



# Uncertainty Propagation Using Differential Algebra

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Workshop on Nonlinear Uncertainty Propagation using Differential Algebra 22nd September 2015

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## Differential Algebra: basic concept

- Uncertainty propagation
  - ODEs flow expansion
  - Linear covariance propagation
  - DA-based Monte Carlo
  - DA-based statistical moments
  - Polynomial bounder
- Applications
  - SST: Impact probability computation
  - High-order filters
- Advanced methods
  - Automatic domain splitting





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Differential Algebra (DA) is an automatic differentiation technique



- Unlike standard automatic differentiation tools, the analytic operations of differentiation and antiderivation are introduced
- DA can be implemented in a computer environment (COSY-Infinity - Berz&Makino 1998, DACE, Jet Transport, JACK)
- Given any sufficiently regular function f of v, DA enables the computation of its Taylor expansion up to an arbitrary order k



# **Uncertainty propagation**





low computational burden

low accuracy



computationally intensive

Monte Carlo

high accuracy

Can we find a compromise technique?

- We need a technique to:
  - Improve accuracy of linearized models
  - Reduce computational cost of classical Monte Carlo



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Linearized models

Differential Algebra

--- Mo

Monte Carlo



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 Reduce computational cost of classical Monte Carlo





Given any dynamics



- Initialize initial conditions as a DA vector:  $[m{x}_0] = m{x}_0 + \deltam{x}_0$
- Perform all operations of the integration scheme in the DA algebra

E.g.: Euler's scheme: 
$$oldsymbol{x}_1 = oldsymbol{x}_0 + oldsymbol{f}(oldsymbol{x}_0) \cdot h$$



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 $[x_1] = \mathcal{T}_{x_1}(\delta x_0)$ : *k*-th order Taylor expansion of the solution at  $t_1$ 



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E.g.: Euler's scheme: 
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 $\sum [x_2] = \mathcal{T}_{x_2}(\delta x_0)$ : *k*-th order Taylor expansion of the solution at  $t_2$ 



Given any dynamics

$$\dot{x} = f(x, t)$$
  
 $x_0$   
 $x_0$   
 $x_f$   
 $\mathcal{T}_{x_f}(\delta x_0)$ 

- Initialize initial conditions as a DA vector:  $[m{x}_0] = m{x}_0 + \deltam{x}_0$
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E.g.: Euler's scheme: 
$$[oldsymbol{x}_f] = [oldsymbol{x}_{f-1}] + oldsymbol{f}([oldsymbol{x}_{f-1}]) \cdot h$$

 $\label{eq:relation} \boxed{[x_f] = \mathcal{T}_{x_f}(\delta x_0): \textit{k-th order Taylor expansion of the solution at } t_f}$ 



# Expansion of the flow of dynamics

- Example: 2-body dynamics
  - Eccentricity: 0.5 Starting point: pericenter
  - Integration scheme: Runge-Kutta (variable step, order 8)
  - DA-based ODE flow expansion order: 6
- Initial box: 0.008 AU in x and 0.08 AU in y





• If all computations are performed to order 1, the final Taylor expansion  $\mathcal{T}_{\boldsymbol{x}_f}^1(\delta \boldsymbol{x}_0)$  coincides with the state-transition matrix  $\Phi(t_f, t_0)$ initial covariance  $C_0$  can be propagated as:  $C_f = \Phi C_0 \Phi^T$ 



• If all computations are performed to order 1, the final Taylor expansion  $\mathcal{T}^1_{x_f}(\delta x_0)$  coincides with the state-transition matrix  $\Phi(t_f, t_0)$ 

 $\Longrightarrow$  Initial covariance  $\ C_0$  can be propagated as:  $\ C_f = \Phi \, C_0 \, \Phi^T$ 

No need of variational equations!



## **DA-based Monte Carlo**



- Any pointwise integration can be replaced by the evaluation of the polynomial  $\mathcal{T}_{m{x}_f}(\delta m{x}_0)$ 



Saving in computational time w.r.t. classical MC



## **DA-based Monte Carlo**



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- Note:
  - The accuracy of Taylor expansion can be tuned with order
  - The same polynomial can be used to map different statistics



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- Given the k-order polynomial map:  $[{m x}_f] = \mathcal{T}_{{m x}_f}(\delta {m x}_0)$
- Moments of the pdf can be  $x_f$  obtained by computing the expectation of the Taylor expansion: for the mean

$$E[\boldsymbol{x}_{f}] = \sum_{p=1}^{k} \frac{1}{p!} \Phi_{1...p} E[\delta x_{0,1} \dots \delta x_{0,p}] \quad \text{computed by Isserlis's formula}$$

 Same approach for high order moments (covariance, skeweness, kurtosis,...)



## **DA-based statistical moments**



(a) Samples distribution after 0.8 orbit



(c) Samples distribution after 10 orbits



(b) Samples distribution after 5 orbits



(d) Samples distribution after 30 orbits



## **DA-based statistical moments**



(a) Uncertainty ellipse after 0.8 orbit



(c) Uncertainty ellipse after 10 orbits



(b) Uncertainty ellipse after 5 orbits



(d) Uncertainty ellipse after 30 orbits



## **Polynomial bounder**





- Polynomial bounders can immediately estimate the range of  $x_f$ 



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- Example: very useful to discard the occurrence of an event
  - Impact of NEOs with Earth







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# **SST: impact probability computation**

### Suppose no impact occurrence for nominal initial states





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- Impact conditions may occur for  $x_0^{sat_i} \neq \overline{x}_0^{sat_i}$  at  $t_{DCA} \neq \overline{t}_{DCA}$





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  - Compute:  $[x_f^{sat_i}] = \mathcal{T}_{x_f^{sat_i}}(\delta x_0^{sat_i}, \delta t_f)$  for i = 1, 2 $\square [d^2] = \mathcal{T}_{d^2}(\delta x_0^{sat_i}, \delta t_f) \square [\partial d^2 / \partial t_f] = \mathcal{T}_{d^2}(\delta x_0^{sat_i}, \delta t_f)$





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  - Invert:

 $[\delta t_f] = \mathcal{T}_{d^2}(\delta x_0^{sat_i}, \partial d^2/\partial t_f)$ 

• Impose  $\partial d^2 / \partial t_f = 0$ :  $[t_{DCA}] = \mathcal{T}_{\partial d^2 = 0}(\delta x_0^{sat_i})$  $[d_{DCA}] = \mathcal{T}_{DCA}(\delta x_0^{sat_i})$ 





# **SST: impact probability computation**

$$[d_{DCA}] = \mathcal{T}_{DCA}(\delta x_0^{sat_i})$$

 DA-based fast Monte Carlo simulations can be run on *T*<sub>DCA</sub> to compute # of samples with

 $\mathcal{T}_{DCA} < \mbox{ threshold}$ 

- The map  $\mathcal{T}_{DCA}(\delta x_0^{sat_i})$  was used for:
  - DA-based Monte Carlo (DAMC)
  - DA-based Line Sampling (DALS)
  - DA-based Subset Simulation (DASS)
- Different orbital regimes: LEO, MEO, GEO



more efficient for

low probabilities



- Efficiency of LS and SS over standard Monte Carlo for low probabilities:
  - Cumulative impact probability distribution for a close conjunction
  - # of samples: 14000





- Efficiency of LS and SS over standard Monte Carlo for low probabilities:
  - Cumulative impact probability distribution for a close conjunction
  - # of samples: 14000
  - No samples generated by standard MC below 5 m distance





- Efficiency of DAMC, DASS, and DALS over standard MC
- Three test cases (Alfano, 2009)
  - Test 5: linear relative motion ( $P_c = 4.440 \text{E-2}$ )
  - Test 6: boundary of linear relative motion ( $P_c = 4.324E-3$ )
  - Test 7: nonlinear relative motion ( $P_c = 1.580E-4$ )

Test case	Method	$P_c$ [-]	$N_T$	$t_c \ [ m s]$
	MC	4.452E-2	1.0E + 5	4.75
5	DAMC-3	4.459E-2	1.0E + 5	0.67
0	DALS-3	4.451E-2	5.0E + 3	2.53
	DASS-3	4.450E-2	$2.0E{+}4$	0.13
	MC	4.339E-3	1.0E + 6	43.21
6	DAMC-3	4.350E-3	1.0E + 6	6.67
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	DAMC-3	1.612E-4	$2.7E{+7}$	179.34
	DALS-3	1.621E-4	5.0E+3	1.43
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## **High-order Kalman filters**

• Given the discrete system model:

$$\mathbf{x}_{k+1} = \boldsymbol{\phi}(t_{k+1}; \mathbf{x}_k, t_k) + \mathbf{w}_k$$
$$\mathbf{z}_{k+1} = \mathbf{h}(\mathbf{x}_{k+1}, t_{k+1}) + \mathbf{v}_{k+1}$$

- Suppose to have the state  $\mathbf{x}_k$ , with mean  $\mathbf{m}_k^+$  and covariance  $\mathbf{P}_k^+$
- General filtering algorithm:
  - Prediction:

$$\mathbf{m}_{k+1}^{-} = E[\boldsymbol{\phi}(t_{k+1}; \mathbf{x}_k, t_k) + \mathbf{w}_k \mid \mathbf{z}_k]$$
$$\mathbf{P}_{k+1}^{-} = E\{[\boldsymbol{\phi}(t_{k+1}; \mathbf{x}_k, t_k) + \mathbf{w}_k][\boldsymbol{\phi}(t_{k+1}; \mathbf{x}_k, t_k) + \mathbf{w}_k]^T \mid \mathbf{z}_k\} - (\mathbf{m}_{k+1}^{-})(\mathbf{m}_{k+1}^{-})^T$$
$$\mathbf{n}_{k+1}^{-} = E[\mathbf{h}(\mathbf{x}_{k+1}, t_{k+1}) + \mathbf{v}_{k+1} \mid \mathbf{z}_k]$$

• Update:



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classical approach: linearization

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• Update:



DA provides high-order polynomials for

$$egin{aligned} \mathbf{x}_{k+1} &= \mathcal{T}_{oldsymbol{\phi}_{k,k+1}}(\delta \mathbf{x}_k) + \mathbf{w}_k \ \mathbf{z}_{k+1} &= \mathcal{T}_{\mathbf{h}_{k+1}}(\delta \mathbf{x}_{k+1}) + \mathbf{v}_{k+1} \end{aligned}$$

Expectations can then be computed on polynomials. E.g.:

$$\mathbf{m}_{k+1}^{-} = \sum_{p=1}^{m} \frac{1}{p!} \phi_{k,k+1}^{\gamma_1 \dots \gamma_p} E(\delta \mathbf{x}_k^{\gamma_1} \dots \delta \mathbf{x}_k^{\gamma_p})$$
coefficients of the polynomials

where  $E(\delta \mathbf{x}_k^{\gamma_1} \dots \delta \mathbf{x}_k^{\gamma_p})$  is computed using Isserlis' formula

- Prediction step is fully nonlinear:
  - Faster convergence
  - Possible reduction of measurements acquisition frequency



# **High-order Kalman filters**

- Example: 2-Body Problem
  - State variables: S/C position and velocity
  - Uncertainty on initial conditions (10% off from true initial state) and no process noise
  - Nonlinear measurements with realistic measurement noise (radial position of the S/C wrt the Earth and the LOS direction to Earth)







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# **Automatic domain splitting**

 In challenging cases, a high-order Taylor expansion is not sufficient to accurately map uncertainties





 In challenging cases, a high-order Taylor expansion is not sufficient to accurately map uncertainties



- An algorithm can be implemented to automatically split the initial domain to guarantee the desired accuracy
- Very useful for: large uncertainty sets, long-term propagations, highly nonlinear dynamics

























![](_page_50_Picture_0.jpeg)

![](_page_50_Picture_1.jpeg)

![](_page_50_Picture_2.jpeg)

![](_page_51_Picture_0.jpeg)

![](_page_51_Picture_1.jpeg)

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![](_page_52_Figure_1.jpeg)

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![](_page_61_Figure_1.jpeg)

![](_page_61_Figure_2.jpeg)

![](_page_62_Picture_0.jpeg)

# **Automatic domain splitting**

Long term propagation of Apophis motion (NEODys in Sept. 2009)

![](_page_62_Figure_3.jpeg)

![](_page_63_Picture_0.jpeg)

# **Automatic domain splitting**

![](_page_63_Figure_2.jpeg)

![](_page_64_Picture_0.jpeg)

## Conclusions

![](_page_64_Picture_2.jpeg)

- improve efficiency of standard nonlinear methods
- improve accuracy of standard linear methods
- Note: method based on Taylor approximations

Size of uncertainty set and order shall guarantee sufficient accuracy

- Polynomials can be manipulated to impose constraints on both initial and final state (e.g. see SST application)
- If linear methods are sufficiently accurate for your application, you may not need to increase order, however...
- ... DA relieves you from the "pain" of writing variational equations

![](_page_65_Picture_0.jpeg)

## References

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![](_page_66_Picture_0.jpeg)

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- High-order robust guidance of interplanetary trajectories based on differential algebra. Journal of Aerospace Engineering, Sciences and Applications

![](_page_67_Picture_0.jpeg)

![](_page_67_Picture_1.jpeg)

![](_page_67_Picture_2.jpeg)

# Uncertainty Propagation Using Differential Algebra

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Workshop on Nonlinear Uncertainty Propagation using Differential Algebra 22nd September 2015

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