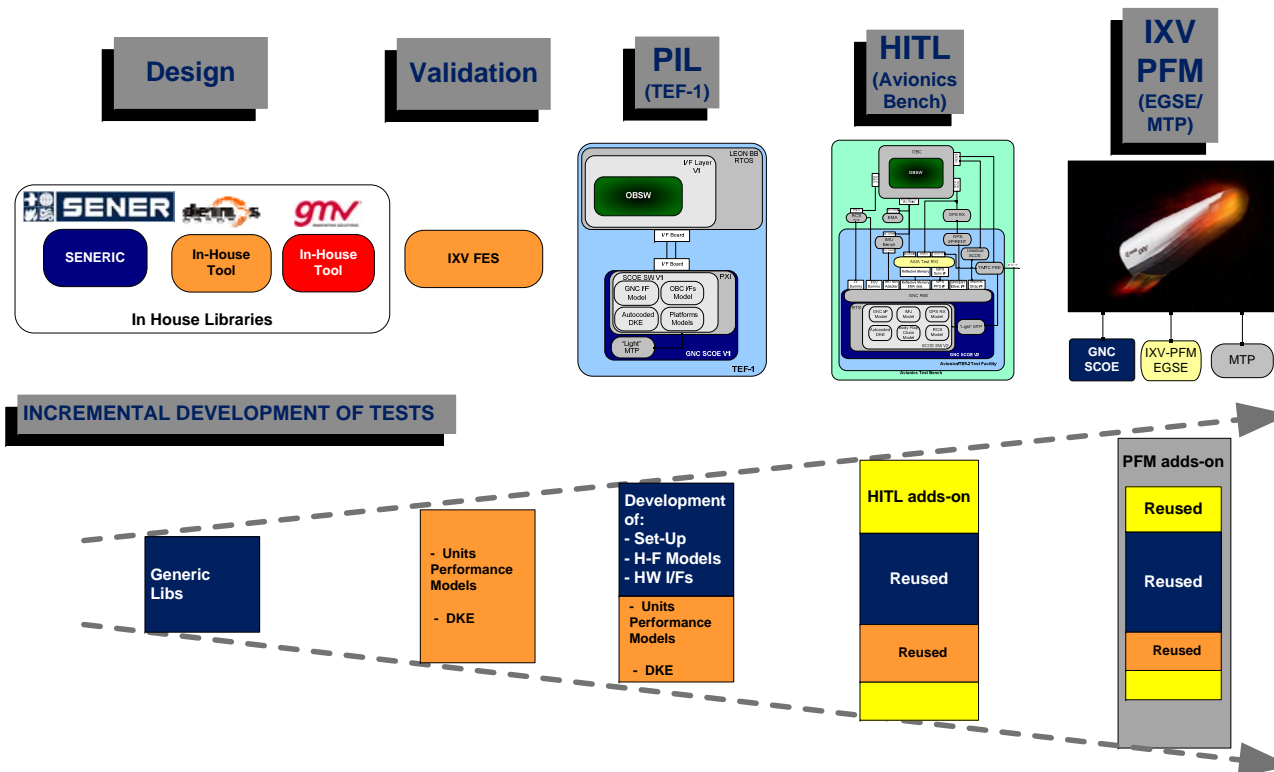
<i>System Properties</i>	<i>Design</i>	<i>Design Validation</i>	<i>Model-Based Verification</i>	<i>HITL Verification</i>
Mass, Centring and Inertia data	X	X	X	X
Control Mode Architecture	X	X	X	X
Equipment Matrix	X	X	X	X(*)
Functional System Properties	X	X	X	X
Measured Mounting data				X(*)
Software parameters		X	X	X
Mission specific performance model data		X	X(**)	X(**)
Calibration data			X(**)	X(**)
TM/TC database			X	X
Electrical ICDs				X

- (\*) Equipment Matrix data are partially replace by Measured Mounting data.
- (\*\*) Mission specific performance model data are mostly replace by Calibration data.



## IXV: V&V Approach to Model-Based Validation

- Incremental Development allow early verification and preparation of the HITL
- Model-Based Verification allow to accommodate for verification in absence of Long Leads Items (Body Flap Chain).





## Proba-3: V&V Approach to Model-Based Validation

- Three Phases are applicable for a design to cost approach:
  - Design
  - Design Validation
  - Model-Based Verification
- The Formation Flying System SW is verified at closed loop level in pure numerical Model-Based simulator facility named SW based test bench (SBTB)

